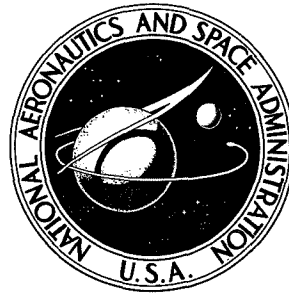


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GENENG II — A PROGRAM FOR CALCULATING
DESIGN AND OFF-DESIGN PERFORMANCE
OF TWO- AND THREE-SPOOL TURBOFANS
WITH AS MANY AS THREE NOZZLES

by Laurence H. Fishbach and Robert W. Koenig

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16. Abstract A computer program titled GENENG II which calculates steady-state design and off-design jet engine performance for two- or three-spool turbofans with one, two, or three nozzles is described. Included in the report are complete FORTRAN IV listings of the program with sample results for nine basic turbofan engines that can be calculated: (1) three-spool, three-stream engine; (2) two-spool, three-stream, boosted-fan engine; (3) two-spool, three-stream, supercharged-compressor engine; (4) three-spool, two-stream engine; (5) two-spool, two-stream engine; (6) three-spool, three-stream, aft-fan engine; (7) two-spool, three-stream, aft-fan engine; (8) two-spool, two-stream, aft-fan engine; (9) three-spool, two-stream, aft-fan engine. The simulation of other engines by using logical variables built into the program is also described. The computer program is available from the authors.					
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SUMMARY

A digital computer program titled GENENG II is described. This program is a derivative of GENENG standing for GENeralized ENGine. GENENG which is capable of calculating steady-state design and off-design performance of turbofan and turbojet engines was evolved from SMOTE (Simulation Of Turbofan Engine) which was developed by the Turbine Engine Division of the Air Force Aero Propulsion Laboratory, Wright-Patterson Air Force Base, Ohio.

GENENG II calculates design and off-design jet engine performance for existing or theoretical turbofan engines with two or three spools and with one, two, or three nozzles. In addition, aft-fan engines can be calculated. Changes to the original SMOTE and GENENG are discussed.

Included in the report are complete FORTRAN IV listings of the program with sample results for nine basic turbofan engines that can be calculated without any programming changes:

- (1) Three-spool, three-stream engine
- (2) Two-spool, three-stream boosted-fan engine
- (3) Two-spool, three-stream, supercharged-compressor engine
- (4) Three-spool, two-stream engine
- (5) Two-spool, two-stream engine
- (6) Three-spool, three-stream, aft-fan engine
- (7) Two-spool, three-stream, aft-fan engine
- (8) Two-spool, two-stream, aft-fan engine
- (9) Three-spool, two-stream, aft-fan engine

The first three of these engines are likely candidates for a STOL aircraft with internally blown flaps. By examining the methods used to simulate these engines, the reader may simulate others. As examples, a boosted aft-fan engine with two streams would simulate a high-bypass-ratio engine where the core and tip portions of the fan have different component performance maps; a boosted-fan, two-stream engine could be simulated (JT9D type); or supercharged-compressor, two-stream engines could be studied. The number of possibilities are too many to enumerate, being determined by the imagination of the user.

INTRODUCTION

For preliminary as well as in-depth studies it is often necessary to study a broad range of engines operating at both design and off-design conditions in order to find an efficient airframe/engine combination. The spectrum of flight conditions through which an engine must operate will strongly affect the optimum design parameters for that engine.

The SMOTE code (SiMulation Of Turbofan Engines), discussed in references 1 and 2, provided a computer program having off-design-point calculation capability for either existing engines or theoretical ones - a major advance. Theoretical engines are simulated by scaling component performance from existing engines to the design conditions of the theoretical engine.

GENENG (GENeralized ENGine), a computer code derived from SMOTE and reported in a companion report to this one (ref. 3), greatly increased the versatility of the original code while retaining the ability to simulate theoretical engines. The most significant change was providing the capability of studying one- and two-spool turbojets as well as turbofans.

Additional changes to SMOTE included generalization of afterburner performance maps, an automatic redesign of the fan and compressor pressure ratios for mixed-flow turbofans (one-stream engines) if the static pressures at the mix point do not match, duct combustor pressure losses, a new method of entering data into the program, and an automatic recall of previously loaded design-point data so that it is only necessary to change what is being varied when studying a series of design engines.

This report describes GENENG II, a derivative program from GENENG. A need has arisen for the capability of calculating the performance of two- or three-spool turbofan engines with as many as three nozzles (or airstreams). An example of this type of engine would be one in which a fan is used to compress all the air, of which some is expanded through a separate nozzle to produce thrust. The remaining air passes through a compressor, after which some air is put into a wing duct and expelled over the wing flaps (an internally blown flap). The remaining air passes through another compressor into a combustor; is heated and expanded through three turbines, each of which drives one of the compressors; and is then expelled out the third (main) nozzle, producing more thrust. This engine type is under consideration for STOL aircraft; and until the development of GENENG II, off-design performance calculations were difficult to attain.

GENENG II was developed to provide the capability to study this engine type. Once this capability had been achieved, it was realized that many other engine types could be simulated by building simple options into the code and modifying the input data to the program. As an example, the fan and first compressor in the engine just described could be physically attached and driven by one turbine (the so-called 'boosted turbofan'),

or the fan could be put at the rear of the engine (an aft fan). Thus GENENG II became a very versatile program with many engine design options built in internally. These are described in the next section ENGINE TYPES.

The GENENG II computer code is available from the authors upon request. This FORTRAN IV program can be used by computer centers having an IBM 7094 Model 2 computer. With modifications, the program can be used on all computers that have a FORTRAN compiler.

ENGINE TYPES

All thermodynamic properties of air and gas are calculated by considering variable specific heats and no dissociation. The air and gas property tables of reference 4 were curve fit and are used herein.

Type a - Three-Spool, Three-Stream Turbofan

The basic engine, a three-spool, three-stream turbofan, of which all other engine types are treated as variations, is shown in figure 1. Free-stream conditions exist at station 1. The conditions at station 2 are determined by flight conditions and inlet recovery. GENENG compressor maps work with corrected values of airflow. At the entrance to the fan, the corrected airflow $WA_{F,c}$ is

$$WA_{F,c} = \frac{WA_F \sqrt{T_2/T_{518.668}}}{P_2/P_{SLS}} \quad (1)$$

where P_2 and P_{SLS} are in atmospheres and P_{SLS} equals 1.0. All symbols are defined in appendix C. Some symbols are formed as the combination of other symbols; thus WA is airflow, F is for fan, and c when following a component symbol means corrected. Station numbers are defined on the appropriate figure.

All the fan air WA_F is compressed by the fan giving rise to conditions at station 22. The power required to do this is

$$\text{Fan power} = WA_F \times (H_{22} - H_2) \quad (2)$$

Some fan air may be lost to the cycle as fan bleed Bl_F , which is expressed as a fraction of the fan airflow

$$Bl_F = PC_{Bl, F} \times WA_F \quad (3)$$

The corrected airflow into the intermediate compressor is

$$WA_{I, c} = \frac{WA_I \sqrt{T_{22}/T_{518.668}}}{P_{22}/1.0} \quad (4)$$

The remaining air goes through the fan duct, where some leakage from the core air may also enter (see eq. (16)).

$$WA_D = WA_F - Bl_F - WA_I + Bl_{DU} \quad (5)$$

This air, which may be heated by a duct burner to a temperature T_{24} , undergoes a pressure drop

$$P_{25} = P_{24} \times \left[1 - \left(\frac{\Delta P}{P} \right)_{DUCT} \right] \quad (6)$$

The air would have been heated by the addition of fuel, which can be expressed as a fuel-air ratio so that

$$WG_{24} = WA_{23} \times \left[1 + (f/a)_{23} \right] \quad (7)$$

The gas is then expanded through a nozzle (station 29) to produce thrust. The bypass ratio is defined by

$$BYPASS = \frac{WA_D}{WA_I} \quad (8)$$

The air going into the intermediate compressor is compressed to the conditions at station 21. The power required is

$$\text{Intermediate-compressor power} = WA_I \times (H_{21} - H_{22}) \quad (9)$$

The conditions at station 21 are the same as those at station 32, which is the entrance to the wing duct as the third streampath is called herein. The airflow entering this duct is

called Bl_I , meaning intermediate bleed flow, and is expressed as a fraction $PC_{Bl,I}$ of the total airflow at station 21.

$$Bl_I = PC_{Bl,I} \times WA_I \quad (10)$$

The remainder of the air enters the core compressor

$$WA_C = WA_I - Bl_I \quad (11)$$

and

$$WA_{C,c} = \frac{WA_C \times \sqrt{T_{21}/T_{518.668}}}{P_{21}^{1.0}} \quad (12)$$

The air entering the wing duct experiences a pressure drop

$$P_{36} = P_{32} \times \left[1 - \left(\frac{\Delta P}{P} \right)_{WING} \right] \quad (13)$$

and then passes through a nozzle (station 39) to produce additional thrust. The air continuing on through the core is compressed to conditions at station 3. The power required is

$$\text{Core compressor power} = WA_C \times (H_3 - H_{21}) = WA_3 \times (H_3 - H_{21}) \quad (14)$$

Some core bleed air Bl_C may be used for turbine cooling. Some of the air is put back into the cycle into each of the three turbines, and some is lost to the cycle as overboard bleed or leakage into the fan duct.

$$Bl_C = PC_{Bl,C} \times WA_3 \quad (15)$$

$$Bl_{DU} = PC_{Bl,DU} \times Bl_C \quad (16)$$

$$Bl_{OB} = PC_{Bl,OB} \times Bl_C \quad (17)$$

$$Bl_{HP} = PC_{Bl,HP} \times Bl_C \quad (18)$$

$$Bl_{IP} = PC_{Bl, IP} \times Bl_C \quad (19)$$

$$Bl_{LP} = PC_{Bl, LP} \times Bl_C \quad (20)$$

Since $Bl_{DU} + Bl_{OB} + Bl_{HP} + Bl_{IP} + Bl_{LP} = Bl_C$, the sum of $PC_{Bl, DU}$, $PC_{Bl, OB}$, $PC_{Bl, HP}$, $PC_{Bl, IP}$, and $PC_{Bl, LP}$ must be equal to 1.
The remaining air is

$$WA_4 = WA_3 - Bl_C \quad (21)$$

and is heated to a turbine inlet temperature T_4 and goes through a combustor pressure drop $(\Delta P/P)_{COMB}$. The fuel required to do this is expressed as a fuel-air ratio $(f/a)_4$ so that the gas entering the first turbine WG_4 can be expressed as

$$WG_4 = WA_4 \times [1 + (f/a)_4] \quad (22)$$

This gas is then expanded through this high-pressure turbine to conditions at station 50. The enthalpy at station 50 is first calculated by making a power balance since this turbine drives the core compressor and supplies any work extracted (HPEXT). By using equation (14)

$$WG_4 \times (H_4 - H_{50}) = WA_3 \times (H_3 - H_{21}) + HPEXT \quad (23)$$

In addition, the physical speeds must match

$$N_{HP, TURBINE} = N_{COMP} \quad (24)$$

If high-pressure-turbine bleed air Bl_{HP} is added back into the cycle at this point, H_{50} must be readjusted

$$H_{50} = \frac{(Bl_{HP} \times H_3) + WG_4 H_{50}}{WG_4 + Bl_{HP}} = \frac{(Bl_{HP} \times H_3) + WG_4 H_{50}}{WG_{50}} \quad (25)$$

Similarly,

$$WG_{50} \times (H_{50} - H_5) = WA_I \times (H_{21} - H_{22}) \quad (26)$$

$$N_{IP, TURBINE} = N_{INT COMP} \quad (27)$$

$$H_5 = \frac{(Bl_{IP} \times H_3) + WG_{50}H_5}{WG_{50} + Bl_{IP}} = \frac{(Bl_{IP} \times H_3) + WG_{50}H_5}{WG_5} \quad (28)$$

$$WG_5 \times (H_5 - H_{55}) = WA \times (H_{22} - H_2) \quad (29)$$

$$N_{LP, TURBINE} = N_{FAN} \quad (30)$$

$$H_{55} = \frac{(Bl_{LP} \times H_3) + WG_5H_{55}}{WG_5 + Bl_{LP}} = \frac{(Bl_{LP} \times H_3) + WG_5H_{55}}{WG_{55}} \quad (31)$$

The gas flow WG_{55} then may be heated by an afterburner to a gas temperature T_7 and may undergo a pressure drop.

$$P_7 = P_6 \left[1 - \left(\frac{\Delta P}{P} \right)_{AFTERBURNER} \right] \quad (32)$$

The gas flow would be increased by any fuel burned.

$$WG_7 = WG_{55} + WFA \quad (33)$$

The gas is then expanded through the nozzle (station 9) to produce the remainder of the total engine thrust.

Type b - Two-Spool, Three-Stream, Boosted-Fan Turbofan

From figure 2 it is immediately apparent why the three-spool, three-stream engine can be modified to represent the other types presented herein. The only difference between engine b and engine a is that the intermediate compressor is physically attached to the fan in terms of speed and the combination is driven by one turbine (the low-pressure turbine). The thermodynamic calculation changes are that the speeds are attached.

$$N_{INT COMP} = N_{FAN} \quad (34)$$

The power of the low-pressure turbine is now

$$WG_{50} \times (H_{50} - H_{55}) = WA_F \times (H_{22} - H_2) + WA_I \times (H_{21} - H_{22}) \quad (35)$$

$PC_{Bl,IP}$ must be zero and H_{55} is readjusted by

$$H_{55} = \frac{(Bl_{LP} \times H_3) + WG_{50}H_{55}}{WG_{50} + Bl_{LP}} \quad (36)$$

This type of engine is of interest because it might be created by adding a new boosted-fan - turbine combination to an existing core. If the third airstream is deleted (see engine e) and ductburner and afterburner are removed, engine b becomes a two-spool, two-stream turbofan of the type represented by the General Electric CF6 and Pratt & Whitney JT9D turbofan, both of which have booster stages on the fan.

Type c - Two-Spool, Three-Stream Supercharged-Compressor Turbofan

Engine c is shown in figure 3. Here, the intermediate and core compressors have been physically attached. For programing reasons, the combination is driven by the intermediate-pressure turbine. The calculation procedure bypasses the routine which calculates high-pressure-turbine performance but transfers the turbine performance data from this routine into that of the intermediate-pressure turbine to represent the turbine performance. Since the intermediate-pressure turbine speed is set by the speed of the intermediate compressor which also sets the speed of the combination of the compressors, this procedure was necessary.

$$N_{COMP} = N_{INT COMP} \quad (37)$$

$$WG_{50} \times (H_{50} - H_5) = WA_I \times (H_{21} - H_{22}) + WA_C \times (H_3 - H_{21}) + HPEXT \quad (38)$$

$PC_{Bl,HP}$ must be zero and H_5 is readjusted by

$$H_5 = \frac{(Bl_{IP} \times H_3) + WG_{50}H_5}{WG_{50} + Bl_{IP}} \quad (39)$$

Type d - Three-Spool, Two-Stream Turbofan

Engine d, shown in figure 4, is presently in existence (Rolls Royce RB 211) and differs from the reference engine in that all the air entering the intermediate compressor also enters the inner compressor. For this reason, the only change necessary to run this engine is to set $PC_{B1,I}$ equal to zero.

Type e - Two-Spool, Two-Stream Turbofan

Engine e is the typical turbofan and is shown in figure 5. To simulate this engine, it is necessary to have the air go through the intermediate compressor at a pressure ratio of 1.0 and an efficiency of 1.0 and to bypass the intermediate-pressure-turbine calculations. A logical control (DUMMYSPOOL) has been built into the program to do this. At the same time, $PC_{B1,I}$ must be set equal to zero. By using this option, GENENG II can be used to replace its original version GENENG (ref. 3) in calculating turbofan performance. It cannot, however do turbojet calculations (two-spool, one-stream or one-spool, one-stream engines). As mentioned earlier, boosted-fan, two-spool, two-stream engines can be calculated by setting $PC_{B1,I}$ equal to zero in engine b.

Type f - Three-Spool, Three-Stream Aft-Fan Turbofan

The three-spool, three-stream aft-fan engine is shown in figure 6. Thermodynamically, the only difference between this and the reference engine is that the intermediate compressor sees the same conditions at its entrance as does the fan (conditions at station 2; both inlets assumed to have the same performance). This is accomplished by setting a logical control variable AFTFAN to be true. The power of the intermediate-pressure turbine would be

$$WG_{50} \times (H_{50} - H_5) = WA_I \times (H_{21} - H_2) \quad (40)$$

Each of the aft-fan engines has a counterpart in the front-fan engines, the only difference being that the intermediate compressor (or in the case of engine h, a two-spool, two-stream aft-fan engine, the compressor) sees free-stream conditions. These engines and their counterparts are described in the following sections.

Type g - Two-Spool, Three-Stream Aft-Fan Turbofan

Engine g, a counterpart of engine c (fig. 3), is shown in figure 7. The power balance would be

$$WG_{50} \times (H_{50} - H_5) = WA_I \times (H_{32} - H_2) + WA_C \times (H_3 - H_{32}) \quad (41)$$

Type h - Two-Spool, Two-Stream Aft-Fan Turbofan

Engine h, a counterpart of engine e (fig. 5), is shown in figure 8. The power balance would be

$$WG_{50} \times (H_{50} - H_5) = WA_C \times (H_3 - H_2) \quad (42)$$

Type i - Three-Spool, Two-Stream Aft-Fan Turbofan

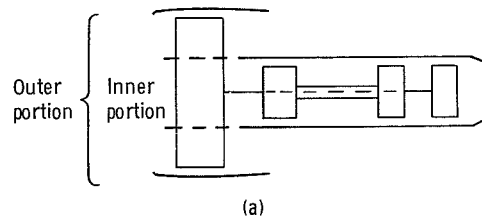
Engine i, a counterpart of engine d (fig. 4), is shown in figure 9. The power balance would be

$$WG_{50} \times (H_{50} - H_5) = WA_I \times (H_{21} - H_2) \quad (43)$$

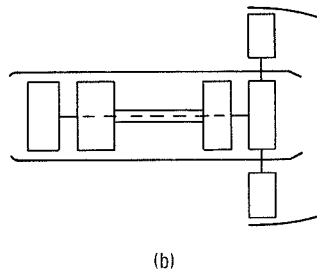
Other Engines

By using his imagination in conjunction with the engines illustrated, the reader can determine other engine types which can be simulated. An obvious one is a supercharged-compressor, two-stream turbofan which is a derivative of engine c, the only change necessary being setting $PC_{B1,I} = 0$. In addition, all engines illustrated could be run as mixed-flow engines eliminating the fan duct nozzle (see ref. 3).

An interesting engine more difficult to be simulated is a high-bypass-ratio turbofan (two streams), where the outer and inner portions of the fan are represented by different performance maps. As can be seen by the following sketches, this engine can be simulated by a boosted aft-fan engine. When AFTFAN is true, the second spool sees free-stream conditions. When the fan and intermediate spool are attached, the physical rotational speeds of the aft fan (outer portion of fan) and the second spool (inner portion of fan) will be the same. Both are driven off the same turbine.



The high-bypass-ratio turbofan (sketch a) can be simulated by a boosted aft-fan engine (sketch b).



BALANCING TECHNIQUE

An off-design engine cycle calculation requires satisfying various matching constraints (rotational speeds, airflows, compressor and turbine work functions, and nozzle flow functions) at each specified operating condition. GENENG II internally searches for compressor and turbine operating points that will satisfy the constraints. It does this by generating differential errors caused by small changes in the independent variables. The program then uses a matrix that is loaded with the differential errors to solve for the zero-error condition. This procedure is known as the Newton-Raphson iteration technique.

For a three-spool engine, a solution for a set of nine simultaneous linear equations is obtained; for other types, less equations are used. The nine independent variables selected are

ZF Ratio of pressure ratios of fan compressor along a speed line,

$$ZF = \frac{(\text{Pressure ratio along speed line}) - (\text{Low pressure ratio on speed line})}{(\text{High pressure ratio on speed line}) - (\text{Low pressure ratio on speed line})}$$

PCNF or T4 Percent fan speed or turbine inlet temperature

ZI	Ratio of pressure ratios of intermediate compressor along a speed line (calculated the same as ZF)
PCNI	Percent intermediate compressor speed
ZC	Ratio of pressure ratios of inner compressor along a speed line (calculated same as ZF)
PCNC or T4	Percent inner compressor speed or turbine inlet temperature
TFFHP	High-pressure-turbine flow function, $WG_4 \sqrt{T_4/P_4}$
TFFIP	Intermediate-pressure-turbine flow function, $WG_{50} \sqrt{T_{50}/P_{50}}$
TFFLP	Low-pressure-turbine flow function, $WG_5 \sqrt{T_5/P_5}$

The program initially selects new (perturbed) values for the variables, based on the design values. It is then possible to proceed through the entire engine cycle calculations, where up to nine errors are generated. The initial values of the nine (or less) variables and nine (or less) errors are base values.

As per reference 1, the partial differential equations for $E = f(V)$ are

$$dE_i = \sum_{j=1}^{j_{\max}} \frac{\partial E_{ij}}{\partial V_j} dV_j \quad (44)$$

for i going from 1 to j_{\max} where j_{\max} is 6, 7, or 9 depending on the engine type being run; and where E is an error, V is a variable, and ∂E_{ij} is the change in E_i caused by a change in V_j .

The assumption of a small change in the variables results in the following approximations (B refers to a base value):

$$dE = E - EB \quad (45)$$

$$dV = V - VB \quad (46)$$

$$\frac{\partial E}{\partial V} = \frac{\Delta E}{\Delta V} \quad (47)$$

With these approximations and the knowledge that E should equal zero for the balanced engine, the set of partial differential equations (eq. (44)) reduces to

$$E_i - EB_i = \sum_{j=1}^{j_{\max}} \frac{\Delta E_{ij}}{\Delta V_j} dV_j = -EB_i \quad (48)$$

for i going from 1 to j_{\max} .

Thus the calculations made with the perturbed variables are used to compute $\Delta E/\Delta V$, and equation (48) is solved for dV_j . The variables V are then given new values from

$$V_j = V_jB + dV_j \quad (49)$$

If the engine cycle calculations were linear functions, the engine would balance (errors within some allowable limit) with the new values of the variables. However, this is not the case, and it is usually necessary to repeat the process of changing each variable by a small amount for each pass. A change in each error because of the small change in the variable is calculated for each pass, where the new values become the base values. This process occurs several times before a balance is obtained.

A subroutine (MATRIX) to determine the solution of a matrix is used to solve the set of differential equations. After each pass through the engine, a matrix array is loaded with the appropriate values; after a number of passes equal to 1 plus the number of independent variables (base value plus up to nine independent variables), the matrix subroutine is used to solve the matrix. The solution of the matrix (E within some allowable limit) yields the correct values of the independent variables and satisfies all the component matching constraints.

The most-often-used independent variables and the differential errors for each of the nine examples of engine types capable of being run on GENENG II are listed in table I.

INPUTS FOR ENGINE PERFORMANCE CALCULATIONS

Two forms of data are supplied to GENENG II. Some data, such as all the constants and component map data, are in the form of BLOCK DATA subprograms. The varying data are supplied at execution time by the use of input data cards.

The FORTRAN listings of GENENG II are presented in appendix A. The function and description of the subroutines follow in the next section.

GENENG II Subroutine Functions and Descriptions

A flow chart of the computer program with the subroutines is shown in figure 10. The functions of the subroutines are listed here and the purpose of each is described.

GEN2	Dummy main program to initiate the calculations and cause the input of the controlled output variables. Because of the looping between subroutines, control is never transferred back to this routine.
ENGBAL	Main subroutine. Controls all engine balancing loops; checks tolerances and number of loops and loads matrix; calls INPUT.
GUESS	Determines initial values of independent variables (see table I) at each point.
MATRIX	Solves error matrix.
PUTIN	Calls input subroutine package. Controls loop on static pressures for mixed-flow turbofan.
ZERO	Zeros nearly all of common and certain controls.
COINLT	Determines ram recovery and performs inlet calculations.
ATMOS	1962 U. S. Standard Atmosphere table.
RAM	Calculates ram recovery defined by MIL-E-5008B specifications.
RAM2	Calculates special cases of input ram recovery as a function of flight Mach number.
COFAN	Uses BLOCK DATA to perform fan calculations.
COINTC	Uses BLOCK DATA to perform intermediate-compressor calculations.
INTDUM	Makes intermediate compressor not change air conditions for engines e and h.
COCOMP	Uses BLOCK DATA to perform inner-compressor calculations.
WDUCT	Performs third-stream (wing) duct calculations (not used in two-stream engines).
COCOMB	Uses BLOCK DATA to perform combustor calculations. May use either T_4 or WFB as the main parameter.
COHPTB	Uses BLOCK DATA to perform inner-turbine calculations (not used in engines c and g).
COIPTB	Uses BLOCK DATA to perform intermediate-turbine calculations (not used in engines b, e, and h).
COLPTB	Uses BLOCK DATA to perform outer-turbine calculations.

CODUCT Performs duct and duct-burning calculations for turbofans. May use either T24 or WFD as main parameters.

COMIX Performs gas-mixing calculations if in mixed-flow mode. At design points it calculates areas either from an input static pressure PS55 or from an input Mach number AM55 if PS55 = 0. At off-design points it calculates static pressures and Mach numbers from the design areas. Calculates ERR (5). Rescales pressure ratios for mixed-flow turbofans to match duct and core static pressures just prior to mixing. COMIX also calculates afterburner entrance area A6 as a function of afterburner entrance Mach number AM6.

COAFBN Performs afterburning calculations. May use either T7 or WFA as the main parameters.

FRTOSD Dummy routine to transfer values from common FRONT to common SIDE.

FASTBK Dummy routine to transfer values from common SIDE to common BACK.

COMNOZ Controls the main nozzle.

ERROR Controls all printouts if an error occurs. Prints names of subroutine where error occurred and also prints the values of all variables in the main commons.

SYG Controls printing from UNIT08. Throughout the program and particularly in ENGBAL, certain messages, variables, and matrix values are written on UNIT08 as an aid in determining why an error occurred or why a point did not balance. These values are printed out if subroutine ERROR is called and IDUMP is greater than zero, or after a good point if IDUMP = 2.

PERF Calculates performance after the engine is balanced.

OUTPUT Prints output except for controlled output. Prints the main commons after the design point.

CONOUT Controls and prints the controlled output variables.

THCOMP Performs isentropic calculations for compressors.

PROCOM Calculates thermodynamic gas properties for either air or a fuel-air mixture based on JP-4 using curve fits of the tables of reference 4.

SEARCH General table lookup and interpolation routine to obtain data from the BLOCK DATA subroutines.

MAPBAC Used when calculations result in values not on the turbine maps. Changes the map value and an independent variable (PCNF, PCNC, or T4) in an attempt to rectify the situation.

CONVRG	Performs nozzle calculations for a convergent nozzle.
CONDIV	Performs nozzle calculations for a convergent-divergent (C-D) nozzle.
THTURB	Performs isentropic calculations for turbines.
THERMO	Provides thermodynamic conditions using PROCOM.
AFQUIR	General quadratic interpolation routine.
PARABO	Parabolic curve-fit routine.
OVERLAY	DUMMY routine to restore working part of program to core when using overlay.
BLKFAN	Performance data for fan map (BLOCK DATA).
BLKINT	Performance data for intermediate-compressor map (BLOCK DATA).
BLKCOMP	Performance data for inner-compressor map (BLOCK DATA).
CMBDAT	BLOCK DATA for combustor.
HPTDAT	Performance data for inner-turbine map (BLOCK DATA).
IPTDAT	Performance data for intermediate-turbine map (BLOCK DATA).
LPTDAT	Performance data for outer-turbine map (BLOCK DATA).
ETAAB	Generalized afterburner performance BLOCK DATA as a function of fuel-air ratio with correction factors for off-design afterburner entrance pressure and Mach number.
INPUT	Package of Huff input subroutines. (The Huff Input Routine is a very versatile input mechanism further detailed in appendix B.)

Entering the Data

The Huff Input Routine, used to enter input data into the program at execution time, is discussed in appendix B. Appendix C presents the individual symbols for component names, station numbers, etc., from which compound names such as WAFCDs (WA + F + C + DS) are formed. Table II and appendix A present the names of the variables, the values of which are supplied on data cards.

Choice of component maps - scaling laws. - Many engines that are studied using GENENG II are theoretical. Therefore, actual component maps for these engines will be nonexistent. The program, however, does require component maps in order to do off-design-point calculations. To alleviate this problem GENENG II uses scaling laws to change data from one component map into a new component map. Hopefully, a component map can be found which could be expected to perform in a similar manner to the

actual map for the engine being studied. In fact, most maps that the authors have obtained are identified as to the range of pressure ratio, airflow, etc., over which they are valid. Thus, a high-bypass-ratio fan map such as that from a CF6 could be used to simulate other high-bypass-ratio fan maps, etc.

The scaling equations used for the compressor maps are

$$PR = \frac{PR_{\text{design}} - 1}{PR_{\text{map, design}} - 1} (PR_{\text{map}} - 1) + 1$$

$$WA = \frac{WA_{\text{design}}}{WA_{\text{map, design}}} \times WA_{\text{map}}$$

$$ETA = \frac{ETA_{\text{design}}}{ETA_{\text{map, design}}} \times ETA_{\text{map}}$$

In the output are printed the correction factors used in scaling the maps. The closer these values are to 1.0, the more reasonable are the simulated maps of the engine. Conversely, however, not being close to 1.0 does not necessarily mean that the simulation is poor since many maps have been shown to be typical over quite large ranges in the variables.

BLOCK DATA input. - The three compressor performance maps are entered into the code as the BLOCK DATA subprograms BLKFAN, BLKINT, and BLKCMP. The subprograms supplied by the authors with the code and shown in appendix A are not to be taken as realistic maps. These maps are of an illustrative nature and are the ones used to run the sample calculations.

Using subprogram BLKFAN as an example (the first nine cards of which are printed here) and referring to a typical compressor map (fig. 11), the data are programmed as follows:

```

$IBFTC BLKFAN DECK
C THIS IS A GENERALIZED FAN MAP FOR UNREALISTIC SUPERSONIC ENGINE 1
  BLOCK DATA 2
  COMMON / FAN/CN(15),PR(15,15),WAC(15,15),ETA(15,15),N,NP(15) 3
  DATA V,NP/10,6,3*7,5*10,8,5*0/ 4
  DATA CV/0.3,0.4,0.5,0.6,0.7,0.8,0.9,1.0,1.1,1.2,5*0./ 5
  DATA (PR( 1,J),WAC( 1,J),ETA( 1,J),J=1, 6)/ 6
1 1.03000, 243.600, 0.75592, 1.01200, 229.800, 0.76120, 7
2 1.02800, 199.800, 0.76648, 1.03840, 166.800, 0.75592, 8
3 1.04480, 133.200, 0.72512, 1.04800, 86.400, 0.64152/ 9

```

Card 1 reminds the reader that these maps are fictitious. Card 2 identifies program as BLOCK DATA. Card 3 identifies common block FAN into which data are to be stored and dimensions the program variable. Card 4 indicates that there are 10 speed lines

N and the number of points NP on each line (six on the lowest speed, seven on the next three lines, etc.). Card 5 assigns the value of speed to each of the 10 lines (low to high). Cards 6 to 9 along the speed line CN=0.3 sets the pressure ratio PR, corrected airflow WAC, and efficiency ETA in sets of three going from low pressure (PR = 1.0) to the surge line (PR = 1.048). Note there are six sets of three values (NP(1) = 6). The rest of the cards (appendix A) set the values for each speed line.

The combustor map is also a BLOCK DATA subprogram (CMBDT). It is a plot of temperature rise across the combustor against efficiency for constant input pressure. Entry to the map is through temperature rise and input pressure with efficiency being output. The cards in the subprogram CMBDT are reproduced here; a typical combustor map shown in figure 12.

\$IBFTC CMBDT DECK	
BLOCK DATA	1
COMMON / COMB/PSI(15),DELT(15,15),ETA(15,15),N,NP(15)	2
DATA N,NP / 15,15*15 /	3
DATA PSI/4.9116,9.8232,14.735,19.646,24.558,29.470,34.381,	4
139.293,44.207,73.674,100.,200.,300.,400.,500./	5
DATA DELT/15*200.,15*300.,15*400.,15*500.,15*600.,15*700.,15*800.,	6
115*900.,15*1000.,15*1100.,15*1200.,15*1300.,15*1400.,15*1500.,	7
215*1600./	8
DATA ETA/	9
1.600,.726,.777,.806,.826,.843,.855,.865,7*.870,	10
2.758,.825,.858,.875,.888,.898,.906,.912,.914,6*.915,	11
3.868,.893,.911,.925,.935,.942,.947,.951,7*.953,	12
4.925,.936,.946,.955,.963,.969,.974,.977,.978,6*.979,	13
5.960,.966,.972,.977,.982,.985,.990,.992,.993,6*.995,	14
6.988,.991,.992,.994,.995,.997,.998,8*.999,	15
78*1.00,7*.999,120*1.00/	16
END	17

Card 1 identifies the subprogram as BLOCK DATA. Card 2 identifies the common block COMB into which data are to be stored and dimensions each variable. Card 3 indicates that there are 15 lines of constant PSI (P3) by the value of N, and that there are 15 values of DELT (DT) and ETA (ETAB) along each line of constant PSI (P3). Cards 4 and 5 assign values to each of the P3 lines from low to high pressure. Cards 6 to 8 assign values of ΔT to each of the P3 lines, starting at low ΔT . The lowest value of ΔT on each of the P3 lines is given starting with the lowest value of ΔT on the lowest value of P3. Next comes the second lowest value of ΔT on each P3, etc. Again, this map is unrealistic, being used for illustrative purposes only. Cards 9 to 16 assign the value of η_B in a one-to-one correspondence with the ΔT values just assigned. The order is the same.

Also entered as BLOCK DATA subprograms are the turbine maps (HPTDAT, IPTDAT, and LPTDAT). To illustrate the entering of turbine data, LPTDAT will be used. A typical turbine map is shown in figure 13; the data are programmed as follows:

```

$IBFTC LPTDAT DECK
BLOCK DATA
COMMON / LTURB/TFF(15),CN(15,15),DH(15,15),ETA(15,15),N,NP(15)
DATA N,NP/11,9*15,12,9,4*0/
DATA TFF / 88.470, 102.795, 116.835, 129.330, 141.045,
1 145.725, 150.000, 153.345, 156.405, 159.780, 163.170,4*0./
DATA (CN( 1,J),DH( 1,J),ETA( 1,J),J=1,15)/
1 0.3582, 0.0018, 0.7120, 0.5336, 0.0026, 0.7300,
2 0.7365, 0.0035, 0.7472, 0.9754, 0.0044, 0.7300,
3 1.2146, 0.0051, 0.7140, 1.4173, 0.0056, 0.7000,
4 1.5201, 0.0059, 0.6850, 1.7673, 0.0061, 0.6730,
5 2.3247, 0.0062, 0.6452, 2.2827, 0.0061, 0.6200,
6 2.4665, 0.0057, 0.6000, 2.6137, 0.0053, 0.5750,
7 2.9166, 0.0044, 0.5310, 2.9456, 0.0035, 0.5000,
8 3.3138, 0.0001, 0.3850/

```

Card 1 identifies subprogram as BLOCK DATA. Card 2 identifies common block into which data are to be loaded and dimensions the program variables. Card 3 indicates the number of constant turbine flow function lines TFF as 11 (N) and the number of points on each line from low to high TFF. Cards 4 and 5 set values of TFF from low to high. Cards 6 to 14 set values of corrected speed CN, work function DH, and efficiency ETA along TFF(1) starting from low CN (0.3682) and ending at high CN (3.3138). The rest of the cards set the values along higher TFF lines.

In many cases, turbine maps for high-performance engines operate at a choked condition (constant TFF). Thus, a turbine map to be represented could possibly have no lines representing constant TFF for a significant portion of the map. For complete map representation, lines of constant TFF may be estimated on the map up to the limit loading line by inputting slight changes for the values of TFF (e. g. , one line for TFF is 62.105, the next may be input as equal to 62.108). This will eliminate computational difficulties which would arise if constant values for TFF lines were input.

Generalized afterburner performance has been programmed into subroutine COAFBN. The afterburner performance map included in the program is shown in a generalized form in figure 14(a). The performance map shows afterburner combustion efficiency as a function of fuel-air ratio. The values of the afterburner combustion efficiency correction factors $\Delta ETAA$ during off-design operation are shown against afterburner entrance Mach number (fig. 14(b)) and afterburner entrance pressure (fig. 14(c)). Other correction factors or performance maps may be added as desired. The afterburner efficiency, fuel-air ratio, inlet total pressure, and Mach number are generalized.

A specific afterburner performance map is generalized by dividing the specific off-design value by the design value as shown below. The symbols shown are the symbols used in the ABETTA subroutine where the generalized and specific values are input. The generalized afterburner values are obtained as follows:

$$\text{Efficiency (ETABRT)} = \frac{\text{ETAA}}{\text{ETAADS}}$$

$$\text{Fuel-air ratio (FART)} = \frac{\text{FART}}{\text{FARTDS}}$$

$$\text{Entrance total pressure (P6T)} = \frac{\text{P6}}{\text{P6DS}}$$

$$\text{Entrance Mach number (EM6T)} = \frac{\text{AM6}}{\text{AM6DS}}$$

However, the correction factor for efficiency ΔETAA is not a generalized value. Also input in ABETTA are

(1) The change in efficiency as a function of EM6T is input as DELM6 (which is really $\Delta\text{ETAA} = f(\text{AM6})$).

(2) The change in efficiency as a function of P6T is DELP6 (which is really $\Delta\text{ETTA} = f(\text{P6})$).

At execution time for the design point, afterburner combustion efficiency ETAADS , exit total temperature T7DS , and entrance Mach number AM6DS design values are input. Then design fuel-air ratio and entrance pressure ratio are calculated from the input values and the other design engine characteristics.

To achieve a reasonable accuracy in cycle calculations when using a generalized component map, the usage of the map should be limited within a certain range of the original design values and configuration changes. Therefore if, for example, an afterburner has a design task that differs significantly from an example used, a new performance map should be used in order to simulate the component more accurately.

GENENG II normally uses a single-point input for the nozzle velocity coefficients (CVMNOZ, CVDWNG, and CVDNOZ) when calculating engine performance. When desired, however, a map of nozzle velocity or thrust coefficients can be readily incorporated.

Inputs required at execution time. - Basically what must be supplied are a list of the desired output variables, design values of any component existing in the engine (compressors, combustion, turbines, etc.), and engine operational controls. The variables that are to be output are selected by the first section of data cards. Any variable that is in one of the main commons (DESIGN, FRONT, SIDE, BACK, SPOOL2, or DUMMYS) may be selected for output by punching, in columns 1 to 6, the name of the variable as it appears in the common. Up to 150 variables (25 lines of six variables) may be chosen for a particular run. During the output phase, the name of the variable is printed out, with its value printed immediately below the name.

Another feature of the controlled output is the ability to change the name of a variable to be output; for example, it may be desired to change a station designation to one more common to a particular programmer. In this case, the variable name would be

punched in columns 1 to 6 as described; but, in addition, the desired name would be punched in columns 13 to 18. Special symbols such as / may be used in the new name. The last card of the controlled output must be a card with "THEEND" punched in columns 1 to 6.

Table II summarizes the design inputs for the nine basic engines shown in this report.

The following control variables should always be supplied at the design point. The value used is independent on how the user wants the engine to operate. The symbols and their purpose are listed in the subroutine PUTIN but are shown here for the reader's convenience. The superscripts (1) to (4) on the symbols have the following meanings: (1) means "automatically returned to zero after each point is calculated, must be re-input if option is again desired"; (2) means "option can be used for design or off-design," whereas the other two MODE's can be used only at off-design; (3) means "these input values remain as input unless changed by a new input value"; (4) means "a setup case must be run where all the components are first matched before these $\neq 0$ options are used; then the identical case may be repeated exercising these options."

IDES = 1 ⁽¹⁾	For calculating the design point.
(2)MODE = 0 ⁽³⁾	Specify T4.
MODE = 1 ⁽³⁾	Specify PCNC.
(2)MODE = 2 ⁽³⁾	Specify WFB.
MODE = 3 ⁽³⁾	Specify PCNF.
INIT = 0 ⁽³⁾	Initializes point.
INIT = 1 ⁽³⁾	Will not initialize point.
IDUMP = 0 ⁽³⁾	No looping write-outs.
IDUMP = 1 ⁽³⁾	Will dump looping write-outs if error occurs.
IDUMP = 2 ⁽³⁾	Will dump looping write-outs after every point.
IAMTP = 0 ⁽³⁾	Will use AM and military-specification ETAR.
IAMTP = 1 ⁽³⁾	Will use input AM and input ETAR.
IAMTP = 2 ⁽³⁾	Will use input T2 as $T1 = T1 + T2$ and standard P1. (T2 value needs to be input at every point or an error will occur whenever used.)

IAMTP = 3⁽³⁾ Will use input P2 and standard T1.
 IAMTP = 4⁽³⁾ Will use input T2 and input P2.
 IAMTP = 5⁽³⁾ Will use specific schedule of ETAR located in subroutine RAMTWO.
 IGASMX = -1⁽³⁾ Separate flow, input A6.
 IGASMX = 0⁽³⁾ Separate flow, A6 = A55. (This control was used on all basic cycles in this report.)
 IGASMX = 1⁽³⁾ Will mix fan duct and main streams, A6 = A25 + A55.
 IGASMX = 2⁽³⁾ Will mix fan duct and main streams, input A6.
 IDBURN = 1⁽¹⁾ For duct burning (fan stream only), input T24.⁽⁴⁾
 IDBURN = 2⁽¹⁾ For duct burning (fan stream only), input WFD.⁽⁴⁾
 IAFTBN = 1⁽¹⁾ For afterburning (main stream or mixed stream of fan and main stream), input T7.⁽⁴⁾
 IAFTBN = 2⁽¹⁾ For afterburning (main stream or mixed stream of fan and main stream), input WFA.⁽⁴⁾
 IDCD = 1⁽³⁾ Fan duct nozzle will be convergent-divergent.
 IMCD = 1⁽³⁾ Main nozzle will be convergent-divergent.
 NOZFLT = 1⁽³⁾ For floating main nozzle exit area.⁽⁴⁾
 NOZFLT = 2⁽³⁾ For floating fan duct nozzle exit area.⁽⁴⁾
 NOZFLT = 3⁽³⁾ For floating fan duct and main nozzle exit area.⁽⁴⁾
 ITRY5 = N Number of passes through engine before quitting.
 TOLALL = X Tolerance which the errors must satisfy before engine is matched.

The following are other input variables for which some value depending on the engine design should be input at the discretion of the user:

DELFG, DELFN, DELSFC Normally input as 1.0 unless a correction is desired.

A6, AM55, AM23, AM6 HPEXT, AM, ALTP, CVMNOZ, CVDNOZ, CVDWNG	See appendix C, Input Symbols.
PCBLC, PCBLDU, PCBLOB, PCBLF	Values for bleed out of cycle; decimal equivalent of percent compressor flow.
PCBLHP, PCBLIP, PCBLLP	Value of total bleed returned to turbines; fractional equivalent of flow. The sum of these variables plus PCBLDU and PCBLOB should equal 1.

Inputs required for additional options to basic cycles. - To run duct-burning (available only in fan stream duct) cases load ETAD, T24 or WFBDS, and DPDUDS. To run afterburning (mixed-flow fan or unmixed fan - available for core and fan stream if mixed, or for core stream if unmixed) cases load T7DS or WAFDS, ETAADS, and DPAFDS. Afterburner operation is the same as in reference 2 with the exception of a generalized afterburner performance map addition. For changing the generalized map to a specific map for a specific engine design, the preceding design values are needed at the design point.

Means of specifying mode of engine operation. - Shown in the section SAMPLE CALCULATIONS (pp. 26 to 53) are the methods of specifying off-design operation points. The most common one and the one used exclusively herein is to select a Mach number, an altitude, and a turbine inlet temperature other than design values. There are, however, several other possibilities which the user may employ. For example, changing the following controls:

- MODE = 0 Specify a new turbine inlet temperature T4.
- MODE = 1 Specify a compressor rotational speed PCNC.
- MODE = 2 Specify a fuel flow rate WFB.
- MODE = 3 Specify a fan rotational speed PCNF.

If the engine has all its nozzles fixed, then an input such as turbine inlet temperature, fuel flow, or speed will set the thrust level. But other means of changing engine operation can be accomplished by varying such nozzle throat areas as

- A8 Main nozzle throat area
- A38 Wing nozzle throat area
- A28 Fan nozzle throat area

For example, an off-design condition may exist where, in an attempt to satisfy continuity of mass flow (one of the component matching requirements), the fan operating point may lie outside the limits of the data that were input for the component map. A

fan nozzle throat area change could be used to return the fan operating condition on the map such that a match would occur. This would indicate a possibility exists that a variable fan nozzle would be required on this engine for operation at the desired condition. The area is changed by inputting (example; $A28 = 1.2 * A28$). Since the design areas are not known prior to running the design point, the Huff Input Routine provides the versatility in which A28 is increased by a factor of 1.2 as was shown. It should be noted that any area remains changed until it is recalculated by a new design case or altered by a new input. The preceding example and statements would apply if changes were made instead to A8 or A38.

If the engine uses thrust augmentation by either duct burning or afterburning, the affected throat area (A28 or A8) will adjust to account for the temperature increase during afterburning. The area adjustment will be such that sonic conditions will exist at the affected throat. The component cycle match point is not affected. The area will then revert to the original design of modified A28 or A8 when the next case is run without afterburning.

The nozzle exit area (A9 or A29) may be fully expanded (if A28 or A8 are sonic) by using the control variables NOZFLT = 1, 2, or 3 for the nonafterburning cases, or it is done automatically for the afterburning cases when the control variables of IAFITBN or IDBURN are used. The significance of these values was explained in the previous section.

Use of Overlay - The Huff Input Routine Subroutine Package

The use of the Huff Input Routine (appendix B) may, as in the case of The Lewis Research Center's 7094, require the use of overlay. Because of the flow of the program, the rules of overlay are violated in that the system will detect that it is possible to call a link which will override the link that called it. The link that is called is the Input Subroutine Package. The authors, however, have provided that as soon as the input is read, the original link is restored so that the program will execute correctly. The computer system might have to be told that the rules of overlay are being violated but to proceed anyway.

In addition, the use of overlay increases the execution time of the program. The sample cases being run herein take about 4 seconds per case without overlay and 8 seconds with overlay.

If the user feels that the aforementioned difficulties are not worth the flexibility of the Huff Input Routine, he may restore the program to the original form, in which NAMELIST was used. The authors will supply directions for this if so desired.

The suggested links when using overlay are as follows:

\$IBLDR GEN2	{	LINK 0
\$IBLDR ENGBOL		
\$IBLDR GUESSS		
\$IBLDR MATRIX		
\$IBLDR PUTIN		
\$IBLDR ZERO		
\$IBLDR COINLT		
\$IBLDR ATMOS		
\$IBLDR RAMS		
\$IBLDR RAMTWO		
\$IBLDR COFAN		
\$IBLDR COINTC		
\$IBLDR INTDUM		
\$IBLDR COCOMP		
\$IBLDR WDUCT		
\$IBLDR COCOMB		
\$IBLDR COHPTB		
\$IBLDR COIPTB		
\$IBLDR COLPTB		
\$IBLDR CODUCT		
\$IBLDR PUTOUT		
\$IBLDR CONOUT		
\$IBLDR THCOMP		
\$IBLDR PROCDM		
\$IBLDR SURCH		
\$IBLDR MAPBAK		
\$IBLDR CONVRG		
\$IBLDR CONDIV		
\$IBLDR THTRB		
\$IBLDR THERMO		
\$IBLDR AFQUER		
\$IBLDR PARABO		
\$IBLDR BLKFAN		
\$IBLDR BLKINT		
\$IBLDR BLCCMP		
\$IBLDR CMBDT		
\$IBLDR HPTDAT		
\$IBLDR IPTDAT		
\$IBLDR LPTDAT		
\$IBLDR ABETTA		
\$IBLDR BLOCK		
\$IBLDR ICHAR4		

\$ORIGIN	HUFF	{	LINK 1
\$IBLDR OVELAY			
\$IBLDR COMIX			
\$IBLDR COAFBN			
\$IBLDR FRTOSD			
\$IBLDR FASTBC			
\$IBLDR COMNOZ			
\$IBLDR ERROR			
\$IBLDR SYGS			
\$IBLDR PERFOR			
\$ORIGIN	HUFF	{	LINK 2
\$IBLDR IARITI			
\$IBLDR ICNVTI			
\$IBLDR IERORI			
\$IBLDR ILOOKI			
\$IBLDR INAMEI			
\$IBLDR INAMEN			
\$IBLDR INMBRI			
\$IBLDR INPUT			
\$IBLDR ITABLI			
\$IBLDR ISUBI			
\$IBLDR IXQTI			
\$IBLDR LOCKX			
\$IBLDR DEBUGX			
\$IBLDR STACKP			
\$IBLDR IFLO			

SAMPLE CALCULATIONS

Examples of nine basic engines capable of being run without changes to the computer code are shown in this section.

Type a - Three-Spool, Three-Stream Turbofan

The first example engine is engine a, the three-spool, three-stream turbofan engine. The first page output from the program is reproduced at the end of this main section. What is shown are the card images of the input cards, which is one of the advantages of using the Huff Input Routine (HIR); they were printed under the control of that routine.

The card numbers have been added by the authors and appear to the right on the output sheets. Card 1 (\$D(1)) is the identification card for the forthcoming block of data. The (1) agrees with the third argument of the CALL INPUT (5, 6, 1, WORD, ITABLE) statements in subroutine PUTIN. The #1 (punched as :1) causes the carriage control to start a new page. Card 2 starts the TABLE relating the location of each of the named variables to the location of the variable "WORD" in common block ALL. These names have been mentioned in the section Inputs required at execution time and are listed alphabetically in appendix C. The first group of variables have been tagged with the label INTEGER, denoting their mode. Cards 3 to 14 complete the table with the REAL and LOGICAL variables. Cards 15 to 17 have been found to be useful in that they provide an easy method for designing an engine for other than a standard day. Since GENENG II uses corrected airflows as design-point inputs, the actual airflow into the fan must be multiplied by $\sqrt{\theta}$ (= 1 for standard day). The $\sqrt{\theta}$ for other than a standard day has been predetermined for the user (TROPICAL DAY = $\sqrt{(518.69 + 31)/518.69}$, etc.) and using the HIR, these factors can be used to multiply the desired airflows.

Cards 18 to 20 set the design point as a "normal" engine at sea-level-static conditions. Card 21, which has been made to stand out by the use of #0, emphasizes the logical control variables which set the engine being run, in this case a three-spool, three-stream turbofan (engine a, fig. 1). Cards 22 to 24 describe the engine to be run and illustrate the use of the HIR to identify what is being done.

In table II are listed the required inputs to run each of the sample engines. Card 25 sets the fan design parameters and illustrates both tropical day and the HIR performing mathematical operations. In addition, the pressure coming out of the second spool was picked (2.2) and thus the pressure ratio of the intermediate compressor determined from this value and the pressure already achieved by the fan (1.4).

Cards 26 to 28 remind the user that the program requires corrected airflows and that the HIR is used to correct the airflow going into the intermediate compressor on card 29. Cards 30 to 36 complete the input data for the design case. The input values of turbine design data correspond to the design point on the turbine maps. The compressor data are the actual values for the components, and the design points on the compressor maps are specified by the percent corrected speeds (PCNFDS, PCNIDS, and PCNCDS) and the values of Z (ZFDS, ZIDS, and ZCDS). Card 34 sets the design day as being a tropical day (recall cards 25 and 29). Execution commences since the next card is a \$D(1), indicating a new block of data. (This card is on page 34.)

The first group of output shows the scaling factors being used on the component maps. For the three compressors these are defined by

$$PR_CF = 1 + \frac{PR_DS - 1}{PR^* - 1} \quad (50)$$

$$ETA_CF = \frac{ETA_DS}{ETA^*} \quad (51)$$

$$WA_CF = \frac{WA_CDS}{WAC^*} \quad (52)$$

where the * values are those defined on the component maps by PCN_DS and Z_DS. Recall that the closer these and the turbine correction factors are to 1.0, the better the performance map data which were input to the program simulates the engine being studied. Even if these values are not close to 1.0, the simulation may still be quite accurate if the maps used were identified as being applicable over a range of design-point specification within which the user's design point falls.

The Mach number, the area of the throat, and the exit area of the wing duct are printed next, followed by the combustor design corrected airflow and combustor correction factors. For the combustor

$$ETABCF = \frac{ETABDS}{1.0} \quad (53)$$

and DTCCOF always equals 1 since the combustor value of DTCODS is calculated internally at the design point and the map is then scaled.

Following the combustor correction factors are the correction factors for each of the turbines, where

$$CN_PCF = CN_PDS \frac{\sqrt{T^*}}{PCN_**} \quad (54)$$

$$TF_PCF = TF_PDS \frac{14.696 \times P^*}{WG^* \times \sqrt{T^*}} \quad (55)$$

$$ET_PCF = \frac{ET_PDS}{ETAT_P} \quad (56)$$

$$DH_PCF = \frac{DHT_C}{DHTC_P} \quad (57)$$

where the $*$ values are at the inlet to the turbine, PCN_** is the percent corrected speed of the compressor being driven by the turbine, DHT_C is the actual work function of the turbine, and $ETAT_P$ and $DHTC_P$ are the values of efficiency and work function, respectively, at the design point chosen on the turbine maps. These correction factors are followed by the areas for the duct nozzle, turbine exit, and core nozzle.

The aforementioned output occurs only at the design point. This is followed by output which was determined by a list of variable names submitted at execution time (see section Input required at execution time). The engine type is spelled out prior to the listing of the variable values. The list of variables printed here with their definitions are found in appendix C.

The type of nozzle is printed and then the number of loops required for convergence (always 1 for design points). Next is printed a line labeled **COMMON** and on this line reading from left to right are the values of **ZF**, **PCNF**, **ZI**, **PCNI**, **ZC**, **PCNC**, **T4**, and **MODE**. The variables in the labeled common **BLOCKS** are printed next according to the following code:

```
COMMON /DESIGN/
1PCNFGU,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC,
2ZFDS ,PCNFDS,PRFDS ,ETA FDS,WAFDS ,PRFCF ,ETAFCF,WAFCF ,
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,
4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCOCF,ETABCF,
5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLP CF,ETLPCF,DHLP CF,T21DS ,
7T24DS ,WFDDS ,DTDUDS,ETADDS,WA23DS,DPDUDS,DTDUCF,ETADCF,
8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF,
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
$PS55 ,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV
```

```

COMMON / FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLOB ,PCBLHP ,PCBLLP
COMMON / SIDE/
1XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 ,
2XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 ,
3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 ,
4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 ,
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS ,DUMS3 ,
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
COMMON / BACK/
1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,
2XWFB ,XWG55 ,XFAR55 ,XWFD ,XWG24 ,XFAR24 ,XXP1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC
COMMON /SPOOL2/
1T22 ,P22 ,H22 ,S22 ,T50 ,P50 ,H50 ,S50 ,
2WA22 ,ZI ,PCNI ,CNI ,PRI ,ETAI ,WACI ,TFFIP ,
3CNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,PCBLIP ,PCNIGU ,ZIDS ,
4PCNIDS ,PRIDS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,TFIPDS ,
5CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETAPCF ,DHIPCF ,WAICDS ,WAI ,
6PCBLI ,BLI ,T22DS ,WA21
COMMON /DUMMYS/ XYZ(21) (not printed)
IWA32 ,DPWGDS ,DPWING ,WA32DS ,A38 ,AM38 ,V38 ,T38 ,
2H38 ,P38 ,TS38 ,PS38 ,T39 ,H39 ,P39 ,TS39 ,
3V39 ,AM39 ,A39 ,BPRINT ,WG37 ,CVDWNG ,FGMWNG ,FGPWNG ,
4FNWING ,FNMAIN ,FWOVFN ,PS39 ,FXFN2M ,FXM2CP ,TRUE ,FALSE ,
5FFOVFN ,FCOVFN ,FMNOFN ,FNOVFD ,AFTFAN ,DUMSPL ,FDES ,PCBLID
6TFFHP ,TFFIP ,TFFLP ,CNHP ,CNIP ,CNLP ,DHTCHP ,DHTC

```

This completes the printout of the design-point engine. Next the engine is run at an altitude of 25 000 feet, a Mach number of 0.6, a turbine inlet temperature of 2260° R (recall MODE = 0), and standard-day conditions (IAMTP = 0).

What follows is the off-design performance for the three-spool engine. Note that the common blocks are not printed for the off-design-point case. The running of the other eight engines is described in the following subsections and appears at the end of this main section.

Type b - Two-Spool, Three-Stream Boosted-Fan Turbofan

Two \$D(1), IDES = 1 will cause the deck to run a new design-point engine. The first of these two cards recalls all the design data of the previous design case. Any variable name (except speed when attaching spools) with a DS on the end (e.g., PRFDS) remains unchanged unless overridden with a new value in input. These changed values must follow the second \$D(1), IDES = 1 card. Therefore, referring to table II, to run engine b (the boosted fan; fig. 2) all that is necessary is to reset the design Mach number and altitude to zero, set up to run on a tropical day ($T_2 = 31$, $IAMTP = 2$), and set $FIXFANTOMIDDLE = .T$. Note that in the output there are no correction factors for the intermediate-pressure turbine since the entire boosted fan is driven by the low-pressure turbine, and that PCNI and CNI are not 100 since the physical speed of the middle spool is set by that of the fan. Note also that TFFIP, CNIP, and DHTCIP are zero since that turbine is not used. The user was reminded of this by the line reading "FAN AND MIDDLE SPOOL ARE ATTACHED, USE INNER AND OUTER TURBINES."

To run the off-design case for all engines proceed as in engine a.

Type c - Two-Spool, Three-Stream, Supercharged-Compressor Turbofan

To run engine c (fig. 3), it is necessary to detach the fan and middle spool ($FIXFANTOMIDDLE = .F.$); to set $FIXMIDDLETOCOMP = .T.$; to reset Mach number, altitude, and tropical day; and to reset PCNIDS to design value (100). Note that the high-pressure turbine has been deleted.

Type d - Three-Spool, Two-Stream Turbofan

To run engine d (fig. 4), it is necessary to detach the middle and inner spools ($FIXMIDDLETOCOMP = .F.$); to reset Mach number, altitude, and tropical day; to reset PCNCDS to design value (100); and to set PCBLIDS, the fraction of air exiting the middle spool and going into the wing, equal to zero. Note that the intermediate duct is deleted and there is no wing thrust.

Type e - Two-Spool, Two-Stream Turbofan

To run engine e (fig. 5), it is necessary to reset Mach number, altitude, and tropical day; to set DUMMYSPOOL = .T.; and to change inner-compressor pressure

ratio to 16/1.4 so that overall pressure ratio is the same. Note that CNI = 0; PRI = 1.0; WACI = WACC; and there is no wing thrust.

Type f - Three-Spool, Three-Stream Aft-Fan Turbofan

To run engine f (fig. 6), it is necessary to set DUMMYSPOOL = .F.; to set AFTAN = .T.; to set WAICDS = 400* TROPICALDAY (middle spool at free-stream conditions; tropical day); to reset PCBLIDS = 0.5; to reset Mach number, altitude, and tropical day; to change airflows to fit new configuration; and to lower inner-compressor pressure ratio to 16/2.2 (since middle compressor is now operating). Note that WAI > WAF since WAF does not enter the middle spool.

Type g - Two-Spool, Three-Stream Aft-Fan Turbofan

To run engine g (fig. 7), it is necessary to set FIXMIDDLETOCOMP = .T.; and to reset Mach number, altitude, and tropical day. Note that AFTFAN continued to be true.

Type h - Two-Spool, Two-Stream Aft-Fan Turbofan

To run engine h (fig. 8), it is necessary to detach the middle and inner spools (FIXMIDDLETOCOMP = .F.); to set DUMMYSPOOL = .T.; to set PCBLIDS = 0 (no airflow into wing); to reset PCNIDS to 100 and change PRCDS to maintain overall pressure ratio of 16; and to reset Mach number, altitude, and tropical day.

Type i - Two-Stream Aft-Fan Turbofan

To run engine i (fig. 9), it is necessary to set DUMMYSPOOL = .F.; to reset Mach number, altitude, and tropical day; to lower the inner-compressor pressure ratio; and to reset PRIDS. The last case is followed by a \$END card to terminate the reading of data.


```

$D(1),#1 THIS CAUSES THIS CARD TO BE PRINTED AT TOP OF PAGE
$TABLE (.INTEGER.,2=IDES,5=MODE,7=IDUMP,8=IAMTP,9=IGASMX,10=IDBURN,11=IAFTBN,
12=IDCD,13=IMCD,16=NOZFLT,17=ITRYS,.LOGICAL.,358=FIXFANTOMIDDLE,
359=FIXMIDDLETOCOMP,366=AFTFAN,367=DUMMYSPOOL,.REAL.22=TOLALL,34=DELFG,35=DELFN,
36=DELSFC,38=PCNFDS,39=PRFDS,40=ETAFDS,46=PCNCDS,47=PRCDS,48=ETACDS,53=T4DS,
54=WFBDS,56=ETABDS,58=DPCDS,63=ETHPDS,71=ETLPDS,82=DPDUS,85=T7DS,88=ETAADS,
90=DPAFDS,95=A6,97=A8,99=A28,101=PS55,102=AM55,103=CVDNOZ,104=CVMNOZ,113=T2,
114=P2,125=T4,144=WAFCD,152=WACCD,172=HPEXT,173=AM,174=ALTP,175=ETAR,177=PCNF,
179=PCNC,180=WFB,183=PCBLF,184=PCBLC,185=PCBLDU,186=PCBLOB,187=PCBLHP,
188=PCBLIP,205=T24,225=ETAD,257=T7,270=WFA,273=ETAA,279=AM6,313=AM23,331=DPWGDS,
334=A38,400=HOTDAY,401=TROPICALDAY,402=COLDAY,403=POLARDAY,433=PCNIDS,
430=PCBLIP,37=ZFDS,45=ZCDS,432=ZIOS,369=PCBLIDS,351=CVDWNG,
61=TFHPDS,62=CNHPDS,440=TFIPDS,441=CNIPDS,69=TFLPDS,70=CNLPDS,
434=PRIDS,435=ETAIDS,442=ETIPDS,447=WAICDS,449=PCBLI) * END OF TABLE
* THE FOLLOWING CONSTANTS ARE HELPFULL IN DESIGNING FOR OTHER THAN STANDARD DAY
* HOTDAY(T2=44),TROPICALDAY(T2=31),COLDAY(T2=-19),POLARDAY(T2=-75)
HOTDAY=1.041553,TROPICALDAY=1.029451,COLDAY=.9815130,POLARDAY=.9248777,
* CONTROL CARDS FOR NORMAL CASE - DESIGN FOR SEALEVEL
AM=0,ALTP=0,IAFTBN=0,IDBURN=0,IAMTP=0,AM6=.24,AM23=.18,AM55=.238,
NOZFLT=0,IDUMP=1,IDCD=0,IMCD=0,IGASMX=0,MODE=0,TOLALL=.010,ITRYS=400,IDES=1,

FIXFANTOMIDDLE=.F.,FIXMIDDLETOCOMP=.F.,AFTFAN=.F.,DUMMYSPOOL=.F. *0

*0 RUN STANDARD 3 SPOOL-3 STREAM TURBOFAN ENGINE AND DESIGN IT FOR A TROP. DAY
* ACTUAL AIRFLOWS DESIRED ARE 600 LB/SEC INTO FAN, 400 LB/SEC INTO INTERMEDIATE
* COMPRESSOR, AND 200 LB/SEC INTO CORE. NOTE - PROGRAM USES CORRECTED AIRFLOWS
PRFDS=1.4,ETAFDS=.88,WAFCD=600*TROPICALDAY,PCNFDS=100,PCBLF=0,PRIDS=2.2/1.4,

*0 AT SLS,SQRT(THETA)/DELTA=SQRT(1.+ (PRFDS**(.4/1.4)-1.1)/ETAFDS)/PRFDS
* SINCE THE PROGRAM REQUIRES CORRECTED AIRFLOWS, THE HIR CAN BE USED TO DO THIS
* THE HIR USES THE FUNCTION PWR(X,Y) TO CALCULATE X**Y
WAICDS=400*TROPICALDAY*SQRT(1.+(PWR(PRFDS,(.4/1.4))-1.1)/ETAFDS)/PRFDS
ETAIDS=.87,PCNIDS=100,PCBLIDS=.5,ETABDS=.983, * 50 PERCENT OF THE AIR THAT IS
* COMING OUT OF INTERMEDIATE COMPRESSOR GOES TO WING,REMAINDER TO CORE
DPCDS=.05,DPWGDS=.10,PRCDS=16/2.2,PCNCDS=100,ETACDS=.86,PCBLC=0,DPDUS=.05,
ETHPDS=.9,ETIPDS=.9,ETLPDS=.9,DELSFC=1,DELFN=1,DELFG=1,PCBLOB=0,T4DS=2560,
IAMTP=2,T2=31, * DESIGN FOR TROP. DAY, INPUT T2 (T1=T1+T2 INPUT), STANDARD P1
TFHPDS=50,CNHPDS=2.0,TFIPDS=120,CNIPDS=2.2,TFLPDS=130,CNLPDS=2.3,
ZFDS=.83333333,ZIOS=.83333333,ZCDS=.81433225,CVDNOZ=.985,CVMNOZ=.985,CVDWNG=.985

FAN DESIGN PRFCF= 0.10000000E+01 ETAFCF= 0.10000000E+01 WAFCF= 0.10294510E+01 T2DS= 0.54966819E+03
MIDDLE SPOOL DESIGN PRICF= 0.95238095E+00 ETAICF= 0.98863635E+00 WAICF= 0.10351161E+01 T22DS= 0.61268656E+03
COMPRESSOR DESIGN PRCCF= 0.89610390E+00 ETACCF= 0.10000000E+01 WACCF= 0.10632227E+01 T21DS= 0.70943958E+03
INTER DUCT DESIGN A38= 0.23859579E+01 AM38= 0.10000000E+01 A39= 0.23859579E+01 AM39= 0.10000000E+01
COMBUSTOR DESIGN WA3CDS= 0.30802648E+02 ETABCF= 0.98300000E+00 DTCOCF= 0.10000000E+01
H.P. TURBINE DESIGN CNHPCF= 0.10119289E+01 TFHPCF= 0.10815634E+01 ETHPCF= 0.10000000E+01 DHMPCF= 0.95721405E+00
I.P. TURBINE DESIGN CNIPCF= 0.99843589E+00 TFIPCF= 0.10150019E+01 ETIPCF= 0.10000000E+01 DHIPCF= 0.10077022E+01
L.P. TURBINE DESIGN CNLPCF= 0.10026477E+01 TFLPCF= 0.78657749E+00 ETLPCF= 0.10201339E+01 DHLPCF= 0.13610839E+01
DUCT NOZZLE DESIGN A28= 0.37508934E+01 AM28= 0.65160510E+00 A29= 0.37508934E+01 AM29= 0.65160510E+00
TURBINE AREA DESIGN A55= 0.79371517E+01 AM55= 0.23826588E+00

NOZZLE DESIGN AB= 0.31020577E+01 AM8= 0.10000000E+01 A9= 0.31020577E+01 AM9= 0.10000000E+01
OUTPUT AM= 0. ALTP= 0. T4= 2560.00 ETAR= 1.0000
THREE SPOOL ENGINE
PCNF 0.1000000E+03 CNF 0.1000000E+01 ZF 0.833333E+00 PRF 0.140000E+01 WAF 0.617671E+03 WAF 0.600000E+03
PCNI 0.100000E+03 CNI 0.100000E+01 ZI 0.833333E+00 PRI 0.157143E+01 WAI 0.310535E+03 WAI 0.400004E+03
PCNC 0.100000E+03 CNC 0.100000E+01 ZC 0.814332E+00 PRC 0.727273E+01 WAC 0.106322E+03 WAC 0.200002E+03
T2 0.549668E+03 P2 0.100000E+01 T2 0.612687E+03 P2 0.140000E+01 T2 0.709440E+03 P2 0.220000E+01
T3 0.131143E+04 P3 0.160000E+02 T3 0.0. P3 0.0. T3 0.0. P3 0.0.
PCBLHP 0.0. BLHP 0.0. PCBLIP 0.0. BLIP 0.0. PCBLLP 0.0. BLLP 0.0.
WA3 0.200002E+03 WFB 0.409714E+01 WG4 0.204099E+03 FAR4 0.204855E-01 T4 0.256000E+04 P4 0.152000E+02
TFHP 0.462294E+02 CNHP 0.197642E+01 DHTCHP 0.600000E-01 DHTC 0.147028E+03 T50 0.205966E+04 P50 0.53312CE+01
TFIP 0.118226E+03 CNIP 0.220345E+01 DHTCIP 0.456615E+02 DHTI 0.190038E+04 P5 0.366320E+01
TFFLP 0.165273E+03 CNLP 0.229393E+01 DHTCLP 0.444244E+02 DHTF 0.174312E+04 T55 0.246301E+01 P55 0.133000E+01
ETAB 0.983000E+00 PCBLOU 0.0. ETAD 0.500000E-01 DPDU 0.612687E+03 T24 0.133000E+01 P24 0.133000E+01
WAD 0.199997E+03 WFD 0.199997E+03 WG24 0.0. FAR24 0.612687E+03 T25 0.133000E+01 P25 0.133000E+01
ETAF 0.0. ETAI 0.0. ETAC 0.0. ETATHP 0.0. ETATIP 0.0. ETATLP 0.0.

```

0.880000E+00	0.870000E+00	0.860000E+00	0.900000E+00	0.900000E+00	0.900000E+00
T6	P6	PS6	AM6	V6	WG6
0.174312E+04	0.246301E+01	0.237252E+01	0.238266E+00	0.472365E+03	0.204099E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.174312E+04	0.	0.204099E+03	0.204855E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9
0.133990E+01	0.100000E+01	0.185277E+04	0.133990E+01	0.100000E+01	0.185277E+04
PS28	AM28	V28	PS29	AM29	V29
0.100000E+01	0.651605E+00	0.759245E+03	0.100000E+01	0.651605E+00	0.759245E+03
BPRINT	DPCOM	DPWING	PS38	AM38	V38
0.100000E+01	0.500000E-01	0.100000E+00	0.104874E+01	0.100000E+01	0.119237E+04
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.499987E+00	0.	0.409714E+01	0.604097E+03	0.	0.
PCBLI	WG37	VJW	PS39	AM39	V39
0.500000E+00	0.200002E+03	0.117448E+04	0.104874E+01	0.100000E+01	0.119237E+04
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.730087E+04	0.246082E+03	0.754695E+04	0.184570E+05	0.133000E+01
FFOVFN	FWDVFN	FCOVFN	FMNOFN	FNOVFD	P38
0.178770E+00	0.290223E+00	0.531006E+00	0.709777E+00	0.100000E+01	0.198000E+01
CVMNOZ	VJM	CVDOZ	VJD	FGM	FGP
0.985000E+00	0.182498E+04	0.985000E+00	0.747856E+03	0.235265E+05	0.247741E+04

MAIN SONIC CONVERGENT NOZZLE
DUCT SUBSONIC CONVERG. NOZZLE

FG= 26003.95

FN= 26003.95

SFC= 0.56721

CONVERGED AFTER 1 LOOPS

COMMON	0.833333E+00	0.100000E+03	0.833333E+00	0.100000E+03	0.814332E+00	0.100000E+03	0.256000E+04	0
0.100000E+03	0.100000E+03	0.256000E+04	0.100000E+03	0.	0.100000E+01	0.100000E+01	0.100000E+01	
0.833333E+00	0.100000E+03	0.140000E+01	0.880000E+00	0.600000E+03	0.100000E+01	0.100000E+01	0.102945E+01	
0.814332E+00	0.100000E+03	0.727273E+01	0.860000E+00	0.200002E+03	0.896104E+00	0.100000E+01	0.106322E+01	
0.256000E+04	0.	0.124857E+04	0.983000E+00	0.308026E+02	0.500000E-01	0.100000E+01	0.983000E+00	
0.500000E+02	0.200000E+01	0.900000E+00	0.108156E+01	0.101193E+01	0.100000E+01	0.957214E+00	0.549668E+03	
0.130000E+03	0.230000E+01	0.900000E+00	0.786577E+00	0.100265E+01	0.102013E+01	0.136108E+01	0.709440E+03	
0.	0.	0.	0.	0.353601E+04	0.500000E-01	0.	0.	
0.	0.	0.	0.	0.345970E+04	0.	0.	0.	
0.793715E+01	0.	0.793715E+01	0.793715E+01	0.310206E+01	0.310206E+01	0.375089E+01	0.375089E+01	
0.237252E+01	0.238266E+00	0.985000E+00	0.985000E+00	0.	0.	0.	0.	
0.549670E+03	0.100000E+01	0.131340E+03	0.160493E+01	0.549668E+03	0.100000E+01	0.131340E+03	0.160493E+01	
0.709440E+03	0.220000E+01	0.169750E+03	0.161219E+01	0.131143E+04	0.160000E+02	0.319790E+03	0.162873E+01	
0.256000E+04	0.152000E+02	0.681178E+03	0.182692E+01	0.190038E+04	0.366320E+01	0.488488E+03	0.183764E+01	
0.174312E+04	0.246301E+01	0.444064E+03	0.184049E+01	0.	0.	0.	0.	
0.100000E+01	0.140000E+01	0.880000E+00	0.617671E+03	0.600000E+03	0.200002E+03	0.204099E+03	0.204855E-01	
0.100000E+01	0.727273E+01	0.860000E+00	0.106322E+03	0.200002E+03	0.983000E+00	0.500000E-01	0.	
0.200000E+01	0.900000E+00	0.574328E-01	0.147028E+03	0.	0.204099E+03	0.204855E-01	0.114970E+04	
0.230000E+01	0.900000E+00	0.233766E-01	0.444244E+02	0.	0.204099E+03	0.204855E-01	0.	
0.	0.	0.100000E+01	0.833333E+00	0.100000E+03	0.814332E+00	0.100000E+03	0.409714E+01	
0.500000E+02	0.130000E+03	0.	0.	0.	0.	0.	0.	
0.100000E+01	0.600000E+03	0.200002E+03	0.	0.	0.319790E+03	0.	0.	
0.709440E+03	0.220000E+01	0.169750E+03	0.161219E+01	0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	
0.199997E+03	0.	0.199997E+03	0.	0.	0.500000E-01	0.499987E+00	0.	
0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	
0.174312E+04	0.246301E+01	0.444064E+03	0.184049E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	
0.409714E+01	0.204099E+03	0.204855E-01	0.	0.199997E+03	0.	0.100000E+01	0.	
0.174312E+04	0.246301E+01	0.444064E+03	0.184049E+01	0.174312E+04	0.246301E+01	0.444066E+03	0.184049E+01	
0.174312E+04	0.246301E+01	0.444066E+03	0.184049E+01	0.174312E+04	0.246301E+01	0.444066E+03	0.184049E+01	
0.204099E+03	0.	0.204099E+03	0.204855E-01	0.	0.	0.472365E+03	0.	
0.237252E+01	0.472365E+03	0.238266E+00	0.172717E+04	0.237232E+01	0.472360E+03	0.238266E+00	0.	
0.149533E+04	0.133940E+01	0.185277E+04	0.100000E+01	0.149533E+04	0.133990E+01	0.185277E+04	0.100000E+01	
0.	0.	0.747856E+03	0.464873E+04	0.182498E+04	0.115769E+05	0.	0.223133E+04	
0.235265E+05	0.247741E+04	0.409714E+01	0.604097E+03	0.682856E-02	0.260040E+05	0.260040E+05	0.567210E+00	
0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	0.205966E+04	0.533120E+01	0.534150E+03	0.183496E+01	
0.400004E+03	0.833333E+00	0.100000E+03	0.100000E+01	0.157143E+01	0.870000E+00	0.310535E+03	0.120000E+03	
0.220000E+01	0.900000E+00	0.221694E-01	0.456615E+02	0.	0.	0.100000E+03	0.833333E+00	
0.100000E+03	0.157143E+01	0.870000E+00	0.400004E+03	0.952381E+00	0.988636E+00	0.103512E+01	0.120000E+03	
0.220000E+01	0.900000E+00	0.101500E+01	0.998436E+00	0.	0.100770E+01	0.310535E+03	0.400004E+03	
0.500000E+00	0.200002E+03	0.612687E+03	0.200002E+03					
0.200002E+03	0.100000E+00	0.100000E+00	0.242141E+04	0.238596E+01	0.100000E+01	0.119237E+04	0.709440E+03	
0.169750E+03	0.198000E+01	0.591600E+03	0.104874E+01	0.709440E+03	0.169750E+03	0.198000E+01	0.591600E+03	
0.119237E+04	0.100000E+01	0.238596E+01	0.100000E+01	0.200002E+03	0.985000E+00	0.730087E+04	0.246082E+03	
0.754695E+04	0.184570E+05	0.290223E+00	0.104874E+01	0.	0.	0.	0.	
0.178770E+00	0.531006E+00	0.709777E+00	0.100000E+01	0.	0.	0.260040E+05	0.500000E+00	
0.462294E+02	0.118226E+03	0.165273E+03	0.197642E+01	0.220345E+01	0.229393E+01	0.600000E-01	0.147028E+03	

\$D(1),AM=.6,ALTP=25000,IAMTP=0,T4=2260, * RUN AT ALTITUDE AT REDUCED T4-STD DAY

OUTPUT AM= 0.600 ALTP= 25000. T4= 2260.00 ETAR= 1.0000

THREE SPOOL ENGINE

PCNF	CNF	ZF	PRF	WAF	WAF
0.942240E+02	0.102935E+01	0.786343E+00	0.140145E+01	0.651681E+03	0.327391E+03
PCNI	CNI	ZI	PRI	WACI	WAI
0.933258E+02	0.101769E+01	0.840261E+00	0.159827E+01	0.318517E+03	0.212025E+03
PCNC	CNC	ZC	PRC	WACC	WAC
0.933807E+02	0.101401E+01	0.830689E+00	0.758977E+01	0.109121E+03	0.107434E+03
T2	P2	T22	P22	T21	P21
0.460573E+03	0.473409E+00	0.515237E+03	0.663457E+00	0.601655E+03	0.106038E+01
T3	P3	PCBLF	BLF	PCBLC	BLC
0.113928E+04	0.804806E+01	0.	0.	0.	0.
PCBLHP	BLHP	PCBLIP	BLIP	PCBLLP	BLLP
0.	0.	0.	0.	0.	0.
WA3	WFB	WG4	FAR4	T4	P4
0.107434E+03	0.190718E+01	0.109341E+03	0.177521E-01	0.226000E+04	0.764753E+01
TFHP	CNHP	DHTCHP	DHTC	T50	P50
0.462507E+02	0.196428E+01	0.600790E-01	0.129697E+03	0.180739E+04	0.267177E+01
TFHP	CNIP	DHTCIP	DHTI	T5	P5
0.118389E+03	0.219521E+01	0.220459E-01	0.401518E+02	0.166336E+04	0.183258E+01
TFFLP	CNLP	DHTCLP	DHTF	T55	P55
0.165582E+03	0.231030E+01	0.172933E-01	0.391541E+02	0.152068E+04	0.122776E+01
ETAB	PCBLDU	ETAD	DPDUC	T24	P24
0.983000E+00	0.	0.	0.558117E-01	0.515237E+03	0.626428E+00
WAD	WFD	WG24	FAR24	T25	P25
0.115366E+03	0.	0.115366E+03	0.	0.515237E+03	0.626428E+00
ETAF	ETAI	ETAC	ETATHP	ETATIP	ETATLP
0.852878E+00	0.854577E+00	0.855369E+00	0.899049E+00	0.899082E+00	0.899963E+00
T6	P6	PS6	AM6	V6	WG6
0.152068E+04	0.122776E+01	0.118420E+01	0.237976E+00	0.442789E+03	0.109341E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.152068E+04	0.	0.109341E+03	0.177521E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9
0.665199E+00	0.100000E+01	0.173415E+04	0.665199E+00	0.100000E+01	0.173415E+04
PS28	AM28	V28	PS29	AM29	V29
0.371092E+00	0.897209E+00	0.926739E+03	0.371092E+00	0.897209E+00	0.926739E+03
BPRINT	DPCOM	DPWING	PS38	AM38	V38
0.973537E+00	0.497678E-01	0.999163E-01	0.504516E+00	0.100000E+01	0.109806E+04
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.544115E+00	0.	0.190718E+01	0.329298E+03	0.609938E+03	0.620650E+04
PCBLI	WG37	VJW	PS39	AM39	V39
0.493296E+00	0.104591E+03	0.108159E+04	0.504516E+00	0.100000E+01	0.109806E+04
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.351602E+04	0.673685E+03	0.220693E+04	0.678512E+04	0.626428E+00
FFOVFN	FQOVFN	FCOVFN	FMNOFN	FNOVFD	P38
0.120785E+00	0.245431E+00	0.633784E+00	0.754569E+00	0.345795E+00	0.954433E+00
CVMNOZ	VJM	CVDNOZ	VJD	FGM	FGP
0.985000E+00	0.170814E+04	0.985000E+00	0.912838E+03	0.125942E+05	0.260439E+04

MAIN SCNIC CONVERGENT NOZZLE FG= 15198.55

FN= 8992.05

SFC= 0.76355

DUCT SUBSONIC CONVERG. NOZZLE

CONVERGED AFTER 25 LOOPS

\$D(1), IDES=1, * NOW GOING TO DESIGN SAME ENGINE WITH ONLY 2 SPOOLS USING
 \$D(1), IDES=1, * BOOSTED FAN (FIG 2), NOTE NEED 2 IDES=1 CARDS TO RUN NEW DESIGN
 AM=0,ALTP=0, FIXFANTOMIDLE=.T., IAMTP=2, T2=31, * NOTE PCBLIDS STILL = .5

FAN DESIGN	PRFCF= 0.10000000E+01	ETAFCF= 0.10000000E+01	WAFCF= 0.10294510E+01	T2DS= 0.54966819E+03
MIDDLE SPOOL DESIGN	PRICF= 0.10443214E+01	ETAICF= 0.97711820E+00	WAICF= 0.11059473E+01	T22DS= 0.61268656E+03
COMPRESSOR DESIGN	PRCCF= 0.89610390E+00	ETACCF= 0.10000000E+01	WACCF= 0.10632227E+01	T21DS= 0.70943958E+03
INTER DUCT DESIGN	A38= 0.23859580E+01	AM38= 0.10000000E+01	A39= 0.23859580E+01	AM39= 0.10000000E+01
COMBUSTOR DESIGN	WA3CDS= 0.30802648E+02	ETABCF= 0.98300000E+00	DTCOCF= 0.10000000E+01	
H.P. TURBINE DESIGN	CNHPCF= 0.10119289E+01	TFHPCF= 0.10815634E+01	ETHPCF= 0.10000000E+01	DHHPHF= 0.957214C5E+00
L.P. TURBINE DESIGN	CVLPCF= 0.10438194E+01	TFLPCF= 0.10995854E+01	ETLPCF= 0.10201339E+01	DHLPCF= 0.254663C5E+01
DUCT NOZZLE DESIGN	A28= 0.37508934E+01	AM28= 0.65160510E+00	A29= 0.37508934E+01	AM29= 0.65160510E+00
TURBINE AREA DESIGN	A55= 0.79688787E+01	AM55= 0.23826598E+00		
NOZZLE DESIGN	A8= 0.31144586E+01	AM8= 0.10000000E+01	A9= 0.31144586E+01	AM9= 0.10000000E+01
OUTPUT	AM= 0.	ALTP= 0.	T4= 2560.00	ETAR= 1.0000

FAN AND MIDDLE SPOOL ARE ATTACHED , USE INNER AND OUTER TURBINES

PCNF	CNF	ZF	PRF	WAF	WAF
0.100000E+03	0.100000E+01	0.833333E+00	0.140000E+01	0.617671E+03	0.600000E+03
PCNI	CNI	ZI	PR1	WACI	WAI
0.947177E+02	0.947177E+00	0.833333E+00	0.157143E+01	0.310535E+03	0.400004E+C3
PCNC	CNC	ZC	PRC	WACC	WAC
0.100000E+03	0.100000E+01	0.814332E+00	0.727273E+01	0.106322E+03	0.200002E+C3
T2	P2	T22	P22	T21	P21
0.549668E+03	0.100000E+01	0.612687E+03	0.140000E+01	0.709440E+03	0.220000E+01
T3	P3	PCBLF	BLF	PCBLC	BLC
0.131143E+04	0.160000E+02	0.	0.	0.	0.
PCBLHP	BLHP	PCBLIP	BLIP	PCBLLP	BLIP
0.	0.	0.	0.	0.	0.
WA3	WFB	WG4	FAR4	T4	P4
0.200002E+03	0.409714E+01	0.204099E+03	0.204855E-01	0.256000E+04	0.152000E+02
TFFHP	CNHP	DHTCHP	DHTC	T50	P50
0.462294E+02	0.197642E+01	0.600000E-01	0.147028E+03	0.205966E+04	0.533120E+01
TFFIP	CNIP	DHTCIP	DHTI	T5	P5
0.	0.	0.	0.	0.205966E+04	0.533120E+01
TFFLP	CNLP	DHTCLP	DHTF	T55	P55
0.118226E+03	0.220345E+01	0.171750E-01	0.900859E+02	0.174312E+04	0.245320E+01
ETAB	PCBLDU	ETAD	DPDUC	T24	P24
0.983000E+00	0.	0.	0.500000E-01	0.612687E+03	0.133000E+01
WAD	WFO	WG24	FAR24	T25	P25
0.199997E+03	0.	0.199997E+03	0.	0.612687E+03	0.133000E+01
ETAF	ETAI	ETAC	ETATHP	ETATIP	ETATLP
0.880000E+00	0.870000E+00	0.860000E+00	0.900000E+00	0.	0.900000E+00
T6	P6	PS6	AM6	V6	WG6
0.174312E+04	0.245320E+01	0.236307E+01	0.238266E+00	0.472365E+03	0.204099E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.174312E+04	0.	0.204099E+03	0.204855E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9
0.133457E+01	0.100000E+01	0.185277E+04	0.133457E+01	0.100000E+01	0.185277E+04
PS28	AM28	V28	PS29	AM29	V29
0.100000E+01	0.651605E+00	0.759245E+03	0.100000E+01	0.651605E+00	0.759245E+03
BPRINT	DPCOM	DPWING	PS38	AM38	V38
0.100000E+01	0.500000E-01	0.100000E+00	0.104874E+01	0.100000E+01	0.119237E+C4
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.499987E+00	0.	0.409714E+01	0.604097E+03	0.	0.
PCBLI	WG37	VJW	PS39	AM39	V39
0.500000E+00	0.200002E+03	0.117448E+04	0.104874E+01	0.100000E+01	0.119237E+C4
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.730087E+04	0.246082E+03	0.754695E+04	0.184308E+05	0.133000E+01
FFOVFN	FWOVFN	FCOVFN	FMNOFN	FNOVFD	P38
0.178951E+00	0.290516E+00	0.530533E+00	0.709483E+00	0.100000E+01	0.198000E+01
CVNNOZ	VJM	CVNNOZ	VJD	FGM	FGP
0.985000E+00	0.182498E+04	0.985000E+00	0.747856E+03	0.235265E+05	0.245117E+04

MAIN SCNIC CONVERGENT NOZZLE
DUCT SUBSONIC CONVERG. NOZZLE

FG= 25977.71

FN= 25977.71

SFC= 0.54778

CONVERGED AFTER 1 LOOPS

COMMON	0.833333E+00	0.100000E+03	0.833333E+00	0.947177E+02	0.814332E+00	0.100000E+03	0.256000E+04	0
0.100000E+03	0.100000E+03	0.256000E+04	0.100000E+03	0.	0.100000E+01	0.100000E+01	0.100000E+01	
0.833333E+00	0.100000E+03	0.140000E+01	0.880000E+00	0.600000E+03	0.100000E+01	0.100000E+01	0.102945E+01	
0.814332E+00	0.100000E+03	0.727273E+01	0.860000E+00	0.200002E+03	0.896104E+00	0.100000E+01	0.106322E+01	
0.256000E+04	0.	0.124857E+04	0.983000E+00	0.308026E+02	0.500000E-01	0.100000E+01	0.983000E+00	
0.500000E+02	0.200000E+01	0.900000E+00	0.108156E+01	0.101193E+01	0.100000E+01	0.957214E+00	0.549668E+03	
0.130000E+03	0.230000E+01	0.900000E+00	0.109959E+01	0.104382E+01	0.102013E+01	0.254663E+01	0.709440E+03	
0.	0.	0.	0.	0.353601E+04	0.500000E-01	0.	0.	
0.	0.	0.	0.	0.347353E+04	0.	0.	0.	
0.796888E+01	0.	0.796888E+01	0.796888E+01	0.311446E+01	0.311446E+01	0.375089E+01	0.375089E+01	
0.236307E+01	0.238266E+00	0.985000E+00	0.985000E+00	0.	0.	0.	0.	
0.549670E+03	0.100000E+01	0.131340E+03	0.160493E+01	0.549668E+03	0.100000E+01	0.131340E+03	0.160493E+01	
0.709440E+03	0.220000E+01	0.169750E+03	0.161219E+01	0.131143E+04	0.160000E+02	0.319790E+03	0.162873E+01	
0.256000E+04	0.152000E+02	0.681178E+03	0.182692E+01	0.205966E+04	0.533120E+01	0.534150E+03	0.183496E+01	
0.174312E+04	0.245320E+01	0.444064E+03	0.184076E+01	0.	0.	0.	0.	
0.100000E+01	0.140000E+01	0.880000E+00	0.617671E+03	0.600000E+03	0.200002E+03	0.204099E+03	0.204855E-01	
0.100000E+01	0.727273E+01	0.860000E+00	0.106322E+03	0.200002E+03	0.983000E+00	0.500000E-01	0.	
0.200000E+01	0.900000E+00	0.574328E-01	0.147028E+03	0.	0.204099E+03	0.204855E-01	0.114970E+04	
0.230000E+01	0.900000E+00	0.437383E-01	0.900859E+02	0.	0.204099E+03	0.204855E-01	0.	
0.	0.	0.100000E+01	0.833333E+00	0.100000E+03	0.814332E+00	0.100000E+03	0.409714E+01	
0.500000E+02	0.130000E+03	0.	0.	0.	0.	0.	0.	
0.100000E+01	0.600000E+03	0.200002E+03	0.	0.	0.319790E+03	0.	0.	
0.709440E+03	0.220000E+01	0.169750E+03	0.161219E+01	0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	
0.612637E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	
0.199997E+03	0.	0.199997E+03	0.	0.	0.500000E-01	0.499987E+00	0.	
0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	

0.174312E+04	0.245320E+01	0.444064E+03	0.184076E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01
0.409714E+01	0.204099E+03	0.204855E-01	0.	0.199977E+03	0.	0.100000E+01	0.
0.174312E+04	0.245320E+01	0.444064E+03	0.184076E+01	0.174312E+04	0.245320E+01	0.444066E+03	0.184076E+01
0.174312E+04	0.245320E+01	0.444066E+03	0.184076E+01	0.174312E+04	0.245320E+01	0.444066E+03	0.184076E+01
0.204099E+03	0.	0.204099E+03	0.204855E-01	0.	0.	0.472365E+03	0.
0.236307E+01	0.472365E+03	0.238266E+00	0.172717E+04	0.236288E+01	0.472361E+03	0.238266E+00	0.
0.149533E+04	0.133457E+01	0.185277E+04	0.100000E+01	0.149533E+04	0.133457E+01	0.185277E+04	0.100000E+01
0.	0.	0.747856E+03	0.464873E+04	0.182498E+04	0.115769E+05	0.	0.220509E+04
0.235265E+05	0.245117E+04	0.409714E+01	0.604097E+03	0.682856E-02	0.259777E+05	0.259777E+05	0.567783E+00
0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	0.205966E+04	0.533120E+01	0.534150E+03	0.183496E+01
0.400004E+03	0.833333E+00	0.947177E+02	0.947177E+00	0.157143E+01	0.870000E+00	0.310535E+03	0.120000E+03
0.219177E+01	0.899082E+00	0.222157E-01	0.401518E+02	0.	0.	0.100000E+03	0.833333E+00
0.947177E+02	0.157143E+01	0.870000E+00	0.400004E+03	0.104432E+01	0.977118E+00	0.110595E+01	0.120000E+03
0.220000E+01	0.900000E+00	0.101500E+01	0.998436E+00	0.	0.100770E+01	0.310535E+03	0.400004E+03
0.500000E+00	0.200002E+03	0.612687E+03	0.200002E+03				
0.200002E+03	0.100000E+00	0.100000E+00	0.242141E+04	0.238596E+01	0.100000E+01	0.119237E+04	0.709440E+03
0.169750E+03	0.198000E+01	0.591600E+03	0.104874E+01	0.109440E+03	0.169750E+03	0.198000E+01	0.591600E+03
0.119237E+04	0.100000E+01	0.238596E+01	0.100000E+01	0.200002E+03	0.985000E+00	0.730087E+04	0.246082E+03
0.754695E+04	0.184308E+05	0.290516E+00	0.104874E+01	0.000000E+38	0.	0.	0.
0.178951E+00	0.530533E+00	0.709483E+00	0.100000E+01	0.	0.	0.259777E+05	0.500000E+00
0.462294E+02	0.	0.118226E+03	0.197642E+01	0.	0.220345E+01	0.600000E-01	0.147028E+03

\$D(1),AM=.6,ALTP=25000,IAMTP=0,T4=2260,† RUN AT ALTITUDE AT REDUCED T4-STD DAY

OUTPUT	AM= 0.600	ALTP= 25000.	T4= 2260.00	ETAR= 1.0000
FAN AND MIDDLE SPOOL ARE ATTACHED , USE INNER AND OUTER TURBINES				
PCNF	CNF	ZF	PRF	WAF
0.943578E+02	0.103081E+01	0.782227E+00	0.140053E+01	0.653515E+03
PCNI	CNI	ZI	PRI	WAI
0.893735E+02	0.974618E+00	0.844122E+00	0.160785E+01	0.213001E+03
PCNC	CNC	ZC	PRC	WACC
0.933772E+02	0.101356E+01	0.830117E+00	0.757917E+01	0.109033E+03
T2	P2	T22	P22	T21
0.460573E+03	0.473409E+00	0.515214E+03	0.663025E+00	0.602139E+03
T3	P3	PCBLF	BLF	PCBLC
0.113958E+04	0.807976E+01	0.	0.	0.
PCBLHP	BLHP	PCBLIP	BLIP	PCBLLP
0.	0.	0.	0.	0.
WA3	WFR	WG4	FAR4	T4
0.107878E+03	0.191458E+01	0.109792E+03	0.177476E-01	0.226000E+04
TFFHP	CNHP	DHTCHP	DHTC	T50
0.462601E+02	0.196420E+01	0.604728E-01	0.129657E+03	0.180753E+04
TFFIP	CNIP	DHTCIP	DHTI	T5
0.	0.	0.	0.	0.180753E+04
TFFLP	CNLP	DHTCLP	DHTF	T55
0.118498E+03	0.221940E+01	0.172633E-01	0.794933E+02	0.152013E+04
ETAB	PCBLDU	ETAD	DPDUC	T24
0.983000E+00	0.	0.	0.558207E-01	0.515214E+03
WAD	WFD	WG24	FAR24	T25
0.115312E+03	0.	0.115312E+03	0.	0.515214E+03
ETAF	ETA1	ETAC	ETATHP	ETATIP
0.851517E+00	0.861143E+00	0.855535E+00	0.898265E+00	0.
T6	P6	PS6	AMS	V6
0.152013E+04	0.122442E+01	0.118073E+01	0.238655E+00	0.443966E+03
T7	WFA	WG7	FAR7	ETAA
0.152013E+04	0.	0.109792E+03	0.177476E-01	0.
PS8	AM8	V8	PS9	AM9
0.665151E+00	0.100000E+01	0.173385E+04	0.665151E+00	0.100000E+01
PS28	AM28	V28	PS29	AM29
0.371092E+00	0.896824E+00	0.926377E+03	0.371092E+00	0.896824E+00
BPRINT	DPQOM	DPWING	PS38	AM38
0.974461E+00	0.497839E-01	0.999307E-01	0.507285E+00	0.100000E+01
BYPASS	HPEXT	WFT	WGT	VA
0.541371E+00	0.	0.191458E+01	0.330227E+03	0.609938E+03
PCBLI	WG37	VJW	PS39	AM39
0.493533E+00	0.105123E+03	0.108202E+04	0.507285E+00	0.100000E+01
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN
0.985000E+00	0.353531E+04	0.687667E+03	0.223012E+04	0.680525E+04
FFOVFN	FNOVFN	FNOVFN	FMNOFN	FNOVFD
0.120008E+00	0.246821E+00	0.633171E+00	0.753179E+00	0.347812E+00
CVMNOZ	VJM	CVDNOZ	VJD	FGM
0.985000E+00	0.170784E+04	0.985000E+00	0.912481E+03	0.126336E+05

MAIN SONIC CONVERGENT NOZZLE
DUCT SUBSONIC CONVERG. NOZZLE

FG= 15259.34

FN= 9035.37

SFC= 0.76283

CONVERGED AFTER 16 LOOPS

\$D(11,IDES=1, * NOW GOING TO RUN SAME ENGINE WITH ONLY 2 SPOOLS USING
 \$D(11,IDES=1, * SUPERCHARGED COMPRESSOR (FIGURE 3)
 FIXFANTOMIDDLE=.F.,FIXMIDDLETOCOMP=.T.,AM=0,ALTP=0,IAMTP=2,T2=31,
 PCNIDS=100, * MUST BE RESET BECAUSE IT PREVIOUSLY WAS DETERMINED BY PCNFDS

FAN DESIGN PRFCF= 0.10000000E+01 ETAFCF= 0.10000000E+01 WAFCF= 0.10294510E+01 T2DS= 0.54966819E+03
 MIDDLE SPOOL DESIGN PRICF= 0.95238095E+00 ETAICF= 0.98863635E+00 WAICF= 0.10351161E+01 T22DS= 0.61268656E+03
 COMPRESSOR DESIGN PRCCF= 0.10340043E+01 ETACCF= 0.10268452E+01 WACCF= 0.12048016E+01 T21DS= 0.70943958E+03
 INTER DUCT DESIGN A38= 0.23859579E+01 AM38= 0.10000000E+01 A39= 0.23859579E+01 AM39= 0.10000000E+01
 COMBUSTOR DESIGN WA3CDS= 0.30802649E+02 ETABCF= 0.98300000E+00 DTCOCF= 0.10000000E+01
 I.P. TURBINE DESIGN CNIPCF= 0.10119288E+01 TFIPCF= 0.10815634E+01 ETIPCF= 0.10000000E+01 DHIPCF= 0.12544893E+01
 L.P. TURBINE DESIGN CNLPCF= 0.10026477E+01 TFLPCF= 0.77781743E+00 ETLPCF= 0.10201339E+01 DHLPCF= 0.13610838E+01
 DUCT NOZZLE DESIGN A28= 0.37508934E+01 AM28= 0.65160510E+00 A29= 0.37508934E+01 AM29= 0.65160510E+00
 TURBINE AREA DESIGN A55= 0.80265361E+01 AM55= 0.23826609E+00
 NOZZLE DESIGN A8= 0.31369934E+01 AM8= 0.10000000E+01 A9= 0.31369934E+01 AM9= 0.10000000E+01
 OUTPUT AM= 0. ALTP= 0. T4= 2560.00 ETAR= 1.0000

MIDDLE AND COMPRESSOR SPOOLS ARE ATTACHED, USE MIDDLE AND OUTER TURBINES

PCNF	CNF	ZF	PRF	WAF	WAF
0.100000E+03	0.100000E+01	0.833333E+00	0.140000E+01	0.617671E+03	0.600000E+03
PCNI	CNI	ZI	PRI	WACI	WAI
0.100000E+03	0.100000E+01	0.833333E+00	0.157143E+01	0.310535E+03	0.400004E+03
PCNC	CNC	ZC	PRC	WACC	WAC
0.929312E+02	0.929312E+00	0.814332E+00	0.727273E+01	0.106322E+03	0.200002E+03
T2	P2	T22	P22	T21	P21
0.549668E+03	0.100000E+01	0.612687E+03	0.140000E+01	0.709440E+03	0.220000E+01
T3	P3	PCBLF	BLF	PCBLC	BLC
0.131143E+04	0.160000E+02	0.	0.	0.	0.
PCBLHP	BLHP	PCBLIP	BLIP	PCBLLP	BLLP
0.	0.	0.	0.	0.	0.
WA3	WFB	WG4	FAR4	T4	P4
0.200002E+03	0.409714E+01	0.204099E+03	0.204855E-01	0.256000E+04	0.152000E+C2
TFFHP	CNHP	DHTCHP	DHTC	T50	P50
0.	0.	0.	0.	0.256000E+04	0.152000E+02
TFFIP	CNIP	DHTCIP	DHTI	T5	P5
0.462294E+02	0.197642E+01	0.600000E-01	0.192690E+03	0.190038E+04	0.36224CE+01
TFFLP	CNLP	DHTCLP	DHTF	T55	P55
0.167134E+03	0.229393E+01	0.171750E-01	0.444244E+02	0.174312E+04	0.243558E+01
ETAB	PCBLDU	ETAD	DPDUC	T24	P24
0.983000E+00	0.	0.	0.500000E-01	0.612687E+03	0.133000E+C1
WAD	WFD	WG24	FAR24	T25	P25
0.199997E+03	0.	0.199997E+03	0.	0.612687E+03	0.133000E+01
ETAF	ETAI	ETAC	ETATHP	ETATIP	ETATLP
0.880000E+00	0.870000E+00	0.860000E+00	0.	0.900000E+00	0.900000E+00
T6	P6	PS6	AM6	V6	WG6
0.174312E+04	0.243558E+01	0.234610E+01	0.238266E+00	0.472365E+03	0.204099E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.174312E+04	0.	0.204099E+03	0.204855E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9
0.132498E+01	0.100000E+01	0.185277E+04	0.132498E+01	0.100000E+01	0.185277E+04
PS28	AM28	V28	PS29	AM29	V29
0.100000E+01	0.651605E+00	0.759245E+03	0.100000E+01	0.651605E+00	0.759245E+03
BPRINT	DPCOM	DPWING	PS38	AM38	V38
0.100000E+01	0.500000E-01	0.100000E+00	0.104874E+01	0.100000E+01	0.119237E+04
BPASS	HPEXT	WFT	WGT	VA	FRD
0.499987E+00	0.	0.409714E+01	0.604097E+03	0.	0.
PCBLI	WG37	VJW	PS39	AM39	V39
0.500000E+00	0.200002E+03	0.117448E+04	0.104874E+01	0.100000E+01	0.119237E+C4
CVOWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.730087E+04	0.246082E+03	0.754695E+04	0.183831E+05	0.133000E+01
FFOVFN	FWOVFN	FCOVFN	FMNOFN	FNOVFD	P38
0.179280E+00	0.291051E+00	0.529669E+00	0.708949E+00	0.100000E+01	0.198000E+C1
CVMNOZ	VJM	CVONDOZ	VJD	FGM	FGP
0.985000E+00	0.182498E+04	0.985000E+00	0.747856E+03	0.235265E+05	0.240348E+04

MAIN SONIC CONVERGENT NOZZLE FG= 25930.02 FN= 25930.02 SFC= 0.56883
 DUCT SUBSONIC CONVERG. NOZZLE
 CONVERGED AFTER 1 LOOPS

COMMON	0.833333E+00	0.100000E+03	0.833333E+00	0.100000E+03	0.814332E+00	0.929312E+02	0.256000E+04	0
0.100000E+03	0.100000E+03	0.256000E+04	0.100000E+03	0.	0.100000E+01	0.100000E+01	0.100000E+01	
0.833333E+00	0.100000E+03	0.140000E+01	0.880000E+00	0.600000E+03	0.100000E+01	0.100000E+01	0.102945E+01	
0.814332E+00	0.929312E+02	0.727273E+01	0.860000E+00	0.200002E+03	0.103400E+01	0.102685E+01	0.120480E+01	
0.256000E+04	0.	0.124857E+04	0.983000E+00	0.308026E+02	0.500000E-01	0.100000E+01	0.983000E+00	
0.500000E+02	0.200000E+01	0.900000E+00	0.	0.	0.	0.	0.549668E+03	
0.130000E+03	0.230000E+01	0.900000E+00	0.777817E+00	0.100265E+01	0.102013E+01	0.136108E+01	0.709440E+03	
0.	0.	0.	0.	0.353601E+04	0.500000E-01	0.	0.	
0.	0.	0.	0.	0.349866E+04	0.	0.	0.	
0.802654E+01	0.	0.802654E+01	0.802654E+01	0.313699E+01	0.313699E+01	0.375089E+01	0.375089E+01	
0.234610E+01	0.238266E+00	0.985000E+00	0.985000E+00	0.	0.	0.	0.	
0.549670E+03	0.100000E+01	0.131340E+03	0.160493E+01	0.549668E+03	0.100000E+01	0.131340E+03	0.160493E+01	
0.709440E+03	0.220000E+01	0.169750E+03	0.161219E+01	0.131143E+04	0.160000E+02	0.319790E+03	0.162873E+01	
0.256000E+04	0.152000E+02	0.681178E+03	0.182692E+01	0.190038E+04	0.362240E+01	0.488488E+03	0.183841E+01	
0.174312E+04	0.243558E+01	0.444064E+03	0.184125E+01	0.	0.	0.	0.	
0.100000E+01	0.140000E+01	0.880000E+00	0.617671E+03	0.600000E+03	0.200002E+03	0.204099E+03	0.204855E-01	
0.929312E+00	0.727273E+01	0.860000E+00	0.106322E+03	0.200002E+03	0.983000E+00	0.500000E-01	0.114970E+04	
0.	0.	0.	0.	0.	0.204099E+03	0.204855E-01	0.	
0.230000E+01	0.900000E+00	0.233766E-01	0.444244E+02	0.	0.204099E+03	0.204855E-01	0.	
0.	0.	0.100000E+01	0.833333E+00	0.100000E+03	0.814332E+00	0.929312E+02	0.409714E+01	
0.500000E+02	0.130000E+03	0.	0.	0.	0.	0.	0.	
0.100000E+01	0.600000E+03	0.200002E+03	0.	0.	0.319790E+03	0.	0.	
0.709440E+03	0.220000E+01	0.169750E+03	0.161219E+01	0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	
0.159997E+03	0.	0.199997E+03	0.	0.	0.500000E-01	0.499987E+00	0.	
0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	
0.174312E+04	0.243558E+01	0.444064E+03	0.184125E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	
0.409714E+01	0.204099E+03	0.204855E-01	0.	0.199997E+03	0.	0.100000E+01	0.	
0.174312E+04	0.243558E+01	0.444064E+03	0.184125E+01	0.174312E+04	0.243558E+01	0.444066E+03	0.184125E+01	
0.174312E+04	0.243558E+01	0.444066E+03	0.184125E+01	0.174312E+04	0.243558E+01	0.444066E+03	0.184125E+01	
0.204099E+03	0.	0.204099E+03	0.204855E-01	0.	0.	0.472365E+03	0.	
0.234610E+01	0.472365E+03	0.238266E+00	0.172717E+04	0.234590E+01	0.472361E+03	0.238266E+00	0.	
0.149533E+04	0.132498E+01	0.185277E+04	0.100000E+01	0.149533E+04	0.132498E+01	0.185277E+04	0.100000E+01	
0.	0.	0.747856E+03	0.464873E+04	0.182498E+04	0.115769E+05	0.	0.215740E+04	
0.235265E+05	0.240348E+04	0.409714E+01	0.604097E+03	0.682856E-02	0.259300E+05	0.259300E+05	0.568827E+00	
0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	0.256000E+04	0.152000E+02	0.681178E+03	0.182692E+01	
0.400004E+03	0.833333E+00	0.100000E+03	0.100000E+01	0.157143E+01	0.870000E+00	0.310535E+03	0.500000E+02	
0.200000E+01	0.900000E+00	0.752694E-01	0.192690E+03	0.	0.	0.100000E+03	0.833333E+00	
0.100000E+03	0.157143E+01	0.870000E+00	0.400004E+03	0.952381E+00	0.988636E+00	0.103512E+01	0.120000E+03	
0.220000E+01	0.900000E+00	0.108156E+01	0.101193E+01	0.	0.125449E+01	0.310535E+03	0.400004E+03	
0.500000E+00	0.200002E+03	0.612687E+03	0.200002E+03					
0.200002E+03	0.100000E+00	0.100000E+00	0.242141E+04	0.238596E+01	0.100000E+01	0.119237E+04	0.709440E+03	
0.169750E+03	0.198000E+01	0.591600E+03	0.104874E+01	0.709440E+03	0.169750E+03	0.198000E+01	0.591600E+03	
0.119237E+04	0.100000E+01	0.238596E+01	0.100000E+01	0.200002E+03	0.985000E+00	0.730087E+04	0.246082E+03	
0.754695E+04	0.183831E+05	0.291051E+00	0.104874E+01	0.	0.000000E-38	0.	0.	
0.179280E+00	0.529669E+00	0.708949E+00	0.100000E+01	0.	0.	0.259300E+05	0.500000E+00	
0.	0.462294E+02	0.167134E+03	0.	0.197642E+01	0.229393E+01	0.	0.	

SD(1),AM=.6,ALTP=25000,IAMTP=0,T4=2260,* RUN AT ALTITUDE AT REDUCED T4-STD DAY

OUTPUT	AM= 0.600	ALTP= 25000.	T4= 2260.00	ETAR= 1.0000
MIDDLE AND COMPRESSOR SPOOLS ARE ATTACHED , USE MIDDLE AND OUTER TURBINES				
PCNF	CNF	ZF	PRF	WAF
0.944038E+02	0.103131E+01	0.782864E+00	0.140127E+01	0.653913E+03
PCNI	CNI	ZI	PRI	WAI
0.936838E+02	0.102152E+01	0.845932E+00	0.160711E+01	0.319862E+03
PCNC	CNC	ZC	PRC	WAC
0.870615E+02	0.944101E+00	0.827728E+00	0.758119E+01	0.109122E+03
T2	P2	T22	P22	T21
0.460573E+03	0.473409E+00	0.515320E+03	0.663373E+00	0.603297E+03
T3	P3	PCBLF	BLF	PCBLC
0.113799E+04	0.808241E+01	0.	0.	0.
PCBLHP	BLHP	PCBLIP	BLIP	PCBLLP
0.	0.	0.	0.	0.
WAF	WFB	WG4	FAR4	T4
0.107869E+03	0.191696E+01	0.109786E+03	0.177712E-01	0.226000E+04
TFHHP	CNHP	DHTCHP	DHTC	T50
0.	0.	0.	0.	0.226000E+04
TFHIP	CNIP	DHTCIP	DHTI	T5
0.462394E+02	0.197065E+01	0.600370E-01	0.169863E+03	0.166333E+04
TFHLP	CNLP	DHTCLP	DHTF	T55
0.167379E+03	0.231473E+01	0.173050E-01	0.391884E+02	0.152052E+04
ETAB	PCBLDU	ETAD	DPDUC	T24
0.983000E+00	0.	0.	0.559541E-01	0.515320E+03
WAD	WFD	WG24	FAR24	T25
0.115637E+03	0.	0.115637E+03	0.	0.515320E+03
ETAF	ETAI	ETAC	ETATHP	ETATIP
0.851255E+00	0.850116E+00	0.861725E+00	0.	0.899548E+00
T6	P6	PS6	AM6	V6
				WG6

0.152052E+04	0.121897E+01	0.117572E+01	0.237976E+00	0.442765E+03	0.109786E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.152052E+04	0.	0.109786E+03	0.177712E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9
0.660433E+00	0.100000E+01	0.173406E+04	0.660433E+00	0.100000E+01	0.173406E+04
PS28	AM28	V28	PS29	AM29	V29
0.371092E+00	0.899116E+00	0.928506E+03	0.371092E+00	0.899116E+00	0.928506E+03
BPRINT	DPCOM	DPWING	PS38	AM38	V38
0.973473E+00	0.497287E-01	0.999108E-01	0.507217E+00	0.100000E+01	0.109955E+04
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.543211E+00	0.	0.191696E+01	0.330429E+03	0.609938E+03	0.622776E+04
PCBLI	WG37	VJW	PS39	AM39	V39
0.493279E+00	0.105007E+03	0.108306E+04	0.507217E+00	0.100000E+01	0.109955E+04
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.353481E+04	0.687321E+03	0.223146E+04	0.679908E+04	0.626255E+00
FDOVFN	FDOVFN	FCOVFN	FMVOFN	FNOVFD	P38
0.121245E+00	0.247102E+00	0.631653E+00	0.752898E+00	0.348266E+00	0.959598E+00
CVMNDZ	VJM	CVDNOZ	VJD	FGM	FGP
0.985000E+00	0.170805E+04	0.985000E+00	0.914579E+03	0.126502E+05	0.260813E+04

MAIN SCNIC CONVERGENT NOZZLE FG= 15258.30 FN= 9030.54 SFC= 0.76416
DUCT SUBSONIC CONVERG. NOZZLE

CONVERGED AFTER 21 LOOPS

\$D(1),IDES=1, * NOW GOING TO RUN 3 SPOOL,2 STREAM ENGINE (FIGURE 4)
\$D(1),FIXMIDDLTODCOMP=F,PCBLIDS=0,AM=0,ALTP=0,IAMTP=2,T2=31,IDES=1,PCNCDS=100,

FAN DESIGN	PRFCF= 0.10000000E+01	ETAFCF= 0.10000000E+01	WAFCF= 0.10294510E+01	T2DS= 0.54966819E+03
MIDDLE SPOOL DESIGN	PRICF= 0.95238095E+00	ETAICF= 0.98863635E+00	WAICF= 0.10351161E+01	T22DS= 0.61268656E+03
COMPRESSOR DESIGN	PRCCF= 0.89610390E+00	ETACCF= 0.10000000E+01	WACCF= 0.21264453E+01	T21DS= 0.70943958E+03
COMBUSTOR DESIGN	WA3CDS= 0.61605297E+02	ETABCF= 0.98300000E+00	DTCCCF= 0.10000000E+01	
H.P. TURBINE DESIGN	CNHPCF= 0.10119289E+01	TFHPCF= 0.54078169E+00	ETHPCF= 0.10000000E+01	DHHPCF= 0.95721405E+00
I.P. TURBINE DESIGN	CNIPCF= 0.99843589E+00	TFIPCF= 0.50750092E+00	ETIPCF= 0.10000000E+01	DHIPCF= 0.50385109E+00
L.P. TURBINE DESIGN	CNLPCF= 0.10235123E+01	TFLPCF= 0.46674602E+00	ETLPCF= 0.10201339E+01	DHLPFCF= 0.6537859E+00
DUCT NOZZLE DESIGN	A28= 0.37508935E+01	AM28= 0.65160510E+00	A29= 0.37508935E+01	AM29= 0.65160510E+00
TURBINE AREA DESIGN	A55= 0.11109352E+02	AM55= 0.23826426E+00		
NOZZLE DESIGN	A8= 0.43403495E+01	AM8= 0.10000000E+01	A9= 0.43403495E+01	AM9= 0.10000000E+01
OUTPUT	AM= 0.	ALTP= 0.	T4= 2560.00	ETAR= 1.0000

THREE SPOOL ENGINE

NO AIRFLOW INTO WING

PCNF	CNF	ZF	PRF	WAF	WAF
0.100000E+03	0.100000E+01	0.833333E+00	0.140000E+01	0.617671E+03	0.600000E+03
PCNI	CNI	ZI	PRI	WACI	WAI
0.100000E+03	0.100000E+01	0.833333E+00	0.157143E+01	0.310535E+03	0.400004E+03
PCNC	CNC	ZC	PRC	WACC	WAC
0.100000E+03	0.100000E+01	0.814332E+00	0.727273E+01	0.212645E+03	0.400004E+03
T2	P2	T22	P22	T21	P21
0.549668E+03	0.100000E+01	0.612687E+03	0.140000E+01	0.709440E+03	0.220000E+01
T3	P3	PCBLF	BLF	PCBLC	BLC
0.131143E+04	0.160000E+02	0.	0.	0.	0.
PCBLHP	BLHP	PCBLIP	BLIP	PCBLLP	BLLP
0.	0.	0.	0.	0.	0.
WA3	WFB	WG4	FAR4	T4	P4
0.400004E+03	0.819427E+01	0.408198E+03	0.204855E-01	0.256000E+04	0.152000E+02
TFFHP	CNH	DHTCHP	DHTC	T50	P50
0.924587E+02	0.197642E+01	0.600000E-01	0.147028E+03	0.205966E+04	0.53312CE+C1
TFFIP	CNIP	DHTCIP	DHTI	T5	P5
0.236453E+03	0.220345E+01	0.220000E-01	0.228307E+02	0.198030E+04	0.443787E+01
TFFLP	CNLP	DHTCLP	DHTF	T55	P55
0.278524E+03	0.224716E+01	0.171750E-01	0.222122E+02	0.190256E+04	0.368588E+01
ETAB	PCBLDU	ETAD	DPDUC	T24	P24
0.983000E+00	0.	0.	0.500000E-01	0.612687E+03	0.133000E+01
WAD	WFD	WG24	FAR24	T25	P25
0.199997E+03	0.	0.199997E+03	0.	0.612687E+03	0.133000E+01
ETAF	ETAI	ETAC	ETATHP	ETATIP	ETATLP
0.880000E+00	0.870000E+00	0.860000E+00	0.900000E+00	0.900000E+00	0.900000E+00
T6	P6	PS6	AM6	V6	WG6
0.190256E+04	0.368588E+01	0.355121E+01	0.238264E+00	0.492279E+03	0.408198E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.190256E+04	0.	0.408198E+03	0.204855E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9

0.200961E+01	0.100000E+01	0.193326E+04	0.200961E+01	0.100000E+01	0.193326E+04
PS28	AM28	V28	PS29	AM29	V29
0.100000E+01	0.651605E+00	0.759245E+03	0.100000E+01	0.651605E+00	0.759245E+03
BPRINT	DPCOM	DPWING	PS38	AM38	V38
0.	0.500000E-01	0.	0.	0.	0.
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.499987E+00	0.	0.819427E+01	0.608194E+03	0.	0.
PCBLI	WG37	VJW	PS39	AM39	V39
0.149012E-07	0.	0.	0.	0.	0.
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.	0.	0.	0.380818E+05	0.133000E+C1
FFOVFN	FWDVFN	FCOVFN	FMNOFN	FNOVFD	P38
0.122072E+00	0.	0.877928E+00	0.100000E+01	0.100000E+01	0.
CVMNDZ	VJM	CVDNDZ	VJD	FGM	FGP
0.985000E+00	0.190426E+04	0.985000E+00	0.747856E+03	0.288084E+05	0.927342E+04

MAIN SCNIC CONVERGENT NOZZLE
DUCT SUBSONIC CONVERG. NOZZLE

FG= 38081.84

FN= 38081.84

SFC= 0.77463

CONVERGED AFTER 1 LOOPS

COMMON	0.833333E+00	0.100000E+03	0.833333E+00	0.100000E+03	0.814332E+00	0.100000E+03	C.256000E+04	0
0.100000E+03	0.100000E+03	0.256000E+04	0.100000E+03	0.	0.100000E+01	0.100000E+01	0.100000E+01	
0.833333E+00	0.100000E+03	0.140000E+01	0.880000E+00	0.600000E+03	0.100000E+01	0.100000E+01	0.102945E+01	
0.814332E+00	0.100000E+03	0.727273E+01	0.860000E+00	0.400004E+03	0.896104E+00	0.100000E+01	0.212645E+01	
0.256000E+04	0.	0.124857E+04	0.983000E+00	0.616053E+02	0.500000E-01	0.100000E+01	0.983000E+00	
0.500000E+02	0.200000E+01	0.900000E+00	0.540782E+00	0.101193E+01	0.100000E+01	0.957214E+00	0.549668E+03	
0.130000E+03	0.230000E+01	0.900000E+00	0.466746E+00	0.102351E+01	0.102013E+01	0.653079E+00	0.709440E+03	
0.	0.	0.	0.	0.353601E+04	0.500000E-01	0.	0.	
0.	0.	0.	0.	0.483057E+04	0.	0.	0.	
0.111094E+02	0.	0.111094E+02	0.111094E+02	0.434035E+01	0.434035E+01	0.375089E+01	0.375089E+01	
0.355121E+01	0.238264E+00	0.985000E+00	0.985000E+00	0.	0.	0.	0.	
0.549670E+03	0.100000E+01	0.131340E+03	0.160493E+01	0.549668E+03	0.100000E+01	0.131340E+03	0.160493E+01	
0.709440E+03	0.220000E+01	0.169750E+03	0.161219E+01	0.131143E+04	0.160000E+02	0.319790E+03	0.162873E+01	
0.256000E+04	0.152000E+02	0.681178E+03	0.182692E+01	0.198030E+04	0.443787E+01	0.511319E+03	0.183624E+01	
0.190256E+04	0.368588E+01	0.489107E+03	0.183754E+01	0.	0.	0.	0.	
0.100000E+01	0.140000E+01	0.880000E+00	0.617671E+03	0.600000E+03	0.400004E+03	0.408198E+03	0.204855E-01	
0.100000E+01	0.727273E+01	0.860000E+00	0.212645E+03	0.400004E+03	0.983000E+00	0.500000E-01	0.	
0.200000E+01	0.900000E+00	0.574328E-01	0.147028E+03	0.	0.408198E+03	0.204855E-01	0.114970E+04	
0.230000E+01	0.900000E+00	0.112166E-01	0.222122E+02	0.	0.408198E+03	0.204855E-01	C.	
0.	0.	0.100000E+01	0.833333E+00	0.100000E+03	0.814332E+00	0.100000E+03	0.819427E+01	
0.500000E+02	0.130000E+03	0.	0.	0.	0.	0.	0.	
0.100000E+01	0.600000E+03	0.400004E+03	0.	0.	0.319790E+03	0.	0.	
0.709440E+03	0.220000E+01	0.169750E+03	0.161219E+01	0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	
0.199997E+03	0.	0.199997E+03	0.	0.	0.500000E-01	0.499987E+00	0.	
0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	
0.190256E+04	0.368588E+01	0.489107E+03	0.183754E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	
0.819427E+01	0.408198E+03	0.204855E-01	0.	0.199997E+03	0.	0.100000E+01	0.	
0.190256E+04	0.368588E+01	0.489107E+03	0.183754E+01	0.190256E+04	0.368588E+01	0.489106E+03	0.183754E+01	
0.190256E+04	0.368588E+01	0.489106E+03	0.183754E+01	0.190256E+04	0.368588E+01	0.489106E+03	0.183754E+01	
0.408198E+03	0.	0.408198E+03	0.204855E-01	0.	0.	0.492279E+03	C.	
0.355121E+01	0.492279E+03	0.238264E+00	0.188551E+04	0.355092E+01	0.492274E+03	0.238264E+00	0.	
0.163715E+04	0.200961E+01	0.193326E+04	0.100000E+01	0.163715E+04	0.200961E+01	0.193326E+04	0.100000E+01	
0.	0.	0.747856E+03	0.464873E+04	0.190426E+04	0.241597E+05	0.	0.927342E+04	
0.288084E+05	0.927342E+04	0.819427E+01	0.608194E+03	0.136571E-01	0.380818E+05	0.380818E+05	0.774631E+00	
0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	0.205966E+04	0.533120E+01	0.534150E+03	0.183496E+01	
0.400004E+03	0.833333E+00	0.100000E+03	0.100000E+01	0.157143E+01	0.870000E+00	0.310535E+03	0.120000E+03	
0.220000E+01	0.900000E+00	0.110847E-01	0.228307E+02	0.	0.	0.100000E+03	0.833333E+00	
0.100000E+03	0.157143E+01	0.870000E+00	0.400004E+03	0.952381E+00	0.988636E+00	0.103512E+01	0.120000E+03	
0.220000E+01	0.900000E+00	0.507501E+00	0.998436E+00	0.	0.503851E+00	0.310535E+03	0.400004E+03	
0.149012E-07	0.	0.612687E+03	0.400004E+03					
0.381470E-05	0.100000E+00	0.	0.	0.	0.	0.	0.	
0.	0.	0.	0.	0.	0.	0.	0.	
0.	0.	0.	0.	0.	0.985000E+00	0.	C.	
0.	0.380818E+05	0.	0.	0.	0.	0.	0.	
0.122072E+00	0.877928E+00	0.100000E+01	0.100000E+01	0.	0.	0.380818E+05	0.	
0.924537E+02	0.236453E+03	0.278524E+03	0.197642E+01	0.220345E+01	0.224716E+01	0.600000E-01	0.147028E+03	

\$D(1),AM=.6,ALTP=25000,IAMTP=0,T4=2260, # RUN AT ALTITUDE AT REDUCED T4-STD DAY

OUTPUT AM= 0.600 ALTP= 25000. T4= 2260.00 ETAR= 1.0000

THREE SPOOL ENGINE

NO AIRFLOW INTO WING

PCNF	CNF	ZF	PRF	WAFc	WAF
0.944120E+02	0.103140E+01	0.775104E+00	0.139736E+01	0.654868E+03	0.328992E+03
PCNI	CNI	ZI	PRI	WACI	WAI
0.936826E+02	0.102192E+01	0.821694E+00	0.159021E+01	0.321998E+03	0.213788E+03
PCNC	CNC	ZC	PRC	WACC	WAC
0.934573E+02	0.101581E+01	0.831314E+00	0.761931E+01	0.218983E+03	0.214090E+03
T2	P2	T22	P22	T21	P21
0.460573E+03	0.473409E+00	0.514897E+03	0.661525E+00	0.600508E+03	0.105196E+01
T3	P3	PCBLF	BLF	PCBLC	BLC
0.113861E+04	0.801523E+01	0.	0.	0.	0.
PCBLHP	BLHP	PCBLIP	BLIP	PCBLLP	BLLP
0.	0.	0.	0.	0.	0.
WA3	WFB	WG4	FAR4	T4	P4
0.214090E+03	0.380269E+01	0.217893E+03	0.177621E-01	0.226000E+04	0.761627E+01
TFFHP	CNHP	DHTCHP	DHTC	T50	P50
0.925457E+02	0.196589E+01	0.605788E-01	0.129798E+03	0.180704E+04	0.265522E+01
TFFIP	CNIP	DHTCIP	DHTI	T5	P5
0.237371E+03	0.220382E+01	0.223231E-01	0.201260E+02	0.173511E+04	0.220700E+01
TFFLP	CNLP	DHTCLP	DHTF	T55	P55
0.279837E+03	0.226654E+01	0.173347E-01	0.195211E+02	0.166447E+04	0.182993E+01
ETAB	PCBLDU	ETAD	DPDUC	T24	P24
0.983000E+00	0.	0.	0.557311E-01	0.514897E+03	0.624657E+00
WAD	WFD	WG24	FAR24	T25	P25
0.114902E+03	0.	0.114902E+03	0.	0.514897E+03	0.624657E+00
ETAF	ETAI	ETAC	ETA THP	ETATIP	ETATLP
0.850443E+00	0.852141E+00	0.855233E+00	0.898165E+00	0.897880E+00	0.900068E+00
T6	P6	PS6	AM6	V6	WG6
0.166447E+04	0.182993E+01	0.176503E+01	0.238501E+00	0.463022E+03	0.217893E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.166447E+04	0.	0.217893E+03	0.177621E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9
0.995600E+00	0.100000E+01	0.181211E+04	0.995600E+00	0.100000E+01	0.181211E+04
PS28	AM28	V28	PS29	AM29	V29
0.371092E+00	0.893783E+00	0.923390E+03	0.371092E+00	0.893783E+00	0.923390E+03
8PRINT	DPCOM	DPWING	PS38	AM38	V38
0.	0.497761E-01	0.	0.	0.	0.
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.538869E+00	0.	0.380269E+01	0.332795E+03	0.609938E+03	0.623685E+04
PCBLI	WG37	VJW	PS39	AM39	V39
-0.141260E-02	0.	0.	0.	0.	0.
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.	0.	0.	0.148299E+05	0.624657E+00
FFOVFN	FWOVFN	FCOVFN	FMNOFN	FNOVFD	P38
0.721207E-01	0.	0.927493E+00	0.999614E+00	0.389572E+00	0.
CVMNOZ	VJM	CVDNOZ	VJD	FGM	FGP
0.985000E+00	0.178492E+04	0.985000E+00	0.909539E+03	0.153363E+05	0.573618E+04

MAIN SCNIC CONVERGENT NOZZLE FG= 21072.47 FN= 14835.61 SFC= 0.92276
 DUCT SUBSONIC CONVERG. NOZZLE

CONVERGED AFTER 24 LOOPS

\$D(1),IDES=1, # NOW GOING TO RUN 2 SPOOL-2 STREAM ENGINE (NORMAL TURBOFAN)
 \$D(1),IDES=1,AM=0,ALTP=0,IAMTP=2,T2=31,DUMMYSPOOL=.T, # (FIGURE 5) (PCBLIDS=0)
 PCDS=16/1.4, # KEEP OVERALL PRESSURE RATIO THE SAME

FAN DESIGN	PRFCF= 0.10000000E+01	ETAFCF= 0.10000000E+01	WAFCF= 0.10294510E+01	T2DS= 0.54966819E+03
MIDDLE SPOOL DESIGN	PRICF= 0.10000000E+01	ETAICF= 0.10000000E+01	WAICF= 0.10000000E+01	T22DS= 0.61268656E+03
COMPRESSOR DESIGN	PRCCF= 0.14897959E+01	ETACCF= 0.10000000E+01	WACCF= 0.31053487E+01	T21DS= 0.61268656E+03
COMBUSTOR DESIGN	WA3CDS= 0.61393844E+02	ETABCF= 0.98300000E+00	DTCCCF= 0.10000000E+01	
H.P. TURBINE DESIGN	CNHPCF= 0.10119289E+01	TFHPCF= 0.54070825E+00	ETHPCF= 0.10000000E+01	DHHPCF= 0.10909047E+01
L.P. TURBINE DESIGN	CNLPFC= 0.10256150E+01	TFLPCF= 0.47233292E+00	ETLPCF= 0.10201339E+01	DHLPCF= 0.65031519E+00
DUCT NOZZLE DESIGN	A28= 0.37508928E+01	AM28= 0.65160510E+00	A29= 0.37508928E+01	AM29= 0.65160510E+00
TURBINE AREA DESIGN	A55= 0.10971928E+02	AM55= 0.23826411E+00		
NOZZLE DESIGN	A8= 0.42865705E+01	AM8= 0.10000000E+01	A9= 0.42865705E+01	AM9= 0.10000000E+01
OUTPUT	AM= 0.	ALTP= 0.	T4= 2560.00	ETAR= 1.0000

MIDDLE SPOOL IS DUMMY

NO AIRFLOW INTO WING

PCNF	CNF	ZF	PRF	WAF	WAF
0.100000E+03	0.100000E+01	0.833333E+00	0.140000E+01	0.617671E+03	0.600000E+C3
PCNI	CNI	ZI	PR	WAI	WAI
0.100000E+01	0.	0.571429E+00	0.100000E+01	0.310535E+03	0.400004E+C3
PCNC	CNC	ZC	PRC	WAC	WAC
0.100000E+03	0.100000E+01	0.814332E+00	0.114286E+02	0.310535E+03	0.400004E+C3
T2	P2	T22	P22	T21	P21
0.549668E+03	0.100000E+01	0.612687E+03	0.140000E+01	0.612687E+03	0.140000E+01
T3	P3	PCBLF	BLF	PCBLC	BLC
0.130245E+04	0.160000E+02	0.	0.	0.	0.
PCBLHP	BLHP	PCBLIP	BLIP	PCBLLP	BLLP
0.	0.	0.	0.	0.	0.
WA3	WFB	WG4	FAR4	T4	P4
0.400004E+03	0.824969E+01	0.408253E+03	0.206240E-01	0.256000E+04	0.152000E+02
TFFHP	CNHP	DHTCHP	DHTC	T50	P50
0.924713E+02	0.197642E+01	0.600000E-01	0.167563E+03	0.198844E+04	0.450082E+01
TFFIP	CNIP	DHTCIP	DHTI	T5	P5
0.	0.	0.	0.	0.198844E+04	0.450082E+01
TFFLP	CNLP	DHTCLP	DHTF	T55	P55
0.275230E+03	0.224256E+01	0.171750E-01	0.222092E+02	0.191079E+04	0.374119E+01
ETAB	PCBLDU	ETAD	DPDUC	T24	P24
0.983000E+00	0.	0.	0.500000E-01	0.612687E+03	0.133000E+01
WAD	WFD	WG24	FAR24	T25	P25
0.199996E+03	0.	0.199996E+03	0.	0.612687E+03	0.133000E+01
ETAF	ETAI	ETAC	ETATHP	ETATIP	ETATLP
0.880000E+00	0.100000E+01	0.860000E+00	0.900000E+00	0.	0.900000E+CC
T6	P6	PS6	AM6	V6	WG6
0.191079E+04	0.374119E+01	0.360454E+01	0.238264E+00	0.493269E+03	0.408253E+C3
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.191079E+04	0.	0.408253E+03	0.206240E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9
0.204005E+01	0.100000E+01	0.193729E+04	0.204005E+01	0.100000E+01	0.193729E+04
PS28	AM28	V28	PS29	AM29	V29
0.100000E+01	0.651605E+00	0.759245E+03	0.100000E+01	0.651605E+00	0.759245E+C3
BPRINT	DPCOM	DPWING	PS38	AM38	V38
0.	0.500000E-01	0.	0.	0.	0.
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.499987E+00	0.	0.824969E+01	0.608250E+03	0.	0.
PCBLI	WG37	VJW	PS39	AM39	V39
-0.	0.	0.	0.	0.	0.
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.	0.	0.	0.382967E+05	0.133000E+01
FFQVFN	FWDVFN	FCQVFN	FMQVFN	FNOVFD	P38
0.121387E+00	0.	0.878613E+00	0.100000E+01	0.100000E+01	0.
CVMNOZ	VJM	CVDNOZ	VJD	FGM	FGP
0.985000E+00	0.190823E+04	0.985000E+00	0.747856E+03	0.288621E+05	0.943459E+04

MAIN SCNIC CONVERGENT NOZZLE
DUCT SUBSONIC CONVERG. NOZZLE

FG= 38296.70

FN= 38296.70

SFC= 0.77545

CONVERGED AFTER 1 LOOPS

COMMON	0.833333E+00	0.100000E+03	0.571429E+00	0.100000E+01	0.814332E+00	0.100000E+03	0.256000E+04	0
0.100000E+03	0.100000E+03	0.256000E+04	0.100000E+03	0.	0.100000E+01	0.100000E+01	0.100000E+01	0.100000E+01
0.833333E+00	0.100000E+03	0.140000E+01	0.880000E+00	0.600000E+03	0.100000E+01	0.100000E+01	0.100000E+01	0.102945E+01
0.814332E+00	0.100000E+03	0.114286E+02	0.860000E+00	0.400004E+03	0.148980E+01	0.100000E+01	0.100000E+01	0.105335E+01
0.256000E+04	0.	0.125755E+04	0.983000E+00	0.613938E+02	0.500000E-01	0.100000E+01	0.100000E+01	0.983000E+00
0.500000E+02	0.200000E+01	0.900000E+00	0.540708E+00	0.101193E+01	0.100000E+01	0.109090E+01	0.549668E+03	0.549668E+03
0.130000E+03	0.230000E+01	0.900000E+00	0.472333E+00	0.102551E+01	0.102013E+01	0.650315E+00	0.612687E+03	0.612687E+03
0.	0.	0.	0.	0.353601E+04	0.500000E-01	0.	0.	0.
0.	0.	0.	0.	0.477008E+04	0.	0.	0.	0.
0.109719E+02	0.	0.109719E+02	0.109719E+02	0.428657E+01	0.428657E+01	0.375089E+01	0.375089E+01	0.375089E+01
0.360454E+01	0.238264E+00	0.985000E+00	0.985000E+00	0.	0.	0.	0.	0.
0.549670E+03	0.100000E+01	0.131340E+03	0.160493E+01	0.549668E+03	0.100000E+01	0.131340E+03	0.160493E+01	0.160493E+01
0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	0.130245E+04	0.160000E+02	0.317471E+03	0.162695E+01	0.162695E+01
0.256000E+04	0.152000E+02	0.681299E+03	0.182700E+01	0.198844E+04	0.450082E+01	0.513736E+03	0.183651E+01	0.183651E+01
0.191079E+04	0.374119E+01	0.491527E+03	0.183781E+01	0.	0.	0.	0.	0.
0.100000E+01	0.140000E+01	0.880000E+00	0.617671E+03	0.600000E+03	0.400004E+03	0.408253E+03	0.206240E-01	0.206240E-01
0.100000E+01	0.114286E+02	0.860000E+00	0.310535E+03	0.400004E+03	0.983000E+00	0.500000E-01	0.	0.
0.200000E+01	0.900000E+00	0.654543E-01	0.167563E+03	0.	0.408253E+03	0.206240E-01	0.114970E+04	0.114970E+04
0.230000E+01	0.900000E+00	0.111691E-01	0.222092E+02	0.	0.408253E+03	0.206240E-01	0.	0.
0.	0.	0.100000E+01	0.833333E+00	0.100000E+03	0.814332E+00	0.100000E+03	0.824969E+01	0.824969E+01
0.500000E+02	0.130000E+03	0.	0.	0.	0.	0.	0.	0.
0.100000E+01	0.600000E+03	0.400004E+03	0.	0.	0.317471E+03	0.	0.	0.
0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	0.160788E+01
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.161140E+01
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.161140E+01
0.159996E+03	0.	0.199996E+03	0.	0.	0.500000E-01	0.499987E+00	0.	0.
0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	0.651605E+00

0.191079E+04	0.374119E+01	0.491527E+03	0.183781E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01
0.824959E+01	0.408253E+03	0.206240E-01	0.	0.199996E+03	0.	0.100000E+01	0.
0.191079E+04	0.374119E+01	0.491527E+03	0.183781E+01	0.191079E+04	0.374119E+01	0.491526E+03	0.183781E+01
0.191079E+04	0.374119E+01	0.491526E+03	0.183781E+01	0.191079E+04	0.374119E+01	0.491526E+03	0.183781E+01
0.408253E+03	0.	0.408253E+03	0.206240E-01	0.	0.	0.493269E+03	0.
0.360454E+01	0.493269E+03	0.238264E+00	0.189369E+04	0.360425E+01	0.493264E+03	0.238264E+00	0.
0.164454E+04	0.204005E+01	0.193729E+04	0.100000E+01	0.164454E+04	0.204005E+01	0.193729E+04	0.100000E+01
0.	0.	0.747856E+03	0.464873E+04	0.190823E+04	0.242134E+05	0.	0.943459E+04
0.288621E+05	0.943459E+04	0.824969E+01	0.608250E+03	0.137495E-01	0.382967E+05	0.382967E+05	0.775494E+00
0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	0.198844E+04	0.450082E+01	0.513736E+03	0.183651E+01
0.400004E+03	0.571429E+00	0.100000E+01	0.	0.100000E+01	0.100000E+01	0.310535E+03	0.120000E+03
0.220037E+01	0.897880E+00	0.112475E-01	0.201260E+02	0.	0.	0.100000E+03	0.833333E+00
0.100000E+03	0.157143E+01	0.870000E+00	0.400004E+03	0.100000E+01	0.100000E+01	0.100000E+01	0.120000E+03
0.220000E+01	0.900000E+00	0.507501E+00	0.998436E+00	0.	0.503851E+00	0.310535E+03	0.400004E+03
-0.	0.	0.612687E+03	0.400004E+03				
0.	0.100000E+00	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.985000E+00	0.	0.
0.	0.382967E+05	0.	0.	0.	0.	0.	0.
0.121397E+00	0.878613E+00	0.100000E+01	0.100000E+01	0.	0.000000E-38	0.382967E+05	0.
0.924713E+02	0.	0.275230E+03	0.197642E+01	0.	0.224256E+01	0.600000E-01	0.167563E+03

\$D(1),AM=.6,ALTP=25000,IAMTP=0,T4=2260, RUN AT ALTITUDE AT REDUCED T4-STD DAY

OUTPUT AM= 0.600 ALTP= 25000. T4= 2260.00 ETAR= 1.0000

MIDDLE SPOOL IS DUMMY

NO AIRFLOW INTO WING

PCNF	CNF	ZF	PRF	WAF	WAF
0.947947E+02	0.103558E+01	0.770832E+00	0.139853E+01	0.659213E+03	0.331175E+03
PCNI	CNI	ZI	PRI	WACI	WAI
0.100000E+01	0.	0.833333E+00	0.100000E+01	0.325217E+03	0.216035E+03
PCNC	CNC	ZC	PRC	WACC	WAC
0.939719E+02	0.102475E+01	0.832435E+00	0.122227E+02	0.325217E+03	0.216035E+03
T2	P2	T22	P22	T21	P21
0.460573E+03	0.473409E+00	0.515231E+03	0.662075E+00	0.515231E+03	0.662075E+00
T3	P3	PCBLF	BLF	PCBLC	BLC
0.113145E+04	0.809237E+01	0.	0.	0.	0.
PCBLHP	BLHP	PCBLIP	BLIP	PCBLLP	BLLP
0.	0.	0.	0.	0.	0.
W43	WFB	W44	FAR4	T4	P4
0.216035E+03	0.386023E+01	0.219895E+03	0.178685E-01	0.226000E+04	0.768966E+C1
TFFHP	CNHP	DHTCHP	DHTC	T50	P50
0.925046E+02	0.197671E+01	0.606063E-01	0.148082E+03	0.174181E+04	0.225796E+C1
TFFIP	CNIP	DHTCIP	DHTI	T5	P5
0.	0.	0.	0.	0.174181E+04	0.225796E+01
TFFLP	CNLP	DHTCLP	DHTF	T55	P55
0.276568E+03	0.227135E+01	0.173802E-01	0.195919E+02	0.167097E+04	0.187221E+01
ETAB	PCBLDU	ETAD	DPDUC	T24	P24
0.983000E+00	0.	0.	0.558184E-01	0.515231E+03	0.625119E+00
W4D	WFD	W424	FAR24	T25	P25
0.115140E+03	0.	0.115140E+03	0.	0.515231E+03	0.625119E+00
ETAF	ETAI	ETAC	ETATHP	ETATIP	ETATLP
0.847442E+00	0.100000E+01	0.855182E+00	0.898798E+00	0.	0.899885E+00
T6	P6	PS6	AM6	V6	W66
0.167097E+04	0.187221E+01	0.180570E+01	0.238717E+00	0.464280E+03	0.219895E+03
T7	WFA	W7	FAR7	ETAA	DPAFT
0.167097E+04	0.	0.219895E+03	0.178685E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9
0.101958E+01	0.100000E+01	0.181553E+04	0.101958E+01	0.100000E+01	0.181553E+04
PS28	AM28	V28	PS29	AM29	V29
0.371092E+00	0.895664E+00	0.925362E+03	0.371092E+00	0.895664E+00	0.925362E+C3
BPRINT	DPCOM	DPWING	PS38	AM38	V38
0.	0.497637E-01	0.	0.	0.	0.
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.532976E+00	0.	0.386023E+01	0.335035E+03	0.609938E+03	0.627824E+C4
PCBLI	W637	VJW	PS39	AM39	V39
-0.	0.	0.	0.	0.	0.
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.	0.	0.	0.150885E+05	0.625119E+00
FFOVFN	FWOVFN	FCOVFN	FMNOFN	FNVOFN	P38
0.715196E-01	0.	0.928480E+00	0.100000E+01	0.393990E+00	0.
CVMNOZ	VJM	CVDNOZ	VJD	FGM	FGP
0.985000E+00	0.178830E+04	0.985000E+00	0.911482E+03	0.154841E+05	0.588267E+C4

MAIN SONIC CONVERGENT NOZZLE FG= 21366.77

FN= 15088.53

SFC= 0.92102

DUCT SUBSONIC CONVERG. NOZZLE

CONVERGED AFTER 15 LOOPS

\$D(1),IDES=1, * NOW RUN 3 SPOOL, 3 STREAM AFTFAN ENGINE
 \$D(1),IDES=1,DUMMYSPOOL=.F.,AFTFAN=.T.,PCBLIDS=.5,IAMTP=2,T2=31,PRCDS=16/2.2,
 WAFCD=200*TROPICALDAY,WAICD=400*TROPICALDAY, * SEE FIG 6 FOR CHANGED AIRFLOWS
 AM=0,ALTP=0,PRIDS=2.2, * EXIT P FROM FAN=1.4,1ST COMP 2.2, 2ND COMP 16

FAN DESIGN PRFCF= 0.10000000E+01 ETAFCF= 0.10000000E+01 WAFCF= 0.34315033E+00 T2DS= 0.54966819E+03
 MIDDLE SPOOL DESIGN PRICF= 0.20000000E+01 ETAICF= 0.98863635E+00 WAICF= 0.13726013E+01 T22DS= 0.54966819E+03
 COMPRESSOR DESIGN PRCCF= 0.89610390E+00 ETACCF= 0.10000000E+01 WACCF= 0.10630393E+01 T21DS= 0.70920736E+03
 INTER DUCT DESIGN A38= 0.23855523E+01 AM38= 0.10000000E+01 A39= 0.23855523E+01 AM39= 0.10000000E+01
 COMBUSTOR DESIGN WA3CDS= 0.30797648E+02 ETABCF= 0.98300000E+00 DTCOCF= 0.10000000E+01
 H.P. TURBINE DESIGN CNHPCF= 0.10119289E+01 TFHPCF= 0.10815664E+01 ETHPCF= 0.10000000E+01 DHHPCF= 0.95650290E+00
 I.P. TURBINE DESIGN CNIPCF= 0.99847737E+00 TFIPCF= 0.10153566E+01 ETIPCF= 0.10000000E+01 DHIPCF= 0.16587348E+01
 L.P. TURBINE DESIGN CNLPCF= 0.97482939E+00 TFLPCF= 0.62291450E+00 ETLPCF= 0.10201339E+01 DHLPCF= 0.47955920E+00
 DUCT NOZZLE DESIGN A28= 0.37509597E+01 AM28= 0.65160510E+00 A29= 0.37509597E+01 AM29= 0.65160510E+00
 TURBINE AREA DESIGN A55= 0.79397937E+01 AM55= 0.23826586E+00
 NOZZLE DESIGN A8= 0.31030816E+01 AM8= 0.10000000E+01 A9= 0.31030816E+01 AM9= 0.10000000E+01
 OUTPUT AM= 0. ALTP= 0. T4= 2560.00 ETAR= 1.0000
 THREE SPOOL ENGINE

AFT-TURBOFAN

PCNF	CNF	ZF	PRF	WAF	WAF
0.100000E+03	0.100000E+01	0.833333E+00	0.140000E+01	0.205890E+03	0.200000E+03
PCNI	CNI	ZI	PRI	WACI	WAI
0.100000E+03	0.100000E+01	0.833333E+00	0.220000E+01	0.411780E+03	0.400000E+03
PCNC	CNC	ZC	PRC	WACC	WAC
0.100000E+03	0.100000E+01	0.814332E+00	0.727273E+01	0.106304E+03	0.200000E+03
T2	P2	T22	P22	T21	P21
0.549668E+03	0.100000E+01	0.612687E+03	0.140000E+01	0.709207E+03	0.220000E+01
T3	P3	PCBLF	BLF	PCBLC	BLC
0.131103E+04	0.160000E+02	0.	0.	0.	0.
PCBLHP	BLHP	PCBLIP	BLIP	PCBLLP	BLLP
0.	0.	0.	0.	0.	0.
WA3	WFB	WG4	FAR4	T4	P4
0.200000E+03	0.409834E+01	0.204098E+03	0.204917E-01	0.256000E+04	0.152000E+02
TFHP	CNHP	DHTCHP	DHTC	T50	P50
0.462292E+02	0.197642E+01	0.600000E-01	0.146980E+03	0.205983E+04	0.533327E+01
TFIP	CNIP	DHTCIP	DHTI	T5	P5
0.118185E+03	0.220335E+01	0.220000E-01	0.751676E+02	0.179639E+04	0.282050E+01
TFPL	CNLP	DHTCLP	DHTF	T55	P55
0.208696E+03	0.235939E+01	0.171750E-01	0.148082E+02	0.174370E+04	0.246262E+01
ETAB	PCBLDU	ETAD	DPDUC	T24	P24
0.983000E+00	0.	0.	0.500000E-01	0.612687E+03	0.133000E+01
WAD	WFD	WG24	FAR24	T25	P25
0.200000E+03	0.	0.200000E+03	0.	0.612687E+03	0.133000E+01
ETAF	ETAI	ETAC	ETATHP	ETATIP	ETATLP
0.880000E+00	0.870000E+00	0.860000E+00	0.900000E+00	0.900000E+00	0.900000E+00
T6	P6	PS6	AM6	V6	WG6
0.174370E+04	0.246262E+01	0.237214E+01	0.238266E+00	0.472437E+03	0.204098E+03
T7	WFA	WG7	FAR7	ETAA	OPAFT
0.174370E+04	0.	0.204098E+03	0.204917E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9
0.133970E+01	0.100000E+01	0.185307E+04	0.133970E+01	0.100000E+01	0.185307E+04
PS28	AM28	V28	PS29	AM29	V29
0.100000E+01	0.651605E+00	0.759245E+03	0.100000E+01	0.651605E+00	0.759245E+03
BPRINT	DPCOM	DPWING	PS38	AM38	V38
0.100000E+01	0.500000E-01	0.100000E+00	0.104873E+01	0.100000E+01	0.119217E+04
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.500000E+00	0.	0.409834E+01	0.604098E+03	0.	0.
PCBLI	WG37	VJW	PS39	AM39	V39
0.500000E+00	0.200000E+03	0.117429E+04	0.104873E+01	0.100000E+01	0.119217E+04
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.729961E+04	0.246010E+03	0.754562E+04	0.184583E+05	0.133000E+01
FFOVFN	FMOVFN	FCOVFN	FMNOFN	FNOVFD	P38
0.178774E+00	0.290172E+00	0.531054E+00	0.709828E+00	0.100000E+01	0.198000E+01
CVMNOZ	VJM	CVDNOZ	VJD	FGM	FGP
0.985000E+00	0.182527E+04	0.985000E+00	0.747856E+03	0.235272E+05	0.247676E+04

MAIN SCNIC CONVERGENT NOZZLE FG= 26003.93

FN= 26003.93

SFC= 0.56738

DUCT SUBSCNIC CONVERG. NOZZLE

CONVERGED AFTER 1 LOOPS

COMMON	0.833333E+00	0.100000E+03	0.833333E+00	0.100000E+03	0.814332E+00	0.100000E+03	0.256000E+04	0
0.100000E+03	0.100000E+03	0.256000E+04	0.100000E+03	0.	0.100000E+01	0.100000E+01	0.100000E+01	
0.833333E+00	0.100000E+03	0.140000E+01	0.880000E+00	0.200000E+03	0.100000E+01	0.100000E+01	0.100000E+01	C.343150E+00
0.814332E+00	0.100000E+03	0.727273E+01	0.860000E+00	0.200000E+03	0.896104E+00	0.100000E+01	0.100000E+01	0.106304E+01
0.256000E+04	0.	0.124897E+04	0.983000E+00	0.307776E+02	0.500000E-01	0.100000E+01	0.983000E+00	0.983000E+00
0.500000E+02	0.200000E+01	0.900000E+00	0.108157E+01	0.101193E+01	0.100000E+01	0.956903E+00	0.549668E+03	0.549668E+03
0.130000E+03	0.230000E+01	0.900000E+00	0.622915E+00	0.974829E+00	0.102013E+01	0.479959E+00	0.709207E+03	0.709207E+03
0.	0.	0.	0.	0.353607E+04	0.500000E-01	0.	0.	0.
0.	0.	0.	0.	0.346081E+04	0.	0.	0.	0.
0.793979E+01	0.	0.793979E+01	0.793979E+01	0.310308E+01	0.310308E+01	0.375096E+01	0.375096E+01	0.375096E+01
0.237214E+01	0.238266E+00	0.985000E+00	0.985000E+00	0.	0.	0.	0.	0.
0.549670E+03	0.100000E+01	0.131340E+03	0.160493E+01	0.549668E+03	0.100000E+01	0.131340E+03	0.160493E+01	0.160493E+01
0.709207E+03	0.220000E+01	0.169694E+03	0.161211E+01	0.131103E+04	0.160000E+02	0.319686E+03	0.162865E+01	0.162865E+01
0.256000E+04	0.152000E+02	0.681183E+03	0.182692E+01	0.179639E+04	0.282050E+01	0.459035E+03	0.183965E+01	0.183965E+01
0.174370E+04	0.246262E+01	0.444227E+03	0.184059E+01	0.	0.	0.	0.	0.
0.100000E+01	0.140000E+01	0.880000E+00	0.205890E+03	0.200000E+03	0.204098E+03	0.204098E+03	0.204917E-01	0.204917E-01
0.100000E+01	0.727273E+01	0.860000E+00	0.106304E+03	0.200000E+03	0.983000E+00	0.500000E-01	C.	C.
0.200000E+01	0.900000E+00	0.574142E-01	0.146980E+03	0.	0.204098E+03	0.204917E-01	0.114970E+04	0.114970E+04
0.230000E+01	0.900000E+00	0.824328E-02	0.148082E+02	0.	0.204098E+03	0.204917E-01	0.	0.
0.	0.	0.100000E+01	0.833333E+00	0.100000E+03	0.814332E+00	0.100000E+03	0.409834E+01	0.409834E+01
0.500000E+02	0.130000E+03	0.	0.	0.	0.	0.	0.	0.
0.100000E+01	0.200000E+03	0.200000E+03	0.	0.	0.319686E+03	0.	0.	0.
0.709207E+03	0.220000E+01	0.169694E+03	0.161211E+01	0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	0.160788E+01
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.161140E+01
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.161140E+01
0.200000E+03	0.	0.200000E+03	0.	0.	0.500000E-01	0.500000E+00	0.	0.
0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	0.651605E+00
0.174370E+04	0.246262E+01	0.444227E+03	0.184059E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.161140E+01
0.409834E+01	0.204098E+03	0.204917E-01	0.	0.200000E+03	0.	0.100000E+01	0.	0.
0.174370E+04	0.246262E+01	0.444227E+03	0.184059E+01	0.174370E+04	0.246262E+01	0.444229E+03	0.184059E+01	0.184059E+01
0.174370E+04	0.246262E+01	0.444229E+03	0.184059E+01	0.174370E+04	0.246262E+01	0.444229E+03	0.184059E+01	0.184059E+01
0.204098E+03	0.	0.204098E+03	0.204917E-01	0.	0.	0.472437E+03	0.	0.
0.237214E+01	0.472437E+03	0.238266E+00	0.172774E+04	0.237194E+01	0.472433E+03	0.238266E+00	0.	0.
0.149584E+04	0.133970E+01	0.185307E+04	0.100000E+01	0.149584E+04	0.133970E+01	0.185307E+04	0.100000E+01	0.100000E+01
0.	0.	0.747856E+03	0.464882E+04	0.182527E+04	0.115787E+05	0.	0.223075E+04	0.223075E+04
0.235272E+05	0.247676E+04	0.409834E+01	0.604098E+03	0.683057E-02	0.260039E+05	0.260039E+05	0.567377E+00	0.567377E+00
0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	0.205983E+04	0.533327E+01	0.534203E+03	0.183496E+01	0.183496E+01
0.400000E+03	0.833333E+00	0.100000E+03	0.100000E+01	0.220000E+01	0.870000E+00	0.411780E+03	0.120000E+03	0.120000E+03
0.220000E+01	0.900000E+00	0.364922E-01	0.751676E+02	0.	0.	0.100000E+03	0.833333E+00	0.833333E+00
0.100000E+03	0.220000E+01	0.870000E+00	0.400000E+03	0.200000E+01	0.988636E+00	0.137260E+01	0.120000E+03	0.120000E+03
0.220000E+01	0.900000E+00	0.101536E+01	0.998477E+00	0.	0.165873E+01	0.411780E+03	0.400000E+03	0.400000E+03
0.500000E+00	0.200000E+03	0.549668E+03	0.200000E+03					
0.200000E+03	0.100000E+00	0.100000E+00	0.242100E+04	0.238555E+01	0.100000E+01	0.119217E+04	0.709207E+03	0.709207E+03
0.169694E+03	0.198000E+01	0.591405E+03	0.104873E+01	0.709207E+03	0.169694E+03	0.198000E+01	0.591405E+03	0.591405E+03
0.119217E+04	0.100000E+01	0.238555E+01	0.100000E+01	0.200000E+03	0.985000E+00	0.725961E+04	0.246010E+03	0.246010E+03
0.754562E+04	0.184583E+05	0.290172E+00	0.104873E+01	0.	0.	0.	0.	0.
0.178774E+00	0.531054E+00	0.709828E+00	0.100000E+01	0.000000E-38	0.	0.260039E+05	0.500000E+00	0.500000E+00
0.462292E+02	0.118185E+03	0.208696E+03	0.197642E+01	0.220335E+01	0.235939E+01	0.600000E-01	0.146980E+03	0.146980E+03

\$D(1),AM=.6,ALTP=25000,IAMTP=0,T4=2260, * RUN AT ALTITUDE AT REDUCED T4-STD DAY

OUTPUT AM= 0.600 ALTP= 25000. T4= 2260.00 ETAR= 1.0000

THREE SPOOL ENGINE

AFT-TURBOFAN

PCNF	CNF	ZF	PRF	WAF	WAF
0.962072E+02	0.105101E+01	0.714685E+00	0.138097E+01	0.225997E+03	0.113536E+03
PCNI	CNI	ZI	PRI	WACI	WAI
0.934717E+02	0.102113E+01	0.836950E+00	0.226038E+01	0.425061E+03	0.213542E+03
PCNC	CNC	ZC	PRC	WACC	WAC
0.933542E+02	0.101276E+01	0.829775E+00	0.756550E+01	0.108849E+03	0.108062E+03
T2	P2	T22	P22	T21	P21
0.460573E+03	0.473409E+00	0.514072E+03	0.653762E+00	0.602600E+03	0.107009E+01
T3	P3	PCBLF	BLF	PCBLC	BLC
0.113975E+04	0.809574E+01	0.	0.	0.	0.
PCBLHP	BLHP	PCBLIP	BLIP	PCBLLP	BLLP
0.	0.	0.	0.	0.	0.
W43	WFB	WG4	FAR4	T4	P4
0.108062E+03	0.191758E+01	0.109980E+03	0.177451E-01	0.226000E+04	0.769271E+01
TFHP	CNHP	DHTCHP	DHTC	T50	P50
0.462476E+02	0.196372E+01	0.599475E-01	0.129591E+03	0.180776E+04	0.269097E+01
TFIP	CNIP	DHTCIP	DHTI	T5	P5
0.118243E+03	0.219841E+01	0.220249E-01	0.660349E+02	0.156967E+04	0.141927E+01
TFPL	CNLP	DHTCLP	DHTF	T55	P55
0.208907E+03	0.242830E+01	0.175304E-01	0.132117E+02	0.152127E+04	0.123532E+01
ETAB	PCBLDU	ETAD	DPDUC	T24	P24
0.983000E+00	0.	0.	0.555770E-01	0.514072E+03	0.617362E+00
WAD	WFD	WG24	FAR24	T25	P25
0.113536E+03	0.	0.113536E+03	0.	0.514072E+03	0.617362E+00
ETAF	ETAI	ETAC	ETATHP	ETATIP	ETATLP
0.831650E+00	0.853040E+00	0.855615E+00	0.899274E+00	0.899467E+00	0.898991E+00
T6	P6	PS6	AM6	V6	WG6

0.152127E+04	0.123532E+01	0.119153E+01	0.237865E+00	0.442666E+03	0.109980E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.152127E+04	0.	0.109980E+03	0.177451E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9
0.669006E+00	0.100000E+01	0.173448E+04	0.669006E+00	0.100000E+01	0.173448E+04
PS28	AM28	V28	PS29	AM29	V29
0.371092E+00	0.883805E+00	0.913767E+03	0.371092E+00	0.883805E+00	0.913767E+03
BPRINT	DPCDM	DPWING	PS38	AM38	V38
0.976098E+00	0.497825E-01	0.999471E-01	0.509290E+00	0.100000E+01	0.109892E+04
BYPASS	HPEXT	WFT	WG7	VA	FRD
0.531681E+00	0.	0.191758E+01	0.328996E+03	0.609938E+03	0.215236E+04
PCBLI	WG37	VJW	PS39	AM39	V39
0.493952E+00	0.105480E+03	0.108244E+04	0.509290E+00	0.100000E+01	0.109892E+04
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.354867E+04	0.697668E+03	0.224671E+04	0.108198E+05	0.617362E+00
FFOVFN	FMOVFN	FCOVFN	FMNOFN	FNOVFD	P38
0.783524E-01	0.171945E+00	0.749703E+00	0.828055E+00	0.502481E+00	0.963134E+00
CVMNOZ	VJM	CVDNOZ	VJD	FGM	FGP
0.985000E+00	0.170846E+04	0.985000E+00	0.900061E+03	0.125648E+05	0.265401E+04

MAIN SCNIC CONVERGENT NOZZLE
DUCT SUBSONIC CONVERG. NOZZLE

FG= 15218.84

FN= 13066.49

SFC= 0.52832

CONVERGED AFTER 24 LOOPS

\$D(1),IDES=1, * NOW RUN SUPERCHARGED COMPRESSOR AFTFAN, SEE FIGURE 7
\$D(1),IDES=1,AM=0,ALTP=0,IAMTP=2,T2=31,FXMIDDLETOCOMP=.T.,* AFTFAN STILL TRUE

FAN DESIGN	PRFCF= 0.10000000E+01	ETAFCF= 0.10000000E+01	WAFCF= 0.34315033E+00	T2DS= 0.54966819E+03
MIDDLE SPOOL DESIGN	PRICF= 0.20000000E+01	ETAICF= 0.98863635E+00	WAICF= 0.13726013E+01	T22DS= 0.54966819E+03
COMPRESSOR DESIGN	PRCCF= 0.10336619E+01	ETACCF= 0.10267859E+01	WACCF= 0.12042486E+01	T21DS= 0.70920736E+03
INTER DUCT DESIGN	A38= 0.23855523E+01	AM38= 0.10000000E+01	A39= 0.23855523E+01	AM39= 0.10000000E+01
COMBUSTOR DESIGN	WA3CDS= 0.30797647E+02	ETABCF= 0.98300000E+00	DTOCF= 0.10000000E+01	
I.P. TURBINE DESIGN	CNIPCF= 0.10119288E+01	TFIPCF= 0.10815664E+01	ETIPCF= 0.10000000E+01	DHIPCF= 0.14462754E+01
L.P. TURBINE DESIGN	CNLPFC= 0.97482939E+00	TFLPCF= 0.61068156E+00	ETLPCF= 0.10201339E+01	DHLPCF= 0.47995920E+00
DUCT NOZZLE DESIGN	A28= 0.37509597E+01	AM28= 0.65160510E+00	A29= 0.37509597E+01	AM29= 0.65160510E+00
TURBINE AREA DESIGN	A55= 0.80988405E+01	AM55= 0.23826586E+00		
NOZZLE DESIGN	A8= 0.31652414E+01	AM8= 0.10000000E+01	A9= 0.31652414E+01	AM9= 0.10000000E+01
CUTPUT	AM= 0.	ALTP= 0.	T4= 2560.00	ETAR= 1.0000

MIDDLE AND COMPRESSOR SPOOLS ARE ATTACHED , USE MIDDLE AND OUTER TURBINES

AFT-TURBOFAN

PCNF	CNF	ZF	PRF	WAF	WAF
0.100000E+03	0.100000E+01	0.833333E+00	0.140000E+01	0.205890E+03	0.200000E+03
PCNI	CNI	ZI	PRI	WACI	WAI
0.100000E+03	0.100000E+01	0.833333E+00	0.220000E+01	0.411780E+03	0.400000E+03
PCNC	CNC	ZC	PRC	WACC	WAC
0.929464E+02	0.929464E+00	0.814332E+00	0.727273E+01	0.106304E+03	0.200000E+03
T2	P2	T22	P22	T21	P21
0.549668E+03	0.100000E+01	0.612687E+03	0.140000E+01	0.709207E+03	0.220000E+01
T3	P3	PCBLF	BLF	PCBLC	BLC
0.131103E+04	0.160000E+02	0.	0.	0.	0.
PCBLHP	BLHP	PCBLIP	BLIP	PCBLLP	BLLP
0.	0.	0.	0.	0.	0.
WA3	WFB	WG4	FAR4	T4	P4
0.200000E+03	0.409834E+01	0.204098E+03	0.204917E-01	0.256000E+04	0.152000E+02
TFFHP	CNHP	DHTCHP	DHTC	T50	P50
0.	0.	0.	0.	0.256000E+04	0.152000E+02
TFFIP	CNIP	DHTCIP	DHTI	T5	P5
0.462292E+02	0.197642E+01	0.600000E-01	0.222148E+03	0.179639E+04	0.276511E+01
TFFLP	CNLP	DHTCLP	DHTF	T55	P55
0.212877E+03	0.235939E+01	0.171750E-01	0.148082E+02	0.174370E+04	0.241426E+01
ETAB	PCBLDU	ETAD	DPDUC	T24	P24
0.983000E+00	0.	0.	0.500000E-01	0.612687E+03	0.133000E+01
WAD	WFD	WG24	FAR24	T25	P25
0.200000E+03	0.	0.200000E+03	0.	0.612687E+03	0.133000E+01
ETAF	ETAI	ETAC	ETATHP	ETATIP	ETATLP
0.880000E+00	0.870000E+00	0.860000E+00	0.	0.900000E+00	0.900000E+00
T6	P6	PS6	AM6	V6	WG6
0.174370E+04	0.241426E+01	0.232556E+01	0.238266E+00	0.472437E+03	0.204098E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.174370E+04	0.	0.204098E+03	0.204917E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9

0.131339E+01	0.100000E+01	0.185307E+04	0.131339E+01	0.100000E+01	0.185307E+04
PS28	AM28	V28	PS29	AM29	V29
0.100000E+01	0.651605E+00	0.759245E+03	0.100000E+01	0.651605E+00	0.759245E+03
BPRINT	DPCOM	DPWING	PS38	AM38	V38
0.100000E+01	0.500000E-01	0.100000E+00	0.104873E+01	0.100000E+01	0.119217E+04
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.500000E+00	0.	0.409834E+01	0.604098E+03	0.	0.
PCBLI	WG37	VJW	PS39	AM39	V39
0.500000E+00	0.200000E+03	0.117429E+04	0.104873E+01	0.100000E+01	0.119217E+04
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.729961E+04	0.246010E+03	0.754562E+04	0.183268E+05	0.133000E+01
FFOVFN	FMOVFN	FCOVFN	FMOVFN	FNOVFD	P38
0.179683E+00	0.291648E+00	0.528670E+00	0.708352E+00	0.100000E+01	0.198000E+01
CVMNOZ	VJM	CVDNOZ	VJD	FGM	FGP
0.985000E+00	0.182527E+04	0.985000E+00	0.747856E+03	0.235272E+05	0.234521E+04

MAIN SCNIC CONVERGENT NOZZLE
DUCT SUBSONIC CONVERG. NOZZLE

FG= 25872.39

FN= 25872.39

SFC= 0.57026

CONVERGED AFTER 1 LOOPS

COMMON	0.833333E+00	0.100000E+03	0.833333E+00	0.100000E+03	0.814332E+00	0.929464E+02	0.256000E+04	0
0.100000E+03	0.100000E+03	0.256000E+04	0.100000E+03	0.	0.100000E+01	0.100000E+01	0.100000E+01	
0.833333E+00	0.100000E+03	0.140000E+01	0.880000E+00	0.200000E+03	0.100000E+01	0.100000E+01	0.343150E+00	
0.814332E+00	0.929464E+02	0.727273E+01	0.860000E+00	0.200000E+03	0.103366E+01	0.102679E+01	0.120425E+01	
0.256000E+04	0.	0.124897E+04	0.983000E+00	0.307976E+02	0.500000E-01	0.100000E+01	0.983000E+00	
0.500000E+02	0.200000E+01	0.900000E+00	0.	0.	0.	0.	0.549668E+03	
0.130000E+03	0.230000E+01	0.900000E+00	0.610682E+00	0.974829E+00	0.102013E+01	0.479959E+00	0.709207E+03	
0.	0.	0.	0.	0.353607E+04	0.500000E-01	0.	0.	
0.	0.	0.	0.	0.353014E+04	0.	0.	0.	
0.809884E+01	0.	0.809884E+01	0.809884E+01	0.316524E+01	0.316524E+01	0.375096E+01	0.375096E+01	
0.232556E+01	0.238266E+00	0.985000E+00	0.985000E+00	0.	0.	0.	0.	
0.549670E+03	0.100000E+01	0.131340E+03	0.160493E+01	0.549668E+03	0.100000E+01	0.131340E+03	0.160493E+01	
0.709207E+03	0.220000E+01	0.169694E+03	0.161211E+01	0.131103E+04	0.160000E+02	0.319686E+03	0.162865E+01	
0.256000E+04	0.152000E+02	0.681183E+03	0.182692E+01	0.179639E+04	0.276511E+01	0.459035E+03	0.184101E+01	
0.174370E+04	0.241426E+01	0.444227E+03	0.184195E+01	0.	0.	0.	0.	
0.100000E+01	0.140000E+01	0.880000E+00	0.205890E+03	0.200000E+03	0.200000E+03	0.204098E+03	0.204917E-01	
0.929464E+00	0.727273E+01	0.860000E+00	0.106304E+03	0.200000E+03	0.983000E+00	0.500000E-01	0.	
0.	0.	0.	0.	0.	0.204098E+03	0.204917E-01	0.114970E+04	
0.230000E+01	0.900000E+00	0.824328E-02	0.148082E+02	0.	0.204098E+03	0.204917E-01	0.	
0.	0.	0.100000E+01	0.833333E+00	0.100000E+03	0.814332E+00	0.929464E+02	0.409834E+01	
0.500000E+02	0.130000E+03	0.	0.	0.	0.	0.	0.	
0.100000E+01	0.200000E+03	0.200000E+03	0.	0.	0.319686E+03	0.	0.	
0.709207E+03	0.220000E+01	0.169694E+03	0.161211E+01	0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	
0.200000E+03	0.	0.200000E+03	0.	0.	0.500000E-01	0.500000E+00	0.	
0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	
0.174370E+04	0.241426E+01	0.444227E+03	0.184195E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	
0.409834E+01	0.204098E+03	0.204917E-01	0.	0.200000E+03	0.	0.100000E+01	0.	
0.174370E+04	0.241426E+01	0.444227E+03	0.184195E+01	0.174370E+04	0.241426E+01	0.444229E+03	0.184195E+01	
0.174370E+04	0.241426E+01	0.444229E+03	0.184195E+01	0.174370E+04	0.241426E+01	0.444229E+03	0.184195E+01	
0.204098E+03	0.	0.204098E+03	0.204917E-01	0.	0.	0.472437E+03	0.	
0.232556E+01	0.472437E+03	0.238266E+00	0.172774E+04	0.232536E+01	0.472433E+03	0.238266E+00	0.	
0.149584E+04	0.131339E+01	0.185307E+04	0.100000E+01	0.149584E+04	0.131339E+01	0.185307E+04	0.100000E+01	
0.	0.	0.747856E+03	0.464882E+04	0.182527E+04	0.115787E+05	0.	0.209920E+04	
0.235272E+05	0.234521E+04	0.409834E+01	0.604098E+03	0.683057E-02	0.258724E+05	0.258724E+05	0.570262E+00	
0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	0.256000E+04	0.152000E+02	0.681183E+03	0.182692E+01	
0.400000E+03	0.833333E+00	0.100000E+03	0.100000E+01	0.220000E+01	0.870000E+00	0.411780E+03	0.500000E+02	
0.200000E+01	0.900000E+00	0.867765E-01	0.222148E+03	0.	0.	0.100000E+03	0.833333E+00	
0.100000E+03	0.220000E+01	0.870000E+00	0.400000E+03	0.200000E+01	0.988636E+00	0.137260E+01	0.120000E+03	
0.220000E+01	0.900000E+00	0.108157E+01	0.101193E+01	0.	0.144628E+01	0.411780E+03	0.400000E+03	
0.500000E+00	0.200000E+03	0.549668E+03	0.200000E+03					
0.200000E+03	0.100000E+00	0.100000E+00	0.242100E+04	0.238555E+01	0.100000E+01	0.119217E+04	0.709207E+03	
0.169694E+03	0.198000E+01	0.591405E+03	0.104873E+01	0.709207E+03	0.169694E+03	0.198000E+01	0.591405E+03	
0.119217E+04	0.100000E+01	0.238555E+01	0.100000E+01	0.200000E+03	0.985000E+00	0.729961E+04	0.246010E+03	
0.754562E+04	0.183268E+05	0.291648E+00	0.104873E+01	0.	0.000000E-38	0.	0.	
0.179683E+00	0.528670E+00	0.708352E+00	0.100000E+01	0.000000E-38	0.	0.258724E+05	0.500000E+00	
0.	0.462292E+02	0.212877E+03	0.	0.197642E+01	0.235939E+01	0.	0.	

SD(1),AM=.6,ALTP=25000,IAMTP=0,T4=2260,† RUN AT ALTITUDE AT REDUCED T4-STD DAY

OUTPUT AM= 0.600 ALTP= 25000. T4= 2260.00 ETAR= 1.0000

MIDDLE AND COMPRESSOR SPOOLS ARE ATTACHED , USE MIDDLE AND OUTER TURBINES

AFT-TURBOFAN

PCNF	CNF	ZF	PRF	WAF	WAF
0.966327E+02	0.105566E+01	0.717809E+00	0.138610E+01	0.227254E+03	0.114168E+C3
PCNI	CNI	ZI	PR	WACI	WAI
0.938762E+02	0.102555E+01	0.835402E+00	0.226957E+01	0.428150E+03	0.215094E+C3
PCNC	CNC	ZC	PRC	WACC	WAC
0.873807E+02	0.946960E+00	0.827554E+00	0.761729E+01	0.109648E+03	0.109184E+C3
T2	P2	T22	P22	T21	P21
0.460573E+03	0.473409E+00	0.514862E+03	0.656193E+00	0.603867E+03	0.107444E+01
T3	P3	PCBLF	BLF	PCBLC	BLC
0.114009E+04	0.818429E+01	0.	0.	0.	0.
PCBLHP	BLHP	PCBLIP	BLIP	PCBLLP	BLLP
0.	0.	0.	0.	0.	0.
WA3	WFB	WG4	FAR4	T4	P4
0.109184E+03	0.193692E+01	0.111120E+03	0.177401E-01	0.226000E+04	0.777702E+01
TFHFP	CNHP	DHTCHP	DHTC	T50	P50
0.	0.	0.	0.	0.226000E+04	0.777702E+01
TFHFP	CNHP	DHTCIP	DHTI	T5	P5
0.462206E+02	0.197470E+01	0.599642E-01	0.195798E+03	0.156904E+04	0.140467E+C1
TFFLP	CNLP	DHTCLP	DHTF	T55	P55
0.213224E+03	0.243953E+01	0.175766E-01	0.133431E+02	0.152015E+04	0.122077E+01
ETAB	PCBLDU	ETAD	OPDUC	T24	P24
0.983000E+00	0.	0.	0.558221E-01	0.514862E+03	0.619563E+00
WAD	WFD	WG24	FAR24	T25	P25
0.114168E+03	0.	0.114168E+03	0.	0.514862E+03	0.619563E+C0
ETAF	ETAI	ETAC	ETATHP	ETATIP	ETATLP
0.829416E+00	0.850199E+00	0.862589E+00	0.	0.899947E+00	0.898741E+00
T6	P6	PS6	AM6	V6	WG6
0.152015E+04	0.122077E+01	0.117732E+01	0.238357E+00	0.443421E+03	0.111120E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.152015E+04	0.	0.111120E+03	0.177401E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9
0.662400E+00	0.100000E+01	0.173386E+04	0.662400E+00	0.100000E+01	0.173386E+C4
PS28	AM28	V28	PS29	AM29	V29
0.371092E+00	0.888725E+00	0.918858E+03	0.371092E+00	0.888725E+00	0.918858E+03
BPRINT	DPCOM	DPWING	PS38	AM38	V38
0.970021E+00	0.497621E-01	0.100054E+00	0.511908E+00	0.100000E+01	0.110007E+04
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.530781E+00	0.	0.193692E+01	0.331198E+03	0.609938E+03	0.216433E+04
PCBLI	WG37	VJW	PS39	AM39	V39
0.492391E+00	0.105910E+03	0.108357E+04	0.511908E+00	0.100000E+01	0.110007E+04
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.356689E+04	0.710889E+03	0.226999E+04	0.109048E+05	0.519563E+C0
FFOVFN	FDOVFN	FCOVFN	FMNOFN	FNOVFD	P38
0.794910E-01	0.172298E+00	0.748211E+00	0.827702E+00	0.509222E+00	0.966935E+00
CVMNOZ	VJM	CVDNOZ	VJD	FGM	FGP
0.985000E+00	0.170785E+04	0.985000E+00	0.905075E+03	0.126770E+05	0.266217E+04

MAIN SCNIC CONVERGENT NOZZLE FG= 15339.13 FN= 13174.80 SFC= 0.52926

DUCT SUBSONIC CONVERG. NOZZLE

CONVERGED AFTER 11 LOOPS

SD(1),IDES=1,† NOW RUN 2 SPOOL-2 STREAM ENGINE WITH ONLY 1 COMPRESSOR
SD(1),IDES=1,PRCDS=16,AM=0,ALTP=0,PCBLIDS=0,† FIXMIDLETQCOMP=.F.,PCNCDS=100,
PCNIDS=100,DUMMYSPOOL=.T.,IAMTP=2,T2=31, † SEE FIGURE 8

FAN DESIGN	PRFCF= 0.10000000E+01	ETAFCF= 0.10000000E+01	WAFCF= 0.34315033E+00	T2DS= 0.54966819E+03
MIDDLE SPOOL DESIGN	PRICF= 0.10000000E+01	ETAICF= 0.10000000E+01	WAICF= 0.10000000E+01	T22DS= 0.54966819E+03
COMPRESSOR DESIGN	PRCCF= 0.21428571E+01	ETACCF= 0.10000000E+01	WACCF= 0.41178042E+01	T21DS= 0.54966819E+03
COMBUSTOR DESIGN	WA3CDS= 0.61230686E+02	ETABCF= 0.98300000E+00	DTCCCF= 0.10000000E+01	
H.P. TURBINE DESIGN	CNHPCF= 0.10119289E+01	TFHPCF= 0.54065683E+00	ETHPCF= 0.10000000E+01	DHHPCF= 0.11758439E+01
L.P. TURBINE DESIGN	CNLPFC= 0.10138291E+01	TFLPCF= 0.42747655E+00	ETLPCF= 0.10201339E+01	DHLPCF= 0.22181950E+00
DUCT NOZZLE DESIGN	A2B= 0.37509597E+01	AM2B= 0.65160510E+00	A29= 0.37509597E+01	AM29= 0.65160510E+00
TURBINE AREA DESIGN	A55= 0.10866958E+02	AM55= 0.23826395E+00		
NOZZLE DESIGN	A8= 0.42454950E+01	AM8= 0.10000000E+01	A9= 0.42454950E+01	AM9= 0.10000000E+01
OUTPUT	AM= 0.	ALTP= 0.	T4= 2560.00	ETAR= 1.0000

MIDDLE SPOOL IS DUMMY

NO AIRFLOW INTO WING , AFT-TURBOFAN

PCNF	CNF	ZF	PRF	WAFc	WAF
0.100000E+03	0.100000E+01	0.833333E+00	0.140000E+01	0.205890E+03	0.200000E+03
PCNI	CNI	ZI	PRi	WACI	WAI
0.100000E+01	0.	0.571429E+00	0.100000E+01	0.411780E+03	0.400000E+03
PCNC	CNC	ZC	PRC	WACC	WAC
0.100000E+03	0.100000E+01	0.814332E+00	0.160000E+02	0.411780E+03	0.400000E+03
T2	P2	T22	P22	T21	P21
0.549668E+03	0.100000E+01	0.612687E+03	0.140000E+01	0.549668E+03	0.100000E+01
T3	P3	PCBLF	BLF	PCBLC	BLC
0.129556E+04	0.160000E+02	0.	0.	0.	0.
PCRLHP	RLHP	PCBLIP	BLIP	PCBLLP	BLLP
0.	0.	0.	0.	0.	0.
WA3	WFB	WG4	FAR4	T4	P4
0.400000E+03	0.829205E+01	0.408292E+03	0.207301E-01	0.256000E+04	0.152000E+02
TFFHP	CNHP	DHTCHP	DHTC	T50	P50
0.924801E+02	0.197642E+01	0.600000E-01	0.180610E+03	0.194300E+04	0.402696E+01
TFFIP	CNIP	DHTCIP	DHTI	T5	P5
0.	0.	0.	0.	0.194300E+04	0.402696E+01
TFFLP	CNLP	DHTCLP	DHTF	T55	P55
0.304110E+03	0.226863E+01	0.171750E-01	0.740236E+01	0.191708E+04	0.378435E+01
ETAB	PCBLDU	ETAD	OPDUC	T24	P24
0.983000E+00	0.	0.	0.500000E-01	0.612687E+03	0.133000E+01
WAD	WFD	WG24	FAR24	T25	P25
0.200000E+03	0.	0.200000E+03	0.	0.612687E+03	0.133000E+01
ETAF	ETAI	ETAC	ETATHP	ETATIP	ETATLP
0.880000E+00	0.100000E+01	0.860000E+00	0.900000E+00	0.	0.900000E+00
T6	P6	PS6	AM6	V6	WG6
0.191708E+04	0.378435E+01	0.364616E+01	0.238264E+00	0.494024E+03	0.408292E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.191708E+04	0.	0.408292E+03	0.207301E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9
0.206379E+01	0.100000E+01	0.194037E+04	0.205379E+01	0.100000E+01	0.194037E+04
PS28	AM28	V28	PS29	AM29	V29
0.100000E+01	0.651605E+00	0.759245E+03	0.100000E+01	0.651605E+00	0.759245E+03
BPRINT	DPCQM	DPWING	PS38	AM38	V38
0.	0.500000E-01	0.	0.	0.	0.
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.500000E+00	0.	0.829205E+01	0.608292E+03	0.	0.
PCBLI	WG37	VJW	PS39	AM39	V39
-0.	0.	0.	0.	0.	0.
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.	0.	0.	0.384605E+05	0.133000E+01
FDOVFN	FWDVFN	FCOVFN	FMNOFN	FNOVFD	P38
0.120872E+00	0.	0.879128E+00	0.100000E+01	0.100000E+01	0.
CVMNOZ	VJM	CVDOZ	VJD	FGM	FGP
0.985000E+00	0.191127E+04	0.985000E+00	0.747856E+03	0.289030E+05	0.955753E+04

MAIN SONIC CONVERGENT NOZZLE
DUCT SUBSONIC CONVERG. NOZZLE

FG= 38460.51

FN= 38460.51

SFC= 0.77616

CONVERGED AFTER 1 LOOPS

COMMON	0.833333E+00	0.100000E+03	0.571429E+00	0.100000E+01	0.814332E+00	0.100000E+03	0.256000E+04	0
0.100000E+03	0.100000E+03	0.256000E+04	0.100000E+03	0.	0.100000E+01	0.100000E+01	0.100000E+01	0.100000E+01
0.833333E+00	0.100000E+03	0.140000E+01	0.880000E+00	0.200000E+03	0.100000E+01	0.100000E+01	0.100000E+01	0.243150E+00
0.814332E+00	0.100000E+03	0.160000E+02	0.860000E+00	0.400000E+03	0.214286E+01	0.100000E+01	0.100000E+01	0.411780E+01
0.256000E+04	0.	0.126444E+04	0.983000E+00	0.612307E+02	0.500000E-01	0.100000E+01	0.100000E+01	0.983000E+00
0.500000E+02	0.200000E+01	0.900000E+00	0.540657E+00	0.101193E+01	0.100000E+01	0.117584E+01	0.117584E+01	0.549668E+03
0.130000E+03	0.230000E+01	0.900000E+00	0.427477E+00	0.101383E+01	0.102013E+01	0.221820E+00	0.221820E+00	0.549668E+03
0.	0.	0.	0.	0.353607E+04	0.500000E-01	0.	0.	0.
0.	0.	0.	0.	0.472389E+04	0.	0.	0.	0.
0.108670E+02	0.	0.108670E+02	0.108670E+02	0.424549E+01	0.424549E+01	0.375096E+01	0.375096E+01	0.375096E+01
0.364616E+01	0.238264E+00	0.985000E+00	0.985000E+00	0.	0.	0.	0.	0.
0.549670E+03	0.100000E+01	0.131340E+03	0.160493E+01	0.549658E+03	0.100000E+01	0.131340E+03	0.160493E+01	0.160493E+01
0.549668E+03	0.100000E+01	0.131340E+03	0.160493E+01	0.129556E+04	0.160000E+02	0.315694E+03	0.162558E+01	0.162558E+01
0.256000E+04	0.152000E+02	0.681393E+03	0.182706E+01	0.194300E+04	0.402696E+01	0.500783E+03	0.183758E+01	0.183758E+01
0.191708E+04	0.378435E+01	0.493381E+03	0.183800E+01	0.	0.	0.	0.	0.
0.100000E+01	0.140000E+01	0.880000E+00	0.205890E+03	0.200000E+03	0.400000E+03	0.408292E+03	0.207301E-01	0.207301E-01
0.100000E+01	0.160000E+02	0.860000E+00	0.411780E+03	0.400000E+03	0.983000E+00	0.500000E-01	0.	0.
0.200000E+01	0.900000E+00	0.705506E-01	0.180610E+03	0.	0.408292E+03	0.207301E-01	0.114970E+04	0.114970E+04
0.230000E+01	0.900000E+00	0.380975E-02	0.740236E+01	0.	0.408292E+03	0.207301E-01	0.	0.
0.	0.	0.100000E+01	0.833333E+00	0.100000E+03	0.814332E+00	0.100000E+03	0.829205E+01	0.829205E+01
0.500000E+02	0.130000E+03	0.	0.	0.	0.	0.	0.	0.
0.100000E+01	0.200000E+03	0.400000E+03	0.	0.	0.315694E+03	0.	0.	0.
0.549668E+03	0.100000E+01	0.131340E+03	0.160493E+01	0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	0.160788E+01
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.161140E+01
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.161140E+01
0.200000E+03	0.	0.200000E+03	0.	0.	0.500000E-01	0.500000E+00	0.	0.
0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	0.564699E+03	0.100000E+01	0.759245E+03	0.651605E+00	0.651605E+00

0.191708E+04	0.378435E+01	0.493381E+03	0.183800E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01
0.829205E+01	0.408292E+03	0.207301E-01	0.	0.200000E+03	0.	0.100000E+01	0.
0.191708E+04	0.378435E+01	0.493381E+03	0.183800E+01	0.191708E+04	0.378435E+01	0.493380E+03	0.183800E+01
0.191708E+04	0.378435E+01	0.493380E+03	0.183800E+01	0.191708E+04	0.378435E+01	0.493380E+03	0.183800E+01
0.408292E+03	0.	0.408292E+03	0.207301E-01	0.	0.	0.494024E+03	0.
0.364616E+01	0.494024E+03	0.238264E+00	0.189994E+04	0.364586E+01	0.494020E+03	0.238264E+00	0.
0.165020E+04	0.206379E+01	0.194037E+04	0.100000E+01	0.165020E+04	0.206379E+01	0.194037E+04	0.100000E+01
0.	0.	0.747856E+03	0.464882E+04	0.191127E+04	0.242542E+05	0.	0.955753E+04
0.289030E+05	0.955753E+04	0.829205E+01	0.608292E+03	0.138201E-01	0.384605E+05	0.384605E+05	0.776157E+00
0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	0.194300E+04	0.402696E+01	0.500783E+03	0.183758E+01
0.400000E+03	0.571429E+00	0.100000E+01	0.	0.100000E+01	0.100000E+01	0.411780E+03	0.120000E+03
0.199826E+01	0.899947E+00	0.867247E-01	0.195798E+03	0.	0.	0.100000E+03	0.833333E+00
0.100000E+03	0.220000E+01	0.870000E+00	0.400000E+03	0.100000E+01	0.100000E+01	0.100000E+01	0.120000E+03
0.220000E+01	0.900000E+00	0.108157E+01	0.101193E+01	0.	0.144628E+01	0.411780E+03	0.400000E+03
-0.	0.	0.549668E+03	0.400000E+03				
0.	0.100000E+00	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.985000E+00	0.	0.
0.	0.384605E+05	0.	0.	0.	0.	0.	0.
0.120872E+00	0.879128E+00	0.100000E+01	0.100000E+01	0.000000E-38	0.000000E-38	0.384605E+05	0.
0.924801E+02	0.	0.304110E+03	0.197642E+01	0.	0.226863E+01	0.600000E-01	0.180610E+03

\$D(1),AM=.6,ALTP=25000,IAMTP=0,T4=2260,† RUN AT ALTITUDE AT REDUCED T4-STD DAY

OUTPUT AM= 0.600 ALTP= 25000. T4= 2260.00 ETAR= 1.0000

MIDDLE SPOOL IS DUMMY

NO AIRFLOW INTO WING , AFT-TURBOFAN

PCNF	CNF	ZF	PRF	WAF	WAF
0.965218E+02	0.105445E+01	0.719983E+00	0.138636E+01	0.226850E+03	0.113965E+03
PCNI	CNI	ZI	PRI	WACI	WAI
0.100000E+01	0.	0.833333E+00	0.100000E+01	0.435673E+03	0.218873E+03
PCNC	CNC	ZC	PRC	WACC	WAC
0.943035E+02	0.103022E+01	0.832563E+00	0.173236E+02	0.435673E+03	0.218873E+03
T2	P2	T22	P22	T21	P21
0.460573E+03	0.473409E+00	0.514835E+03	0.656317E+00	0.460573E+03	0.473409E+00
T3	P3	PCBLF	BLF	PCBLC	BLC
0.112613E+04	0.820116E+01	0.	0.	0.	0.
PCBLHP	BLHP	PCBLIP	BLIP	PCBLLP	BLLP
0.	0.	0.	0.	0.	0.
WA3	WFB	WG4	FAR4	T4	P4
0.218873E+03	0.392827E+01	0.222802E+03	0.179477E-01	0.226000E+04	0.779304E+01
TFFHP	CNHP	DHTCHP	DHTC	T50	P50
0.924839E+02	0.198369E+01	0.604594E-01	0.159594E+03	0.170054E+04	0.204749E+01
TFFIP	CNIP	DHTCIP	DHTI	T5	P5
0.	0.	0.	0.	0.170054E+04	0.204749E+01
TFFLP	CNLP	DHTCLP	DHTF	T55	P55
0.305344E+03	0.234063E+01	0.176003E-01	0.663962E+01	0.167661E+04	0.192098E+01
ETAB	PCBLDU	ETAD	DPDUC	T24	P24
0.983000E+00	0.	0.	0.557109E-01	0.514835E+03	0.619753E+00
WAD	WFD	WG24	FAR24	T25	P25
0.113965E+03	0.	0.113965E+03	0.	0.514835E+03	0.619753E+00
ETAF	ETAI	ETAC	ETATHP	ETATIP	ETATLP
0.830336E+00	0.100000E+01	0.855328E+00	0.899537E+00	0.	0.899171E+00
T6	P6	PS6	AM6	V6	WG6
0.167661E+04	0.192098E+01	0.185293E+01	0.238418E+00	0.464430E+03	0.222802E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.167661E+04	0.	0.222802E+03	0.179477E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9
0.104502E+01	0.100000E+01	0.181849E+04	0.104502E+01	0.100000E+01	0.181849E+04
PS28	AM28	V28	PS29	AM29	V29
0.371092E+00	0.887317E+00	0.917579E+03	0.371092E+00	0.887317E+00	0.917579E+03
BPRINT	DPCOM	DPWING	PS38	AM38	V38
0.	0.497637E-01	0.	0.	0.	0.
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.520689E+00	0.	0.392827E+01	0.336767E+03	0.609938E+03	0.216048E+04
PCBLI	WG37	VJW	PS39	AM39	V39
-0.	0.	0.	0.	0.	0.
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.	0.	0.	0.194998E+05	0.619753E+00
FFOVFN	FWOVFN	FCOVFN	FMNOFN	FNOVFD	P38
0.533830E-01	0.	0.946617E+00	0.100000E+01	0.507007E+00	0.
CVMNQZ	VJM	CVDNQZ	VJD	FGM	FGP
0.985000E+00	0.179122E+04	0.985000E+00	0.903816E+03	0.156054E+05	0.605485E+04

MAIN SCNIC CONVERGENT NOZZLE
DUCT SUBSONIC CONVERG. NOZZLE

FG= 21660.25

FN= 19499.77

SFC= 0.72523

CONVERGED AFTER 10 LOOPS

\$D(1),IDES=1,* NOW RUN 3 SPOOL- 2 STREAM AFTFAN, FIGURE 9
 \$D(1),IDES=1,IAMTP=2,T2=31,AM=0,ALTP=0,DUMMYSPOOL=.F.,PCNIDS=100,
 PRIDS=2,2,PRCDS=16/2,2,

FAN DESIGN PRFCF= 0.10000000E+01 ETAFCF= 0.10000000E+01 WAFCF= 0.34315033E+00 T2DS= 0.54966819E+03
 MIDDLE SPOOL DESIGN PRICF= 0.20000000E+01 ETAICF= 0.98863635E+00 WAICF= 0.13726013E+01 T22DS= 0.54966819E+03
 COMPRESSOR DESIGN PRCCF= 0.89610390E+00 ETACCF= 0.10000000E+01 WACCF= 0.21260785E+01 T21DS= 0.70920736E+03
 COMBUSTOR DESIGN WA3CDS= 0.61595295E+02 ETABCF= 0.98300000E+00 DTCOCF= 0.10000000E+01
 H.P. TURBINE DESIGN CNHPCF= 0.10119289E+01 TFHPCF= 0.54078318E+00 ET1PCF= 0.10000000E+01 DHHPCF= 0.95690290E+00
 I.P. TURBINE DESIGN CNIPCF= 0.99847737E+00 TFIPCF= 0.50767828E+00 ET1PCF= 0.10000000E+01 DHIPCF= 0.82936738E+00
 L.P. TURBINE DESIGN CNLPCF= 0.10101420E+01 TFLPCF= 0.41838630E+00 ET1PCF= 0.10201339E+01 DHLPCF= 0.22349441E+00
 DUCT NOZZLE DESIGN A28= 0.37509597E+01 AM28= 0.65160510E+00 A29= 0.37509597E+01 AM29= 0.65160510E+00
 TURBINE AREA DESIGN A55= 0.11104701E+02 AM55= 0.23826402E+00
 NOZZLE DESIGN A8= 0.43385243E+01 AM8= 0.10000000E+01 A9= 0.43385243E+01 AM9= 0.10000000E+01
 OUTPUT AM= 0. ALTP= 0. T4= 2560.00 ETAR= 1.0000
 THREE SPOOL ENGINE

NO AIRFLOW INTO WING , AFT-TURBOFAN
 PCNF CNF ZF PRF WAF WAF
 0.100000E+03 0.100000E+01 0.833333E+00 0.140000E+01 0.205890E+03 0.200000E+03
 PCNI CNF ZI PRI WACI WAI
 0.100000E+03 0.100000E+01 0.833333E+00 0.220000E+01 0.411780E+03 0.400000E+03
 PCNC CNC ZC PRC WACC WAC
 0.100000E+03 0.100000E+01 0.814332E+00 0.727273E+01 0.212608E+03 0.400000E+03
 T2 P2 T22 P22 T21 P21
 0.549668E+03 0.100000E+01 0.612687E+03 0.140000E+01 0.709207E+03 0.220000E+01
 T3 P3 PCBLF BLF PCBLC BLC
 0.131103E+04 0.160000E+02 0. PCBLIP BLIP PCBLLP BLLP
 0. WFB WG4 FAR4 T4 P4
 0.400000E+03 0.819668E+01 0.408197E+03 0.204917E-01 0.256000E+04 0.152000E+02
 TFFHP CNHP DHTCHP DHTC T50 P50
 0.924585E+02 0.197642E+01 0.600000E-01 0.146980E+03 0.205983E+04 0.533327E+C1
 TFFIP CNIP DHTCIP DHTI T5 P5
 0.23637CE+03 0.220335E+01 0.220000E-01 0.375838E+02 0.192890E+04 0.392608E+01
 TFFLP CNLP DHTCLP DHTF T55 P55
 0.310718E+03 0.227691E+01 0.171750E-01 0.740409E+01 0.190292E+04 0.36878CE+01
 ETAB PCBLDU ETAD DPDU T24 P24
 0.983000E+00 0. WFD WG24 FAR24 T25 P25
 0.200000E+03 0. ETAT ETAC ETATHP ETATIP ETATLP
 0.880000E+00 0.870000E+00 0.860000E+00 0.900000E+00 0.900000E+00 0.900000E+00
 T6 P6 PS6 AM6 V6 WG6
 0.190292E+04 0.368780E+01 0.355306E+01 0.238264E+00 0.492323E+03 0.408197E+03
 T7 WFA WG7 FAR7 ETAA DPAFT
 0.190292E+04 0. 0.408197E+03 0.204917E-01 0. 0.
 PS8 AM8 V8 PS9 AM9 V9
 0.201067E+01 0.100000E+01 0.193344E+04 0.201067E+01 0.100000E+01 0.193344E+04
 PS28 AM28 V28 PS29 AM29 V29
 0.100000E+01 0.651605E+00 0.759245E+03 0.100000E+01 0.651605E+00 0.759245E+C3
 BPRINT DPCOM DPWING PS38 AM38 V38
 0. 0.500000E-01 0. WFT WGT VA FRD
 0.500000E+00 0. HPEXT 0.819668E+01 0.608197E+03 0. AM39 V39
 0.149012E-07 0. WG37 VJW PS39 0. FNWING FNMAIN P28
 0.985000E+00 0. FGMWNG FGPWNG FMNOFN FNOVFD P38
 FFOVFN FCOVFN FMNOFN FNOVFD P38
 0.122049E+00 0. 0.877952E+00 0.100000E+01 0.100000E+01 0.
 CVMNOZ VJM CVDNOZ VJD FGM FGP
 0.985000E+00 0.190444E+04 0.985000E+00 0.747856E+03 0.288107E+05 0.927924E+04

MAIN SCNIC CONVERGENT NOZZLE FG= 38089.93 FN= 38089.93 SFC= 0.7745
 DUCT SUBSONIC CONVERG. NOZZLE
 CONVERGED AFTER 1 LOOPS

COMMON	0.833333E+00	0.100000E+03	0.833333E+00	0.100000E+03	0.814332E+00	0.100000E+03	0.256000E+04	0
0.100000E+03	0.100000E+03	0.256000E+04	0.100000E+03	0.	0.100000E+01	0.100000E+01	0.100000E+01	0.100000E+01
0.833333E+00	0.100000E+03	0.140000E+01	0.880000E+00	0.200000E+03	0.100000E+01	0.100000E+01	0.100000E+01	0.343150E+00
0.814332E+00	0.100000E+03	0.727273E+01	0.860000E+00	0.400000E+03	0.896104E+00	0.100000E+01	0.100000E+01	0.212608E+01
0.256000E+04	0.	0.124897E+04	0.983000E+00	0.615953E+02	0.500000E-01	0.100000E+01	0.100000E+01	0.983000E+00
0.500000E+02	0.200000E+01	0.900000E+00	0.540783E+00	0.101193E+01	0.100000E+01	0.956903E+00	0.956903E+00	0.549668E+03
0.130000E+03	0.230000E+01	0.900000E+00	0.418386E+00	0.101014E+01	0.102013E+01	0.223494E+00	0.223494E+00	0.709207E+03
0.	0.	0.	0.	0.353607E+04	0.500000E-01	0.	0.	0.
0.	0.	0.	0.	0.482851E+04	0.	0.	0.	0.
0.111047E+02	0.	0.111047E+02	0.111047E+02	0.433852E+01	0.433852E+01	0.375096E+01	0.375096E+01	0.375096E+01
0.355306E+01	0.238264E+00	0.985000E+00	0.985000E+00	0.	0.	0.	0.	0.
0.549670E+03	0.100000E+01	0.131340E+03	0.160493E+01	0.549668E+03	0.100000E+01	0.131340E+03	0.160493E+01	0.160493E+01
0.709207E+03	0.220000E+01	0.169694E+03	0.161211E+01	0.131103E+04	0.160000E+02	0.319686E+03	0.319686E+03	0.162865E+01
0.256000E+04	0.152000E+02	0.681183E+03	0.182692E+01	0.192890E+04	0.392608E+01	0.496619E+03	0.496619E+03	0.183713E+01
0.190292E+04	0.368780E+01	0.489215E+03	0.183756E+01	0.	0.	0.	0.	0.
0.100000E+01	0.140000E+01	0.880000E+00	0.205890E+03	0.200000E+03	0.400000E+03	0.408197E+03	0.408197E+03	0.204917E-01
0.100000E+01	0.727273E+01	0.860000E+00	0.212608E+03	0.400000E+03	0.983000E+00	0.500000E-01	0.500000E-01	0.
0.200000E+01	0.900000E+00	0.574142E-01	0.146980E+03	0.	0.408197E+03	0.204917E-01	0.204917E-01	0.114970E+04
0.230000E+01	0.900000E+00	0.383851E-02	0.740409E+01	0.	0.408197E+03	0.204917E-01	0.204917E-01	0.
0.	0.	0.100000E+01	0.833333E+00	0.100000E+03	0.814332E+00	0.100000E+03	0.100000E+03	0.819668E+01
0.500000E+02	0.130000E+03	0.	0.	0.	0.	0.	0.	0.
0.100000E+01	0.200000E+03	0.400000E+03	0.	0.	0.319686E+03	0.	0.	0.
0.709207E+03	0.220000E+01	0.169694E+03	0.161211E+01	0.612687E+03	0.140000E+01	0.146452E+03	0.146452E+03	0.160788E+01
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.146451E+03	0.161140E+01
0.612687E+03	0.133000E+01	0.146451E+03	0.161140E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.146451E+03	0.161140E+01
0.200000E+03	0.	0.200000E+03	0.	0.	0.500000E-01	0.500000E+00	0.500000E+00	0.
0.564639E+03	0.100000E+01	0.759245E+03	0.651605E+00	0.564699E+03	0.100000E+01	0.759245E+03	0.759245E+03	0.651605E+00
0.190292E+04	0.368780E+01	0.489215E+03	0.183756E+01	0.612687E+03	0.133000E+01	0.146451E+03	0.146451E+03	0.161140E+01
0.819668E+01	0.408197E+03	0.204917E-01	0.	0.200000E+03	0.	0.100000E+01	0.100000E+01	0.
0.190292E+04	0.368780E+01	0.489215E+03	0.183756E+01	0.190292E+04	0.368780E+01	0.489214E+03	0.489214E+03	0.183756E+01
0.190292E+04	0.368780E+01	0.489214E+03	0.183756E+01	0.190292E+04	0.368780E+01	0.489214E+03	0.489214E+03	0.183756E+01
0.408197E+03	0.	0.408197E+03	0.204917E-01	0.	0.	0.492323E+03	0.492323E+03	0.
0.355306E+01	0.492323E+03	0.238264E+00	0.188588E+04	0.355277E+01	0.492318E+03	0.238264E+00	0.238264E+00	0.
0.163748E+04	0.201067E+01	0.193344E+04	0.100000E+01	0.163748E+04	0.201067E+01	0.193344E+04	0.193344E+04	0.100000E+01
0.	0.	0.747856E+03	0.464882E+04	0.190444E+04	0.246119E+05	0.	0.	0.927924E+04
0.288107E+05	0.927924E+04	0.819668E+01	0.608197E+03	0.136611E-01	0.380899E+05	0.380899E+05	0.380899E+05	0.774694E+00
0.612687E+03	0.140000E+01	0.146452E+03	0.160788E+01	0.205983E+04	0.533327E+01	0.534203E+03	0.534203E+03	0.183496E+01
0.400000E+03	0.833333E+00	0.100000E+03	0.100000E+01	0.220000E+01	0.870000E+00	0.411780E+03	0.411780E+03	0.120000E+03
0.220000E+01	0.900000E+00	0.182461E-01	0.375838E+02	0.	0.	0.100000E+03	0.100000E+03	0.833333E+00
0.100000E+03	0.220000E+01	0.870000E+00	0.400000E+03	0.200000E+01	0.988636E+00	0.137260E+01	0.137260E+01	0.120000E+03
0.220000E+01	0.900000E+00	0.507678E+00	0.998477E+00	0.	0.829367E+00	0.411780E+03	0.411780E+03	0.400000E+03
0.149012E-07	0.	0.549668E+03	0.400000E+03	0.	0.	0.	0.	0.
0.381470E-05	0.100000E+00	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.985000E+00	0.	0.	0.
0.	0.380899E+05	0.	0.	0.	0.	0.	0.	0.
0.122048E+00	0.877952E+00	0.100000E+01	0.100000E+01	0.000000E-38	0.	0.380899E+05	0.380899E+05	0.
0.924585E+02	0.236370E+03	0.310718E+03	0.197642E+01	0.220335E+01	0.227691E+01	0.600000E-01	0.600000E-01	0.146980E+03

SD(1),AM=.6,ALTP=25000,IAMTP=0,T4=2260,† RUN AT ALTITUDE AT REDUCED T4-STD DAY

OUTPUT AM= 0.600 ALTP= 25000. T4= 2260.00 ETAR= 1.0000

THREE SPOOL ENGINE

NO AIRFLOW INTO WING , AFT-TURBOFAN

PCNF	CNF	ZF	PRF	WAF	WAF
0.962024E+02	0.105096E+01	0.714546E+00	0.138085E+01	0.225986E+03	0.113531E+03
PCNI	CNI	ZI	PRI	WACI	WAI
0.937649E+02	0.102433E+01	0.822534E+00	0.224690E+01	0.428468E+03	0.215254E+03
PCNC	CNC	ZC	PRC	WACC	WAC
0.933548E+02	0.101339E+01	0.830442E+00	0.757939E+01	0.217951E+03	0.215216E+03
T2	P2	T22	P22	T21	P21
0.460573E+03	0.473409E+00	0.514057E+03	0.653708E+00	0.601857E+03	0.106370E+01
T3	P3	PCBLF	BLF	PCBLC	BLC
0.113915E+04	0.806220E+01	0.	0.	0.	0.
PCBLHP	BLHP	PCBLIP	BLIP	PCBLLP	BLLP
0.	0.	0.	0.	0.	0.
W43	WFR	WG4	FAR4	T4	P4
0.215216E+03	0.382096E+01	0.219037E+03	0.177540E-01	0.226000E+04	0.766098E+01
TFFHP	CNHP	DHTCHP	DHTC	T50	P50
0.924888E+02	0.196373E+01	0.599702E-01	0.129616E+03	0.180768E+04	0.267912E+01
TFFIP	CNIP	DHTCIP	DHTI	T5	P5
0.236531E+03	0.220536E+01	0.221711E-01	0.332470E+02	0.168858E+04	0.196577E+01
TFFLP	CNLP	DHTCLP	DHTF	T55	P55
0.311565E+03	0.234113E+01	0.175597E-01	0.663149E+01	0.166464E+04	0.184363E+01
ETAB	PCBLDU	ETAD	DPDUC	T24	P24
0.983000E+00	0.	0.	0.556780E-01	0.514057E+03	0.617311E+00
W40	WFD	WG24	FAR24	T25	P25
0.113531E+03	0.	0.113531E+03	0.	0.514057E+03	0.617311E+00
ETAF	ETAI	ETAC	ETATHP	ETATIP	FTATLP
0.831663E+00	0.850485E+00	0.855426E+00	0.899232E+00	0.899013E+00	0.899308E+00
T6	P6	PS6	AMS	V6	WG6

0.166464E+04	0.184363E+01	0.177848E+01	0.238056E+00	0.462190E+03	0.219037E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.166464E+04	0.	0.219037E+03	0.177540E-01	0.	0.
PS8	AM8	V8	PS9	AM9	V9
0.100131E+01	0.100000E+01	0.181220E+04	0.100131E+01	0.100000E+01	0.181220E+04
PS28	AM28	V28	PS29	AM29	V29
0.371092E+00	0.883755E+00	0.913709E+03	0.371092E+00	0.883755E+00	0.913709E+03
BPRINT	DPCOM	DPWING	PS38	AM38	V38
0.	0.497663E-01	0.	0.	0.	0.
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.527427E+00	0.	0.382096E+01	0.332605E+03	0.609938E+03	0.215225E+04
PCBLI	WG37	VJW	PS39	AM39	V39
0.172958E-03	0.	0.	0.	0.	0.
CVDWNG	FGMWNG	FGPWNG	FNWING	FNMAIN	P28
0.985000E+00	0.	0.	0.	0.189626E+05	0.617311E+00
FFOVFN	FWDVFN	FCOVFN	FMVOFN	FNOVFD	P38
0.539786E-01	0.	0.946059E+00	0.100004E+01	0.497819E+00	0.
CVMNOZ	VJM	CVDNOZ	VJD	FGM	FGP
0.985000E+00	0.178502E+04	0.985000E+00	0.900003E+03	0.153280E+05	0.578615E+04

MAIN SCNIC CONVERGENT NOZZLE
DUCT SUBSONIC CONVERG. NOZZLE

FG= 21114.14

FN= 18961.89

SFC= 0.72543

CONVERGED AFTER 24 LOOPS

\$END, #0 THIS CARD TERMINATES THE READING IN OF THE DATA, A \$D(1) CARD WITHOUT
ANY DATA FOLLOWING WOULD DO THE SAME, WITHOUT ONE OF THESE THE LAST CASE
WOULDNT RUN

01 UNIT05, EOF.

REC= 00000 FIL= 00002

CONCLUDING REMARKS

The computer code (GENENG II) presented herein has proved to be an indispensable tool for steady-state cycle analysis of various types of jet engines. Nine basic engine types have been illustrated:

- (1) Three-spool, three-stream turbofan
- (2) Two-spool, three-stream, boost-fan turbofan
- (3) Two-spool, three-stream, supercharged-compressor turbofan
- (4) Three-spool, two-stream turbofan
- (5) Two-spool, two-stream turbofan
- (6) Three-spool, three-stream aft-fan turbofan
- (7) Two-spool, three-stream, aft-fan turbofan
- (8) Two-spool, two-stream, aft-fan turbofan
- (9) Three-spool, two-stream aft-fan turbofan

Some of these engines are candidates for STOL aircraft propulsion.

This program is valuable for many applications because it has the capability of studying a broad range of engine types with different design characteristics, while it also has low-execution-time characteristics. By appropriately choosing among the options built into the program, other engine types can be simulated.

The program has proven itself to be easy to use especially in terms of input requirements. It is felt that with a minimum of effort, the reader can become proficient in using the computer code. The code is available to be reproduced on the requestor's tape upon application to the authors.

Lewis Research Center,
National Aeronautics and Space Administration,
Cleveland, Ohio, October 1, 1971,
132-15.

APPENDIX A

PROGRAM LISTING

```

$IBFTC GEN2    DECK
COMMON/LOOPPR/KKGO,PRFNEW,PRCNEW
DATAIII/O/
COMMON/TERBHI/DHHISV,TFHISV,CNHISV,ETHISV,DHHPDS
COMMON/TERBLO/DHLOSV,TFLOSV,CNLOSV,ETLOSV,DHLPDS
COMMON/TERBMD/DHMDSV,TFMDSV,CNMDSV,ETMDSV,DHMDDS
DIMENSION XLO(5),XMD(5),XHI(5)
EQUIVALENCE (XLO,DHLOSV),(XMD,DHMDSV),(XHI,DHHISV)
COMMON/ALL/X(28)/DESIGN/Y(80)/FRONT/Z(80)/SIDE/W(48)/BACK/V(72)
COMMON/DUMMYS/DUMMY(100)
COMMON/SPOOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI
1,ETAI,WACI,TFFIP,CNIP,ETATIP,DHTCIP,DHTI,BLIP,PCBLIP,PCNIGU,ZIDS,
2PCNIDS,PRIDS,ETAIDS,WAIDS,PRICF,ETAICF,WAICF,TFIPDS,CNIPDS,ETIPDS,
3TFIPCF,CNIPCF,ETAPCF,DHIPCF,WAICDS,WAI,PCBLI,BLI ,T22DS,WA2I
EQUIVALENCE (ERR,DUMMY(11))
DIMENSION ERR(9)
LOGICAL ERRER,CLEAR
DATA CLEAR/.TRUE./
COMMON/ERER/ERRER
ERRER=.FALSE.
IF (.NOT.CLEAR) CALL ENGBAL
CLEAR=.FALSE.
DO 1 J=1,452
1 X(J)=0.
C SET ARBITRARY VALUES FOR INTERMEDIATE SPOOL DESIGN PARAMETERS TO
C AVOID ERROR WHEN RUNNING A DUMMYSPOOL ENGINE
PRIDS=1.5
ETAIDS=1.0
PCNIDS=100.
ZIDS=.75
IF (III.EQ.0) KKGO=0
CALL CONOUT (1)
DO 2 I=1,5
XLO(I)=100.
XMD(I)=100.
2 XHI(I)=100.
DO 3 I=7,10
3 DUMMY(I)=1.0
CALL ENGBAL
STOP
END

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```

$IBFTC ENGBOL DECK
SUBROJTINE ENGBAL
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASMK,IDBURN,IAFTBV,IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,
3ITRYS ,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ARR(6)
COMMON /DESIGN/
1PCNFGJ,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELEFG ,DELEFN ,DELSFC,

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2ZFDS ,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WAFCF ,      8
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,      9
4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCOCF,ETABCF,     10
5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,     11
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPFC,ETLPFC,DHLPFC,T21DS ,     12
7T24DS ,WFDDSD,DTUDSD,ETADSD,WA23DS,DPDUSD,DTUDCF,ETADCF,     13
8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF,     14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,     15
$PS55 ,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV     16
COMMON / FRONT/      17
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,      18
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,      19
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,      20
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,      21
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,      22
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,      23
7CNHP ,ETATHP,DHTCHP,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,      24
8CNLP ,ETATLP,DHTCLP,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,      25
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,      26
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU,PCBLOB,PCBLHP,PCBLLP      27
COMMON / SIDE/      28
1XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 ,      29
2XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 ,      30
3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 ,      31
4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 ,      32
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS,DUMS3 ,      33
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29 ,      34
COMMON / BACK/      35
1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,      36
2XWFB ,XWG55 ,XFAR55,XWFD ,XWG24 ,XFAR24,XXP1 ,DUMB ,      37
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,      38
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,      39
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,      40
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,      41
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,      42
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,      43
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC      44
COMMON/DUMMYS/DUMMY(100)      45
COMMON/SPOOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI      46
1,ETAI,WACI,TFFIP,CNIP,ETATIP,DHTCIP,DHTI,BLIP,PCBLIP,PCNIGU,ZIDS,      47
2PCNIDS,PRIDS,ETAIDS,WAIDS,PRICF,ETAICF,WAICF,TFIPDS,CNIPDS,ETIPDS,      48
3TFIPCF,CNIPCF,ETAPCF,DHIPCF,WAICDS,WAI,PCBLI,BLI ,T22DS,WA21      49
DIMENSION ERR(9)      50
EQUIVALENCE (ERR,DUMMY(11)),(DUMSPL,DUMMY(59))      51
EQUIVALENCE (FXFN2M,DUMMY(50)),(FXM2CP,DUMMY(51))      52
LOGICAL ERRER,FXFN2M,FXM2CP,DUMSPL      53
DIMENSION DELSAV(9)      54
COMMON/ERER/ERRER      55
DIMENSIONVAR(9),DEL(9),ERRB(9),DELVAR(9),EMAT(9,9),VMAT(9),AMAT(9)      56
DATA AWORD/6HENGBAL/      57
DATA VDELTA,VLIM,VCHNGE,NOMISX/      58
1 0.001,0.100,0.850,4/      59
DATA DEL/9*0./      60
DATA DELSAV/9*.001/      61
CALL PUTIN      62
IF (INIT.EQ.1) GO TO 1      63
TFFHP=TFHPDS      64
TFFIP=TFIPDS      65
IF (FXM2CP) TFFIP=TFHPDS      66
TFFLP=TFLPDS      67
1 LOOPER=0      68
NUMMAP=0      69
NOMISS=0      70

```

2	LOOP=0	71
	MISMAT=0	72
	NOMAP=0	73
	IGO=2	74
	DO 3 I=1,9	75
	VMAT(I)=0.	76
	AMAT(I)=0.	77
	DELVAR(I)=0.	78
	DO 3 L=1,9	79
3	EMAT(I,L)=0.	80
4	LOOPER=LOOPER+1	81
	CALL CDFAN	82
	WORD=AWORD	83
	IF (LOOPER.GT.ITRYS) ERRER=.TRUE.	84
	IF (LOOPER.GT.ITRYS) GO TO 26	85
	IF (NOMAP.GT.0) GO TO 2	86
	NUMMAP=0	87
5	VAR(1)=ZF*100.	88
	IF (MODE.NE.3) VAR(2)=PCNF	89
	IF (MODE.EQ.3) VAR(2)=T4/10.	90
	VAR(3)=ZC*100.	91
	IF (MODE.NE.1) VAR(4)=PCNC	92
	IF (MODE.EQ.1) VAR(4)=T4/10.	93
	VAR(5)=TFFHP	94
	VAR(6)=TFFLP	95
	VAR(7)=ZI*100.	96
	VAR(8)=PCNI	97
	VAR(9)=TFFIP	98
	NMAX=9	99
	IF (.NOT.FXFN2M.AND.(.NOT.DUMSPL)) GO TO 6	100
	NMAX=7	101
	IF (DJMSPL) NMAX=6	102
6	IF (.NOT.FXM2CP) GO TO 7	103
	NMAX=7	104
	VAR(4)=PCNI	105
	VAR(5)=TFFIP	106
7	CONTINUE	107
	DO 8 I=1,NMAX	108
8	IF (ABS(ERR(I)).GT.TOLALL) GO TO 9	109
	CALL PERF	110
	CALL ERROR	111
9	IF (LJOP.GT.0) GO TO 11	112
	MAPELG=0	113
	MAPSET=0	114
	DO 10 I=1,NMAX	115
	ERRB(I)=ERR(I)	116
10	DEL(I)=VDELTA*VAR(I)	117
	GO TO 14	118
11	IF (MISMAT.GT.0) GO TO 29	119
	IF (MAPELG.EQ.0) GO TO 12	120
	MAPELG=0	121
	MAPSET=1	122
	VAR(LJOP)=VAR(LOOP)+2.*DEL(LOOP)	123
	GO TO 15	124
12	IF (MAPSET.EQ.0) VAR(LOOP)=VAR(LOOP)+DEL(LOOP)	125
	IF (MAPSET.EQ.1) VAR(LOOP)=VAR(LOOP)-DEL(LOOP)	126
	MAPSET=0	127
	DO 13 I=1,NMAX	128
	IF (DEL(LOOP).NE.0.) DELSAV(LOOP)=DEL(LOOP)	129
	IF (DEL(LOOP).EQ.0.) DEL(LOOP)=DELSAV(LOOP)	130
	EMAT(I,LOOP)=(ERRB(I)-ERR(I))/DEL(LOOP)	131
13	CONTINUE	132

14	LOOP=LJOP+1	133
	IF (LJOP.GT.NMAX) GO TO 17	134
	IF (LJOP.GT.9) GO TO 17	135
	VAR(LJOP)=VAR(LOOP)-DEL(LOOP)	136
15	ZF=VAR(1)/100.	137
	IF (MJDE.NE.3) PCNF=VAR(2)	138
	IF (MJDE.EQ.3) T4=VAR(2)*10.	139
	ZC=VAR(3)/100.	140
	IF (MJDE.NE.1) PCNC=VAR(4)	141
	IF (MJDE.EQ.1) T4=VAR(4)*10.	142
	TFFHP=VAR(5)	143
	TFFLP=VAR(6)	144
	ZI=VAR(7)/100.	145
	PCNI=VAR(8)	146
	TFFIP=VAR(9)	147
	IF (.NOT.FXM2CP) GO TO 16	148
	PCNI=VAR(4)	149
	TFFIP=VAR(5)	150
16	CONTINUE	151
	IF (ZI.LT.0.) ZI=0.05	152
	IF (ZF.LT.0.) ZF=0.05	153
	IF (ZC.LT.0.) ZC=0.05	154
	GO TO (2,4),IGD	155
17	DO 18 I=1,NMAX	156
18	AMAT(I)=-ERRB(I)	157
	DO 20 I=1,NMAX	158
	IZERO=0	159
	DO 19 LOOP=1,NMAX	160
19	IF (EMAT(I,LOOP).EQ.0.) IZERO=IZERO+1	161
	IF (IZERO.LT.NMAX) GO TO 20	162
	WRITE (6,32) I	163
	LOOPER=ITRYS+100	164
	GO TO 26	165
20	CONTINUE	166
	DO 22 LOOP=1,NMAX	167
	IZERO=0	168
	DO 21 I=1,NMAX	169
21	IF (EMAT(I,LOOP).EQ.0.) IZERO=IZERO+1	170
	IF (IZERO.LT.NMAX) GO TO 22	171
	WRITE (6,33) LOOP	172
	LOOPER=ITRYS+100	173
	GO TO 26	174
22	CONTINUE	175
23	CALL MATRIX (EMAT,VMAT,AMAT,NMAX)	176
	LBIG=0	177
	VARBIG=0.	178
	DO 24 L=1,NMAX	179
	ABSVAR=ABS(VMAT(L))	180
	IF (ABSVAR.LE.VLIM*VAR(L)) GO TO 24	181
	IF (ABSVAR.LE.VARBIG) GO TO 24	182
	LBIG=L	183
	VARBIG=ABSVAR	184
24	CONTINUE	185
	VRATIO=1.0	186
	IF (LBIG.GT.0) VRATIO=VLIM*VAR(LBIG)/VARBIG	187
	ERRAVE=0.0	188
	VMTAVE=0.0	189
	DELAVE=0.0	190
	DO 25 L=1,NMAX	191
	DELVAR(L)=VRATIO*VMAT(L)	192
	ERRAVE=ERRAVE+ABS(AMAT(L))/FLOAT(NMAX)	193
	VAR(L)=VAR(L)+DELVAR(L)	194
	VMTAVE=VMTAVE+ABS(VMAT(L))/FLOAT(NMAX)	195

25	DELAVE=DELAVE+ABS(DELVAR(L))/FLOAT(NMAX)	195
	IF (MISMAT.GT.0) GO TO 31	197
	IF (NOMISS.EQ.0) MISMAT=1	198
	IF (MISMAT.EQ.0) IGO=1	199
26	WRITE (8,34) LOOPER	200
	DO 27 I=1,NMAX	201
27	WRITE (8,35) AMAT(I),(EMAT(I,L),L=1,9),VMAT(I),DELVAR(I),VAR(I)	202
	WRITE (8,36) ERRAVE,VMTAVE,DELAVE	203
28	IF (LJOPER.LT.ITRYS) GO TO 15	204
	CALL ERROR	205
	RETURN	206
29	VMTAVX=VMTAVE	207
	DO 30 I=1,NMAX	208
30	AMAT(I)=-ERR(I)	209
	GO TO 23	210
31	WRITE (8,37) AMAT,ERRAVE,DELVAR,DELAVE,VMAT,VMTAVE,VAR	211
	MISMAT=MISMAT+1	212
	IF (VMTAVE.LT.VCHNGE*VMTAVX) GO TO 28	213
	WRITE (8,38)	214
	IF (MISMAT.LT.NOMISX) NOMISS=1	215
	MISMAT=0	216
	LOOP=3	217
	IGO=2	218
	GO TO 5	219
C		220
C		221
32	FORMAT (4H0ROW,I2,16H IS ZERO IN EMAT)	222
33	FORMAT (7H0COLUMN,I2,16H IS ZERO IN EMAT)	223
34	FORMAT (8HB ERRB,28X23HERROR MATRIX AFTER LOOP,I4,29X4HVMAT,5X5H	224
	1DELVAR,7X14HVARIAABLE\$\$\$\$\$)	225
35	FORMAT (1H0,F8.4,10F9.3,2F11.4,6H\$\$\$\$\$)	226
36	FORMAT (1H0,F8.4,32X14HAVERAGE VALUES,31X,2F11.4,6H\$\$\$\$\$)	227
37	FORMAT (12H0----- AMAT,10F11.6,6H\$\$\$\$\$,/,12H -----DELVAR,10F11.6	228
	1,6H\$\$\$\$\$,/,12H ----- VMAT,10F11.6,6H\$\$\$\$\$,/,12H ----- VAR,9F1	229
	21.6,6+\$\$\$\$\$)	230
38	FORMAT (1H0,50X22HCHANGE TOO SMALL\$\$\$\$\$)	231
	END	232

\$IBFTC GUESSS DECK	
FUNCTION GUESS(M,T,TD,P,PD,W,WD,D,DD,VD)	1
IF (M.EQ.0) GUESS=VD*((T/TD)**1.60)*((DD/D)**0.50)	2
IF (M.EQ.1) GUESS=VD*((P/PD)**1.80)*((DD/D)**0.33)	3
IF (M.EQ.2) GUESS=VD*((W/WD)**0.33)*((DD/D)**1.00)	4
IF (M.EQ.3) GUESS=VD*((W/WD)**0.00)*((P/PD)**0.50)	5
IF (M.EQ.4) GUESS=VD*((W/WD)**0.00)*((P/PD)**0.50)	6
IF (M.EQ.5) GUESS=VD*((T/TD)**1.1)*((DD/D)**.7)	7
IF (M.EQ.6) GUESS=VD*((P/PD)**1.00)*((D/DD)**0.25)	8
IF (M.EQ.7) GUESS=VD*((P/PD)**0.62)*((D/DD)**0.31)	9
IF (M.EQ.8) GUESS=VD*((T/TD)**1.2)*DD/D	10
IF (M.EQ.9) GUESS=VD*P/PD*((D/DD)**1.5)	11
RETURN	12
END	13

\$IBFTC MATRIX DECK	
SUBROUTINE MATRIX (E,V,A,N)	1
DIMENSION E(9,9),V(9),A(9),PIV(10),T(9,10)	2
NN=N+1	3
NM=N-1	4
DO 1 I=1,N	5
T(I,NN)=A(I)	6
DO 1 J=1,N	7
1 T(I,J)=E(I,J)	8
DO 7 I=1,N	9
TEMP=0.	10
DO 2 J=I,N	11
IF (TEMP.GT.ABS(T(J,I))) GO TO 2	12
TEMP=ABS(T(J,I))	13
IPIV=J	14
2 CONTINUE	15
IP1=I+1	16
DO 3 J=IP1,NN	17
3 PIV(J)=T(IPIV,J)/T(IPIV,I)	18
IFROM=N	19
ITO=N	20
4 IF (IFROM.EQ.IPIV) GO TO 6	21
RM=-T(IFROM,I)	22
DO 5 J=IP1,NN	23
5 T(ITO,J)=T(IFROM,J)+RM*PIV(J)	24
ITO=ITO-1	25
6 IFROM=IFROM-1	26
IF (IFROM.GE.I) GO TO 4	27
DO 7 J=IP1,NN	28
7 T(I,J)=PIV(J)	29
DO 8 I=1,NM	30
J=NN-I	31
K=N-I	32
DO 8 L=J,N	33
8 T(K,NN)=T(K,NN)-T(K,L)*T(L,NN)	34
DO 9 I=1,N	35
9 V(I)=T(I,NN)	36
RETURN	37
END	38

\$IBFTC PUTIN DECK	
SUBROUTINE PUTIN	1
COMMON / ALL/	2
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
2IGASMX,IDBURN,IAFTBV,IDCD ,IMCD ,IDSHQC,IMSHQC,NOZFLT,	4
3ITRYS ,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ARR(6)	5
COMMON /DESIGN/	6
1PCNFGJ,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELFQ ,DELFN ,DELSFC,	7
2ZFDS ,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WAFCF ,	8
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,	9
4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCCCF,ETABCF,	10
5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,	11
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPCF,ETLPCF,DHLPCF,T21DS ,	12
7T24DS ,WFDDDS ,DTDUUS,ETADDS,WA23DS,DPOUDS,DTDUCCF,ETADCF,	13
8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF,	14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,	15
\$PS55 ,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV	16
COMMON / FRONT/	17
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,	18

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2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 , 19
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 , 20
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB , 21
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 , 22
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,JPCDM ,DUMF , 23
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS , 24
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT , 25
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB , 26
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLJB ,PCBLHP ,PCBLLP 27
COMMON / SIDE/ 28
1XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 , 29
2XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 , 30
3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 , 31
4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 , 32
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS ,DUMS3 , 33
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29 34
COMMON / BACK/ 35
1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 , 36
2XWFB ,XWG55 ,XFAR55 ,XWFD ,XWG24 ,XFAR24 ,XFP1 ,DUMB , 37
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 , 38
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 , 39
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,JPAFT ,V55 ,V25 , 40
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 , 41
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 , 42
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM , 43
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC 44
COMMON/DUMMYS/DUMMY(100) 45
COMMON/SPOOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI 46
1,ETAI,WACI,TFFIP,CNIP,ETATIP,DHTCIP,DHTI,BLIP,PCBLIP,PCNIGU,ZIDS, 47
2PCNIDS,PRIDS,ETAIDS,WAIDS,PRICF,ETAICF,WAICF,TFIPDS,CNIPDS,ETIPDS, 48
3TFIPCF,CNIPCF,ETAPCF,DHIPC,F,WAICDS,WAI,PCBLI,BLI ,T22DS,WA21 49
DIMENSION ERR(9) 50
EQUIVALENCE (ERR,DUMMY(11)) 51
COMMON/LOOPPR/KKGO,PRFNEW,PRCNEW 52
DIMENSION XSAVE(308),XFILL(1) 53
EQUIVALENCE (XFILL,WORD) 54
LOGICAL ERROR 55
COMMON/ERER/ERRER 56
C *** IDES =1 FOR CALCULATING DESIGN POINT 57
C *** MODE =0 FOR CONSTANT T4 58
C *** MODE =1 FOR CONSTANT PCNC 59
C *** MODE =2 FOR CONSTANT WFB 60
C *** MODE=3 FOR CONSTANT PCNF 61
C *** INIT =1 WILL NOT INITIALIZE POINT 62
C *** IDUMP =1 WILL DUMP LOOPING WRITE-OUTS IF ERROR OCCURS 63
C *** IDUMP =2 WILL DUMP LOOPING WRITE-OUTS AFTER EVERY POINT 64
C *** IAMTP =0 WILL USE INPUT AM AND MIL SPEC ETAR 65
C *** IAMTP =1 WILL USE INPUT AM AND INPUT ETAR 66
C *** IAMTP =2 WILL USE T2 AS T1=T1+T2 AND STANDARD P1 67
C *** IAMTP =3 WILL USE P2 AND STANDARD T1 68
C *** IAMTP =4 WILL USE T2 AND P2 69
C *** IAMTP =5 WILL USE RAM2 FOR SPECIAL RECOVERY 70
C *** IGASMX=-1 SEPARATE FLOW, INPUT A6 71
C *** IGASMX=0 SEPARATE FLOW, A6=A55 72
C *** IGASMX=1 WILL MIX DUCT AND MAIN STREAMS, A6=A25+A55 73
C *** IGASMX=2 WILL MIX DUCT AND MAIN STREAMS, INPUT A6 74
C *** IDBURN=1 FOR DUCT BURNING, INPUT T24 75
C *** IDBURN=2 FOR DUCT BURNING, INPUT WFD 76
C *** IAFTBV=1 FOR AFTERBURNING, INPUT T7 77
C *** IAFTBV=2 FOR AFTERBURNING, INPUT WFA 78
C *** IDC0 =1 DUCT NOZZLE WILL BE C-D 79
C *** IMCD =1 MAIN NOZZLE WILL BE C-D 80
C *** NOZFLT=1 FOR FLOATING MAIN NOZZLE 81

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C *** NOZFLT=2 FOR FLOATING DUCT NOZZLE	82
C *** NOZFLT=3 FOR FLOATING MAIN AND DUCT NOZZLES	83
C *** ITRY5 =N NUMBER OF PASSES THRU ENGINE BEFORE QUITTING	84
DIMENSION ITABLE (500)	85
DATA (ITABLE(I),I=1,3)/0,500,0/	86
1 CALL ZERO	87
IF (K<GO.EQ.1) GO TO 5	88
IDES=0	89
CALL INPUT (5,6,1,WORD,ITABLE)	90
IF (ERRER.AND.IAFTBN.GT.0) GO TO 1	91
IF (ERRER.AND.IDBURN.GT.0) GO TO 1	92
ERRER=.FALSE.	93
C TABLE IS REFERENCED TO COMMON/ALL/FIRST ENTRY	94
IF (IDES.EQ.0) GO TO 7	95
IF (K<GO.NE.2) GO TO 3	96
DO 2 I=1,308	97
2 XFILL(I)=XSAVE(I)	98
CALL INPUT (5,6,1,WORD,ITABLE)	99
3 CONTINUE	100
C SAVE INPUT IN CASE OF LOOP ON PRESSURE RATIOS	101
DO 4 I=1,308	102
4 XSAVE(I)=XFILL(I)	103
GO TO 7	104
5 DO 6 I=1,308	105
6 XFILL(I)=XSAVE(I)	106
WRITE (6,8) PRFDS,PRFNEW,PRCDS,PRCNEW	107
PRCDS=PRCNEW	108
PRFDS=PRFNEW	109
7 CONTINUE	110
KKGO=2	111
IF (IAFTBN.GT.0.OR.IDBURN.GT.0) INIT=1	112
IF (MODE.EQ.0) WRITE (8,9) IDES,AM,ALTP,T4,T24,T7	113
IF (MODE.EQ.1) WRITE (8,10) IDES,AM,ALTP,PCNC,T24,T7	114
IF (MODE.EQ.2) WRITE (8,11) IDES,AM,ALTP,WFB,T24,T7	115
C DUMMY ROUTINE TO RESTORE WORKING PART OF PROGRAM TO CORE	116
CALL JVLAY	117
CALL COINLT	118
RETURN	119
C	120
C	121
8 FORMAT (18H0CHANGE PRFDS FROM,F9.3,4H TO,F9.3,16H AND PRCDS FROM	122
1,F10.3,4H TO,F10.3)	123
9 FORMAT (1H0,7H IDES=,I3,10X7H AM=,F7.3,6X7H ALTP=,F7.0,6X7H	124
1 T4=,F8.2,5X7H T24=,F8.2,5X7H T7=,F8.2,6H\$\$\$\$\$)	125
10 FORMAT (1H0,7H IDES=,I3,10X7H AM=,F7.3,6X7H ALTP=,F7.0,6X7H	126
1PCNC=,F8.3,5X7H T24=,F8.2,5X7H T7=,F8.2,6H\$\$\$\$\$)	127
11 FORMAT (1H0,7H IDES=,I3,10X7H AM=,F7.3,6X7H ALTP=,F7.0,6X7H	128
1 WFB=,F8.4,5X7H T24=,F8.2,5X7H T7=,F8.2,6H\$\$\$\$\$)	129
END	130

\$IBFTC ZERO DECK	
SUBROJTIME ZERO	1
COMMON / ALL/	2
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
2IGASM, IDBURN, IAFTBN, IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,	4
3ITRYS ,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ARR(6)	5
COMMON/DESIGN/QXQ(80)	6
COMMON / FRONT/	7

1T1	,P1	,H1	,S1	,T2	,P2	,H2	,S2	,	8
2T21	,P21	,H21	,S21	,T3	,P3	,H3	,S3	,	9
3T4	,P4	,H4	,S4	,T5	,P5	,H5	,S5	,	10
4T55	,P55	,H55	,S55	,BLF	,BLC	,BLDU	,BLOB	,	11
5CNF	,PRF	,ETAF	,WAF	,WAF	,WA3	,WG4	,FAR4	,	12
6CNC	,PRC	,ETAC	,WACC	,WAC	,ETAB	,DPCOM	,DUMF	,	13
7CNHP	,ETATHP	,DHTCHP	,DHTC	,BLHP	,WG5	,FAR5	,CS	,	14
8CNLP	,ETATLP	,DHTCLP	,DHTF	,BLLP	,WG55	,FAR55	,HPEXT	,	15
9AM	,ALTP	,ETAR	,ZF	,PCNF	,ZC	,PCNC	,WFB	,	16
\$TFFHP	,TFFLP	,PCBLF	,PCBLC	,PCBLDU	,PCBLOB	,PCBLHP	,PCBLLP	,	17
COMMON / SIDE/									18
1XP1	,XWAF	,XWAC	,XBLLF	,XBLDU	,XH3	,DUMS1	,DUMS2	,	19
2XT21	,XP21	,XH21	,XS21	,T23	,P23	,H23	,S23	,	20
3T24	,P24	,H24	,S24	,T25	,P25	,H25	,S25	,	21
4T28	,P28	,H28	,S28	,T29	,P29	,H29	,S29	,	22
5WAD	,WFD	,WG24	,FAR24	,ETAD	,JPDUC	,BYPASS	,DUMS3	,	23
6TS28	,PS28	,V28	,AM28	,TS29	,PS29	,V29	,AM29	,	24
COMMON / BACK/									25
1XT55	,XP55	,XH55	,XS55	,XT25	,XP25	,XH25	,XS25	,	26
2XWFB	,XWG55	,XFAR55	,XWFD	,XWG24	,XFAR24	,XFP1	,DUMB	,	27
3T6	,P6	,H6	,S6	,T7	,P7	,H7	,S7	,	28
4T8	,P8	,H8	,S8	,T9	,P9	,H9	,S9	,	29
5WG6	,WFA	,WG7	,FAR7	,ETAA	,DPAFT	,V55	,V25	,	30
6PS6	,V6	,AM6	,TS7	,PS7	,V7	,AM7	,AM25	,	31
7TS8	,PS8	,V8	,AM8	,TS9	,PS9	,V9	,AM9	,	32
8VA	,FRD	,VJD	,FGMD	,VJM	,FGMM	,FGPD	,FGPM	,	33
9FGM	,FGP	,WFT	,WGT	,FART	,FG	,FN	,SFC	,	34
COMMON/DUMMYS/DUMMY(100)									35
COMMON/SPOOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI									36
1,ETAI,WACI,TFFIP,CNIP,ETATIP,DHTCIP,DHTI,BLIP,PCBLIP,PCNIGU,ZIDS,									37
2PCNIDS,PRIDS,ETAIDS,WAIDS,PRICF,ETAICF,WAICF,TFIPDS,CNIPDS,ETIPDS,									38
3TFIPCF,CNIPCF,ETAPCF,DHPCF,WAICDS,WAI,PCBLI,BLI ,T22DS,WA21									39
DIMENSION Z1(63),Z2(48),Z3(72)									40
EQUIVALENCE (Z1,T1),(Z2,XP1),(Z3,XT55)									41
IDES=0									42
JDES=0									43
INIT=0									44
IDBURN=0									45
IAFTBN=0									46
IDSHOC=3									47
IMSHOC=3									48
T2Q=T2									49
P2Q=P2									50
T4Q=T4									51
DO 1 I=1,63									52
1 Z1(I)=0.									53
DO 2 I=1,48									54
2 Z2(I)=0.									55
DO 3 I=1,72									56
3 Z3(I)=0.									57
T2=T2Q									58
P2=P2Q									59
T4=T4Q									60
CALL SYG (1)									61
RETURN									62
END									63


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$IBFTC COINLT DECK
SUBROJTINE COINLT
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASMX ,IDBURN ,IAFTBV ,IDCD ,IMCD ,IDSHJC ,IMSHOC ,NOZFLT ,
3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ARR(6)
COMMON /DESIGN/
1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
4T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,DPCODS ,DTCOCF ,ETABCF ,
5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS ,
7T24DS ,WFDDSD ,DTDUDS ,ETADDS ,WA23DS ,DPDUDS ,DTDUCF ,ETADCF ,
8T7DS ,WFAADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
$PS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON / FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,JPCOM ,DUMF ,
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLOB ,PCBLHP ,PCBLLP
COMMON/SIDE/ZYX(48)/BACK/YZX(72)
COMMON/DUMMYS/DUMMY(100)
COMMON/SPOOL2/TWOSPL(44)
EQUIVALENCE (ERR,DUMMY(11))
DIMENSION ERR(9)
DATA AWORD/6HCOINLT/
WORD=AWORD
AJ=779.26
G=32.174049
ALT=ALTP*2.0855531E+07/(2.0855531E+07-ALTP)
CALL AFMOS (ALT,T1,XX1,XX2,XX3,P1,CS,XX4,IIER)
IF (IAMTP.EQ.2) T1=T1+T2
IF (IAMTP.EQ.5) CALL RAM2 (AM,ETAR)
IF (IAMTP.NE.1.AND.IAMTP.NE.5) CALL RAM (AM,ETAR)
FAR=0.0
CALL PROCOM (FAR,T1,CS,XX2,XX3,R1,PHI1,H1)
S1=PHI1-R1*ALOG(P1)
H2=H1+(AM*CS)**2/(2.*AJ*G)
P2T=1.
DO 1 I=1,10
CALL THERMO (P2T,H2,T2T,S2T,AW,0,0,0,1)
IF (ABS(S2T-S1).LE.0.0001*S1) GO TO 2
1 P2T=P1*EXP((AW/1.986375)*((S2T-S1)+(1.986375/AW)*ALOG(P2T/P1)))
CALL ERROR
RETURN
2 IF (IAMTP.EQ.3.OR.IAMTP.EQ.4) ETAR=P2/P2T
P2=ETAR*P2T
IF (IAMTP.NE.4) CALL THERMO (P2,H2,T2,S2,XX5,0,0,0,1)
IF (IAMTP.EQ.4) CALL THERMO (P2,H2,T2,S2,XX5,0,0,0,0)
IF (INIT.EQ.1) GO TO 5
IF (IDES.EQ.1) GO TO 3
IF (MODE.EQ.3) GO TO 4
PCNF=GUESS(MODE,T4,T4DS,PCNC,PCNCDS,WFB,WFBDS,T2,T2DS,PCNFDS)
PCNFGU=PCNF
GO TO 4

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3	PCNF=PCNFDS	63
	PCNFGJ=PCNF	64
	T2DS=T2	65
4	ZF=ZFDS	66
5	RETURN	67
	END	68

\$IBFTC ATMOS DECK

	SUBROUTINE ATMOS (ZFT, TM, SIGMA, RHO, THETA, DELTA, CA, AMU, K)	1
C	THIS IS A SUBROUTINE TO COMPUTE CERTAIN ELEMENTS OF THE 1962	2
C	U.S. STANDARD ATMOSPHERE UP TO 90 KILOMETERS.	3
C	CALLING SEQUENCE...	4
C		5
C	CALL ATMOS (ZFT, TM, SIGMA, RHO, THETA, DELTA, CA, AMU, K)	6
C	ZFT = GEOMETRIC ALTITUDE (FEET)	7
C	TM = MOLECULAR SCALE TEMPERATURE (DEGREES RANKINE)	8
C	SIGMA = RATIO OF DENSITY TO THAT AT SEA LEVEL	9
C	RHO = DENSITY (LB-SEC**2-FT**(-4) OR SLUGS-FT**3)	10
C	THETA = RATIO OF TEMPERATURE TO THAT AT SEA LEVEL	11
C	DELTA = RATIO OF PRESSURE TO THAT AT SEA LEVEL	12
C	CA = SPEED OF SOUND (FT/SEC)	13
C	AMU = VISCOSITY COEFFICIENT (LB-SEC/FT**2)	14
C		15
C	K = 1 NORMAL	16
C	= 2 ALTITUDE LESS THAN -5000 METERS OR GREATER THAN 90 KM	17
C	= 3 FLOATING POINT OVERFLOW	18
C		19
C	ALL DATA AND FUNDAMENTAL CONSTANTS ARE IN THE METRIC SYSTEM AS	20
C	THESE QUANTITIES ARE DEFINED AS EXACT IN THIS SYSTEM.	21
C		22
C	THE RADIUS OF THE EARTH (REFT59) IS THE VALUE ASSOCIATED WITH THE	23
C	1959 ARDC ATMOSPHERE SO THAT PROGRAMS CURRENTLY USING THE LIBRARY	24
C	ROUTINE WILL NOT REQUIRE ALTERATION TO USE THIS ROUTINE.	25
	DIMENSION HB(10), TMB(10), DELTAB(10), ALM(10)	26
	DATA (HB(I), TMB(I), DELTAB(I), ALM(I), I=1,10) /	27
1	-5.0, 320.65, 1.75363E 00, -6.5,	28
2	0.0, 288.15, 1.00000E 00, -6.5,	29
3	11.0, 216.65, 2.23361E-01, 0.0,	30
4	20.0, 216.65, 5.40328E-02, 1.0,	31
5	32.0, 228.65, 8.56663E-03, 2.8,	32
6	47.0, 270.65, 1.09455E-03, 0.0,	33
7	52.0, 270.65, 5.82289E-04, -2.0,	34
8	61.0, 252.65, 1.79718E-04, -4.0,	35
9	79.0, 180.65, 1.0241 E-05, 0.0,	36
\$	88.743, 180.65, 1.6223 E-06, 0.0/	37
	DATA REFT59/2.0855531E 07/, GZ /9.80665/,	38
1	AMZ /28.9644 /, RSTAR /8.31432/,	39
2	FTTOKM/3.048E-04 /, S /110.4 /,	40
3	AMUZ /1.2024E-05 /, CAZ /1116.45/,	41
4	RHOZ /0.076474 /, GZENG /32.1741/	42
C	CONVERT GEOMETRIC ALTITUDE TO GEOPOTENTIAL ALTITUDE	43
	HFT=(REFT59/(REFT59+ZFT))*ZFT	44
C	CONVERT HFT AND ZFT TO KILOMETERS	45
	Z=FTTOKM*ZFT	46
	H=FTTOKM*HFT	47
	K=1	48
	TMZ=TMB(2)	49
	IF (H.LT.-5.0.OR.Z.GT.90.0) GO TO 7	50

	DO 1 M=1,10	51
	IF (H-HB(M)) 2,3,1	52
1	CONTINUE	53
	GO TO 7	54
2	M=M-1	55
3	DELH=H-HB(M)	56
	IF (ALM(M).EQ.0.0) GO TO 4	57
	TMK=TM3(M)+ALM(M)*DELH	58
C	GRADIENT IS NON ZERO, PAGE 10, EQUATION I.2.10-(3)	59
	DELTA=DELTAB(M)*((TMB(M)/TMK)**(GZ*AMZ/(RSTAR*ALM(M))))	60
	GO TO 5	61
4	TMK=TMB(M)	62
C	GRADIENT IS ZERO, PAGE 10, EQUATION I.2.10-(4)	63
	DELTA=DELTAB(M)*EXP(-GZ*AMZ*DELH/(RSTAR*TMB(M)))	64
5	THETA=TMK/TMZ	65
	SIGMA=DELTA/THETA	66
	ALPHA=SQRT(THETA**3)*((TMZ+S)/(TMK+S))	67
C	CONVERSION TO ENGLISH UNITS	68
	TM=1.3*TMK	69
	RHO=R4JZ*SIGMA/GZENG	70
	CA=CAZ*SQRT(THETA)	71
	AMU=AMJZ*ALPHA/GZENG	72
	CALL OVERFL (J)	73
	GO TO (6,8),J	74
6	K=K+2	75
	GO TO 8	76
7	K=2	77
8	RETURN	78
	END	79

\$IBFTC RAMS	DECK	
	SUBROUTINE RAM (AM,ETAR)	1
	IF (AM.GT.1.) GO TO 2	2
	ETAR=1.	3
1	RETURN	4
2	IF (AM.GT.5.) GO TO 3	5
	ETAR=1.-0.075*((AM-1.)**1.35)	6
	GO TO 1	7
3	ETAR=300./((AM**4)+935.)	8
	GO TO 1	9
	END	10

\$IBFTC RAMT#J	DECK	
	SUBROUTINE RAM2 (AM,ETAR)	1
	DIMENSION PRINLT(15),FMN(15)	2
	DIMENSION Y(3),X(3)	3
	DATA FMN/0.,.1,.2,.3,.4,.5,.8,1.1,1.2,1.4,1.6,1.8,2.2,2.4,2.7/	4
	DATA PRINLT/.9,.932,.95,.961,.968,.97,.9701,.97,.9681,.958,.94,	5
	1.9181,.858,.8201,.75/	6
	M=0	7
	DO 1 J=1,15	8
1	IF (AM.GE.FMN(J)) M=J-1	9
	IF (M.EQ.0) M=1	10

IF (M.GE.14) M=13	11
DO 2 I=1,3	12
MM=M-1+I	13
X(I)=FMN(MM)	14
Y(I)=PRINLT(MM)	15
CALL PARABO (X,Y,AM,ETAR)	16
RETURN	17
END	18

\$IBFTC COFAN DECK	
SUBROJTINE COFAN	1
COMMON / ALL/	2
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
2IGASMX ,IDBURN ,IAFTBV ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,	4
3ITRYS ,LOOPER ,NDMAP ,NUMMAP ,MAPEDG ,TDLALL ,ARR(6)	5
COMMON /DESIGN/	6
1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC ,	7
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,	8
3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,	9
4T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,DPCODS ,DTCOCF ,ETABCF ,	10
5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,	11
6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS ,	12
7T24DS ,WFDDDS ,DTDUUS ,ETADDS ,WA23DS ,DPDUUS ,DTDUCF ,ETADCF ,	13
8T7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,	14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,	15
\$PS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV	16
COMMON / FRONT/	17
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,	18
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,	19
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,	20
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,	21
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,	22
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,	23
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,	24
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,	25
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,	26
\$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLOB ,PCBLHP ,PCBLLP	27
COMMON/SIDE/ZYX(48)/BACK/YZX(72)	28
COMMON/DUMMYS/DUMMY(100)	29
COMMON/SPOOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI	30
1,ETAI,WACI,TFFIP,CNIP,ETATIP,DHTCIP,DHTI,BLIP,PCBLIP,PCNIGU,ZIDS,	31
2PCNIDS,PRIDS,ETAIDS,WAIDS,PRICF,ETAICF,WAICF,TFIPDS,CNIPDS,ETIPDS,	32
3TFIPCF,CNIPCF,ETAPCF,DHIPCFC,WAICDS,WAI,PCBLI,BLI ,T22DS,WA21	33
DIMENSION ERR(9)	34
EQUIVALENCE (ERR,DUMMY(11)),(FXM2CP,DUMMY(51))	35
LOGICAL FXM2CP	36
COMMON / FAN/CNX(15),PRX(15,15),WACX(15,15),ETAX(15,15),	37
1NCN,NPT(15)	38
DIMENSION WLH(2)	39
DATA AWORD,WLH/6H COFAN,6H (LO) ,6H (HI) /	40
WORD=AWORD	41
THETA=SQRT(T2/518.668)	42
IF (IDES.NE.1) GO TO 1	43
THETA0=THETA	44
WAFDS=WAFC*P2/THETA	45

1	CNF=PCNF*THETAD/(100.*THETA)	46
	IF (ZF.LT.0.) ZF=0.	47
	IF (ZF.GT.1.) ZF=1.	48
	CNFS=CNF	49
	CALL SEARCH (ZF,CNF,PRF,WAFC,ETAF,CNX(1),NCN,PRX(1,1),WACX(1,1),ET	50
	1AX(1,1),NPT(1),15,15,IGO)	51
	IF ((CNF-CNFS).GT.0.0005*CNF) MAPEDG=1	52
	IF (IGO.EQ.1.OR.IGO.EQ.2) WRITE (8,12) CNFS,WLH(IGO)	53
	WAF=WAFC*P2/THETA	54
	IF (IDES.NE.1) GO TO 2	55
	PRFCF=(PRFDS-1.)/(PRF-1.)	56
	ETAFCF=ETAFFS/ETAF	57
	WAFCF=WAFFS/WAF	58
	WRITE (6,13) PRFCF,ETAFCF,WAFCF,T2DS	59
2	PRF=PRFCF*(PRF-1.)+1.	60
	ETAF=ETAFCF*ETAF	61
	WAF=WAFCF*WAF	62
	WAFC=WAFC*WAFCF	63
	PCNF=100.*THETA*CNF/THETAD	64
	DUMD1=PCNF	65
	CALL THCOMP (PRF,ETAF,T2,H2,S2,P2,T22,H22,S22,P22)	66
	IF (PCBLF.GT.0.) BLF=PCBLF*WAF	67
	IF (JDES.EQ.1) GO TO 9	68
	JDES=1	69
	IF (INIT.EQ.1) GO TO 8	70
	IF (IDES.EQ.1) GO TO 6	71
	IF (MODE.NE.2) GO TO 3	72
	T4=GUESS(3,Y1,Y2,PCNF,PCNFDS,WFB,WFBDS,Y7,Y8,T4DS)	73
	PCNI=GUESS(8,T4,T4DS,Y3,Y4,Y5,Y6,T22,T22DS,PCNIDS)	74
	PCNC=GUESS(4,Y1,Y2,PCNI,PCNIDS,WFB,WFBDS,Y7,Y8,PCNCDS)	75
	GO TO 7	76
3	IF (MODE.EQ.1) GO TO 5	77
	IF (MODE.EQ.0) GO TO 4	78
	T4=GUESS(7,Y1,Y2,PCNF,PCNFDS,Y5,Y6,T2,T2DS,T4DS)	79
4	CONTINUE	80
	PCNC=GUESS(5,T4,T4DS,Y3,Y4,Y5,Y6,T22,T22DS,PCNCDS)	81
	IF (FXM2CP) PCNC=PCNCDS*.99	82
	PCNCG1=PCNC	83
	PCNCG2=PCNCDS	84
	PCNI=GUESS(9,Y1,Y2,PCNCG1,PCNCG2,Y5,Y6,T22,T22DS,PCNIDS)	85
	GO TO 7	86
5	T4=GUESS(6,Y1,Y2,PCNC,PCNCDS,Y5,Y6,T22,T22DS,T4DS)	87
	PCNI=GUESS(8,T4,T4DS,Y3,Y4,Y5,Y6,T22,T22DS,PCNIDS)	88
	GO TO 7	89
6	PCNC=PCNCDS	90
	PCNI=PCNIDS	91
	T4=T4DS	92
	WFB=WFBDS	93
	T21DS=T21	94
7	ZC=ZCDS	95
	ZI=ZIDS	96
	PCNIGJ=PCNI	97
	PCNCGJ=PCNC	98
	T4GU=T4	99
8	INIT=0	100
9	IF (MODE.NE.3) GO TO 10	101
	IF (ABS(CNF-CNFS).LE.0.001*CNFS) GO TO 11	102
	WRITE (8,14) CNFS,CNF	103
	CALL ERROR	104
10	PCNF=100.*THETA*CNF/THETAD	105
11	CALL COINTC	106
	RETURN	107
C		108

C		109
12	FORMAT (19H0* * * CNF OFF MAP,F10.4,2XA6,11H* * *\$\$\$\$\$)	110
13	FORMAT (11HOFAN DESIGN,13X8H PRFCF=,E15.8,8H ETAFCF=,E15.8,8H WA	111
	1FCF=,E15.8,8H T2DS=,E15.8)	112
14	FORMAT (10HOCNF WAS= ,E15.8,11H AND NOW= ,E15.8,24H CHECK PCNF I	113
	INPUT\$\$\$\$\$)	114
	END	115

\$IBFTC COINTC DECK		
SUBROUTINE COINTC		1
COMMON / ALL/		2
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,		3
2IGASMX,IDBURN,IAFTBN,IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,		4
3ITRYS ,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ARR(6)		5
COMMON /DESIGN/		6
1PCNFGJ,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELFQ ,DELFN ,DELSFC,		7
2ZFDS ,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WAFCF ,		8
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,		9
4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCCCF,ETABCF,		10
5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,		11
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPDF,ETLPDF,DHLPDF,T21DS ,		12
7T24DS ,WFDDSD,DTDUODS,ETADDS,WA23DS,DPOUDS,DTDUCF,ETADCF,		13
8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF,		14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,		15
\$PS55 ,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV		16
COMMON / FRONT/		17
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,		18
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,		19
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,		20
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,		21
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,		22
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,		23
7CNHP ,ETATHP,DHTCHP,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,		24
8CNLP ,ETATLP,DHTCLP,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,		25
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,		25
\$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU,PCBLDB,PCBLHP,PCBLLP		27
COMMON/SIDE/ZYX(48)/BACK/YZX(72)		28
COMMON/DUMMYS/DUMMY(100)		29
COMMON/SPOOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI		30
1,ETAI,WACI,TFFIP,CNIP,ETATIP,DHTCIP,DHTI,BLIP,PCBLIP,PCNIGU,ZIDS,		31
2PCNIDS,PRIDS,ETAIDS,WAIDS,PRICF,ETAICF,WAICF,TFIPDS,CNIPDS,ETIPDS,		32
3TFIPCF,CNIPCF,ETAPCF,DHPCF,WAICDS,WAI,PCBLI,BLI ,T22DS,WA21		33
COMMON/INT/CNX(15),PRX(15,15),WACX(15,15),ETAX(15,15),		34
1NCN,NPT(15)		35
COMMON/DUMINT/CNXX(15),PRXX(15,15),WACXX(15,15),ETAXX(15,15),		36
1NCNX,NPTX(15)		37
EQUIVALENCE (FXFN2M,DUMMY(50)),(FXM2CP,DUMMY(51)),		38
1 (AFTFAN,DUMMY(58)),(DUMSPL,DUMMY(59))		39
2 , (PCBLID,DUMMY(61))		40
LOGICAL FXFN2M,FXM2CP,AFTFAN,DUMSPL		41
DIMENSION ERR(9)		42
EQUIVALENCE (ERR,DUMMY(11))		43
DIMENSION WLH(2)		44
DATA AWORD,WLH/6HCOINTC,6H (LO) ,6H (HI) /		45
WORD=AWORD		45
IF (.NOT.AFTFAN) GO TO 1		47
T22S=T22		48
H22S=H22		49

	S22S=S22	50
	P22S=P22	51
	T22=T2	52
	H22=H2	53
	S22=S2	54
	P22=P2	55
1	THETA=SQRT(T22/518.668)	56
	IF (IDES.NE.1) GO TO 2	57
	PRI=PRIDS	58
	PCBLI=PCBLID	59
	WACI=WAICDS	60
	THETAD=THETA	61
	WAIDS=WACI*P22/THETA	62
	ETAI=ETAIDS	63
2	IF (.NOT.FXFN2M) GO TO 3	64
C	FAN AND MIDDLE SPOOL ROTATE AT SAME SPEED	65
	SPDFAN=CNF*SQRT(T2/518.668)	66
	CNI=SPDFAN/THETA	67
	PCNI=100.*CNI*THETA/THETAD	68
	IF (IDES.EQ.1) PCNIDS=PCNI	69
3	CNI=PCNI*THETAD/(100.*THETA)	70
	ZI=AMAX1(ZI,0.)	71
	ZI=AMIN1(ZI,1.)	72
	CNIS=CNI	73
	IF (.NOT.DUMSPL) GO TO 4	74
	CALL INDUMY (CNI,ZI,WAICDS,IDES)	75
	CALL SEARCH (ZI,CNI,PRI,WACI,ETAI,CNXX,NCNX,PRXX,WACXX,ETAXX,NPTX,	76
	115,15,IGO)	77
	GO TO 5	78
4	CONTINUE	79
	CALL SEARCH (ZI,CNI,PRI,WACI,ETAI,CNX(1),NCN,PRX(1,1),WACX(1,1),ET	80
	1AX(1,1),NPT(1),15,15,IGO)	81
5	CONTINUE	82
	IF ((CNI-CNIS).GT..0005*CNI) MAPEDG=1	83
	IF (ISJ.EQ.1.OR.IGO.EQ.2) WRITE (8,12) CNIS,WLH(IGO)	84
	WAI=WACI*P22/THETA	85
	WA22=WAI	86
	IF (IDES.NE.1) GO TO 7	87
	T22DS=T22	88
	IF (.NOT.DUMSPL) PRICF=(PRIDS-1.)/(PRI-1.)	89
	ETAICF=ETAIDS/ETAI	90
	WAICF=WAIDS/WAI	91
	IF (.NOT.DUMSPL) GO TO 6	92
	PRICF=1.	93
	ETAICF=1.	94
	WAICF=1.	95
6	CONTINUE	96
	WRITE (6,13) PRICF,ETAICF,WAICF,T22DS	97
7	PRI=PRICF*(PRI-1.)+1.	98
	ETAI=ETAICF*ETAI	99
	WAI=WAICF*WAI	100
	WACI=WACI*WAICF	101
	WA22=WAI	102
	CALL THCOMP (PRI,ETAI,T22,H22,S22,P22,T21,H21,S21,P21)	103
	IF (.NOT.DUMSPL) GO TO 8	104
	PRI=1.	105
	ETAI=1.	106
	T21=T22	107
	H21=H22	108
	S21=S22	109
	P21=P22	110
8	CONTINUE	111
	IF (IDES.NE.1) GO TO 9	112
	BLI=PCBLI*WAI	113

	WA21=WA22-BLI	114
	WA32=BLI	115
	WAC=WA21	116
9	CONTINUE	117
	IF (ABS(CNI-CNIS).LE.0.001*CNIS) GO TO 10	118
	WRITE (8,14) CNIS,CNI	119
	CALL ERROR	120
	PCNI=100.*THETA*CNI/THETAD	121
10	IF (.NOT.AFTFAN) GO TO 11	122
	T22=T22S	123
	H22=H22S	124
	S22=S22S	125
	P22=P22S	126
11	CALL COCOMP	127
	RETURN	128
C		129
C		130
C		131
12	FORMAT (19H0* * * CNI OFF MAP,F10.4,2XA6,11H* * *\$\$\$\$\$)	132
13	FORMAT (20H/MIDDLE SPOOL DESIGN,4X8H PRICF=,E15.8,8H ETAICF=,E15.	133
	18,8H WAICF=,E15.8,8H T22DS=,E15.8)	134
14	FORMAT (10HOCNI WAS= ,E15.8,11H AND NOW= ,E15.8,24H CHECK PCNI I	135
	INPUT\$\$\$\$\$)	136
	END	137

\$IBFTC	INTDUM DECK	
	SUBROUTINE INDUMY (CNI,ZI,WACI,IDES)	1
	COMMON/DUMINT/CNXX(15),PRXX(15,15),WACXX(15,15),ETAXX(15,15),	2
	1NCNX,VPTX(15)	3
	DIMENSION WACAR(15),XCNXX(15)	4
	DATA XCNXX/.001,.1,.2,.3,.5,.8,1.,1.5,2.0,3.0,4.0,5.0,6.,7.,9./	5
	DATA WACAR/5.,4.5,4.,3.5,3.,2.5,2.,1.5,1.,.8,.6,.4,.25,.1,.05/	6
	IF (IDES.NE.1) GO TO 1	7
	WAIDS=WACI	8
	CNIDS=CNI	9
	ZI=2./3.5	10
1	NCNX=15	11
	DO 2 I=1,15	12
	NPTX(I)=15	13
	CNXX(I)=XCNXX(I)*CNIDS	14
	DO 2 J=1,15	15
	PRXX(I,J)=FLOAT(J+3)/4.	16
	ETAXX(I,J)=1.	17
2	WACXX(J,I)=WACAR(I)*(.993+.001*FLOAT(J))*WAIDS	18
	RETURN	19
	END	20

\$IBFTC	COCOMP DECK	
	SUBROUTINE COCOMP	1
	COMMON / ALL/	2
	1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
	2IGASM, IDBURN, IAFTBV, IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,	4


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3ITRYS ,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ARR(6) 5
COMMON /DESIGN/ 6
1PCNFGJ,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELF6 ,DELFN ,DELSFC, 7
2ZFDS ,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WAFCF , 8
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF , 9
4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCOCF,ETABCF, 10
5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS , 11
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPDF,ETLPDF,DHLPDF,T21DS , 12
7T24DS ,WFODS ,DTODS,ETADDS,WA23DS,DPDUDS,DTODCF,ETADCF, 13
8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF, 14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 , 15
$PS55 ,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV 16
COMMON / FRONT/ 17
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 , 18
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 , 19
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 , 20
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB , 21
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 , 22
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF , 23
7CNHP ,ETATHP,DHTCHP,DHTC ,BLHP ,WG5 ,FAR5 ,CS , 24
8CNLP ,ETATLP,DHTCLP,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT , 25
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB , 26
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU,PCBLOB,PCBLHP,PCBLLP 27
COMMON/SIDE/ZYX(48)/BACK/YZX(72) 28
COMMON/DUMMYS/DUMMY(21),WA32,DPWGDS,DPWING,WA32DS,A38,AM38,V38, 29
1T38,H38,P38,TS38,PS38,T39,H39,P39,TS39,V39,AM39,A39,BPRINT,WG37, 30
2CVDWNG,FGMWNG,FGPWNG,FNWING,FNMAIN,FWDVFN,DIMMY(52) 31
COMMON/SPOOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI 32
1,ETAI,WACI,TFFIP,CNIP,ETATIP,DHTCIP,DHTI,BLIP,PCBLIP,PCNIGU,ZIDS, 33
2PCNIDS,PRIDS,ETAIDS,WAIDS,PRICF,ETAICF,WAICF,TFIPDS,CNIPDS,ETIPDS, 34
3TFIPCF,CNIPCF,ETAPCF,DHPCF,WAICDS,WAI,PCBLI,BLI ,T22DS,WA21 35
EQUIVALENCE(FXFN2M,DUMMY(50)),(FXM2CP,DUMMY(51)) 36
EQUIVALENCE (PCBLID,DUMMY(61)),(DUMSPL,DUMMY(59)) 37
LOGICAL FXFN2M,FXM2CP,DUMSPL 38
DIMENSION ERR(9) 39
EQUIVALENCE (ERR,DUMMY(11)) 40
COMMON / COMP/CNX(15),PRX(15,15),WACX(15,15),ETAX(15,15), 41
1NCN,NPT(15) 42
DIMENSION WLH(2) 43
DATA AWORD,WLH/6HCUCOMP,6H (LO) ,6H (HI) / 44
WORD=AWORD 45
THETA=SQRT(T21/518.668) 46
IF (IDES.NE.1) GO TO 1 47
THETA=THETA 48
WACDS=WAC 49
WACC=WAC*THETA/P21 50
IF (.NOT.FXM2CP) PCNC=PCNCDS 51
1 IF (.NOT.FXM2CP) GO TO 2 52
C SPEEDS OF MIDDLE AND INNER SPOOL ARE THE SAME 53
SPDMID=CNI*SQRT(T22/518.668) 54
CNC=SPDMID/THETA 55
PCNC=100.*CNC*THETA/THETA 56
IF (IDES.EQ.1) PCNCDS=PCNC 57
2 CNC=PCNC*THETA/(100.*THETA) 58
IF (IDES.NE.1) GO TO 3 59
3 CONTINUE 60
IF (ZC.LT.0.) ZC=0. 61
IF (ZC.GT.1.) ZC=1. 62
CNC=CNC 63
CALL SEARCH (ZC,CNC,PRC,WACC,ETAC,CNX(1),NCN,PRX(1,1),WACX(1,1),ET 64
1AX(1,1),NPT(1),15,15,IGD) 65
IF (MODE.EQ.1) GO TO 4 66
IF ((CNC-CNC).GT.0.0005*CNC) MAPEDG=1 67

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4	IF (IGJ.EQ.1.OR.IGJ.EQ.2) WRITE (8,9) CNCS,WLH(IGJ)	68
	WAC=WACC*P21/THETA	69
	IF (IDES.NE.1) GO TO 5	70
	T21DS=T21	71
	PRCCF=(PRCDS-1.)/(PRC-1.)	72
	ETACCF=ETACDS/ETAC	73
	WACCF=WACDS/WAC	74
	WRITE (6,10) PRCCF,ETACCF,WACCF,T21DS	75
5	PRC=PRCCF*(PRC-1.)+1.	76
	ETAC=ETACCF*ETAC	77
	WAC=WACCF*WAC	78
	IF (.NOT.DUMSPL.OR.PCBLID.NE.0.) GO TO 6	79
	WA22=WAC	80
	WAI=WA22	81
	WACI=WACC*WACCF	82
6	WA32=WA22-WAC	83
	WA21=WAC	84
	WACC=WACC*WACCF	85
	PCBLI=1.-WA21/WA22	86
	CALL WDUCTI	87
	IF (PCBLID.EQ.0.) ERR(7)=(WAC-WAI)/WAC	88
	IF (IDES.EQ.1.AND.PCBLID.EQ.0.) ERR(7)=1.E-4	89
	CALL THCOMP (PRC,ETAC,T21,H21,S21,P21,T3,H3,S3,P3)	90
	IF (PCBLC.GT.0.) BLC=PCBLC*WAC	91
	WA3=WAC-BLC	92
	BLDU=PCBLDU*BLC	93
	BLOB=PCBLOB*BLC	94
	BLHP=PCBLHP*BLC	95
	BLIP=PCBLIP*BLC	96
	BLLP=PCBLIP*BLC	97
	IF (MODE.NE.1) GO TO 7	98
	IF (ABS(CNC-CNC).LE.0.001*CNC) GO TO 8	99
	WRITE (8,11) CNCS,CNC	100
	CALL ERROR	101
7	PCNC=100.*THETA*CNC/THETA0	102
8	CALL CCOMB	103
	RETURN	104
C		105
C		106
C		107
9	FORMAT (19H0* * * CNC OFF MAP,F10.4,2XA6,11H* * *\$\$\$\$\$)	108
10	FORMAT (18HCOMPRESSOR DESIGN,6X8H PRCCF=,E15.8,8H ETACCF=,E15.8,	109
	18H WACCF=,E15.8,8H T21DS=,E15.8)	110
11	FORMAT (10HCNC WAS= ,E15.8,11H AND NOW= ,E15.8,24H CHECK PCNC I	111
	INPUT\$\$\$\$\$)	112
	END	113

\$IBFTC	WDUCT	DECK	
	SUBROUTINE	WDUCTI	1
	COMMON /	ALL/	2
1	WORD	,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
2	IGASM	,IDBURN,IAFTBN,IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,	4
3	ITRYS	,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ARR(6)	5
	COMMON /	DESIGN/	5
1	PCNFGJ	,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC,	7
2	ZFDS	,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WAFCF ,	8
3	ZCDS	,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,	9
4	T4DS	,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCOCF,ETABCF,	10

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5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPDF,T2DS      11
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPCF,ETLPCF,DHLPDF,T21DS      12
7T24DS ,WFDDSD ,DTDDSD ,ETAADDSD ,WA23DS ,DPDDSD ,DTDDCF ,ETADCF , 13
8T7DS ,WFADSD ,DTAFDSD ,ETAADDSD ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF , 14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 , 15
$PS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV 15
COMMON / FRONT/ 17
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 , 18
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 , 19
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 , 20
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB , 21
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 , 22
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF , 23
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS , 24
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT , 25
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB , 26
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLOB ,PCBLHP ,PCBLLP 27
COMMON/SIDE/ZYX(48)/BACK/YZX(72) 28
COMMON/DUMMYS/DUMMY(21),WA32,DPWGDS,DPWING,WA32DS,A38,AM38,V38, 29
1T38,H38,P38,TS38,PS38,T39,H39,P39,TS39,V39,AM39,A39,BPRINT,WG37, 30
2CVDWNG,FGMWNG,FGPWNG,FNWING,FNMAIN,FWOVEN,PS39,DIMMY(51) 31
COMMON/SPOOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI 32
1,ETAI,WACI,TFFIP,CNIP,ETATIP,DHTCIP,DHTI,BLIP,PCBLIP,PCNIGU,ZIDS, 33
2PCNIDS,PRIDS,ETAIDS,WAIDS,PRICF,ETAICF,WAICF,TFIPDS,CNIPDS,ETIPDS, 34
3TFIPCF,CNIPCF,ETAPCF,DHPCF,WAICDS,WAI,PCBLI,BLI ,T22DS,WA21 35
DIMENSION ERR(9) 36
EQUIVALENCE (ERR,DUMMY(11)),(PCBLID,DUMMY(61)) 37
DATA AWORD/6HWDUCTI/ 38
WORD=AWORD 39
IF (PCBLID.GT.0.) GO TO 3 40
DO 1 I=24,42 41
1 DUMMY(I)=0. 42
DO 2 I=44,49 43
2 DUMMY(I)=0. 44
RETURN 45
3 CONTINUE 46
P32=P21 47
H32=H21 48
T32=T21 49
BPRINT=WA32/WAC 50
WA32C=WA32*SQR(T32)/P32 51
IF (IDES.EQ.1) WA32DS=WA32C 52
DPWING=DPWGDS*WA32C/WA32DS 53
DPWING=AMIN1(1.0,DPWING) 54
P36=P32*(1.-DPWING) 55
T36=T32 56
H36=H32 57
CALL THERMO (P36,H36,T36,S36,XX2,1,0.0,0) 58
WG37=WA32 59
T37=T36 60
P37=P36 61
H37=H36 62
S37=S36 63
NOZD=0 64
CALL CONVRG (T37,H37,P37,S37,0.0,WG37,P1,IDES,A38,P38R,T38,H38,P38 65
1,S38,TS38,PS38,V38,AM38,ICON) 66
GO TO (5,5,5,4),ICON 67
4 CALL ERROR 68
5 T39=T38 69
H39=H38 70
P39=P38 71
S39=S38 72

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	TS39=TS38	73
	V39=V38	74
	AM39=AM38	75
	A39=A38	76
	PS39=PS38	77
	IDSHOC=ICON+3	78
	ERR(7)=(P38R-P38)/P38R	79
	IF (IDES.EQ.1) WRITE (6,6) A38,AM38,A39,AM39	80
	RETURN	81
C		82
C		83
6	FORMAT (18H0INTER DUCT DESIGN,5X,8H A38=,E15.8,8H AM38=,E15.8	84
	1,8H A39=,E15.8,8H AM39=,E15.8)	85
	END	86

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$IBFTC COCOMB DECK
SUBROUTINE COCOMB
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASM ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,
3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ARR(6)
COMMON /DESIGN/
1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELF ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
4T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,DPCODS ,DTCOCF ,ETABCF ,
5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS ,
7T24DS ,WFDDDS ,DTDUDS ,ETADDS ,WA23DS ,DPDUDS ,DTDUCF ,ETADCF ,
8T7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
$PS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON / FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLOB ,PCBLHP ,PCBLLP
COMMON /SIDE/ZYX(48)/BACK/YZX(72)
COMMON /DUMMYS/DIMMY(100)
COMMON /SPOOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI
1,ETAI ,WACI ,TFFIP ,CNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,PCBLIP ,PCNIGU ,ZIDS ,
2PCNIDS ,PRIDS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,TFIPDS ,CNIPDS ,ETIPDS ,
3TFIPCF ,CNIPCF ,ETAPCF ,DHIPCF ,WAICDS ,WAI ,PCBLI ,BLI ,T22DS ,WA21
EQUIVALENCE (FAR50,DIMMY(21)) , (WG50,DIMMY(20)) , (FXFN2M,DIMMY(50)) ,
1(FXM2CP,DIMMY(51))
EQUIVALENCE (HTF,DIMMY(62)) , (HCN,DIMMY(65)) , (HDH,DIMMY(68)) ,
1(HDHC,DIMMY(69)) , (HETA,DIMMY(74))
LOGICAL FXFN2M ,FXM2CP
DIMENSION ERR(9)
EQUIVALENCE (ERR,DIMMY(11))
COMMON / COMB/PSI(15),DELT(15,15),ETA(15,15),NPS,NPT(15)

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	DIMENSION Q(9),DUMBO(15,15)	42
	DATA AWORD/6HCOCOMB/	43
	WORD=AWORD	44
	Q(2)=0.	45
	Q(3)=0.	46
	P3PSI=14.696*P3	47
	WA3C=WA3*SQRT(T3)/P3PSI	48
	IF (IDES.EQ.1) WA3CDS=WA3C	49
	DPCUM=DPCODS*(WA3C/WA3CDS)	50
	IF (DPCOM.GT.1.) DPCOM=1.	51
	P4=P3*(1.-DPCOM)	52
1	IF (T4.GT.4000.) T4=4000.	53
	IF (T4.GE.1000.) GO TO 2	54
	T4=1000.	55
	IF (MODE.EQ.1) MAPEDG=1	56
2	DTCO=T4-T3	57
	IF (IDES.NE.1) GO TO 3	58
	DTCODS=DTCO	59
	DTCOCF=DTCODS/DTCO	60
3	DTCO=DTCOCF*DTCO	61
	P3PSIN=P3PSI	62
	CALL SEARCH (-1.,P3PSIN,DTCO,ETAB,DUMMY,PSI(1),NPS,DELT(1,1),ETA(1	63
	1,1),DJMBO(1,1),NPT(1),15,15,IGO)	64
	IF (IGO.EQ.7) CALL ERROR	65
	IF (IDES.NE.1) GO TO 4	66
	ETABCF=ETABDS/ETAB	67
4	ETAB=ETABCF*ETAB	68
	HV=((((-4594317E-19*T4)-.2034116E-15)*T4+.2783643E-11)*T4+.2051	69
	1501E-07)*T4-.2453116E-03)*T4-.9433296E-01)*T4+.1845537E+05	70
	CALL THERMO (P4,HA,T4,XX1,XX2,0,0,0,0)	71
	FAR4=(HA-H3)/(HV*ETAB)	72
	IF (FAR4.LT.0.) FAR4=0.	73
	WFBX=FAR4*WA3	74
	IF (MODE.NE.2) GO TO 7	75
	ERRW=(WFB-WFBX)/WFB	76
	DIR=SQRT(WFB/WFBX)	77
	CALL AFQUIR (Q(1),T4,ERRW,0.,20.,0.0001,DIR,T4T,IGO)	78
	GO TO (5,8,6),IGO	79
5	T4=T4T	80
	GO TO 1	81
6	CALL ERROR	82
7	WFB=WFBX	83
8	CALL THERMO (P4,H4,T4,S4,XX2,1,FAR4,0)	84
	WG4=WFB+WA3	85
	IF (IDES.EQ.1) WRITE (6,10) WA3CDS,ETABCF,DTCOCF	86
	IF (FXM2CP) GO TO 9	87
	CALL CJHPTB	88
	RETURN	89
9	P50=P4	90
	H50=H4	91
	T50=T4	92
	S50=S4	93
	FAR50=FAR4	94
	WG50=WG4	95
C	SET HIGH PRESSURE TURBINE PARAMETERS TO ZERO, NOT USED	96
	HTF=0.	97
	HCN=0.	98
	HDH=0.	99
	HDHC=0.	100
	HETA=0.	101
	CALL CJIPTB	102
	RETURN	103
C		104

C		105
C		106
10	FORMAT (17HOCOMBUSTOR DESIGN,7X8H WA3CDS=,E15.8,8H ETABCF=,E15.8,8	107
	1H DTCJCF=,E15.8)	108
	END	109

\$IBFTC	COHPTB	DECK	
	SUBROUTINE	COHPTB	1
	COMMON /	ALL/	2
	1WORD	,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
	2IGASM	X,IDBURN,IAFTBN,IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,	4
	3ITRYS	,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ARR(6)	5
	COMMON /	DESIGN/	6
	1PCNFGJ	,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC,	7
	2ZFDS	,PCNFDS,PRFDS ,ETAFDS,WAFFDS ,PRFCF ,ETAFCF,WAFCF ,	8
	3ZCDS	,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,	9
	4T4DS	,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCOCF,ETABCF,	10
	5TFHPDS	,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,	11
	6TFLPDS	,CNLPDS,ETLPDS,TFLPCF,CNLPDF,ETLPCF,DHLPDF,T21DS ,	12
	7T24DS	,WFDDDS ,DTDUDS,ETADDSD,WA23DS,DPDUDS,DTDUCF,ETADCF,	13
	8T7DS	,WFADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF,	14
	9A55	,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,	15
	\$PS55	,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV	16
	COMMON /	FRONT/	17
	1T1	,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,	18
	2T21	,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,	19
	3T4	,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,	20
	4T55	,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,	21
	5CNF	,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,	22
	6CNC	,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,	23
	7CNHP	,ETATHP,DHTCHP,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,	24
	8CNLP	,ETATLP,DHTCLP,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,	25
	9AM	,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,	26
	\$TFFHP	,TFFLP ,PCBLF ,PCBLC ,PCBLDU,PCBLOB,PCBLHP,PCBLLP	27
	COMMON /	SIDE/QXQ(48)/BACK/QWQ(72)	28
	COMMON /	DUMMYS/DUMMY(100)	29
	COMMON /	SPOOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI	30
	1,ETAI	,JACI,TFFIP,CNIP,ETATIP,DHTCIP,DHTI,BLIP,PCBLIP,PCNIGU,ZIDS,	31
	2PCNIDS	,PRIDS,ETAIDS,WAIDS,PRICF,ETAICF,WAICF,TFIPDS,CNIPDS,ETIPDS,	32
	3TFIPCF	,CNIPCF,ETAPCF,DHPCF,WAICDS,IAI,PCBLI,BLI ,T22DS,WA21	33
	DIMENSION	ERR(9)	34
	EQUIVALENCE	(ERR,DUMMY(11)),(WG50,DUMMY(20)),(FAR50,DUMMY(21))	35
	EQUIVALENCE	(FXFN2M,DUMMY(50)),(FXM2CP,DUMMY(51))	36
	EQUIVALENCE	(DUMSPL,DUMMY(59))	37
	EQUIVALENCE	(TFFACT,DUMMY(62)),(CNACT,DUMMY(65)),(CHCACT,DUMMY(68))	38
	1,(DHTACT	,DUMMY(69)),(ETAACD,DUMMY(74))	39
	EQUIVALENCE	(ITF,DUMMY(63)),(ICN,DUMMY(66)),(IDH,DUMMY(70)),	40
	1(IDHC	,DUMMY(71)),(IETA,DUMMY(75))	41
	LOGICAL	FXFN2M,FXM2CP,DUMSPL	42
	COMMON /	HTURB/TFFX(15),CNX(15,15),DHTCX(15,15),ETATX(15,15),	43
	1NTFFS	,NPTTFF(15)	44
	DATA	4WORD,WLO,WHI/6HCOHPTB,6H (LO) ,6H (HI) /	45
	WORD=	4WORD	46
	IF (IDES	.EQ.0) GO TO 1	47
	CNHPCF	=CNHPDS*SQRT(T4)/PCNC	48
1	CNHPCF	=CNHPDS*PCNC/SQRT(T4)	49
	CNHPS	=CNHP	50
	TFFHPS	=TFFHP	51

	CALL SEARCH (-1.,TFFHP,CNHP,DHTCHP,ETATHP,TFFX(1),NTFFS,CNX(1,1),D	52
	1HTCX(1,1),ETATX(1,1),NPTTFF(1),15,15,IGJ)	53
	IF (IGJ.EQ.1.OR.IGJ.EQ.11.OR.IGJ.EQ.21) WRITE (8,9) TFFHPS,WLO	54
	IF (IGJ.EQ.2.OR.IGJ.EQ.12.OR.IGJ.EQ.22) WRITE (8,9) TFFHPS,WHI	55
	IF (IGJ.EQ.10.OR.IGJ.EQ.11.OR.IGJ.EQ.12) WRITE (8,10) CNHPS,WLO	56
	IF (IGJ.EQ.20.OR.IGJ.EQ.21.OR.IGJ.EQ.22) WRITE (8,10) CNHPS,WHI	57
	IF (IGJ.NE.7) GO TO 2	58
	CALL ERROR	59
	RETURN	60
2	MAPGO=0	61
	IF (ABS(TFFHPS-TFFHP).LE.0.001*TFFHPS) GO TO 3	62
	MAPGO=1	63
	IF (ABS(CNHPS-CNHP).GT.0.001*CNHPS) MAPGO=3	64
	GO TO 4	65
3	IF (ABS(CNHPS-CNHP).GT.0.001*CNHPS) MAPGO=2	66
4	IF (MAPGO.GT.0) CALL MAPBAC (1,MAPGO,TFFHPS,TFFHP,CNHPS,CNHP,PCNC,	67
	1T4,MODE,NOMAP,NUMMAP)	68
	IF (NOMAP.GT.0) RETURN	69
	TFHCAL=WG4*SQR(T4)/(14.696*P4)	70
	BTUEXT=0.706705*HPEXT	71
	DHTCC=(BTUEXT+WAC*(H3-H21))/(WG4*T4)	72
	IF (IDES.EQ.0) GO TO 5	73
	TFHPCF=TFHPDS/TFHCAL	74
	DHHPCF=DHTCC/DHTCHP	75
	ETHPCF=ETHPDS/ETATHP	76
	WRITE (6,11) CNHPCF,TFHPCF,ETHPCF,DHHPCF	77
5	TFHCAL=TFHPCF*TFHCAL	78
	DHTCHP=DHHPCF*DHTCHP	79
	ETATHP=ETHPCF*ETATHP	80
	DHTC=DHTCC*T4	81
	TFFACT=TFHCAL/TFHPCF	82
	CNACT=CNHP/CNHPCF	83
	DHCACT=DHTCHP/DHHPCF	84
	DHTACT=DHTC	85
	ETAECT=ETATHP	86
	ERR(1)=(TFHCAL-TFFHP)/TFHCAL	87
	ERR(2)=(DHTCC-DHTCHP)/DHTCC	88
	CALL THTURB (DHTC,ETATHP,FAR4,H4,S4,P4,T50,H50,S50,P50)	89
	IF (BLHP.LE.0.) GO TO 6	90
	FAR5=WFB/(WA3+BLHP)	91
	FAR5=WFB/(WA3+BLHP)	92
	WG50=W34+BLHP	93
	H50=(BLHP*H3+WG4*H50)/WG50	94
	CALL THERMO (P50,H50,T50,S50,XX2,1,FAR50,1)	95
	GO TO 7	96
6	FAR50=FAR4	97
	WG50=W34	98
7	IF (FXFN2M.OR.DUMSPL) GO TO 8	99
	CALL CCIPT8	100
	RETURN	101
8	P5=P50	102
	H5=H50	103
	T5=T50	104
	S5=S50	105
	FAR5=FAR50	106
	WG5=W350	107
C	SET MIDDLE TURBINE PARAMETERS TO ZERO, NOT USED	108
	ITF=0	109
	ICN=0	110
	IDH=0	111
	IDHC=0	112
	IETA=0	113

CALL COIPTB	114
RETURN	115
C	116
C	117
C	118
9 FORMAT (19H0*****TFFHP OFF MAP,F10.4,2XA6,11H*****\$\$\$\$\$\$)	119
10 FORMAT (19H0***** CVHP OFF MAP,F10.4,2XA6,11H*****\$\$\$\$\$\$)	120
11 FORMAT (20H0H.P. TURBINE DESIGN,5X7HCNHPCF=,E15.8,8H TFHPCF=,E15.8	121
1,8H ETHPCF=,E15.8,8H DHHPCF=,E15.8)	122
END	123

\$IBFTC COIPTB DECK	
SUBROJTINE COIPTB	1
COMMON / ALL/	2
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
2IGASMX, IDBURN, IAFTBN, IDCD ,IMCD ,IDSHJC,IMSHOC,NOZFLT,	4
3ITRYS ,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ARR(6)	5
COMMON /DESIGN/	6
1PCNFGJ,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC,	7
2ZFDS ,PCNFDS,PRFDS ,ETAADS,WAADS ,PRFCF ,ETAFCF,WAFCF ,	8
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,	9
4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCOCF,ETABCF,	10
5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,	11
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPDF,ETLPDF,DHLPDF,T21DS ,	12
7T24DS ,WFDDSD,DTDUDS,ETADDS,WA23DS,DPDUDS,DTDUCF,ETADCF,	13
8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF,	14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,	15
\$PS55 ,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV	16
COMMON / FRONT/	17
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,	18
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,	19
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,	20
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,	21
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,	22
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,	23
7CNHP ,ETATHP,DHTCHP,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,	24
8CNLP ,ETATLP,DHTCLP,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,	25
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,	26
\$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU,PCBLOB,PCBLHP,PCBLLP	27
COMMON/SIDE/QXQ(48)/BACK/QWQ(72)	28
COMMON/DUMMYS/DUMMY(100)	29
COMMON/SPOOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI	30
1,ETAI,WACI,TFFIP,CNIP,ETATIP,DHTCIP,DHTI,BLIP,PCBLIP,PCNIGU,ZIDS,	31
2PCNIDS,PRIDS,ETAIDS,WAIDS,PRICF,ETAICF,WAICF,TFIPDS,CNIPDS,ETIPDS,	32
3TFIPCF,CNIPCF,ETAPCF,DHIPCFC,WAICDS,WAI,PCBLI,BLI ,T22DS,WA21	33
COMMON/TERBMD/DHMDSV,TFMDSV,CNMDSV,ETMDSV,DHMDSD	34
COMMON/ITURB/TFFX(15),CNX(15,15),DHTCX(15,15),ETATX(15,15),	35
1NTFFS,NPTTFF(15)	36
EQUIVALENCE (FXFN2M,DUMMY(50)),(FXM2CP,DUMMY(51))	37
LOGICAL FXFN2M,FXM2CP	38
EQUIVALENCE (AFTFAN,DUMMY(58))	39
LOGICAL AFTFAN	40
COMMON/HTURB/TFFY(15),CNY(15,15),DHTCY(15,15),ETATY(15,15),NTFYS,	41
1NPTTSF(15)	42
DIMENSION ERR(9)	43
EQUIVALENCE (ERR,DUMMY(11)),(WG50,DUMMY(20)),(FAR50,DUMMY(21))	44
EQUIVALENCE(TFFACT,DUMMY(63)),(CNACT,DUMMY(66)),(DHTACT,DUMMY(70))	45
1,(DHTACT,DUMMY(71)),(ETAACF,DUMMY(75))	46

	DATA AWORD,WLO,WHI/6HCOIPT8,6H (LO) ,6H (HI) /	47
	H22SAV=H22	48
	IF (AFTFAN) H22=H2	49
	WORD=AWORD	50
	IF (IDES.EQ.0) GO TO 1	51
	CNIPCF=CNIPDS*SQRT(T50)/PCNI	52
	IF (FXM2CP) CNIPCF=CNHPDS*SQRT(T50)/PCNI	53
1	CNIP=CNIPCF*PCNI/SQRT(T50)	54
	CNIPS=CNIP	55
	TFFIPS=TFFIP	56
	IF (FXM2CP) GO TO 2	57
	CALL SEARCH (-1.,TFFIP,CNIP,DHTCIP,ETATIP,TFFX(1),NTFFS,CNX(1,1),D	58
	1HTCX(1,1),ETATX(1,1),NPTTFF(1),15,15,IGD)	59
2	IF (FXM2CP) CALL SEARCH (-1.,TFFIP,CNIP,DHTCIP,ETATIP,TFFY(1),NTFY	60
	1S,CNY(1,1),DHTCY(1,1),ETATY(1,1),NPTTSF(1),15,15,IGD)	61
	IF (IGD.EQ.1.OR.IGD.EQ.11.OR.IGD.EQ.21) WRITE (8,9) TFFIPS,WLO	62
	IF (IGD.EQ.2.OR.IGD.EQ.12.OR.IGD.EQ.22) WRITE (8,9) TFFIPS,WHI	63
	IF (IGD.EQ.10.OR.IGD.EQ.11.OR.IGD.EQ.12) WRITE (8,9) CNIPS,WLO	64
	IF (IGD.EQ.20.OR.IGD.EQ.21.OR.IGD.EQ.22) WRITE (8,10) CNIPS,WHI	65
	IF (IGD.NE.7) GO TO 3	66
	CALL ERROR	67
	RETURN	68
3	MAPGO=0	69
	IF (ABS(TFFIPS-TFFIP).LE.0.001*TFFIPS) GO TO 4	70
	MAPGO=1	71
	IF (ABS(CNIPS-CNIP).GT.0.001*CNIPS) MAPGO=3	72
	GO TO 5	73
4	IF (ABS(CNIPS-CNIP).GT.0.001*CNIPS) MAPGO=2	74
5	IF (MAPGO.GT.0) CALL MAPBAC (3,MAPGO,TFFIPS,TFFIP,CNIPS,CNIP,PCNI,	75
	1T4,MOJE,NOMAP,NUMMAP)	76
	IF (NOMAP.GT.0) RETURN	77
	TFICAL=WG50*SQRT(T50)/(14.696*P50)	78
	DHTIC=(WAI*(H21-H22))/(WG50*T50)	79
	IF (FXM2CP) DHTIC=(.706705*HPEXT+WAC*(H3-H21)+WAI*(H21-H22))/(WG50	80
	1*T50)	81
	IF (IDES.EQ.0) GO TO 6	82
	TFIPCF=TFIPDS/TFICAL	83
	DHPCF=DHTIC/DHTCIP	84
	ETPCF=ETIPDS/ETATIP	85
	IF (FXM2CP) TFIPCF=TFHPDS/TFICAL	86
	IF (FXM2CP) ETPCF=ETHPDS/ETATIP	87
	WRITE (6,11) CNIPCF,TFIPCF,ETPCF,DHPCF	88
6	TFICAL=TFIPCF*TFICAL	89
	DHTCIP=DHPCF*DHTCIP	90
	ETATIP=ETPCF*ETATIP	91
	DHTI=DHTIC*T50	92
	TFFACT=TFICAL/TFIPCF	93
	CNACT=CNIP/CNIPCF	94
	DHCACT=DHTCIP/DHPCF	95
	DHTACT=DHTI	96
	ETAFACT=ETATIP	97
	N1=8	98
	N2=9	99
	IF (FXM2CP) N1=1	100
	IF (FXM2CP) N2=2	101
	ERR(N1)=(TFICAL-TFFIP)/TFICAL	102
	ERR(N2)=(DHTIC-DHTCIP)/DHTIC	103
	CALL THTUR8 (DHTI,ETATIP,FAR50,H50,S50,P50,T5,H5,S5,P5)	104
	IF (BLIP.LE.0.) GO TO 7	105
	FAR5=WFB/(WA3+BLHP+BLIP)	106
	WG5=WG50+BLIP	107
	H5=(BLIP*H3+WG50*H5)/WG5	108

	CALL THERMO (P5,H5,T5,S5,XX2,1,FAR5,1)	109
	GO TO 3	110
7	FAR5=FAR50	111
	WG5=W350	112
	H22=H22SAV	113
8	CALL COLPTB	114
	RETURN	115
C		116
C		117
C		118
9	FORMAT (19H0*****TFFIP OFF MAP,F10.4,2XA6,11H*****\$\$\$\$\$)	119
10	FORMAT (19H0***** CNIP OFF MAP,F10.4,2XA6,11H*****\$\$\$\$\$)	120
11	FORMAT (20H0I.P. TURBINE DESIGN,5X7HCNIPCF=,E15.8,8H TFIPCF=,E15.8	121
	1,8H ETIPCF=,E15.8,8H DHIPCF=,E15.8)	122
	END	123

\$IBFTC COLPTB DECK		
SUBROUTINE COLPTB		1
COMMON / ALL/		2
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,		3
2IGASMX,IDBURN,IAFTBN,IDCD ,IMCD ,IDSHJC,IMSHOC,NOZFLT,		4
3ITRYS ,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ARR(6)		5
COMMON /DESIGN/		6
1PCNFGJ,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC,		7
2ZFDS ,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WAFCF ,		8
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,		9
4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCOCF,ETABCF,		10
5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,		11
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPDF,ETLPDF,DHLPDF,T21DS ,		12
7T24DS ,WFDDDS ,DTDDDS,ETADDS,WA23DS,DPDUDS,DTDUCF,ETADCF,		13
8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF,		14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,		15
\$PS55 ,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV		16
COMMON / FRONT/		17
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,		18
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,		19
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,		20
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLDB ,		21
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,		22
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,		23
7CNHP ,ETATHP,DHTCHP,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,		24
8CNLP ,ETATLP,DHTCLP,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,		25
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,		26
\$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU,PCBLOB,PCBLHP,PCBLLP		27
COMMON/SIDE/QXQ(48)/BACK/QWQ(72)		28
COMMON/DUMMYS/DUMMY(100)		29
EQUIVALENCE (FXFN2M,DUMMY(50)),(FXM2CP,DUMMY(51))		30
LOGICAL FXFN2M,FXM2CP		31
EQUIVALENCE (AFTFAN,DUMMY(58))		32
EQUIVALENCE(TFFACT,DUMMY(64)),(CNACT,DUMMY(67)),(DHCACT,DUMMY(72))		33
1,(DHTACT,DUMMY(73)),(ETAACF,DUMMY(76))		34
LOGICAL AFTFAN		35
COMMON/SPOOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI		36
1,ETAI,WACI,TFFIP,CNIP,ETATIP,DHTCIP,DHTI,BLIP,PCBLIP,PCNIGU,ZIDS,		37
2PCNIDS,PRIDS,ETAIDS,WAIDS,PRICF,ETAICF,WAICF,TFIPDS,CNIPDS,ETIPDS,		38
3TFIPCF,CNIPCF,ETAPCF,DHPCF,WAICDS,WAI,PCBLI,BLI ,T22DS,WA21		39
COMMON /LTURB/TFFX(15),CNX(15,15),DHTCX(15,15),ETATX(15,15),		40
1NTFFS,NPTTFF(15)		41

	DIMENSION ERR(9)	42
	EQUIVALENCE (ERR,DUMMY(11))	43
	DATA AWORD,WLO,WHI/6HCOLPTB,6H (LO) ,6H (HI) /	44
	WORD=AWORD	45
	IF (IDES.EQ.0) GO TO 1	46
1	CNLP _{CF} =CNLP _{DS} *SQRT(T5)/PCNF	47
	CNLP=CNLP _{CF} *PCNF/SQRT(T5)	48
	CNLP _S =CNLP	49
	TFFLP _S =TFFLP	50
	CALL SEARCH (-1.,TFFLP,CNLP,DHTCLP,ETATLP,TFFX(1),NTFFS,CNX(1,1),D	51
	IHTCX(1,1),ETATX(1,1),NPTTFF(1),15,15,IGO)	52
	IF (IGO.EQ.1.OR.IGO.EQ.11.OR.IGO.EQ.21) WRITE (8,8) TFFLP _S ,WLO	53
	IF (IGO.EQ.2.OR.IGO.EQ.12.OR.IGO.EQ.22) WRITE (8,8) TFFLP _S ,WHI	54
	IF (IGO.EQ.10.OR.IGO.EQ.11.OR.IGO.EQ.12) WRITE (8,9) CNLP _S ,WLO	55
	IF (IGO.EQ.20.OR.IGO.EQ.21.OR.IGO.EQ.22) WRITE (8,9) CNLP _S ,WHI	56
	IF (IGO.NE.7) GO TO 2	57
	CALL ERROR	58
	RETURN	59
2	MAPGO=0	60
	IF (ABS(TFFLP _S -TFFLP).LE.0.001*TFFLP _S) GO TO 3	61
	MAPGO=1	62
	IF (ABS(CNLP _S -CNLP).GT.0.001*CNLP _S) MAPGO=3	63
	GO TO 4	64
3	IF (ABS(CNLP _S -CNLP).GT.0.001*CNLP _S) MAPGO=2	65
4	IF (MAPGO.GT.0) CALL MAPBAC (2,MAPGO,TFFLP _S ,TFFLP,CNLP _S ,CNLP,PCNF,	66
	IT4,MODE,NOMAP,NUMMAP)	67
	IF (NOMAP.GT.0) RETURN	68
	TFLCAL=WG5*SQRT(T5)/(14.696*P5)	69
	DHTCF=WAF*(H22-H2)/(WG5*T5)	70
	IF (FXFN2M) DHTCF=(WAF*(H22-H2)+WAI*(H21-H22))/(WG5*T5)	71
	IF (FXFN2M.AND.AFTFAN) DHTCF=(WAF*(H22-H2)+WAI*(H21-H2))/(WG5*T5)	72
	IF (IDES.EQ.0) GO TO 5	73
	TFLPCF=TFLPDS/TFLCAL	74
	DHLPCF=DHTCF/DHTCLP	75
	ETLPCF=ETLPDS/ETATLP	76
	WRITE (6,10) CNLP _{CF} ,TFLPCF,ETLPCF,DHLPCF	77
5	TFLCAL=TFLPCF*TFLCAL	78
	DHTCLP=DHLPCF*DHTCLP	79
	ETATLP=ETLPCF*ETATLP	80
	DHTF=DHTCF*T5	81
	TFFACT=TFLCAL/TFLPCF	82
	CNACT=CNLP/CNLP _{CF}	83
	DHCACT=DHTCLP/DHLPCF	84
	DHTACT=DHTF	85
	ETAACF=ETATLP	86
	ERR(3)=(TFLCAL-TFFLP)/TFLCAL	87
	ERR(4)=(DHTCF-DHTCLP)/DHTCF	88
	CALL T4TURB (DHTF,ETATLP,FAR5,H5,S5,P5,T55,H55,S55,P55)	89
	IF (BLLP.LE.0.) GO TO 6	90
	FAR55=WFB/(WA3+BLHP+BLLP)	91
	WG55=WG5+BLLP	92
	H55=(BLLP*H3+WG5*H55)/WG55	93
	CALL THERMO (P55,H55,T55,S55,XX2,1,FAR55,1)	94
	GO TO 7	95
6	FAR55=FAR5	96
	WG55=WG5	97
7	CALL FRTOSD	98
	RETURN	99
C		100
C		101
8	FORMAT (19H0*****TFFLP OFF MAP,F10.4,2XA6,11H*****\$\$\$\$\$\$)	102
9	FORMAT (19H0***** CNLP OFF MAP,F10.4,2XA6,11H*****\$\$\$\$\$\$)	103

10	FORMAT (20HOL.P. TURBINE DESIGN,5X7HCNLPCF=,E15.8,8H TFLPCF=,E15.8	104
	1,8H ETLPCF=,E15.8,8H DHLPCF=,E15.8)	105
	END	106

\$IBFTC CODUCT DECK

SUBROJTINE CODUCT

COMMON / ALL/

1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,

2IGASMX ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,

3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ARR(6)

COMMON /DESIGN/

1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC ,

2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,

3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,

4T4DS ,WF8DS ,DTCODS ,ETABDS ,WA3CDS ,DPCODS ,DTCOCF ,ETABCF ,

5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,

6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS ,

7T24DS ,WFDDDS ,DTDUDS ,ETAADS ,WA23DS ,DPDUDS ,DTDUCF ,ETADCF ,

8T7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,

9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,

\$PS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV

COMMON /FRONT /XX(80)

COMMON / SIDE/

1P1 ,WAF ,WAC ,BLF ,BLDU ,H3 ,DUMS1 ,DUMS2 ,

2T21 ,P21 ,H21 ,S21 ,T23 ,P23 ,H23 ,S23 ,

3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 ,

4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 ,

5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS ,DUMS3 ,

6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29

COMMON /BACK /ZZ(72)

COMMON /DUMMYS /DUMMY(100)

COMMON /SPOOL2 /T22 ,P22 ,H22 ,S22 ,T50 ,P50 ,H50 ,S50 ,WA22 ,ZI ,PCNI ,CNI ,PRI

1 ,ETAI ,WACI ,TFFIP ,CNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,PCBLIP ,PCNIGU ,ZIDS ,

2PCNIDS ,PRIDS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,TFIPDS ,CNIPDS ,ETIPDS ,

3TFIPCF ,CNIPCF ,ETAPCF ,DHIPCF ,WAICDS ,WAI ,PCBLI ,BLI ,T22DS ,WA21

DIMENSION ERR(9)

EQUIVALENCE (ERR ,DUMMY(11))

EQUIVALENCE (A24 ,DUMMY(4)) , (AM23 ,DUMMY(5))

EQUIVALENCE (AFTFAN ,DUMMY(58)) , (PCBLID ,DUMMY(61))

LOGICAL AFTFAN

DIMENSION Q(9)

DATA AWORD1 ,AWORD2 /6H CODUCT ,6HDNOZZL /

WORD=AWORD1

Q(2)=0.

Q(3)=0.

AJ=779.26

CAPSF=2116.2170

GOGO=0.0

G=32.174049

WAX=WAF-WAI-BLF

IF (PCBLID.EQ.0.) WAX=WAF-WAC-BLF

IF (AFTFAN) WAX=WAF-BLF

WAD=WAX+BLDU

P23=P22

C*** DRY L3SS

H23=(3LDU*H3+WAX*H22)/WAD

CALL THERMO (P23 ,H23 ,T23 ,S23 ,XX2 ,1 ,0.0 ,1)

WA23C=WAD*SQRT(T23)/P23

	IF (IDES.EQ.1) WA23DS=WA23C	54
	BYPASS=(WAF-WAI)/WAI	55
	IF (A=TFAN) BYPASS=WAF/WAI	56
	DPDUC=DPDUDS*(WA23C/WA23DS)	57
	IF (DPDUC.GT.1.) DPDUC=1.0	58
	P24=P23*(1.-DPDUC)	59
	CALL PROCOM (0.,T23,XX1,XX2,XX3,XX4,PHI23,XX6)	60
	IF (IGASM.GT.0) IDBURN=0	61
	IF (SJB.FAN.GT.0.) GO TO 7	62
	AM24=AM23	63
	TS24=T23*0.875	64
1	DO 2 I=1,15	65
	CALL PROCOM (0.,TS24,CS24,AK24,CP24,REX24,PHIS24,HS24)	66
	V24=AM24*CS24	67
	HSCAL=+23-V24**2/(2.*G*AJ)	68
	DELHS=+SCAL-HS24	69
	IF (ABS(DELHS).LE.0.001*HSCAL) GO TO 3	70
2	TS24=TS24+DELHS/CP24	71
	GO TO 11	72
3	C1=P24*SQR(TG/(T23*AJ))*CAPSF	73
	IF (IDES.NE.1) GO TO 4	74
	IF (GJGO.GT.0.) GO TO 4	75
	ASTOA=((AK24+1.)/2.)*((AK24+1.)/(2.*(AK24-1.)))*AM24*(1.+(((AK24-11.)/2.)*AM24**2))*((AK24+1.)/(2.*(AK24-1.)))	76
	EQWCR=SQR(TG*AK24/REX24/AJ)/(SQR(T518.69)/2116.2)*(2.0/(AK24+1.))*1*((AK24+1.)/2./((AK24-1.)))	77
	WA23CC=WA23C/SQR(T518.69)	78
	A24=L./ASTOA*WA23CC/EQWCR	79
	GOGO=1.0	80
4	WQA=WAD/A24	81
	WQAT=C1*SQR(TAK24/REX24)*AM24/(1.+(AK24-1.)*AM24**2/2.)*((AK24+1.1)/(2.*(AK24-1.)))	82
	DIR=WQA/WQAT	83
	EW=(WQA-WQAT)/WQA	84
	CALL AFQUIR (Q(1),AM24,EW,0.,30.,0.001,DIR,AM24T,IGO)	85
	GO TO (5,6,11),IGO	86
5	AM24=AM24T	87
	IF (AM24.GT.1.0) AM24=0.5	88
	GO TO 1	89
6	PS24=P24/EXP((PHI23-PHIS24)/REX24)	90
7	IF (IDBURN.GT.0) GO TO 8	91
C***	NON-DJCT BURNING	92
	T24=T23	93
	WFD=0.	94
	FAR24=0	95
	GO TO 17	96
8	IF (IDBURN.EQ.2) T24=T23+2000.	97
9	IF (T24.GT.4000.) T24=4000.	98
	IF (T24.LT.T23) T24=T23	99
C***	DUCT BURNING	100
	RHD42=CAPSF*PS24/(AJ*REX24*TS24)	101
	PS42=PS24	102
	V42=V24	103
	Q(2)=0.	104
	Q(3)=0.	105
C ***	IF DESIRED, ENTER CALCULATIONS FOR ETAD HERE	106
	HV=((((((-.4594317E-19*T24)-.2034116E-15)*T24+.2783643E-11)*T24+.21051501E-07)*T24-.2453116E-03)*T24-.9433296E-01)*T24+.1845537E+05	107
	CALL THERMO (P24,HA,T24,XX1,XX2,0,0.0,0)	108
	FAR24=(HA-H23)/(HV*ETAD)	109
	IF (FAR24.LT.0.) FAR24=0.	110
	WFDX=FAR24*WAD	111
	IF (IDBURN.NE.2) GO TO 12	112

	ERRW=(WFD-WFDX)/WFD	117
	DIR=SQRT(WFD/WFDX)	118
	CALL AFQUIR (Q(1),T24,ERRW,0.,20.,0.0001,DIR,T24T,IGO)	119
	GO TO (10,13,11),IGO	120
10	T24=T24T	121
	GO TO 9	122
11	CALL ERROR	123
12	WFD=WFDX	124
13	CONTINUE	125
C***	MOMENTUM LOSS	126
	WG24=WFD+WAD	127
	CALL PROCOM (FAR24,T24,XX1,XX2,XX3,REX24,PHI24,H24)	128
	RHO24=CAPSF*P24/(AJ*REX24*T24)	129
	V24=WG24/(RHO24*A24)	130
	Q(2)=0.	131
	Q(3)=0.	132
	PS24=PS42-0.01	133
14	RHO24=WG24/(V24*A24)	134
	HS24=H24-V24**2/(2.*G*AJ)	135
	CALL THERMO (1.0,HS24,TS24,PHIS24,XX2,1,FAR24,1)	136
	IF (TS24.GE.301.) GO TO 15	137
	CALL THERMO (1.0,HS24,400.,PHIS24,XX2,1,FAR24,1)	138
	V24=SQRT(2.*G*AJ*(H24-HS24))	139
	GO TO 14	140
15	PS24=RHO24*AJ*REX24*TS24/CAPSF	141
	PS24A=PS42+(RHO42*V42**2-RHO24*V24**2)/(G*CAPSF)	142
	DIR=SQRT(ABS(PS24/PS24A))	143
	EP=(PS24-PS24A)/PS24	144
	CALL AFQUIR (Q(1),V24,EP,0.,50.,0.001,DIR,V24T,IGO)	145
	V24=V24T	146
	IF (V24.LT.25.) V24=25.	147
	GO TO (14,16,11),IGO	148
16	P24=PS24*EXP((PHI24-PHIS24)/REX24)	149
	CALL PROCOM (FAR24,TS24,CS24,XX2,XX3,XX4,XX5,XX6)	150
	AM24=V24*CS24	151
17	CALL THERMO (P24,H24,T24,S24,XXI,1,FAR24,0)	152
	WG24=WFD+WAD	153
	T25=T24	154
	P25=P24	155
	H25=H24	156
	S25=S24	157
	IF (IGASMX.GT.0) GO TO 21	158
	WORD=AWORD2	159
	A28SAV=A28	160
	A29SAV=A29	161
	NOZD=0	162
	IDNOZ=0	163
	IF (NOZFLT.EQ.2.OR.NOZFLT.EQ.3) NOZD=1	164
	IF (IDES.EQ.1.OR.IDBURN.GT.0.OR.NOZD.EQ.1) IDNOZ=1	165
	IF (IDCD.EQ.1) GO TO 18	166
	CALL CONVRG (T25,H25,P25,S25,FAR24,WG24,P1,IDNOZ,A28,P25R,T28,H28,	167
	1P28,S28,TS28,PS28,V28,AM28,ICON)	168
	GO TO (19,19,19,11),ICON	169
18	CALL CONDIV (T25,H25,P25,S25,FAR24,WG24,P1,IDNOZ,A28,A29,P25R,T28,	170
	1H28,P28,S28,T29,H29,P29,S29,TS28,TS29,PS28,PS29,V28,V29,AM28,AM29,	171
	2ICON)	172
	IDSHOC=ICON	173
	GO TO (20,20,20,11),ICON	174
19	T29=T28	175
	H29=H28	175
	P29=P28	177
	S29=S28	178

	TS29=TS28	179
	PS29=PS28	180
	V29=V28	181
	AM29=AM28	182
	A29=A28	183
	IDSHOC=ICON+3	184
20	ERR(5)=(P25R-P25)/P25R	185
	IF (IDNOZ.EQ.1) WRITE (6,22) A28,AM28,A29,AM29	186
21	CALL FASTBK	187
	RETURN	188
C		189
C		190
22	FORMAT (19HODUCT NOZZLE DESIGN,5X8H A28=,E15.8,8H AM28=,E15.8	191
	1,8H A29=,E15.8,8H AM29=,E15.8)	192
	END	193

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$IBFTC COMIX  DECK
SUBROUTINE COMIX
COMMON /  ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASMX,IDBURN,IAFTBN,IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,
3ITRYS ,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ARR(6)
COMMON /DESIGN/
1PCNFGJ,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC,
2ZFDS ,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WAFCF ,
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,
4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCOCF,ETABCF,
5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPFC,ETLPCF,DHLPCF,T21DS ,
7T24DS ,WFDDDS ,DTODDS,ETADDS,WA23DS,DPDUDS,DTDUCF,ETADCF,
8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF,
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
$PS55 ,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV
COMMON/FRONT/QZQ(80)/SIDE/QWQ(48)
COMMON /  BACK/
1T55 ,P55 ,H55 ,S55 ,T25 ,P25 ,H25 ,S25 ,
2WFB ,WG55 ,FAR55 ,WFD ,WG24 ,FAR24 ,P1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC
COMMON/DUMMYS/DUMMY(100)
COMMON/SPOOL2/TWOSPL(44)
EQUIVALENCE (ERR,DUMMY(11))
EQUIVALENCE (ZF,QZQ(68)),(PCNF,QZQ(69))
DIMENSION ERR(9)
COMMON/LOOPPR/KKGO,PRFNEW,PRCNEW
DATA AWORD/6H COMIX/
DIMENSION QQ(9)
WORD=AWORD
AJ=773.26
CAPSF=2116.2170
G=32.174049
CALL PROCOM (FAR55,T55,XX1,XX2,XX3,XX4,PHI55,XX5)

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	CALL PROCOM (FAR24,T25,XX1,XX2,XX3,XX4,PHI25,XX5)	41
	IF (IDES.EQ.0) GO TO 12	42
C ***	CALCULATE A55 AND A25 WITH PS25=PS55	43
	IF (PS55.EQ.0.) GO TO 3	44
	TS55=T55*(PS55/P55)**0.286	45
	DO 1 I=1,15	46
	CALL PROCOM (FAR55,TS55,CS55,AK55,CP55,REX55,PHIS55,HS55)	47
	PHIS=PHI55-REX55*ALOG(P55/PS55)	48
	DELPHI=PHIS-PHIS55	49
	IF (ABS(DELPHI).LE.0.0001*PHIS) GO TO 6	50
1	TS55=TS55*EXP(4.0*DELPHI)	51
2	CALL ERROR	52
	RETURN	53
3	TS55=0.875*T55	54
	DO 4 I=1,15	55
	CALL PROCOM (FAR55,TS55,CS55,AK55,CP55,REX55,PHIS55,HS55)	56
	V55=AM55*CS55	57
	HSCAL=H55-V55**2/(2.*G*AJ)	58
	DELHS=HSCAL-HS55	59
	IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 5	60
4	TS55=TS55+DELHS/CP55	61
	GO TO 2	62
5	PS55=P55/EXP((PHI55-PHIS55)/REX55)	63
	IF (PS55.GT.P25.AND.IDES.EQ.1.AND.IGASMXX.GT.0) GO TO 45	64
6	IF (H55.GT.HS55) GO TO 7	65
	WRITE (8,46) P55,PS55,T55,TS55,H55,HS55	66
	CALL ERROR	67
7	V55=SQRT(2.*G*AJ*(H55-HS55))	68
	RHO=CAPSF*PS55/(AJ*REX55*TS55)	69
	A55=WG55/(RHO*V55)	70
	AM55=V55/CS55	71
	IF (IGASMXX.GT.0) GO TO 8	72
	WRITE (6,47) A55,AM55	73
	GO TO 41	74
8	PS25=PS55	75
	TS25=T25*(PS25/P25)**0.286	76
	DO 9 I=1,15	77
	CALL PROCOM (FAR24,TS25,CS25,AK25,CP25,REX25,PHIS25,HS25)	78
	PHIS=PHI25-REX25*ALOG(P25/PS25)	79
	DELPHI=PHIS-PHIS25	80
	IF (ABS(DELPHI).LE.0.0001*PHIS) GO TO 10	81
9	TS25=TS25*EXP(4.0*DELPHI)	82
	GO TO 2	83
10	IF (H25.GT.HS25) GO TO 11	84
	WRITE (8,48) P25,PS25,T25,TS25,H25,HS25	85
	CALL ERROR	86
11	V25=SQRT(2.*G*AJ*(H25-HS25))	87
	RHO=CAPSF*PS25/(AJ*REX25*TS25)	88
	A25=WG24/(RHO*V25)	89
	AM25=V25/CS25	90
	WRITE (6,49) A55,AM55,A25,AM25	91
	GO TO 27	92
C ***	CALCULATE PS55 AND PS25	93
12	WQA=WG55/A55	94
	C1=P55*SQRT(G/(T55*AJ))*CAPSF	95
	MCON=0	96
	QQ(2)=0.	97
	QQ(3)=0.	98
	AM55=0.50	99
	TS55=0.875*T55	100
13	DO 14 I=1,15	101
	CALL PROCOM (FAR55,TS55,CS55,AK55,CP55,REX55,PHIS55,HS55)	102
	V55=AM55*CS55	103

	HSCAL=+55-V55**2/(2.*G*AJ)	104
	DELHS=HSCAL-HS55	105
	IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 15	106
14	TS55=TS55+DELHS/CP55	107
	GO TO 2	108
15	WQAT=C1*SQRT(AK55/REX55)*AM55/(1.+(AK55-1.)*AM55**2/2.)*((AK55+1.1)/(2.*(AK55-1.)))	109
	AMX=AM55	110
	IGOGO=0	111
16	DIR=WQA/WQAT	112
	EW=(WQA-WQAT)/WQA	113
	CALL AFQUIR (QQ(1),AMX,EW,0.,30.,0.0005,DIR,AMXT,ICON)	114
	GO TO (17,22,2),ICON	115
17	IF (AMXT.LE.1.0) GO TO 20	116
	AMXT=0.7	117
	MCON=MCON+1	118
	IF (MCON.LE.1) GO TO 20	119
	IF (MODE.EQ.3) GO TO 19	120
	PCNF=0JMD1	121
	WRITE (8,50) PCNF,AMX,P55,PS55,P25,PS25	122
	PCNF=1.01*PCNF	123
	DUMD1=PCNF	124
18	NOMAP=7	125
	RETURN	126
19	WRITE (8,51) ZF,AMX,P55,PS55,P25,PS25	127
	ZF=0.99*ZF	128
	GO TO 18	129
20	IF (I30GO.EQ.1) GO TO 21	130
	AM55=AMXT	131
	GO TO 13	132
21	AM25=AMXT	133
	GO TO 23	134
22	IF (I30GO.EQ.1) GO TO 26	135
	PS55=P55/EXP((PHI55-PHIS55)/REX55)	136
	IF (I34SMX.LE.0) GO TO 41	137
	WQA=W324/A25	138
	C1=P25*SQRT(G/(T25*AJ))*CAPSF	139
	MCON=0	140
	QQ(2)=0.	141
	QQ(3)=0.	142
	AM25=0.25	143
	TS25=0.875*T25	144
23	DO 24 I=1,15	145
	CALL PROCOM (FAR24,TS25,CS25,AK25,CP25,REX25,PHIS25,HS25)	146
	V25=AM25*CS25	147
	HSCAL=+25-V25**2/(2.*G*AJ)	148
	DELHS=+HSCAL-HS25	149
	IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 25	150
24	TS25=TS25+DELHS/CP25	151
	GO TO 2	152
25	WQAT=C1*SQRT(AK25/REX25)*AM25/(1.+(AK25-1.)*AM25**2/2.)*((AK25+1.1)/(2.*(AK25-1.)))	153
	AMX=AM25	154
	IGOGO=1	155
	GO TO 16	156
26	PS25=P25/EXP((PHI25-PHIS25)/REX25)	157
27	WG6=W324+WG55	158
	ERR(5)=(PS25-PS55)/PS25	159
	WF6=WFD+WFB	160
	FAR6=WFD/(WG6-WF6)	161
	H6=(W324*H25+WG55*H55)/WG6	162
	CALL THERMD (1.,H6,T6,PHI6,AMX,1,FAR6,1)	163
	C1=PS55*A55*(1.+AK55*AM55**2)+PS25*A25*(1.+AK25*AM25**2)	164
		165

	TS6=0.833*T6	167
	DO 32 I=1,15	168
	CALL PROCOM (FAR6,TS6,CS6,AK6,CP6,REX6,PHIS6,HS6)	169
	C2=WG5*SQRT(AJ*REX6*T6/(AK6*G))	170
	C3=C2/(CAPSF*C1)	171
	C4=(AK5-1.)/2.-(C3*AK6)**2	172
	C5=1.-2.*AK6*C3**2	173
	C6=C5**2+4.*C4*C3**2	174
	IF (C5) 28,29,30	175
28	CALL ERROR	176
	RETURN	177
29	AM62G=-C5/(2.*C4)	178
	GO TO 31	179
30	AM62G=(SQRT(C6)-C5)/(2.*C4)	180
31	IF (AM62G.LE.0.) GO TO 28	181
	AM6G=SQRT(AM62G)	182
	V6=AM5G*CS6	183
	HSCAL=H6-V6**2/(2.*G*AJ)	184
	DELHS=HSCAL-HS6	185
	IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 33	186
32	TS6=TS5+DELHS/CP6	187
	GO TO 28	188
33	IF (IGASM.GT.0) GO TO 34	189
34	A6G=A25+A55	190
	C7=SQRT(1.+(AK6-1.)*AM62G/2.)	191
	PS6=C2/(CAPSF*A6G*AM6G*C7)	192
	P6=PS5*EXP((PHI6-PHIS6)/REX6)	193
	CALL THERMO (P6,H6,T6,S6,XX1,1,FAR6,0)	194
	S6AVE=(WG24*S25+WG55*S55)/WG6	195
	IF (S5.GE.S6AVE) GO TO 35	196
	S6=S6AVE	197
	P6=EXP(AMX*(PHI6-S6)/1.986375)	198
35	IF (IGASM.EQ.1) GO TO 43	199
	IF (IES.EQ.0) GO TO 38	200
C**	CALCULATES A6 AS A FUNCTION OF INPUT AM6	201
	TS6=TS/(1.0+(((AK6-1.0)/2.0)*AM6**2))	202
	DO 36 JJ=1,15	203
	AK6P=AK6	204
	CALL PROCOM (FAR6,TS6,CS6,AK6,CP6,REX6,PHIS6,HS6)	205
	V6=AM5*CS6	206
	DELA6=AK6P-AK6	207
	IF (ABS(DELA6).LE.0.0005*AK6) GO TO 37	208
36	TS6=TS/(1.0+(((AK6-1.0)/2.0)*AM6**2))	209
	GO TO 28	210
37	PS6=P5/((1.0+(((AK6-1.0)/2.0)*AM6**2))**((AK6/(AK6-1.0))))	211
	AM6ABD=AM6	212
	RHO=CAPSF*PS6/(AJ*REX6*TS6)	213
	A6=WG5/(RHO*V6)	214
	WRITE (6,52) A6	215
	GO TO 44	216
C	CALCULATES M6=F(A6DESIGN)	217
38	TS6P=TS/(1.0+(((AK6-1.0)/2.0)*AM6ABD**2))	218
	DO 39 I=1,15	219
	CALL PROCOM (FAR6,TS6P,CS6,AK6,CP6,REX6,PHIS6,HS6)	220
	PS6P=PS6*(TS6P/TS6)**((AK6/(AK6-1.0))	221
	RHO6=CAPSF*PS6P/(AJ*REX6*TS6P)	222
	V6=SQRT(2.*G*AJ*(H6-HS6))	223
	IF ((H6-HS6).LT.0.0) GO TO 42	224
	A6P=WG5/(RHO6*V6)	225
	DELA6=A6P-A6	226
	V6=WG5/(RHO6*A6)	227
	AM6=V5/CS6	228

	AM62=AM6**2	229
	IF (ABS(DELA6).LE.00.002*A6) GO TO 40	230
39	TS6P=T6/(1.0+(((AK6-1.0)/2.0)*AM62))	231
	GO TO 28	232
40	TS6=TS5P	233
	PS6=PS5P	234
	GO TO 44	235
41	T6=T55	236
	P6=P55	237
	H6=H55	238
	S6=S55	239
	WG6=WG55	240
	PS6=PS55	241
	V6=V55	242
	AM6=AM55	243
	IF (IGASM*.EQ.0) A6=A55	244
	GO TO 44	245
42	WRITE (6,53) H6,HS6	246
	GO TO 28	247
43	AM62=AM62G	248
	AM6=AM6G	249
	A6=A25+A55	250
44	CALL COAFBN	251
	RETURN	252
45	KKGO=1	253
	OPRDS=PRFDS*PRCDS	254
	PRFNEW=PRFDS*PS55/P25*1.02	255
	PRCNEW=OPRDS/PRFNEW	256
	CALL ENGBAL	257
	RETURN	258
C		259
C		260
46	FORMAT (22HOSQRT OF H55-HS55 NEG ,6E15.6,6H\$\$\$\$\$)	261
47	FORMAT (20HOTURBINE AREA DESIGN,6X6H A55=,E15.8,8H AM55=,E15.8)	262
48	FORMAT (22HOSQRT OF H25-HS25 NEG ,6E15.6,6H\$\$\$\$\$)	263
49	FORMAT (25HOTURBINE/DUCT AREA DESIGN,7H A55=,E15.8,8H AM55=,E1	264
	15.8,8H A25=,E15.8,8H AM25=,E15.8)	265
50	FORMAT (12HOCOMIX PCNF=,F7.4,4H AM=,F8.6,5H P55=,F9.5,6H PS55=,F9.	266
	15.5H P25=,F9.5,6H PS25=,F9.5,6H\$\$\$\$\$)	267
51	FORMAT (10HOCOMIX ZF=,F8.5,4H AM=,F8.6,5H P55=,F9.5,6H PS55=,F9.5,	268
	15H P25=,F9.5,6H PS25=,F9.5,6H\$\$\$\$\$)	269
52	FORMAT (3X,27HAFTERBURNER DESIGN AREA A6 F8.3)	270
53	FORMAT (3X,18HNEG.HS6 FACTOR H6 F9.4,3X,4HHS6 F9.4)	271
	END	272

\$IBFTC COAFBN DECK	
SUBROUTINE COAFBN	1
COMMON / ALL/	2
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
2IGASM*,IDBURN,IAFTBN,IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,	4
3ITRYS ,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ARR(6)	5
COMMON /DESIGN/	6
1PCNFGJ,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC,	7
2ZFDS ,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WAFCF ,	8
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,	9
4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCOCF,ETABCF,	10
5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,	11
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPDF,ETLPDF,DHLPDF,T21DS ,	12

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7T24DS ,WFDDSD,DTDDSD,ETADDS,WA23DS,DPDDSD,DTDDCF,ETADCF,      13
8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF,      14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,      15
$PS55 ,AM55 ,CVDNDZ,CVMNDZ,A8SAV ,A9SAV ,A28SAV,A29SAV      16
COMMON/FRONT/QXQ(80)/SIDE/QYQ(48)      17
COMMON / BACK/      18
1T55 ,P55 ,H55 ,S55 ,T25 ,P25 ,H25 ,S25 ,      19
2WFB ,WG55 ,FAR55 ,WFD ,WG24 ,FAR24 ,P1 ,DUMB ,      20
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,      21
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,      22
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,      23
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,      24
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,      25
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,      26
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC      27
COMMON/DUMMYS/DUMMY(100)      28
COMMON/SPDOL2/TWOSPL(44)      29
EQUIVALENCE (ERR,DUMMY(11))      30
DIMENSION ERR(9)      31
EQUIVALENCE (P6DSAV,DUMMY(7)),(AM6DSV,DUMMY(8)),(ETAASV,DUMMY(9)),      32
1(FARTSV,DUMMY(10))      33
DIMENSION Q(9)      34
DATA AWORD/6HCOAFBN/      35
WORD=AWORD      36
Q(2)=0.      37
Q(3)=0.      38
AJ=778.26      39
CAPSF=2116.2170      40
G=32.174049      41
C*** P6DS AND AM6DS ARE SET FOR GENERALIZATION OF AFTERBURNER      42
C*** EFFICIENCY MAP GENERALIZATION      43
IF (IDES.EQ.1) P6DS=P6*14.696      44
IF (IDES.EQ.1) AM6DS=AM6      45
WF6=WFB      46
IF (IGASM.GT.0) WF6=WF6+WFD      47
WA6=WG5-WF6      48
C *** DRY LOSS      49
WG6C=WG6*SQR(T6)/P6      50
IF (IDES.EQ.1) WG6CDS=WG6C      51
DPAFT=DPAFDS*(WG6C/WG6CDS)      52
IF (DPAFT.GT.1.) DPAFT=1.      53
P7=P6*(1.-DPAFT)      54
A7=A6      55
FAR6=WF6/WA6      56
CALL PROCOM (FAR6,T6,XX1,XX2,XX3,XX4,PHI6,XX6)      57
WQA=WG6/A7      58
C1=P7*SQR(G/(T6*AJ))*CAPSF      59
AM7=AM6      60
TS7=0.875*T6      61
1 DO 2 I=1,15      62
CALL PROCOM (FAR6,TS7,CS7,AK7,CP7,REX7,PHI7,HS7)      63
V7=AM7*CS7      64
HSCAL=H6-V7**2/(2.*G*AJ)      65
DELHS=HSCAL-HS7      66
IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 3      67
2 TS7=TS7+DELHS/CP7      68
GO TO 14      69
3 WQAT=C1*SQR(AK7/REX7)*AM7/((1.+(AK7-1.)*AM7**2/2.))*((AK7+1.)/(2.*      70
1(AK7-1.)))      71
DIR=WQA/WQAT      72
EW=(WQA-WQAT)/WQA      73
CALL AFQUIR (Q(1),AM7,EW,0.,40.,.001,DIR,AM7T,IGO)      74
GO TO (4,5,14),IGO      75

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4	AM7=AM7T	76
	IF (AM7.GE.1.0) AM7=0.9	77
	GO TO 1	78
5	PS7=P7/EXP((PHI6-PHIS7)/REX7)	79
	IF (IAFTBN.GT.0) GO TO 7	80
C ***	NON-AFTERBURNING	81
6	T7=T6	82
	WFA=0.0	83
	FAR7=FAR6	84
	WG7=WG6	85
	IF (IDES.EQ.1.AND.T7DS.NE.0.) GO TO 7	86
	GO TO 20	87
C ***	AFTERBURNING	88
7	IF (IAFTBN.EQ.2) T7=T6+2000.	89
	IF (IDES.EQ.1) T7=T7DS	90
	IF (T7.LE.T6) GO TO 6	91
	RHO65=CAPSF*PS7/(AJ*REX7*TS7)	92
	PS65=PS7	93
	V65=V7	94
	Q(2)=0.	95
	Q(3)=0.	96
8	IF (T7.GT.4000.) T7=4000.	97
	HV=((((-.4594317E-19*T7)-.2034116E-15)*T7+.2783643E-11)*T7+.2051	98
	1501E-07)*T7-.2453115E-03)*T7-.9433296E-01)*T7+.1845537E+05	99
	CALL THERMO (P7,HA,T7,XX1,XX2,1,FAR6,0)	100
C ***	TO ALTER DESIGN ABETAA MAP FROM GENERAL TO SPECIFIC MAP	101
	IF (IDES.NE.1) GO TO 9	102
	FAR7DS=(HA-H6)/(HV*ETAADS)	103
	CALL ETAAB (0.,0.,0.,0.,ETAADS,ETAASV,P6DS,P6DSAV,AM6DS,AM6DSV,IDE	104
	1S,FAR7DS,FAR7SV)	105
	T7=T6	106
	GO TO 20	107
9	P6GS=26*14.696	108
	FAR7DS=(HA-H6)/(HV*ETAADS)	109
	DO 10 II=1,15	110
	CALL ETAAB (FAR7GS,AM6,P6GS,ETAA,ETAADS,ETAASV,P6DS,P6DSAV,AM6DS,A	111
	1M6DSV,IDES,FAR7DS,FAR7SV)	112
	FAR7=(HA-H6)/(HV*ETAA)	113
	DELFA7=ABS(FAR7-FAR7GS)	114
	IF (DELFA7.LE.0.01*FAR7) GO TO 11	115
10	FAR7GS=FAR7	116
11	CONTINJE	117
	IF (FAR7.GT.0.) GO TO 12	118
	CALL ERROR	119
12	WFAX=FAR7*WG6	120
	IF (IAFTBN.EQ.1) GO TO 15	121
	ERRW=(WFA-WFAX)/WFA	122
	DIR=SQRT(WFA/WFAX)	123
	CALL AFQUIR (Q(1),T7,ERRW,0.,30.,.0005,DIR,T7T,IGD)	124
	GO TO (13,16,14),IGD	125
13	T7=T7T	126
	GO TO 8	127
14	CALL ERROR	128
15	WFA=WFAX	129
16	FAR7=(WF6+WFA)/WA6	130
	WG7=WG6+WFA	131
C ***	MOMENTUM LOSS	132
	CALL PROCOM (FAR7,T7,XX1,XX2,XX3,REX7,PHI7,H7)	133
	RHO7=CAPSF*P7/(AJ*REX7*T7)	134
	V7=WG7/(RHO7*A7)	135
	Q(2)=0.	136
	Q(3)=0.	137
	PS7=PS65-0.01	138

17	RHO7=#G7/(V7*A7)	139
	HS7=H7-V7**2/(2.*G*AJ)	140
	CALL THERMO (1.0,HS7,TS7,PHIS7,XX2,1,FAR7,1)	141
	IF (TS7.GE.301.) GO TO 18	142
	CALL THERMO (1.0,HS7,400.,PHIS7,XX2,1,FAR7,0)	143
	V7=SQRT(2.*G*AJ*(H7-HS7))	144
	GO TO 17	145
18	PS7=RHO7*AJ*REX7*TS7/CAPSF	146
	PS7A=PS65+(RHO65*V65**2-RHO7*V7**2)/(G*CAPSF)	147
	DIR=SQRT(ABS(PS7/PS7A))	148
	EP=(PS7-PS7A)/PS7	149
	CALL AFQUIR (Q(1),V7,EP,0.,50.,.001,DIR,V7T,IGO)	150
	V7=V7T	151
	IF (V7.LT.100.) V7=100.	152
	GO TO (17,19,14),IGO	153
19	P7=PS7*EXP((PHI7-PHIS7)/REX7)	154
	CALL PROCOM (FAR7,TS7,CS7,XX2,XX3,XX4,XX5,XX6)	155
	AM7=V7/CS7	156
20	CALL THERMO (P7,H7,T7,S7,XX2,1,FAR7,0)	157
	CALL COMNOZ	158
	RETURN	159
C		160
C		161
	END	162

```

$IBFTC FRTOSD DECK
SUBROJTINE FRTOSD
COMMON/ALL/XX(28)/DESIGN/YY(80)
COMMON / FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLDB ,PCBLHP ,PCBLLP
COMMON / SIDE/
1XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 ,
2XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 ,
3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 ,
4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 ,
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPOUC ,BYPASS ,DUMS3 ,
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
COMMON/BACK/ZZ(72)
COMMON/DUMMYS/DUMMY(100)
COMMON/SPOOL2/TWOSPL(44)
XP1=P1
XWAF=#AF
XWAC=#AC
XBLF=BLF
XBLDU=BLDU
XH3=H3
XT21=T21
XP21=P21

```

XH21=I21	32
XS21=S21	33
CALL CDDUCT	34
RETURN	35
END	36

\$IBFTC FASTBC DECK	
SUBROUTINE FASTBK	1
COMMON/ALL/XX(28)/DESIGN/YY(80)	2
COMMON / FRONT/	3
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,	4
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,	5
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,	6
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,	7
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,	8
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,	9
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,	10
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,	11
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,	12
\$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLDB ,PCBLHP ,PCBLLP	13
COMMON / SIDE/	14
1XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 ,	15
2XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 ,	16
3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 ,	17
4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 ,	18
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS ,DUMS3 ,	19
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29	20
COMMON / BACK/	21
1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,	22
2XWFB ,XWG55 ,XFAR55 ,XWFD ,XWG24 ,XFAR24 ,XXP1 ,DUMB ,	23
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,	24
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,	25
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,	26
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,	27
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,	28
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,	29
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC	30
COMMON/DUMMYS/DUMMY(100)	31
COMMON/SPOOL2/TWOSPL(44)	32
XT55=T55	33
XP55=P55	34
XH55=I55	35
XS55=S55	36
XT25=T25	37
XP25=P25	38
XH25=I25	39
XS25=S25	40
XWFB=IWB	41
XWG55=WG55	42
XFAR55=FAR55	43
XWFD=IFD	44
XWG24=WG24	45
XFAR24=FAR24	46
XXP1=P1	47
CALL CMIX	48
RETURN	49
END	50

```

$IBFTC COMNOZ DECK
SUBROJTINE COMNOZ
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASMX ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,
3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDEG ,TOLALL ,ARR(6)
COMMON /DESIGN/
1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
4T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,DPCODS ,DTCOCF ,ETABCF ,
5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS ,
7T24DS ,WFDDDS ,DTODDS ,ETADDS ,WA23DS ,DPDUDS ,DTDUCF ,ETADCF ,
8T7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
$PS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON /FRONT/QXQ(80)/SIDE/QYQ(48)
COMMON / BACK/
1T55 ,P55 ,H55 ,S55 ,T25 ,P25 ,H25 ,S25 ,
2WFB ,WG55 ,FAR55 ,WFD ,WG24 ,FAR24 ,P1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC
COMMON /DUMMYS/DUMMY(100)
COMMON /SPOOL2/TWOSPL(44)
EQUIVALENCE (ERR,DUMMY(11))
DIMENSION ERR(9)
DATA AWORD/6HMNZZL/
WORD=AWORD
A8SAV=A8
A9SAV=A9
NOZM=0
IMNOZ=0
IF (NOZFLT.EQ.1.OR.NOZFLT.EQ.3) NOZM=1
IF (IDES.EQ.1.OR.IAFTBN.GT.0.OR.NOZM.EQ.1) IMNOZ=1
IF (IMCD.EQ.1) GO TO 1
CALL CONVRG (T7,H7,P7,S7,FAR7,WG7,P1,IMNOZ,A8,P7R,T8,H8,P8,S8,TS8,
1PS8,V8,AM8,ICON)
GO TO (3,3,3,2),ICON
1 CALL CONDIV (T7,H7,P7,S7,FAR7,WG7,P1,IMNOZ,A8,A9,P7R,T8,H8,P8,S8,T
19,H9,P9,S9,TS8,TS9,PS8,PS9,V8,V9,AM8,AM9,ICON)
IMSHOC=ICON
GO TO (4,4,4,2),ICON
2 CALL ERROR
3 T9=T8
H9=H8
P9=P8
S9=S8
TS9=TS8
PS9=PS8
V9=V8
AM9=AM8
A9=A8
IMSHOC=ICON+3
4 ERR(6)=(P7R-P7)/P7R
IF (IMNOZ.EQ.1) WRITE (6,5) A8,AM8,A9,AM9
RETURN

```


C										62
C										63
5	FORMAT (14HONNOZZLE DESIGN,10X8H	A8=,E15.8,8H	AM8=,E15.8,8H							64
1	A9=,E15.8,8H	AM9=,E15.8)								65
	END									66

\$IBFTC ERROR	DECK									
	SUBROJTINE ERROR									1
	COMMON / ALL/									2
1	WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,									3
2	IGASMK ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,									4
3	ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ARR(6)									5
	COMMON / DESIGN/									6
1	PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC ,									7
2	ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,									8
3	ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,									9
4	T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,DPCODS ,DTCOCF ,ETABCF ,									10
5	TFLPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,									11
6	TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS ,									12
7	T24DS ,WFDD ,DTDUDS ,ETADD ,WA23DS ,DPDUDS ,DTDUCF ,ETADCF ,									13
8	T7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,									14
9	A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,									15
\$	PS55 ,AM55 ,CVDNOZ ,CVMNOZ ,ABSAV ,A9SAV ,A28SAV ,A29SAV									15
	COMMON / FRONT/									17
1	T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,									18
2	T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,									19
3	T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,									20
4	T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,									21
5	CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,									22
6	CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,									23
7	CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,									24
8	CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,									25
9	AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,									26
\$	TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLOB ,PCBLHP ,PCBLLP									27
	COMMON / SIDE/									28
1	XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 ,									29
2	XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 ,									30
3	T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 ,									31
4	T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 ,									32
5	WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS ,DUMS3 ,									33
6	TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29									34
	COMMON / BACK/									35
1	XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,									36
2	XWFB ,XWG55 ,XFAR55 ,XWFD ,XWG24 ,XFAR24 ,XP1 ,DUMB ,									37
3	T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,									38
4	T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,									39
5	WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,CPAFT ,V55 ,V25 ,									40
6	PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,									41
7	TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,									42
8	VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,									43
9	FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC									44
	COMMON/DUMMYS/DUMMY(100)									45
	COMMON/SPDOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI									46
1	ETAI ,WACI ,TFFIP ,CNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,PCBLIP ,PCNIGU ,ZIDS ,									47
2	PCNIDS ,PRIOS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,TFIPDS ,CNIPDS ,ETIPDS ,									48
3	TFIPCF ,CNIPCF ,ETAPCF ,DHIPCF ,WAICDS ,WAI ,PCBLI ,BLI ,T22DS ,WA21									49
	LOGICAL ERROR									50
	COMMON/ERER/ERRER									51

	DIMENSION TRASH1(80),TRASH2(80),TRASH3(48),TRASH4(72)	52
	DIMENSION TRASH5(44),TRASH6(48)	53
	EQUIVALENCE (TRASH5,T22),(TRASH6,DUMMY(22))	54
	EQUIVALENCE (TRASH1,PCNFGU),(TRASH2,T1),(TRASH3,XP1),(TRASH4,XT55)	55
	DATA AWORD/6HCOMMON/	56
	ERRER=.TRUE.	57
	WRITE (6,2) WORD	58
	WORD=AWORD	59
	WRITE (6,3) WORD,ZF,PCNF,ZI,PCNI,ZC,PCNC,T4,MODE	60
	WRITE (6,4)	61
	WRITE (6,5) (TRASH1(I),I=1,80)	62
	WRITE (6,6)	63
	WRITE (6,5) (TRASH2(I),I=1,80)	64
	WRITE (6,4)	65
	WRITE (6,5) (TRASH3(I),I=1,48)	66
	WRITE (6,4)	67
	WRITE (6,5) (TRASH4(I),I=1,72)	68
	WRITE (6,4)	69
	WRITE (6,5) (TRASH5(I),I=1,44)	70
	WRITE (6,4)	71
	WRITE (6,5) (TRASH6(I),I=1,48)	72
	WRITE (6,4)	73
	WRITE (6,7) LOOPER	74
	IF (IDUMP.EQ.0) GO TO 1	75
	WRITE (6,6)	75
	CALL SYG (2)	77
1	CALL ENGBAL	78
	RETURN	79
C		80
C		81
2	FORMAT (28H0AN ERROR HAS BEEN FOUND IN ,A6)	82
3	FORMAT (1H0,A6,9X,7E15.6,I4)	83
4	FORMAT (2H0)	84
5	FORMAT (1H ,8E15.6)	85
6	FORMAT (1H1)	86
7	FORMAT (25H0FAILED TO CONVERGE AFTER,I4,6H LOOPS)	87
	END	88

\$IBFTC	SYGS	DECK	
	SUBROUTINE SYG (ICON)		1
	DIMENSION WORD(132)		2
	DATA ONEDOL/6H\$ /		3
	GO TO (1,2),ICON		4
1	END FILE 8		5
	REWIND 8		6
	RETURN		7
C	TERMINATE THE FILE		8
2	WRITE (8,10)		9
	END FILE 8		10
	REWIND 8		11
C	READ RECORD		12
3	READ (8,11) (WORD(I),I=1,132)		13
C	CHECK FOR 12 LEADING DOLLAR SIGNS		14
	DO 4 I=1,12		15
	IF (WORD(I)-ONEDOL) 5,4,5		16
4	CONTINUE		17
	RETURN		18
C	CHECK FOR 6 TRAILING DOLLAR SIGNS		19

5	DO 8 I=1,132	20
	I=I	21
	IF (WORD(I)-ONEDOL) 8,6,8	22
6	K=I+5	23
	DO 7 J=I,K	24
	IF (WORD(J)-ONEDOL) 8,7,8	25
7	CONTINUE	26
	GO TO 9	27
8	CONTINUE	28
	WRITE (6,12)	29
	RETURN	30
C	PRINT LINE	31
9	I=I-1	32
	WRITE (6,11) (WORD(M),M=1,I)	33
	GO TO 3	34
C		35
C		36
10	FORMAT (12H\$\$\$\$\$\$\$\$\$\$\$)	37
11	FORMAT (132A1)	38
12	FORMAT (1H0,12HERROR IN SYG)	39
	END	40

\$IBFTC PERFJR DECK

SUBROUTINE PERF

COMMON / ALL/

1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASM ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,
3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ARR(6)

COMMON /DESIGN/

1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELF6 ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
4T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,DPCODS ,DTCOCF ,ETABCF ,
5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS ,
7T24DS ,WFDDDS ,DTDUDS ,ETADDS ,WA23DS ,DPDUDS ,DTDUCF ,ETADCF ,
8T7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
\$PS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV

COMMON / FRONT/

1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,
\$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLOB ,PCBLHP ,PCBLLP

COMMON / SIDE/

1XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 ,
2XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 ,
3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 ,
4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 ,
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS ,DUMS3 ,
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29

COMMON / BACK/

1XT55	,XP55	,XH55	,XS55	,XT25	,XP25	,XH25	,XS25		36
2XWFB	,XWG55	,XFAR55	,XWFD	,XWG24	,XFAR24	,XXP1	,DUMB		37
3T6	,P6	,H6	,S6	,T7	,P7	,H7	,S7		38
4T8	,P8	,H8	,S8	,T9	,P9	,H9	,S9		39
5WG6	,WFA	,WG7	,FAR7	,ETAA	,DPAFT	,V55	,V25		40
6PS6	,V6	,AM6	,TS7	,PS7	,V7	,AM7	,AM25		41
7TS8	,PS8	,V8	,AM8	,TS9	,PS9	,V9	,AM9		42
8VA	,FRD	,VJD	,FGMD	,VJM	,FGMM	,FGPD	,FGPM		43
9FGM	,FGP	,WFT	,WGT	,FART	,FG	,FN	,SFC		44
COMMON/DUMMYS/DUMMY(21),WA32,DPWGDS,DPWING,WA32DS,A38,AM38,V38,									45
1T38,H38,P38,TS38,PS38,T39,H39,P39,TS39,V39,AM39,A39,8PRINT,WG37,									46
2CVDWNG,FGMWNG,FGPWNG,FNWIN,FMNAIN,FNOVFN,PS39,DIMMY(51)									47
COMMON/SPOOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI									48
1,ETAI,ACI,TFIP,CNIP,ETATIP,DHTCIP,DHTI,BLIP,PCBLIP,PCNIGU,ZIDS,									49
2PCNIDS,PRIDS,ETAIDS,WAIDS,PRICF,ETAICF,WAICF,TFIPDS,CNIPDS,ETIPDS,									50
3TFIPCF,CNIPCF,ETAPCF,DHPCF,WAICDS,WAI,PCBLI,BLI,T22DS,WA21									51
DIMENSION ERR(9)									52
EQUIVALENCE (AFTFAN,DUMMY(58)),(DUMSPL,DUMMY(59))									53
LOGICAL DUMSPL									54
LOGICAL AFTFAN									55
EQUIVALENCE (ERR,DUMMY(11)),(FDES,DUMMY(60)),(PCBLID,DUMMY(61))									56
1,(VJWING,DUMMY(77))									57
EQUIVALENCE (FFOVFN,DUMMY(54)),(FCOVFN,DUMMY(55)),									58
1(FMNOFN,DUMMY(56)),(FNOVFD,DUMMY(57))									59
DATA AWORD/6H PERF/									60
WORD=AWORD									61
G=32.174049.									62
CAPSF=2116.2170									63
WFT=WFB+WFD+WFA									64
WAT=WAF-BLOB									65
IF (AFTFAN) WAT=WAT+WAI									66
WGT=WAT+WFT									67
FART=WFT/WAT									68
VA=AM*CS									69
FRD=VA*WAF/G									70
VJM=CVMNOZ*V9									71
FGMM=VJM*WG7/G									72
FGPM=CAPSF*(PS9-P1)*A9									73
IF (IGASMGT.0) GO TO 1									74
VJD=CVDNOZ*V29									75
FGMD=VJD*WG24/G									76
FGPD=CAPSF*(PS29-P1)*A29									77
1	VJWING=0.								78
FGMWNG=0.									79
FGPWNG=0.									80
FGWING=0.									81
FNWING=0.									82
IF (PCBLID.EQ.0.) GO TO 2									83
VJWING=CVDWNG*V39									84
FGMWNG=VJWING*WG37/G									85
FGPWNG=CAPSF*(PS39-P1)*A39									86
FGWING=FGMWNG+FGPWNG									87
FNWING=FGWING-VA*WA32/G									88
2	FGM=FGMM+FGMD+FGMWNG								89
FGP=FGPM+FGPD+FGPWNG									90
FNMAIN=(FGMM+FGMD+FGPM+FGPD)-VA*(WAF-WA32)/G									91
FG=FGM+FGP									92
FN=FG-FRD									93
SFC=3500.*WFT/FN									94
FG=DELFG*FG									95
FN=DELFN*FN									96
SFC=DELSFC*SFC									97
FFAN=FGMD+FGPD-VA*WAD/G									98

FCORE=FNMAIN-FFAN	99
FFOVFN=FFAN/FN	100
FCOVFN=FCORE/FN	101
FWOVFN=FNWING/FN	102
FMNOFN=FNMAIN/FN	103
IF (IDES.EQ.1) FDES=FN	104
FNOVFD=FN/FDES	105
IF (.NOT.DUMSPL) GO TO 3	106
PCNI=1.0	107
CNI=0.	108
3 CALL JJTPUT	109
CALL ERROR	110
RETURN	111
END	112

\$IBFTC PUTOUT DECK

SUBROUTINE OUTPUT	1
COMMON / ALL/	2
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
2IGASM ,IDBURN,IAFTBV, IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,	4
3ITRYS ,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ARR(6)	5
COMMON /DESIGN/	5
1PCNFGJ,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC,	7
2ZFDS ,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WAFCF ,	8
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,	9
4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCCCF,ETABCF,	10
5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,	11
6TFLPDS,CNLPDS,ETLPDS,TELPCF,CNLPCF,ETLPCF,DHLPCF,T21DS ,	12
7T24DS ,WFDDDS ,DTDUDS,ETAADS,WA23DS,DPDUDS,DTDUCF,ETADCF,	13
8T7DS ,WFAADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF,	14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,	15
\$PS55 ,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV	16
COMMON / FRONT/	17
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,	18
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,	19
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,	20
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,	21
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,	22
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,	23
7CNHP ,ETATHP,DHTCHP,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,	24
8CNLP ,ETATLP,DHTCLP,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,	25
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,	26
\$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU,PCBLDB,PCBLHP,PCBLLP	27
COMMON / SIDE/	28
1XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 ,	29
2XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 ,	30
3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 ,	31
4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 ,	32
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS,DUMS3 ,	33
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29	34
COMMON / BACK/	35
1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,	36
2XWFB ,XWG55 ,XFAR55,XWFD ,XWG24 ,XFAR24,XP1 ,DUMB ,	37
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,	38
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,	39
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,	40
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,	41
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,	42

	8VA	,FRD	,VJD	,FGMD	,VJM	,FGMM	,FGPD	,FGPM		43
	9FGM	,FGP	,WFT	,WGT	,FART	,FG	,FN	,SFC		44
	COMMON/DUMMYS/DUMMY(21),WA32,DPWGDS,DPWING,WA32DS,A38,AM38,V38,									45
	1T38,H38,P38,TS38,PS38,T39,H39,P39,TS39,V39,AM39,A39,BPRINT,WG37,									46
	2CVDWNG,FGMWNG,FGPWNG,FNWING,FNMAIN,FWDVFN,DIMMY(52)									47
	COMMON/SPOOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI									48
	1,ETAI,ACI,TFFIP,CNIP,ETATIP,DHTCIP,DHTI,BLIP,PCBLIP,PCNIGU,ZIDS,									49
	2PCNIDS,PRIDS,ETAIDS,WAIDS,PRICF,ETAICF,WAICF,TFIPDS,CNIPDS,ETIPDS,									50
	3TFIPCF,CNIPCF,ETAPCF,DHPCF,WAICDS,WAI,PCBLI,BLI,T22DS,WA21									51
	EQUIVALENCE (T4PBL,DUMMY(2)),(T41,DJMMY(3))									52
	DIMENSION W(5,4),ANS1(80),ANS2(80),ANS3(48),ANS4(72)									53
	EQUIVALENCE (ANS1,PCNFGU),(ANS2,T1),(ANS3,XP1),(ANS4,XT55)									54
	EQUIVALENCE (ANS5,T22)									55
	DIMENSION ANS5(44)									56
	DIMENSION ANS6(48)									57
	EQUIVALENCE (ANS6,WA32)									58
	EQUIVALENCE (FXFN2M,DUMMY(50)),(FXM2CP,DUMMY(51)),									59
1	(AFTFAN,DUMMY(58)),(DUMSPL,DUMMY(59))									60
	EQUIVALENCE (PCBLID,DUMMY(61))									61
	LOGICAL FXFN2M,FXM2CP,AFTFAN,DUMSPL									62
	DATA AWORD1,AWORD2/6HOUTPUT,6HCOMMON/									63
	DATA (W(1,I),I=1,4)/6HSUBSON,6HIC C-D,6H NOZZL,6HE /									64
	DATA (W(2,I),I=1,4)/6HSHOCK ,6HINSIDE,6H C-D N,6HOZZLE /									65
	DATA (W(3,I),I=1,4)/6HSHOCK ,6HOUTSID,6HE C-D ,6HNOZZLE/									66
	DATA (W(4,I),I=1,4)/6HSUBSON,6HIC CON,6HVERG. ,6HNOZZLE/									67
	DATA (W(5,I),I=1,4)/6HSONIC ,6HCONVER,6HAGENT N,6HOZZLE /									68
	WORD=AWORD1									69
	IF (IDBURN.GT.0) GO TO 2									70
	IF (IAFTBN.GT.0) GO TO 1									71
	WRITE (6,7) WORD,AM,ALTP,T4,ETAR									72
	GO TO 3									73
1	WRITE (6,8) WORD,AM,ALTP,T4,T7,ETAR									74
	GO TO 3									75
2	WRITE (6,9) WORD,AM,ALTP,T4,T24,ETAR									76
3	IF (FXFN2M) WRITE (6,17)									77
	IF (FXM2CP) WRITE (6,18)									78
	IF (.NOT.FXFN2M.AND.(.NOT.FXM2CP).AND.(.NOT.DUMSPL)) WRITE (6,19)									79
	IF (DUMSPL) WRITE (6,23)									80
	IF (PCBLID.EQ.0.) WRITE (6,20)									81
	IF (PCBLID.EQ.0..AND.AFTFAN) WRITE (6,21)									82
	IF (PCBLID.NE.0..AND.AFTFAN) WRITE (6,22)									83
	CALL CONOUT (2)									84
	WRITE (6,10) (W(IMSHOC,I),I=1,4),FG,FN,SFC									85
	IF (IGASM.X.GT.0) GO TO 4									86
	WRITE (6,11) (W(IDSHOC,I),I=1,4)									87
4	WRITE (6,12) LOOPER									88
	IF (IDES.NE.1) GO TO 5									89
	WORD=AWORD2									90
	WRITE (6,13) WORD,ZF,PCNF,ZI,PCNI,ZC,PCNC,T4,MODE									91
	WRITE (6,14)									92
	WRITE (6,15) (ANS1(I),I=1,80)									93
	WRITE (6,14)									94
	WRITE (6,15) (ANS2(I),I=1,80)									95
	WRITE (6,14)									96
	WRITE (6,15) (ANS3(I),I=1,48)									97
	WRITE (6,14)									98
	WRITE (6,15) (ANS4(I),I=1,72)									99
	WRITE (6,14)									100
	WRITE (6,15) (ANS5(I),I=1,44)									101
	WRITE (6,14)									102
	WRITE (6,15) (ANS6(I),I=1,48)									103

	WRITE (6,16)	104
	IF (IDES.EQ.1) GO TO 6	105
5	CONTINUE	106
	A8=A8SAV	107
	A9=A9SAV	108
	A28=A28SAV	109
	A29=A29SAV	110
	IF (IDUMP.NE.2) GO TO 6	111
	WRITE (6,16)	112
	CALL SYG (2)	113
6	CALL ENGBAL	114
	RETURN	115
C		116
C		117
C		118
7	FORMAT (1HB,A6,14X7H AM=,F7.3,6X7H ALTP=,F7.0,6X7H T4=,F8.2	119
	1,25X7H ETAR=,F7.4)	120
8	FORMAT (1HB,A6,14X7H AM=,F7.3,6X7H ALTP=,F7.0,6X7H T4=,F8.2	121
	1,5X7H T7=,F8.2,5X7H ETAR=,F7.4)	122
9	FORMAT (1HB,A6,14X7H AM=,F7.3,6X7H ALTP=,F7.0,6X7H T4=,F8.2	123
	1,5X7H T24=,F8.2,5X7H ETAR=,F7.4)	124
10	FORMAT (6HMAIN ,4A6,9X3HFG=,F9.2,18X3HFN=,F9.2,18X4HSFC=,F8.5)	125
11	FORMAT (6H DUCT ,4A6)	126
12	FORMAT (16HCONVERGED AFTER,14,6H LOOPS,/,1H1)	127
13	FORMAT (1H ,A6,9X,7E15.6,14)	128
14	FORMAT (1H)	129
15	FORMAT (1H ,8E15.6)	130
16	FORMAT (1H1)	131
17	FORMAT (65HOFAN AND MIDDLE SPOOL ARE ATTACHED , USE INNER AND DUTE	132
	1R TURBINES)	133
18	FORMAT (74HOMIDDLE AND COMPRESSOR SPOOLS ARE ATTACHED , USE MIDDLE	134
	1 AND JJTER TURBINES)	135
19	FORMAT (19HOTHREE SPOOL ENGINE)	136
20	FORMAT (21HONO AIRFLOW INTO WING)	137
21	FORMAT (1H+22X,14H, AFT-TURBOFAN)	138
22	FORMAT (14H0 AFT-TURBOFAN)	139
23	FORMAT (22HOMIDDLE SPOOL IS DUMMY)	140
	END	141

\$IBFTC CONOJT DECK	
SUBROUTINE CONOUT (ICON)	1
COMMON / ALL/	2
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
2IGASMK ,IDBURN,IAFTBN,IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,	4
3ITRYS ,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ARR(6)	5
COMMON /DESIGN/	6
1PCNFGJ,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,JELF3 ,DELFN ,DELSFC,	7
2ZFDS ,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WACF ,	8
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,	9
4T4DS ,WFBDS ,DTCDS,ETABDS,WA3CDS,DPCJDS,DTCCF,ETABCF,	10
5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,	11
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPDF,ETLPDF,DHLPDF,T21DS ,	12
7T24DS ,WFDS ,DTUDS,ETADS,WA23DS,DPUUDS,DTUDCF,ETADCF,	13
8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF,	14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,	15
\$PS55 ,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV	16
COMMON / FRONT/	17
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,	18

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2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 , 19
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 , 20
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB , 21
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 , 22
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF , 23
7CNHPM ,ETATHM,DHTCHM,DHTCM ,BLHP ,WG5 ,FAR5 ,CS , 24
8CNLPM ,ETATLM,DHTCLM,DHTFM ,BLLP ,WG55 ,FAR55 ,HPEXT , 25
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB , 26
$TFFHPM,TFFLPM,PCBLF ,PCBLC ,PCBLDU,PCBLOB,PCBLHP,PCBLLP 27
COMMON / SIDE/ 28
1XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 , 29
2XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 , 30
3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 , 31
4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 , 32
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS,DUMS3 , 33
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29 34
COMMON / BACK/ 35
1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 , 36
2XWFB ,XWG55 ,XFAR55,XWFD ,XWG24 ,XFAR24,XXP1 ,DUMB , 37
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 , 38
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 , 39
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 , 40
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 , 41
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 , 42
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM , 43
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC 44
COMMON/DUMMYS/DUMMY(21),WA32,DPWGD,DPWING,WA32DS,A38,AM38,V38, 45
1T38,H33,P38,TS38,PS38,T39,H39,P39,TS39,V39,AM39,A39,BPRINT,WG37, 46
2CVDWN3,FGMWNG,FGPWNG,FNWIN,FMMAIN,FWOVFN,DIMMY(52) 47
COMMON/SPOOL2/T22,P22,H22,S22,T50,P50,H50,S50,WA22,ZI,PCNI,CNI,PRI 48
1,ETAI,ACI,TFFIPM,CNIPM,ETATIM,DHTCIM,DHTIM,BLIP,PCBLIP,PCNIGU, 49
2ZIDS, 50
3PCNIDS,PRIDS,ETAIDS,WAIDS,PRICF,ETAICF,WAICF,TFIPDS,CNIPDS,ETIPDS, 51
4TFIPCF,CNIPCF,ETAPCF,DHPCF,WAICDS,WAI,PCBLI,BLI ,T22DS,WA21 52
DIMENSION PARAM(424),WORDY(424),IOUT(150),ADUT(6),WOUT(6) 53
EQUIVALENCE (FFOVFN,DUMMY(54)) , (FCOVFN,DUMMY(55)), 54
1 (FMNOFN,DUMMY(56)) , (FNOVFN,DUMMY(57)) 55
EQUIVALENCE (PARAM,PCNFGU) 56
DATA (WORDY(I),I=1,98)/ 57
16HPCNFGU,6HPCNCGU,6HT4GU ,6HDUMD1 ,6HDUMD2 ,6HDELF ,6HDELFN , 58
26HDELSFC,6HZFDS ,6HPCNFDS,6HPRFDS ,6HETAFDS,6HWAFDS ,6HPRFCF , 59
36HETACF,6HWACF ,6HZCDS ,6HPCNCDS,6HPRCDS ,6HETACDS,6HWACDS , 60
46HPRCCF ,6HETACCF,6HWACCF ,6HT4DS ,6HWFBD,6HDTCD,6HETABDS, 61
56HWA3DS,6HDPD,6HDTCCF,6HETABCF,6HTFHPDS,6HCNHPDS,6HETHPDS, 62
66HTFHPCF,6HCNHPCF,6HETHPCF,6HDHHPCF,6HT2DS ,6HTFLPDS,6HCNLPDS, 63
76HETLPDS,6HTFLPCF,6HCNLP,6HETLPCF,6HDHLP,6HT21DS ,6HT24DS , 64
86HWFDD,6HDTDD,6HETADD,6HWA23DS,6HDPD,6HDTDCF,6HETDCF, 65
96HT7DS ,6HWFADS ,6HDTAFDS,6HETAADS,6HWG6CDS,6HDPAFDS,6HDTAFCF, 66
$6HETAACF,6HA55 ,6HA25 ,6HA6 ,6HA7 ,6HA8 ,6HA9 , 67
$6HA28 ,6HA29 ,6HPS55 ,6HAM55 ,6HCVNOZ,6HCVMNOZ,6HABSAV , 68
$6HA9SAV ,6HA28SAV,6HA29SAV,6HT1 ,6HP1 ,6HH1 ,6HS1 , 69
$6HT2 ,6HP2 ,6HH2 ,6HS2 ,6HT21 ,6HP21 ,6HH21 , 70
$6HS21 ,6HT3 ,6HP3 ,6HH3 ,6HS3 ,6HT4 ,6HP4 , 71
DATA (WORDY(I),I=99,189)/ 72
16HH4 ,6HS4 ,6HT5 ,6HP5 ,6HH5 ,6HS5 ,6HT55 , 73
26HP55 ,6HH55 ,6HS55 ,6HBLF ,6HBLC ,6HBLDU ,6HBLDB , 74
36HCNF ,6HPRF ,6HETAF ,6HWAF ,6HWAF ,6HWA3 ,6HWG4 , 75
46HFAR ,6HCNC ,6HPRC ,6HETAC ,6HWACC ,6HWAC ,6HETAB , 76
56HDP ,6HDUMP ,6HCNHPM ,6HETATHM,6HDHTCHM,6HDHTCM ,6HBLHP , 77
66HWG5 ,6HFAR5 ,6HCS ,6HCNLP ,6HETALM ,6HDTCLM,6HDHTFM , 78
76HBLLP ,6HWG55 ,6HFAR55 ,6HHPEXT ,6HAM ,6HALTP ,6HETAR , 79
86HZF ,6HPCNF ,6HZC ,6HPCNC ,6HWF ,6HTFFHPM,6HTFFLPM, 80
96HPCBLF ,6HPCBLC ,6HPCBLDU,6HPCBLOB,6HPCBLHP,6HPCBLLP,6HXP1 , 81

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\$6HXWAF	,6HXWAC	,6HXBLF	,6HXBLDU	,6HXH3	,6HDUMS1	,6HDUMS2	,	82
\$6HXT21	,6HXP21	,6HXH21	,6HXS21	,6HT23	,6HP23	,6HH23	,	83
\$6HS23	,6HT24	,6HP24	,6HH24	,6HS24	,6HT25	,6HP25	,	84
\$6HH25	,6HS25	,6HT28	,6HP28	,6HH28	,6HS28	,6HT29	/	85
DATA (WORDY(I),I=190,280)/								86
16HP29	,6HH29	,6HS29	,6HWAD	,6HWFD	,6HWG24	,6HFAR24	,	87
26HETAJ	,6HDPDUC	,6HBYPASS	,6HDUMS3	,6HTS28	,6HPS28	,6HV28	,	88
36HAM28	,6HTS29	,6HPS29	,6HV29	,6HAM29	,6HXT55	,6HXP55	,	89
46HXH55	,6HXS55	,6HXT25	,6HXP25	,6HXH25	,6HXS25	,6HXWFB	,	90
56HXWG55	,6HXFAR55	,6HXWFD	,6HXWG24	,6HXFAR24	,6HXXP1	,6HDUMB	,	91
66HT6	,6HP6	,6HH6	,6HS6	,6HT7	,6HP7	,6HH7	,	92
76HS7	,6HT8	,6HP8	,6HH8	,6HS8	,6HT9	,6HP9	,	93
86HH9	,6HS9	,6HWG6	,6HWFA	,6HWG7	,6HFAR7	,6HETAA	,	94
96HDP AFT	,6HV55	,6HV25	,6HPS6	,6HV6	,6HAM6	,6HTS7	,	95
\$6HPS7	,6HV7	,6HAM7	,6HAM25	,6HTS8	,6HPS8	,6HV8	,	96
\$6HAM8	,6HTS9	,6HPS9	,6HV9	,6HAM9	,6HVA	,6HFRJ	,	97
\$6HVJD	,6HFGMD	,6HVJM	,6HFGMM	,6HFGPD	,6HFGPM	,6HFGM	,	98
\$6HFGP	,6HWFT	,6HWGT	,6HFART	,6HFG	,6HFN	,6HSFC	/	99
DATA (WORDY(I),I=281,424)/								100
121*0.	,6HWA32	,6HDPWGDS	,6HDPWING	,6HWA32DS	,6HA38	,6HAM38	,	101
26HV38	,6HT38	,6HH38	,6HP38	,6HTS38	,6HPS38	,6HT39	,	102
36HH39	,6HP39	,6HTS39	,6HV39	,6HAM39	,6HA39	,6HBPRT	,	103
46HWG37	,6HCVDWNG	,6HFGMWNG	,6HFGPWNG	,6HFNWING	,6HFNMAIN	,6HFWQVFN	,	104
56HPS39	,4*0.	,6HFFOVFN	,6HFCJVFN	,6HFVNOFN	,6HFNOVFD	,4*0.	,	105
66HTFFIP	,6HTFFIP	,6HTFFLP	,6HCNHP	,6HCNIP	,6HCNLP	,	,	106
76HDHTCP	,6HDHTC	,6HDHTCIP	,6HDHTI	,6HDHTCLP	,6HDHTF	,	,	107
86HETATIP	,6HETATIP	,6HETATLP	,6HVJW	,23*0.	,	,	,	108
96HT22	,6HP22	,6HH22	,6HS22	,6HT50	,6HP50	,6HH50	,	109
\$6HS50	,6HWA22	,6HZI	,6HPCNI	,6HCNI	,6HPRI	,6HETAI	,	110
\$6HWACI	,6HTFFIPM	,6HCNIPM	,6HETATIM	,6HDHTCIM	,6HDHTIM	,6HBLIP	,	111
\$6HPCBLIP	,6HPCNIGU	,6HZIDS	,6HPCNIDS	,6HPRIDS	,6HETAIDS	,6HWAIDS	,	112
\$6HPRICF	,6HETAICF	,6HWAICF	,6HTFIPDS	,6HCNIPDS	,6HETIPDS	,6HTFIPCF	,	113
\$6HCNIPCF	,6HETAPCF	,6HDHPCF	,6HWAICDS	,6HWAI	,6HPCBLI	,6HBLI	,	114
\$6HT22DS	,6HWA21	/						115
DATA THEEND,BLANK,LIMIT/6HTHEEND,6H								116
GO TO (1,6),ICON								117
C ***	INPUT SECTION							118
1	DO 4 N=1,150							119
	NUM=N							120
	READ (5,11) AIN,CHANGE							121
	IF (AIN.EQ.THEEND) GO TO 5							122
	DO 2 J=1,LIMIT							123
	JJ=J							124
	IF (AIN.EQ.WORDY(J)) GO TO 3							125
2	CONTINUE							126
	WRITE (6,12) AIN							127
	GO TO 4							128
3	IOUT(NJM)=JJ							129
	IF (CHANGE.NE.BLANK) WORDY(JJ)=CHANGE							130
4	CONTINUE							131
	WRITE (6,13)							132
5	NUM=NJM-1							133
	RETURN							134
C ***	OUTPUT SECTION							135
6	IF (NJM.EQ.1) GO TO 10							136
	N=NUM							137
	J=6							138
	DO 9 I=1,NUM,6							139
	IF (N.GT.6) GO TO 7							140
	J=N							141
7	N=N-6							142
	DO 8 K=1,J							143
	L=I+K-1							144

	M=IDUT(L)	145
	WOUT(<)=WORDY(M)	146
8	ADUT(K)=PARAM(M)	147
	WRITE (6,14) (WOUT(<),K=1,J)	148
	WRITE (6,15) (ADUT(K),K=1,J)	149
	IF (N.LE.0) GO TO 10	150
9	CONTINJE	151
10	RETURN	152
C		153
C		154
C		155
11	FORMAT (A6,6X,A6)	156
12	FORMAT (10H0THE WORD ,A6,26H NOT FOUND IN COMMON ARRAY)	157
13	FORMAT (22H0ERROR IN CONOUT INPUT)	158
14	FORMAT (26X,A6,5(9XA6))	159
15	FORMAT (1H ,20X6E15.6)	160
	END	161

\$IBFTC THCOMP DECK		
	SUBROJTINE THCOMP (PR,ETA,T,H,S,P,TO,HO,SO,PO)	1
	PD=P*PR	2
	TP=T*PR**0.28572	3
	DO 1 I=1,25	4
	CALL THERMO (PD,HP,TP,SP,X1,0,X2,0)	5
	DELS=SP-S	6
	IF (ABS(DELS).LE.0.00005*S) GO TO 2	7
1	TP=TP/EXP(4.*DELS)	8
	CALL ERROR	9
2	HO=H+((HP-H)/ETA)	10
	CALL THERMO (PD,HO,TO,SO,X1,0,X2,1)	11
	RETURN	12
	END	13

\$IBFTC PROCDM DECK		
	SUBROJTINE PROCDM (FARX,TEX,CSEX,AKEX,CPEX,REX,PHI,HEX)	1
	IF (FARX.LE.0.067623) GO TO 1	2
	FARX=0.067623	3
1	IF (TEX.GE.300.) GO TO 2	4
	TEX=300.	5
2	IF (TEX.LE.4000.) GO TO 3	6
	TEX=4000.	7
3	IF (FARX.GE.0.0) GO TO 4	8
	FARX=0.0	9
C	AIR PATH	10
4	CPA=(((((1.0115540E-25*TEX-1.4526770E-21)*TEX+7.6215767E-18)*TEX-11.5123259E-14)*TEX-6.7178376E-12)*TEX+6.5519486E-08)*TEX-5.15358792E-05)*TEX+2.5020051E-01	11
	HEA=(((((1.2644425E-26*TEX-2.0752522E-22)*TEX+1.2702630E-18)*TEX-1-3.0255518E-15)*TEX-1.6794594E-12)*TEX+2.1839826E-08)*TEX-2.57684420E-05)*TEX+2.5020051E-01)*TEX-1.7558886E+00	12
	SEA=+2.5020051E-01*ALOG(TEX)+((((1.4450767E-26*TEX-2.4211288E-22)*TEX+1.5243153E-18)*TEX-3.7820648E-15)*TEX-2.2392790E-12)*TEX+3.2759743E-08)*TEX-5.1576879E-05)*TEX+4.5432300E-02	13
		14
		15
		16
		17
		18
		19

	IF (FARX.LE.0.0) GO TO 5	20
C	FUEL/AIR PATH	21
	CPF=(((((7.2678710E-25*TEX-1.3335668E-20)*TEX+1.0212913E-16)*TEX-	22
	14.2051104E-13)*TEX+9.9686793E-10)*TEX-1.3771901E-06)*TEX+1.2258530	23
	2E-03)*TEX+7.3816638E-02	24
	HEF=(((((9.0848388E-26*TEX-1.9050949E-21)*TEX+1.7021525E-17)*TEX	25
	1-8.4102208E-14)*TEX+2.4921698E-10)*TEX-4.5906332E-07)*TEX+6.129315	26
	20E-04)*TEX+7.3816638E-02)*TEX+3.0581530E+01	27
	SEF=+7.3816638E-02*ALOG(TEX)+((((1.0382670E-25*TEX-2.2226118E-21	28
	1)*TEX+2.0425826E-17)*TEX-1.0512776E-13)*TEX+3.3228928E-10)*TEX-6.8	29
	2859505E-07)*TEX+1.2258630E-03)*TEX+6.483398E-01	30
5	CPEX=(CPA+FARX*CPF)/(1.+FARX)	31
	HEX=(1EA+FARX*HEF)/(1.+FARX)	32
	PHI=(SEA+FARX*SEF)/(1.+FARX)	33
	AMW=28.97-.946186*FARX	34
	REX=1.986375/AMW	35
	AKEX=CPEX/(CPEX-REX)	36
	CSEX=SQRT(AKEX*REX*TEX*25031.37)	37
	RETURN	38
	END	39

\$IBFTC	SURCH	DECK	
	SUBROJTIME SEARCH (P,A,B,C,D,AX,NA,BX,CX,DX,NO,NAM,NOM,NCODE)		1
	DIMENSION AX(NAM),BX(NAM,NOM),CX(NAM,NOM),DX(NAM,NOM),NO(NAM),Q(9)		2
C ***	NEEDS SUBROUTINE AFQUIR		3
C ***	AX AND BX MUST BE STORED LO TO HI		4
C ***	P=INPJT PROPORTION BETWEEN 0.0 AND 1.0		5
C	IF NOT INPUT, P MUST EQUAL -1.		6
C ***	NCODE=00 OK		7
C	NCODE=01 A LO		8
C	NCODE=02 A HI		9
C	NCODE=07 ERROR		10
C	NCODE=10 B LO		11
C	NCODE=20 B HI		12
	NCODE=0		13
	C=0.		14
	D=0.		15
C ***	FIND A		16
	DO 1 I=1,NA		17
	IH=I		18
	IF (A.LT.AX(I)) GO TO 2		19
1	CONTINUE		20
	IF (A.GT.AX(IH)) NCODE=2		21
	A=AX(IH)		22
	GO TO 3		23
2	IF (IH.GT.1) GO TO 3		24
	NCODE=1		25
	IH=2		26
	A=AX(1)		27
3	IL=IH-1		28
	LIMH=NO(IH)		29
	LIML=NO(IL)		30
C ***	FIND B		31
	PRM=(A-AX(IL))/(AX(IH)-AX(IL))		32
	PP=P		33
	IF (P.GE.0.) GO TO 6		34
	BL=BX(IL,1)+PRM*(BX(IH,1)-BX(IL,1))		35
	BH=BX(IL,LIML)+PRM*(BX(IH,LIMH)-BX(IL,LIML))		35

	IF (B.GE.BL) GO TO 4	37
	NCODE=NCODE+10	38
	B=BL	39
	GO TO 5	40
4	IF (B.LE.BH) GO TO 5	41
	NCODE=NCODE+20	42
	B=BH	43
5	PP=0.5	44
	Q(2)=0.	45
	Q(3)=0.	46
6	BH=PP*(BX(IH,LIMH)-BX(IH,1))+BX(IH,1)	47
	BL=PP*(BX(IL,LIML)-BX(IL,1))+BX(IL,1)	48
	DO 7 J=2,LIMH	49
	JH=J	50
	IF (B+.LT.BX(IH,J)) GO TO 8	51
7	CONTINUE	52
8	JL=JH-1	53
	DO 9 K=2,LIML	54
	KH=K	55
	IF (B+.LT.BX(IL,K)) GO TO 10	56
9	CONTINUE	57
10	KL=KH-1	58
	PR=(BX(IH,JL)-BH)/(BX(IH,JH)-BX(IH,JL))	59
	CH=CX(IH,JL)-PR*(CX(IH,JH)-CX(IH,JL))	60
	DH=DX(IH,JL)-PR*(DX(IH,JH)-DX(IH,JL))	61
	PR=(BX(IL,KL)-BL)/(BX(IL,KH)-BX(IL,KL))	62
	CL=CX(IL,KL)-PR*(CX(IL,KH)-CX(IL,KL))	63
	DL=DX(IL,KL)-PR*(DX(IL,KH)-DX(IL,KL))	64
	BT=BL+PR*(BH-BL)	65
	CT=CL+PR*(CH-CL)	66
	DT=DL+PR*(DH-DL)	67
	IF (P.GE.0.) GO TO 13	68
	DIR=SQR(B/BT)	69
	ERR=(3-BT)/B	70
	CALL AFQUIR (Q(1),PP,ERR,0.,25.,0.001,DIR,PT,ICON)	71
	GO TO (11,13,12),ICON	72
11	PP=PT	73
	IF (P>.LT.0.) PP=0.	74
	IF (P>.GT.1.) PP=1.	75
	GO TO 5	76
12	NCODE=7	77
13	B=BT	78
	C=CT	79
	D=DT	80
	RETURN	81
	END	82

\$IBFTC MAPBAK DECK		
	SUBROUTINE MAPBAC (MAP,MAPGO,TFFS,TFF,CNS,CN,PCN,T,MODE,IGJ,NUM)	1
	DATA NH,WL,WT,WS/6H H.P. ,6H L.P. ,6H TFF ,6HSPEED /	2
	DATA NM/6H I.P. /	3
	MAPS=MAP	4
	IF (MAP.EQ.3) MAP=2	5
	IF (NJM.GT.0) GO TO 1	6
	NUMH=0	7
	NUML=0	8
1	IGO=MAPGO+3*(MAP-1)	9
	GO TO (2,3,5,6,7,9),IGO	10

C ***	HIGH PRESSURE TURBINE	11
2	TFF=TFF+0.1*(TFF-TFFS)	12
	WRITE (8,10) WH,WT,TFFS,TFF	13
	RETURN	14
3	CN=CN+0.05*(CN-CNS)	15
	IF (MJDE.NE.1) PCN=PCN*(CN/CNS)	16
	IF (MJDE.EQ.1) T=T*(CNS/CN)**2	17
	WRITE (8,10) WH,WS,CNS,CN	18
	IF (NJMH.GT.2) GO TO 4	19
	NUM=1	20
	NUMH=VJMH+1	21
	RETURN	22
4	DELCN=CN-CNS	23
	IF (DELCN.GE.0.) RETURN	24
	TFF=TFF*(1.+DELCN/CN)	25
	WRITE (8,11) WH,WT,TFFS,TFF	26
	RETURN	27
5	TFF=TFF+0.1*(TFF-TFFS)	28
	WRITE (8,10) WH,WT,TFFS,TFF	29
	GO TO 3	30
C ***	LOW PRESSURE TURBINE	31
6	TFF=TFF+0.1*(TFF-TFFS)	32
	IF (MAPS.EQ.2) WRITE (8,10) WL,WT,TFFS,TFF	33
	IF (MAPS.EQ.3) WRITE (8,10) WM,WT,TFFS,TFF	34
	MAP=MAPS	35
	RETURN	36
7	CN=CN+0.05*(CN-CNS)	37
	IF (MJDE.NE.3) PCN=PCN*(CN/CNS)	38
	IF (MJDE.EQ.3) T=T*(CNS/CN)	39
	IF (MAPS.EQ.2) WRITE (8,10) WL,WS,CNS,CN	40
	IF (MAPS.EQ.3) WRITE (8,10) WM,WS,CNS,CN	41
	MAP=MAPS	42
	IF (NJML.GT.2) GO TO 8	43
	NUM=1	44
	NUML=VJML+1	45
	RETURN	46
8	DELCN=CN-CNS	47
	IF (DELCN.GE.0.) RETURN	48
	TFF=TFF*(1.+DELCN/CN)	49
	IF (MAPS.EQ.2) WRITE (8,11) WL,WT,TFFS,TFF	50
	IF (MAPS.EQ.3) WRITE (8,11) WM,WT,TFFS,TFF	51
	MAP=MAPS	52
	RETURN	53
9	TFF=TFF+0.1*(TFF-TFFS)	54
	IF (MAPS.EQ.2) WRITE (8,10) WL,WT,TFFS,TFF	55
	IF (MAPS.EQ.3) WRITE (8,10) WM,WT,TFFS,TFF	56
	MAP=MAPS	57
	GO TO 7	58
C		59
C		60
10	FORMAT (1H0,A6,12HTURBINE MAP ,A6,4HWAS=,E13.6,10H AND NOW=,E13.6	61
	1,6H\$\$\$\$\$)	62
11	FORMAT (1H0,A6,A6,22HWAS ALSO CHANGED FROM ,E13.6,5H TO ,E13.6,5H	63
	1\$\$\$\$\$)	64
	END	65

\$IBFTC	CONVRG	DECK		
	SUBROUTINE	CONVRG	(TI,HI,PI,SI,FAR,WG,PA,IDES,AO,PR,TO,HO,PO,SO,TS	1
			IO,PSO,VO,AMO,ICON)	2
C	ICON=1	SUBSONIC, COMPARE PI WITH PR		3
C	ICON=2	SONIC, COMPARE PI WITH PR		4
C	ICON=4	ERROR		5
	AJ=773.26			6
	CAPSF=2116.217			7
	G=32.174049			8
	CALL PROCOM (FAR,TI,XX1,XX2,XX3,XX4,PHII,XX6)			9
C ***	SONIC CALCULATIONS			10
	J=0			11
	TSS=0.833*TI			12
1	J=J+1			13
	CALL PROCOM (FAR,TSS,CSS,AKS,CP,REXS,PHISS,HSS)			14
	HSCAL=HI-CSS**2/(2.*G*AJ)			15
	DELHS=HSCAL-HSS			16
	IF (ABS(DELHS)-0.0005*HSCAL) 4,4,2			17
2	TSS=TSS+DELHS/CP			18
	IF (J-15) 1,1,3			19
3	ICON=4			20
	RETURN			21
4	IF (IDES) 12,12,5			22
C ***	ISENTROPIC EXPANSION CALCULATIONS			23
5	J=0			24
	TSI=TI*(PA/PI)**0.286			25
6	J=J+1			26
	CALL THERMO (PA,HSI,TSI,SSI,XX1,1,FAR,0)			27
	IF (ABS(SSI-SI)-0.0001*SI) 8,8,7			28
7	TSI=TSI/EXP(4.*(SSI-SI))			29
	IF (J-30) 6,6,3			30
8	VIS=SQRT(2.*G*AJ*(HI-HSI))			31
	IF (VIS-CSS) 9,11,11			32
C ***	SUBSONIC DESIGN, CALCULATE AO			33
9	VO=VIS			34
	TSO=TSI			35
	PSO=PA			36
	CALL PROCOM (FAR,TSO,CSO,XX2,XX3,REX,PHISO,HSO)			37
	RHO=CAPSF*PSO/(AJ*REX*TSO)			38
	AO=WG/(RHO*VO)			39
	AMO=VO/CSO			40
	PR=PI			41
	ICON=1			42
10	TO=TI			43
	HO=HI			44
	PO=PI			45
	SO=SI			46
	RETURN			47
C ***	SONIC DESIGN, CALCULATE AO			48
11	VO=CSS			49
	TSO=TSS			50
	PSO=PI*(TSO/TI)**(AKS/(AKS-1.))			51
	RHO=CAPSF*PSO/(AJ*REXS*TSO)			52
	AO=WG/(RHO*VO)			53
	AMO=1.0			54
	PR=PI			55
	ICON=2			56
	GO TO 10			57
C ***	NON-DESIGN, CALCULATE CRITICAL CONDITIONS			58
12	VO=CSS			59
	TSO=TSS			60
	PSO=PA			61
	RHO=CAPSF*PSO/(AJ*REXS*TSO)			62

AOCRIT=WG/(RHO*VO)	63
AMO=1.0	64
PR=PSJ*(TI/TSD)**(AKS/(AKS-1.))	65
IF (AJ-AOCRIT) 13,13,14	65
C *** NON-DESIGN, CRITICAL AND SUPERCRITICAL CONDITIONS	57
13 PSO=PSJ*AOCRIT/AO	68
PR=PR*AOCRIT/AO	69
ICON=2	70
GO TO 10	71
C *** NON-DESIGN, SUBSONIC CALCULATIONS	72
14 PSO=PA	73
J=0	74
TSD=0.833*TSD	75
15 J=J+1	76
CALL PROCOM (FAR,TSD,CSO,AKD,CP,REX,PHISO,HSD)	77
RHO=CAPSF*PSO/(AJ*REX*TSD)	78
VO=WG/(RHO*AO)	79
HSCAL=HI-VO**2/(2.*G*AJ)	80
DELHS=HSCAL-HSD	81
IF (ABS(DELHS)-0.0005*HSCAL) 17,17,16	82
16 TSD=TSD+DELHS/CP	83
IF (J-15) 15,15,3	84
17 AMO=VO/CSO	85
PR=PSJ*(TI/TSD)**(AKD/(AKD-1.))	85
ICON=1	87
GO TO 10	88
END	89

\$IBFTC CONDIV DECK

SUBRODTINE CONDIV (TI,HI,PI,SI,FAR,WG,PA,IDES,AT,AD,PIR,TT,HT,PT,S	1
1T,TO,HJ,PO,SO,TST,TSD,PST,PSO,VT,VO,AMT,AMO,ICON)	2
C ICON=1 SUBSONIC, COMPARE PIR WITH PI	3
C ICON=2 SONIC, SHOCK INSIDE NOZZLE, COMPARE PIR WITH PI	4
C ICON=3 SONIC, SHOCK OUTSIDE NOZZLE, COMPARE PIR WITH PI	5
C ICON=4 ERROR	6
DIMENSION Q(9)	7
Q(2)=0.	8
Q(3)=0.	9
AJ=778.26	10
CAPSF=2116.2170	11
G=32.174049	12
CALL PROCOM (FAR,TI,XX1,XX2,XX3,XX4,PHI1,XX6)	13
C *** SONIC CALCULATIONS	14
J=0	15
TSS=0.833*TI	15
1 J=J+1	17
CALL PROCOM (FAR,TSS,CSS,AK,CP,REXS,PHISS,HSS)	18
HSCAL=HI-CSS**2/(2.*G*AJ)	19
DELHS=HSCAL-HSS	20
IF (ABS(DELHS)-0.0005*HSCAL) 4,4,2	21
2 TSS=TSS+DELHS/CP	22
IF (J-15) 1,1,3	23
3 ICON=4	24
RETURN	25
4 IF (IDES) 11,11,5	26
C *** SONIC DESIGN, CALCULATE AT	27
5 VT=CSS	28
TST=TSS	29

	PST=PI*(TST/TI)**(AK/(AK-1.))	30
	RHO=CAPSF*PST/(AJ*REXS*TST)	31
	AT=WG/(RHO*VT)	32
	AMT=1.0	33
C ***	IDEAL EXPANSION DESIGN, CALCULATE AO	34
	PSO=PA	35
	J=0	36
	TSO=TI*(PSO/PI)**.286	37
6	J=J+1	38
	CALL PROCOM (FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)	39
	PHICAL=PHII-REX*ALOG(PI/PSO)	40
	DELPHI=PHICAL-PHISO	41
	IF (ABS(DELPHI)-0.0001*PHICAL) 8,8,7	42
7	TSO=TSO*EXP(4.*DELPHI)	43
	IF (J-15) 6,6,3	44
8	VO=SQRT(2.*G*AJ*(HI-HSO))	45
	AMO=VO/CSO	46
	AO=(AT/AMO)*(2.*(1.+(AK-1.)*AMO**2/2.)/(AK+1.))**((AK+1.)/(2.*(AK-	47
	11.)))	48
	PIR=PI	49
	ICON=3	50
9	TO=TI	51
	HO=HI	52
	PO=PI	53
	SO=SI	54
10	TT=TI	55
	HT=HI	56
	PT=PI	57
	ST=SI	58
	RETURN	59
C ***	ASSUME SONIC THROAT AND ISENTROPIC EXPANSION TO AO	60
11	VT=CSS	61
	AMT=1.0	62
	TST=TSS	63
	RHO=WG/(AT*VT)	64
	PST=RHO*AJ*REXS*TST/CAPSF	65
	PIR=PST*(TI/TST)**(AK/(AK-1.))	66
	IF (PST-PA) 12,27,27	67
12	TSO=0.95*TI	68
	MAM=0	69
13	CALL PROCOM (FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)	70
	AMO=SQRT(2.*((TI/TSO)-1.)/(AK-1.))	71
	AOCAL=(AT/AMO)*(2.*(1.+(AK-1.)*AMO**2/2.)/(AK+1.))**((AK+1.)/(2.*(72
	AK-1.)))	73
	EA=(AJ-AOCAL)/AO	74
	DIR=SQRT(AO/AOCAL)	75
	CALL AFQUIR (Q(1),TSO,EA,0.,100.,0.0001,DIR,TSOT,JCON)	76
	GO TO (14,18,3),JCON	77
14	TSO=TSOT	78
	IF (TSO-TI) 15,13,16	79
15	TSC=2.*TI/(AK+1.)	80
	IF (TSO.GT.TSC) GO TO 17	81
16	TSO=0.98*TI	82
	GO TO 13	83
17	IF (Q(2).LT.30.0.OR.AMO.LT.0.95.OR.MAM.EQ.1) GO TO 13	84
	TSO=2.*TI/(2.+0.98*(AK-1.))	85
	MAM=1	86
	GO TO 13	87
18	PSO=PIR*(TSO/TI)**(AK/(AK-1.))	88
	IF (PSO-PA) 20,19,27	89
C ***	CRITICAL FLOW, ISENTROPIC EXPANSION TO PA	90

19	VO=AMJ*CSO	91
	ICON=1	92
	GO TO 9	93
C ***	SUBSONIC FLOW	94
20	PSO=PA	95
	Q(2)=0.	96
	Q(3)=0.	97
	J=0	98
	TSO=0.833*TI	99
21	J=J+1	100
	CALL PROCOM (FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)	101
	RHO=CAPSF*PSO/(AJ*REX*TSO)	102
	VO=WG/(RHO*AO)	103
	HSCAL=HI-VO**2/(2.*G*AJ)	104
	DELHS=HSCAL-HSO	105
	IF (ABS(DELHS)-0.0005*HSCAL) 23,23,22	106
22	TSO=TSO+DELHS/CP	107
	IF (J-15) 21,21,3	108
23	AMO=VO/CSO	109
	PIR=PSO*(TI/TSO)**(AK/(AK-1.))	110
	TST=TSO	111
24	CALL PROCOM (FAR,TST,CST,AK,CP,REX,PHIST,HST)	112
	PST=PIR*(TST/TI)**(AK/(AK-1.))	113
	RHO=PST*CAPSF/(AJ*REX*TST)	114
	VT=WG/(RHO*AT)	115
	HSCAL=HI-VT**2/(2.*G*AJ)	116
	EH=(HSCAL-HST)/HSCAL	117
	DIR=1.+(HSCAL-HST)/(CP*TST)	118
	CALL AFQUIR (Q(1),TST,EH,0.,20.,0.0005,DIR,TSTT,JCON)	119
	GO TO (25,26,3),JCON	120
25	TST=TSTT	121
	GO TO 24	122
26	AMT=VT/CST	123
	ICON=1	124
	GO TO 9	125
C ***	SUPERCRITICAL FLOW, ISENTROPIC EXPANSION TO PA	126
27	PSO=PA	127
	J=0	128
	TSO=TI*(PSO/PIR)**.286	129
28	J=J+1	130
	CALL PROCOM (FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)	131
	PHICAL=PHII-REX*ALOG(PIR/PSO)	132
	DELPHI=PHICAL-PHISO	133
	IF (ABS(DELPHI)-0.0001*PHICAL) 30,30,29	134
29	TSO=TSO*EXP(4.0*DELPHI)	135
	IF (J-15) 28,28,3	136
30	VO=SQRT(2.*G*AJ*(HI-HSO))	137
	AMO=VO/CSO	138
	AOID=(AT/AMO)*(2.*(1.+(AK-1.)*AMO**2/2.)/(AK+1.))**((AK+1.)/(2.*(AK-1.)))	139
	ICON=3	141
	N=0	142
	IF (AJ-AOID) 31,9,32	143
C ***	SUPERCRITICAL FLOW, ISENTROPIC EXPANSION TO AO	144
31	N=1	145
32	TSO=0.833*TI	146
	J=0	147
33	J=J+1	148
	CALL PROCOM (FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)	149
	AMO=SQRT(2.*((TI/TSO)-1.)/(AK-1.))	150
	AOCAL=(AT/AMO)*(2.*(1.+(AK-1.)*AMO**2/2.)/(AK+1.))**((AK+1.)/(2.*(AK-1.)))	151
		152

	DELA=AO-AOCAL	153
	IF (ABS(DELA)-0.0001*AO) 35,35,34	154
34	TSO=TSO*SQRT(AOCAL/AO)	155
	IF (J-50) 33,33,3	156
35	IF (N) 37,37,36	157
C ***	UNDEREXPANDED, SHOCK OUTSIDE NOZZLE	158
36	PSQ=PIR*(TSO/TI)**(AK/(AK-1.))	159
	VO=AMJ*CSO	160
	GO TO 9	161
C ***	OVEREXPANDED, FIND SHOCK POSITION	162
37	PSX=PIR*(TSO/TI)**(AK/(AK-1.))	163
	PSY=PSX*(2.*AK*AMO**2/(AK+1.)-(AK-1.)/(AK+1.))	164
	IF (PA-PSY) 38,39,39	165
C ***	OVEREXPANDED, SHOCK OUTSIDE NOZZLE	166
38	PSO=PSX	167
	VO=AMJ*CSO	168
	GO TO 9	169
C ***	OVEREXPANDED, SHOCK INSIDE NOZZLE	170
39	PSO=PA	171
	J=0	172
	TSO=0.833*TI	173
40	J=J+1	174
	CALL PROCOM (FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)	175
	RHO=CAPSF*PSO/(AJ*REX*TSO)	176
	VO=WG/(RHO*AO)	177
	HSCAL=HI-VO**2/(2.*G*AJ)	178
	DELHS=HSCAL-HSO	179
	IF (ABS(DELHS)-0.0005*HSCAL) 42,42,41	180
41	TSO=TSO+DELHS/CP	181
	IF (J-15) 40,40,3	182
42	AMO=VJ/CSO	183
	TO=TI	184
	HO=HI	185
	PO=PSJ*(TO/TSO)**(AK/(AK-1.))	186
	SO=PHI I-REX*ALOG(PO)	187
	ICON=2	188
	GO TO 10	189
	END	190

\$IBFTC	THTRB	DECK	
	SUBROJTINE	THTURB (DH,ETA,FAR,H,S,P,TO,HO,SO,PO)	1
	HO=H-DH		2
	HOP=H-DH/ETA		3
	PT=P/2.		4
	DO 1 I=1,25		5
	CALL THERMO (PT,HOP,TT,ST,AMWT,1,FAR,1)		6
	DELS=ST-S		7
	IF (ABS(DELS).LE.0.00005*S) GO TO 2		8
1	PT=P*EXP(DELS*AMWT/1.986375+ALOG(PT/P))		9
	CALL ERROR		10
2	PO=PT		11
	CALL THERMO (PO,HO,TO,SO,X1,1,FAR,1)		12
	RETURN		13
	END		14

\$IBFTC THERMO DECK	
SUBROUTINE THERMO (PX,HX,TX,SX,AMX,L,FAR,K)	1
FX=0.	2
IF (L.EQ.1) FX=FAR	3
IF (K.EQ.1) GO TO 1	4
CALL PROCOM (FX,TX,CS,AK,CP,R,PHI,HX)	5
GO TO 3	6
1 TX=4.*HX	7
DO 2 I=1,15	8
CALL PROCOM (FX,TX,CS,AK,CP,R,PHI,H)	9
DELH=HX-H	10
IF (ABS(DELH).LE.0.00001*HX) GO TO 3	11
2 TX=TX+4.*DELH	12
WRITE (8,4)	13
3 SX=PHI-R*ALOG(PX)	14
AMX=1.986375/R	15
RETURN	16
C	17
C	18
4 FORMAT (31HONO CONVERGENCE IN THERMO\$\$\$\$\$)	19
END	20

\$IBFTC SERCH DECK	
SUBROUTINE SEARCH (P,A,B,C,D,AX,NA,BX,CX,DX,NO,NAM,NOM,NCODE)	1
DIMENSION AX(NAM),BX(NAM,NOM),CX(NAM,NOM),DX(NAM,NOM),NO(NAM),Q(9)	2
C *** NEEDS SUBROUTINE AFQUIR	3
C *** AX AND BX MUST BE STORED LO TO HI	4
C *** P=INPJT PROPORTION BETWEEN 0.0 AND 1.0	5
C *** IF NOT INPUT, P MUST EQUAL -1.	6
C *** NCODE=00 OK	7
C NCODE=01 A LO	8
C NCODE=02 A HI	9
C NCODE=07 ERROR	10
C NCODE=10 B LO	11
C NCODE=20 B HI	12
NCODE=0	13
C=0.	14
D=0.	15
C *** FIND A	16
DO 1 I=1,NA	17
IH=I	18
IF (A.LT.AX(I)) GO TO 2	19
1 CONTINUE	20
IF (A.GT.AX(IH)) NCODE=2	21
A=AX(IH)	22
GO TO 3	23
2 IF (I+.GT.1) GO TO 3	24
NCODE=1	25
IH=2	26
A=AX(1)	27
3 IL=IH-1	28
LIMH=ND(IH)	29
LIML=ND(IL)	30
C *** FIND B	31
PRM=(A-AX(IL))/(AX(IH)-AX(IL))	32
PP=P	33
IF (P.GE.0.) GO TO 6	34
BL=BX(IL,1)+PRM*(BX(IH,1)-BX(IL,1))	35

	BH=BX(IL,LIML)+PRM*(BX(IH,LIMH)-BX(IL,LIML))	36
	IF (B.GE.BL) GO TO 4	37
	NCODE=NCODE+10	38
	B=BL	39
	GO TO 5	40
4	IF (B.LE.BH) GO TO 5	41
	NCODE=NCODE+20	42
	B=BH	43
5	PP=0.5	44
	Q(2)=0.	45
	Q(3)=0.	46
6	BH=PP*(BX(IH,LIMH)-BX(IH,1))+BX(IH,1)	47
	BL=PP*(BX(IL,LIML)-BX(IL,1))+BX(IL,1)	48
	DO 7 J=2,LIMH	49
	JH=J	50
	IF (B+.LT.BX(IH,J)) GO TO 8	51
7	CONTINUE	52
8	JL=JH-1	53
	DO 9 K=2,LIML	54
	KH=K	55
	IF (BL.LT.BX(IL,K)) GO TO 10	56
9	CONTINUE	57
10	KL=KH-1	58
	PR=(BX(IH,JL)-BH)/(BX(IH,JH)-BX(IH,JL))	59
	CH=CX(IH,JL)-PR*(CX(IH,JH)-CX(IH,JL))	60
	DH=DX(IH,JL)-PR*(DX(IH,JH)-DX(IH,JL))	61
	PR=(BX(IL,KL)-BL)/(BX(IL,KH)-BX(IL,KL))	62
	CL=CX(IL,KL)-PR*(CX(IL,KH)-CX(IL,KL))	63
	DL=DX(IL,KL)-PR*(DX(IL,KH)-DX(IL,KL))	64
	BT=BL+PRM*(BH-BL)	65
	CT=CL+PRM*(CH-CL)	66
	DT=DL+PRM*(DH-DL)	67
	IF (P.GE.0.) GO TO 13	68
	DIR=SQRT(B/BT)	69
	ERR=(B-BT)/B	70
	CALL AFQUIR (Q(1),PP,ERR,0.,25.,0.001,DIR,PT,ICON)	71
	GO TO (11,13,12),ICON	72
11	PP=PT	73
	IF (P>.LT.0.) PP=0.	74
	IF (P>.GT.1.) PP=1.	75
	GO TO 5	76
12	NCODE=7	77
13	B=BT	78
	C=CT	79
	D=DT	80
	RETURN	81
	END	82

\$IBFTC AFQUER	DECK	
SUBROUTINE AFQUIR	(X,AIND,DEPEND,ANS,AJ,TOL,DIR,ANEW,ICON)	1
DIMENSION	X(9)	2
C X(1)=NAME	OF ARRAY TO USE	3
C AIND=INDEPENDANT	VARIABLE	4
C DEPEND=	DEPENDANT VARIABLE	5
C ANS=ANSWER	UPON WHICH TO CONVERGE	6
C AJ=MAX	NUMBER OF TRYs	7
C TOL=PERCENT	TOLERANCE FOR CONVERGENCE	8
C DIR=DIRECTION	AND PERCENTAGE FOR FIRST GUESS	9

C	ANEW=CALCULATED VALUE OF NEXT TRY AT INDEPENDANT VARIABLE	10
C	ICON=CONTRL =1 GO THRU LOOP AGAIN	11
C	=2 YOU HAVE REACHED THE ANSWER	12
C	=3 COUNTER HAS HIT LIMITS	13
C	X(2)=COUNTER STORAGE	14
C	X(3)=CHOOSES METHOD OF CONVERGENCE	15
C	X(4)=THIRD DEPEND VAR	16
C	X(5)=THIRD IND VAR	17
C	X(6)=SECOND DEPEND VAR	18
C	X(7)=SECOND IND VAR	19
C	X(8)=FIRST DEPEND VAR	20
C	X(9)=FIRST IND VAR	21
C	X(3) MUST BE ZERO UPON FIRST ENTRY TO ROUTINE	22
	Y=0.	23
	IF (ANS) 1,2,1	24
1	DEP=DEPEND-ANS	25
	TOLANS=TOL*ANS	26
	GO TO 3	27
2	DEP=DEPEND	28
	TOLANS=TOL	29
3	IF (ABS(DEP)-TOLANS) 5,5,4	30
4	IF (X(2)-AJ) 8,8,7	31
5	ANEW=AIND	32
	X(2)=0.	33
	ICON=2	34
	RETURN	35
6	ANEW=Y	36
	X(2)=X(2)+1.	37
	ICON=1	38
	RETURN	39
7	ANEW=Y	40
	X(2)=0.	41
	ICON=3	42
	RETURN	43
8	IF (X(3)) 9,9,12	44
C ***	FIRST GUESS USING DIR	45
9	X(3)=1.	46
	X(8)=DEP	47
	X(9)=AIND	48
	IF (AIND) 10,11,10	49
10	Y=DIR*AIND	50
	GO TO 5	51
11	Y=DIR	52
	GO TO 6	53
12	IF (X(3)-1.) 13,13,16	54
C ***	LINEAR GUESS	55
13	X(3)=2.	56
	X(6)=DEP	57
	X(7)=AIND	58
	IF (X(8)-X(6)) 14,9,14	59
14	IF (X(9)-X(7)) 15,9,15	60
15	A=(X(9)-X(7))/(X(8)-X(6))	61
	Y=X(9)-A*X(8)	62
	IF (ABS(10.*X(9))-ABS(Y)) 9,9,6	63
C ***	QUADRATIC GUESS	64
16	X(4)=DEP	65
	X(5)=AIND	66
	IF (X(7)-X(5)) 18,17,18	67
17	IF (X(6)-X(4)) 13,9,13	68
18	IF (X(6)-X(4)) 19,13,19	69
19	IF (X(9)-X(5)) 23,20,23	70
20	IF (X(8)-X(4)) 21,22,21	71

21	X(9)=X(7)	72
	X(8)=X(6)	73
	GO TO 13	74
22	X(9)=X(7)	75
	X(8)=X(6)	76
	X(3)=1.	77
	IF (X(9)) 10,11,10	78
23	IF (X(8)-X(4)) 24,21,24	79
24	F=(X(5)-X(4))/(X(7)-X(5))	80
	A=(X(8)-X(4)-F*(X(9)-X(5)))/(X(9)-X(7))*X(9)-X(5))	81
	B=F-A*(X(5)+X(7))	82
	C=X(4)+X(5)*(A*X(7)-F)	83
	IF (A) 27,25,27	84
25	IF (B) 26,7,26	85
26	Y=-C/B	86
	GO TO 47	87
27	IF (B) 32,28,32	88
28	IF (C) 30,29,30	89
29	Y=0.	90
	GO TO 47	91
30	G=-C/A	92
	IF (G) 7,7,31	93
31	Y=SQRT(G)	94
	YY=-SQRT(G)	95
	GO TO 37	96
32	IF (C) 34,33,34	97
33	Y=-B/A	98
	YY=0.	99
	GO TO 37	100
34	D=4.*A*C/B**2	101
	IF (1.-D) 13,35,36	102
35	Y=-B/(2.*A)	103
	GO TO 47	104
36	E=SQRT(1.-D)	105
	Y=(-B/(2.*A))*(1.+E)	106
	YY=(-B/(2.*A))*(1.-E)	107
37	J=4	108
	DEPMIN=ABS(X(4))	109
	DO 39 I=6,8,2	110
	IF (DEPMIN-ABS(X(I))) 39,39,38	111
38	J=I	112
	DEPMIN=ABS(X(I))	113
39	CONTINUE	114
	K=J+1	115
	IF ((X(K)-Y)*(X(K)-YY)) 42,42,40	116
40	IF (ABS(X(K)-Y)-ABS(X(K)-YY)) 47,47,41	117
41	Y=YY	118
	GO TO 47	119
42	IF (J-6) 43,44,44	120
43	JJ=J+2	121
	KK=K+2	122
	GO TO 45	123
44	JJ=J-2	124
	KK=K-2	125
45	SLOPE=(X(KK)-X(K))/(X(JJ)-X(J))	126
	IF (SLOPE*X(J)*(X(K)-Y)) 46,46,47	127
46	Y=YY	128
47	X(9)=X(7)	129
	X(8)=X(6)	130
	X(7)=X(5)	131
	X(6)=X(4)	132
	GO TO 6	133
	END	134

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$IBFTC PARABO DECK
SUBROJTIME PARABO (X,Y,XD,YANS)
DIMENSION X(3),Y(3)
A=((X(1)-X(2))*(Y(1)-Y(3))-(X(1)-X(3))*(Y(1)-Y(2)))/((X(1)-X(2))*(
1X(1)-X(3))*(X(3)-X(2)))
B=((X(1)**2-X(2)**2)*(Y(1)-Y(3))-(X(1)**2-X(3)**2)*(Y(1)-Y(2)))/((
1X(1)-X(2))*(X(1)-X(3))*(X(2)-X(3)))
D=(Y(1)*X(2)**2-Y(2)*X(1)**2-B*X(2)*X(1)*(X(2)-X(1)))/(X(2)**2-X(1
1)**2)
YANS=(A*XD+B)*XD+D
RETURN
END

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$IBFTC OVELAY DECK
C DUMMY ROUTINE TO RESTORE ALL OF WORKING PROGRAM TO CORE AT 1 TIME
SUBROJTIME OVLAY
X=X
RETURN
END

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$IBFTC BLKFAN DECK
C THIS IS A GENERALIZED FAN MAP FOR UNREALISTIC SUPERSONIC ENGINE
BLOCK DATA
COMMON / FAN/CN(15),PR(15,15),WAC(15,15),ETA(15,15),N,NP(15)
DATA N,NP/10,6,3*7,5*10,8,5*0/
DATA CN/0.3,0.4,0.5,0.6,0.7,0.8,0.9,1.0,1.1,1.2,5*0./
DATA (PR( 1,J),WAC( 1,J),ETA( 1,J),J=1, 6)/
1 1.00000, 243.600, 0.75592, 1.01200, 229.800, 0.76120,
2 1.02800, 199.800, 0.76648, 1.03840, 166.800, 0.75592,
3 1.04480, 133.200, 0.72512, 1.04800, 86.400, 0.64152/
DATA (PR( 2,J),WAC( 2,J),ETA( 2,J),J=1, 7)/
1 1.00000, 286.800, 0.75592, 1.02000, 270.000, 0.77616,
2 1.04000, 253.200, 0.79200, 1.05840, 233.400, 0.79728,
3 1.07520, 209.400, 0.80256, 1.09200, 183.600, 0.77616,
4 1.10000, 156.600, 0.74008/
DATA (PR( 3,J),WAC( 3,J),ETA( 3,J),J=1, 7)/
1 1.00000, 333.600, 0.75064, 1.02560, 322.800, 0.77616,
2 1.05120, 310.200, 0.80256, 1.08000, 291.600, 0.82808,
3 1.11600, 259.800, 0.84392, 1.13200, 240.000, 0.82808,
4 1.14800, 213.600, 0.77616/
DATA (PR( 4,J),WAC( 4,J),ETA( 4,J),J=1, 7)/
1 1.00000, 383.400, 0.74536, 1.03680, 376.200, 0.77616,
2 1.03800, 358.200, 0.82808, 1.12400, 340.200, 0.85448,
3 1.15000, 313.200, 0.88000, 1.18960, 276.600, 0.82808,
4 1.19520, 266.400, 0.80784/
DATA (PR( 5,J),WAC( 5,J),ETA( 5,J),J=1,10)/
1 1.00000, 439.800, 0.72512, 1.06400, 436.800, 0.77616,
2 1.11840, 428.400, 0.82808, 1.14800, 420.600, 0.85448,
3 1.13400, 406.800, 0.88000, 1.20960, 393.600, 0.90112,
4 1.21760, 388.200, 0.90376, 1.22400, 383.400, 0.90112,
5 1.24400, 368.400, 0.88000, 1.26720, 342.600, 0.82808/

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DATA (PR(6,J),WAC(6,J),ETA(6,J),J=1,10)/							31
1	1.00000,	499.800,	0.68816,	1.10000,	499.800,	0.77616,	32
2	1.15000,	493.200,	0.82808,	1.20000,	485.400,	0.85448,	33
3	1.22800,	476.400,	0.88000,	1.25520,	466.800,	0.90112,	34
4	1.27200,	456.600,	0.91080,	1.28640,	448.200,	0.90112,	35
5	1.30240,	433.200,	0.88000,	1.33200,	406.800,	0.82720/	36
DATA (PR(7,J),WAC(7,J),ETA(7,J),J=1,10)/							37
1	1.00000,	566.400,	0.64152,	1.07600,	566.400,	0.72512,	38
2	1.15200,	566.400,	0.77616,	1.21920,	559.800,	0.82808,	39
3	1.25000,	553.200,	0.85888,	1.28960,	544.800,	0.88000,	40
4	1.33120,	528.600,	0.90112,	1.36160,	509.400,	0.88000,	41
5	1.39120,	483.600,	0.82808,	1.40000,	474.000,	0.81752/	42
DATA (PR(8,J),WAC(8,J),ETA(8,J),J=1,10)/							43
1	1.00000,	633.600,	0.60016,	1.04400,	633.600,	0.64152,	44
2	1.13520,	633.600,	0.72512,	1.22080,	633.000,	0.77616,	45
3	1.29440,	625.800,	0.82808,	1.34000,	616.800,	0.85888,	46
4	1.40000,	600.000,	0.88000,	1.42800,	586.800,	0.85888,	47
5	1.44800,	576.600,	0.82808,	1.48000,	553.200,	0.78672/	48
DATA (PR(9,J),WAC(9,J),ETA(9,J),J=1,10)/							49
1	1.00000,	700.200,	0.56936,	1.10400,	700.200,	0.64152,	50
2	1.22000,	700.200,	0.72512,	1.32400,	700.200,	0.77616,	51
3	1.40000,	700.200,	0.80256,	1.44800,	698.400,	0.80784,	52
4	1.50000,	693.600,	0.80256,	1.53360,	683.400,	0.77616,	53
5	1.55800,	666.600,	0.74536,	1.58400,	656.400,	0.72512/	54
DATA (PR(10,J),WAC(10,J),ETA(10,J),J=1, 8)/							55
1	1.00000,	750.000,	0.51744,	1.16320,	750.000,	0.64152,	56
2	1.31200,	750.000,	0.72512,	1.40000,	750.000,	0.75592,	57
3	1.43000,	750.000,	0.76120,	1.54000,	750.000,	0.75064,	58
4	1.58000,	749.400,	0.72512,	1.66000,	736.800,	0.64152/	59
END							60

\$IBFTC BLKINT DECK

C	THIS IS A GENERALIZED FAN MAP FOR UNREALISTIC SUPERSONIC ENGINE	1
	BLOCK DATA	2
	COMMON / INT / CN(15),PR(15,15),WAC(15,15),ETA(15,15),N,NP(15)	3
	DATA N,NP(10,6,3*7,5*10,8,5*0/	4
	DATA CN(0.3,0.4,0.5,0.6,0.7,0.8,0.9,1.0,1.1,1.2,5*0./	5
	DATA (PR(1,J),WAC(1,J),ETA(1,J),J=1, 6)/	6
1	1.00000, 121.800, 0.75592, 1.01800, 114.900, 0.76120,	7
2	1.04200, 99.900, 0.76648, 1.05760, 83.400, 0.75592,	8
3	1.05720, 66.600, 0.72512, 1.07200, 43.200, 0.64152/	9
	DATA (PR(2,J),WAC(2,J),ETA(2,J),J=1, 7)/	10
1	1.00000, 143.400, 0.75592, 1.03000, 135.000, 0.77616,	11
2	1.05000, 126.600, 0.79200, 1.08760, 116.700, 0.79728,	12
3	1.11280, 104.700, 0.80256, 1.13800, 91.800, 0.77616,	13
4	1.15000, 78.300, 0.74008/	14
	DATA (PR(3,J),WAC(3,J),ETA(3,J),J=1, 7)/	15
1	1.00000, 166.800, 0.75064, 1.03840, 161.400, 0.77616,	16
2	1.07680, 155.100, 0.80256, 1.12000, 145.800, 0.82808,	17
3	1.17400, 129.900, 0.84392, 1.19800, 120.000, 0.82808,	18
4	1.22200, 106.800, 0.77616/	19
	DATA (PR(4,J),WAC(4,J),ETA(4,J),J=1, 7)/	20
1	1.00000, 191.700, 0.74536, 1.05520, 188.100, 0.77616,	21
2	1.13200, 179.100, 0.82808, 1.18600, 170.100, 0.85448,	22
3	1.24000, 156.600, 0.88000, 1.28440, 138.300, 0.82808,	23
4	1.29280, 133.200, 0.80784/	24

DATA (PR(5,J),WAC(5,J),ETA(5,J),J=1,10)/							25
1	1.00000,	219.900,	0.72512,	1.09600,	218.400,	0.77616,	26
2	1.17760,	214.200,	0.82808,	1.22200,	210.300,	0.85448,	27
3	1.27600,	203.400,	0.88000,	1.31440,	196.800,	0.90112,	28
4	1.32640,	194.100,	0.90376,	1.33600,	191.700,	0.90112,	29
5	1.35500,	184.200,	0.88000,	1.40080,	171.300,	0.82808/	30
DATA (PR(6,J),WAC(6,J),ETA(6,J),J=1,10)/							31
1	1.00000,	249.900,	0.68816,	1.15000,	249.900,	0.77616,	32
2	1.24000,	246.600,	0.82808,	1.30000,	242.700,	0.85448,	33
3	1.34200,	238.200,	0.88000,	1.38280,	233.400,	0.90112,	34
4	1.40900,	228.300,	0.91080,	1.42960,	224.100,	0.90112,	35
5	1.45360,	216.600,	0.88000,	1.49800,	203.400,	0.82720/	36
DATA (PR(7,J),WAC(7,J),ETA(7,J),J=1,10)/							37
1	1.00000,	283.200,	0.64152,	1.11400,	283.200,	0.72512,	38
2	1.22800,	283.200,	0.77616,	1.32880,	279.900,	0.82808,	39
3	1.39000,	276.600,	0.85888,	1.43440,	272.400,	0.88000,	40
4	1.49580,	264.300,	0.90112,	1.54240,	254.700,	0.88000,	41
5	1.59580,	241.800,	0.82808,	1.60000,	237.000,	0.81752/	42
DATA (PR(8,J),WAC(8,J),ETA(8,J),J=1,10)/							43
1	1.00000,	316.800,	0.60016,	1.06600,	316.800,	0.64152,	44
2	1.20280,	316.800,	0.72512,	1.33120,	316.500,	0.77616,	45
3	1.44160,	312.900,	0.82808,	1.51000,	308.400,	0.85888,	46
4	1.60000,	300.000,	0.88000,	1.64200,	293.400,	0.85888,	47
5	1.67200,	288.300,	0.82808,	1.72000,	276.600,	0.78672/	48
DATA (PR(9,J),WAC(9,J),ETA(9,J),J=1,10)/							49
1	1.00000,	350.100,	0.56936,	1.15600,	350.100,	0.64152,	50
2	1.33000,	350.100,	0.72512,	1.48600,	350.100,	0.77616,	51
3	1.60000,	350.100,	0.80256,	1.67200,	349.200,	0.80784,	52
4	1.75000,	346.800,	0.80256,	1.80040,	341.700,	0.77616,	53
5	1.85200,	333.300,	0.74536,	1.87600,	328.200,	0.72512/	54
DATA (PR(10,J),WAC(10,J),ETA(10,J),J=1, 8)/							55
1	1.00000,	375.000,	0.51744,	1.24480,	375.000,	0.64152,	56
2	1.45800,	375.000,	0.72512,	1.60000,	375.000,	0.75592,	57
3	1.72000,	375.000,	0.76120,	1.81000,	375.000,	0.75064,	58
4	1.87000,	374.700,	0.72512,	1.99000,	368.400,	0.64152/	59
END							60

\$IBFTC BLKCMF DECK

C	THIS IS GENERALIZED COMP. MAP FOR UNREALISTIC SUPERSONIC ENGINE	1
	BLOCK DATA	2
	COMMON / COMP/CN(15),PR(15,15),WAC(15,15),ETA(15,15),N,NP(15)	3
	DATA N,NP/10,2*6,2*8,4*10,2*8,5*0/	4
	DATA CV/.562,.674,.787,.899,1.,1.034,1.067,1.124,1.236,1.292,5*0./	5
	DATA (PR(1,J),WAC(1,J),ETA(1,J),J=1, 6)/	6
1	1.00000, 51.000, 0.59082, 1.84000, 50.200, 0.62178,	7
2	2.42800, 49.500, 0.64242, 2.86900, 48.800, 0.65274,	8
3	3.83500, 46.700, 0.67338, 4.54900, 44.500, 0.64242/	9
	DATA (PR(2,J),WAC(2,J),ETA(2,J),J=1, 6)/	10
1	1.00000, 59.300, 0.59082, 1.96600, 59.300, 0.64242,	11
2	3.09300, 58.800, 0.69402, 3.93300, 57.900, 0.72498,	12
3	4.68900, 56.700, 0.74562, 5.52900, 55.000, 0.72498/	13
	DATA (PR(3,J),WAC(3,J),ETA(3,J),J=1, 8)/	14
1	1.00000, 70.000, 0.58566, 1.84000, 70.000, 0.64242,	15
2	2.68000, 70.000, 0.68370, 3.40800, 69.500, 0.72498,	16
3	4.52100, 68.800, 0.77744, 5.44500, 67.900, 0.79292,	17
4	6.31300, 66.400, 0.77744, 6.52300, 65.700, 0.76970/	18

DATA (PR(4,J),WAC(4,J),ETA(4,J),J=1, 8)/	19
1 1.00000, 84.800, 0.58050, 2.00800, 84.800, 0.64242,	20
2 3.42900, 84.800, 0.72498, 4.60500, 84.800, 0.77744,	21
3 5.69700, 84.000, 0.80840, 6.61400, 83.300, 0.82904,	22
4 7.53800, 81.700, 0.80840, 7.95800, 80.500, 0.79292/	23
DATA (PR(5,J),WAC(5,J),ETA(5,J),J=1,10)/	24
1 1.00000, 101.700, 0.57190, 2.51900, 101.700, 0.64242,	25
2 3.93200, 101.700, 0.72498, 5.27700, 101.700, 0.77744,	25
3 6.43300, 101.200, 0.80840, 7.20200, 101.000, 0.83936,	27
4 8.00000, 100.000, 0.86000, 8.56700, 99.500, 0.83936,	28
5 9.38500, 98.100, 0.80840, 9.59600, 97.400, 0.80582/	29
DATA (PR(6,J),WAC(6,J),ETA(6,J),J=1,10)/	30
1 1.00000, 108.100, 0.57018, 2.85500, 108.100, 0.64242,	31
2 4.29700, 108.100, 0.72498, 5.61300, 108.100, 0.77744,	32
3 6.93500, 107.600, 0.80840, 7.62200, 107.100, 0.83936,	33
4 8.54500, 106.700, 0.86000, 9.13400, 106.000, 0.83936,	34
5 9.92500, 104.500, 0.80840, 10.21900, 104.000, 0.80410/	35
DATA (PR(7,J),WAC(7,J),ETA(7,J),J=1,10)/	36
1 1.00000, 114.500, 0.55986, 3.26100, 114.500, 0.64242,	37
2 4.75900, 114.500, 0.72498, 6.11700, 114.500, 0.77744,	38
3 7.45400, 114.500, 0.80840, 8.30800, 114.300, 0.83936,	39
4 9.21800, 113.600, 0.84968, 9.63800, 113.300, 0.83936,	40
5 10.51300, 112.600, 0.80840, 10.99600, 112.400, 0.79808/	41
DATA (PR(8,J),WAC(8,J),ETA(8,J),J=1,10)/	42
1 1.00000, 122.900, 0.53922, 1.68600, 122.900, 0.57018,	43
2 3.84900, 122.900, 0.64242, 5.46600, 122.900, 0.72498,	44
3 6.85500, 122.900, 0.77744, 8.37100, 122.900, 0.80840,	45
4 8.95500, 122.600, 0.82388, 9.88300, 122.100, 0.83936,	46
5 10.91200, 121.700, 0.80840, 11.81500, 120.700, 0.77744/	47
DATA (PR(9,J),WAC(9,J),ETA(9,J),J=1, 8)/	48
1 1.00000, 139.800, 0.47644, 4.35300, 139.800, 0.60114,	49
2 7.62200, 139.800, 0.72498, 10.21900, 139.800, 0.77744,	50
3 11.05900, 139.800, 0.78260, 11.89900, 139.500, 0.77744,	51
4 13.15900, 139.300, 0.72498, 13.65600, 139.000, 0.69918/	52
DATA (PR(10,J),WAC(10,J),ETA(10,J),J=1, 8)/	53
1 1.00000, 146.200, 0.46612, 3.76500, 146.200, 0.57018,	54
2 6.43100, 146.200, 0.64242, 9.17600, 146.200, 0.72498,	55
3 10.21900, 146.200, 0.75078, 11.47900, 146.200, 0.75078,	56
4 12.71100, 146.200, 0.72498, 14.41200, 146.200, 0.64242/	57
END	58

\$IBFTC CMBDT DECK

BLOCK DATA	1
COMMON / COMB/PSI(15),DELT(15,15),ETA(15,15),N,NP(15)	2
DATA N,NP / 15,15*15 /	3
DATA PSI/4.9116,9.8232,14.735,19.646,24.558,29.470,34.381,	4
139.293,44.207,73.674,100.,200.,300.,400.,500./	5
DATA DELT/15*200.,15*300.,15*400.,15*500.,15*600.,15*700.,15*800.,	6
115*900.,15*1000.,15*1100.,15*1200.,15*1300.,15*1400.,15*1500.,	7
215*1600./	8
DATA ETA/	9
1.600, .726, .777, .806, .826, .843, .855, .865, 7*.870,	10
2.758, .825, .858, .875, .888, .898, .906, .912, .914, 6*.915,	11
3.868, .893, .911, .925, .935, .942, .947, .951, 7*.953,	12
4.925, .936, .946, .955, .963, .969, .974, .977, .978, 6*.979,	13
5.960, .966, .972, .977, .982, .985, .990, .992, .993, 6*.995,	14
6.988, .991, .992, .994, .995, .997, .998, 8*.999,	15
78*1.00, 7*.999, 120*1.00/	16
END	17

\$IBFTC HPTDAT DECK

BLOCK DATA

COMMON / HTURB/TFF(15),CN(15,15),DH(15,15),ETA(15,15),N,NP(15)

DATA N,NP/10,9*15,12,5*0/

DATA TFF / 39.670, 42.990, 47.460, 48.610, 49.175,
1 49.600, 50.000, 50.425, 50.920, 51.575, 5*0./

DATA {CN(1,J),DH(1,J),ETA(1,J),J=1,15)/

1 0.1872, 0.0032, 0.6219, 0.3372, 0.0057, 0.7078,
2 0.5156, 0.0084, 0.7868, 0.7128, 0.0108, 0.8090,
3 0.9382, 0.0133, 0.8090, 1.1442, 0.0152, 0.7963,
4 1.3138, 0.0164, 0.7779, 1.5382, 0.0174, 0.7422,
5 1.7264, 0.0179, 0.7078, 1.9324, 0.0176, 0.7635,
6 2.1500, 0.0167, 0.6068, 2.4058, 0.0144, 0.5309,
7 2.5892, 0.0120, 0.4773, 2.7862, 0.0082, 0.4045,
8 2.9460, 0.0034, 0.3034/

DATA {CN(2,J),DH(2,J),ETA(2,J),J=1,15)/

1 0.1872, 0.0038, 0.6068, 0.3942, 0.0080, 0.7078,
2 0.5914, 0.0113, 0.8090, 0.7128, 0.0136, 0.8292,
3 0.9342, 0.0156, 0.8363, 0.9804, 0.0176, 0.8393,
4 1.1368, 0.0192, 0.8368, 1.2754, 0.0212, 0.8302,
5 1.4450, 0.0228, 0.8254, 1.7068, 0.0248, 0.8090,
6 1.9596, 0.0260, 0.7696, 2.2706, 0.0261, 0.7078,
7 2.5370, 0.0241, 0.6068, 3.0960, 0.0188, 0.5056,
8 3.3774, 0.0128, 0.4197/

DATA {CN(3,J),DH(3,J),ETA(3,J),J=1,15)/

1 0.1872, 0.0046, 0.5764, 0.4362, 0.0100, 0.7078,
2 0.5568, 0.0144, 0.8090, 0.8726, 0.0184, 0.8494,
3 1.0596, 0.0216, 0.8543, 1.2382, 0.0240, 0.8515,
4 1.4538, 0.0268, 0.8494, 1.6882, 0.0292, 0.8409,
5 1.9596, 0.0316, 0.8262, 2.2138, 0.0331, 0.8090,
6 2.5520, 0.0344, 0.7579, 2.8050, 0.0346, 0.7078,
7 3.0392, 0.0340, 0.6652, 3.2648, 0.0324, 0.6068,
8 3.3774, 0.0312, 0.5865/

DATA {CN(4,J),DH(4,J),ETA(4,J),J=1,15)/

1 0.1872, 0.0052, 0.5643, 0.2550, 0.0068, 0.6068,
2 0.4784, 0.0120, 0.7078, 0.6942, 0.0164, 0.8090,
3 0.9148, 0.0204, 0.8494, 1.1442, 0.0244, 0.8596,
4 1.3982, 0.0280, 0.8596, 1.5618, 0.0304, 0.8575,
5 1.9010, 0.0336, 0.8535, 1.9794, 0.0356, 0.8494,
6 2.2794, 0.0388, 0.8363, 2.5138, 0.0412, 0.8262,
7 2.8334, 0.0441, 0.8090, 3.1422, 0.0472, 0.7797,
8 3.3774, 0.0494, 0.7584/

DATA {CN(5,J),DH(5,J),ETA(5,J),J=1,15)/

1 0.1872, 0.0056, 0.5562, 0.3000, 0.0088, 0.6068,
2 0.5254, 0.0144, 0.7078, 0.7500, 0.0192, 0.8090,
3 0.9754, 0.0236, 0.8494, 1.2754, 0.0288, 0.8697,
4 1.4824, 0.0321, 0.8696, 1.7638, 0.0360, 0.8662,
5 2.0450, 0.0400, 0.8615, 2.3362, 0.0444, 0.8555,
6 2.5450, 0.0496, 0.8520, 2.8706, 0.0540, 0.8494,
7 3.0764, 0.0596, 0.8494, 3.1520, 0.0640, 0.8532,
8 3.1618, 0.0661, 0.8570/

DATA {CN(6,J),DH(6,J),ETA(6,J),J=1,15)/

1 0.1872, 0.0068, 0.5309, 0.3568, 0.0120, 0.6068,
2 0.5196, 0.0192, 0.7078, 0.8628, 0.0252, 0.8090,
3 1.0932, 0.0300, 0.8494, 1.2852, 0.0340, 0.8697,
4 1.5010, 0.0384, 0.8819, 1.6882, 0.0421, 0.8899,
5 1.9138, 0.0472, 0.8940, 2.1246, 0.0524, 0.8969,
6 2.2706, 0.0564, 0.8975, 2.4226, 0.0612, 0.8976,
7 2.4950, 0.0640, 0.8968, 2.5372, 0.0668, 0.8937,
8 2.5558, 0.0698, 0.8896/

DATA (CN(7,J),DH(7,J),ETA(7,J),J=1,15)/						60	
1	0.1872,	0.0080,	0.5062,	0.4314,	0.0164,	0.6068,	61
2	0.6844,	0.0236,	0.7078,	0.9568,	0.0308,	0.8090,	62
3	1.2010,	0.0372,	0.8494,	1.3834,	0.0415,	0.8697,	63
4	1.5108,	0.0448,	0.8797,	1.6186,	0.0476,	0.8899,	64
5	1.7450,	0.0510,	0.8954,	1.8618,	0.0544,	0.9000,	65
6	1.9558,	0.0576,	0.9010,	2.0000,	0.0600,	0.9000,	66
7	2.0450,	0.0624,	0.8980,	2.0824,	0.0660,	0.8925,	67
8	2.1010,	0.0700,	0.8793/				68
DATA (CN(8,J),DH(8,J),ETA(8,J),J=1,15)/						69	
1	0.1872,	0.0088,	0.5051,	0.4834,	0.0196,	0.6068,	70
2	0.7314,	0.0272,	0.7078,	0.8814,	0.0316,	0.7665,	71
3	1.0226,	0.0356,	0.8090,	1.1442,	0.0392,	0.8292,	72
4	1.2804,	0.0432,	0.8494,	1.3696,	0.0460,	0.8596,	73
5	1.4538,	0.0488,	0.8697,	1.5950,	0.0528,	0.8808,	74
6	1.5746,	0.0560,	0.8848,	1.7450,	0.0596,	0.8848,	75
7	1.3010,	0.0640,	0.8788,	1.8156,	0.0664,	0.8697,	76
8	1.8196,	0.0693,	0.8590/				77
DATA (CN(9,J),DH(9,J),ETA(9,J),J=1,15)/						78	
1	0.1872,	0.0093,	0.4909,	0.3372,	0.0159,	0.5380,	79
2	0.5344,	0.0232,	0.6068,	0.6754,	0.0284,	0.6573,	80
3	0.8068,	0.0330,	0.7078,	0.9196,	0.0368,	0.7463,	81
4	1.0128,	0.0400,	0.7776,	1.1254,	0.0442,	0.8090,	82
5	1.2196,	0.0480,	0.8191,	1.3138,	0.0524,	0.8302,	83
6	1.3696,	0.0556,	0.8347,	1.4068,	0.0580,	0.8363,	84
7	1.4450,	0.0612,	0.8322,	1.4638,	0.0640,	0.8241,	85
8	1.4576,	0.0668,	0.8090/				86
DATA (CN(10,J),DH(10,J),ETA(10,J),J=1,12)/						87	
1	0.1872,	0.0132,	0.4257,	0.2814,	0.0180,	0.4747,	88
2	0.3804,	0.0228,	0.5056,	0.4686,	0.0268,	0.5359,	89
3	0.5528,	0.0314,	0.5683,	0.6382,	0.0352,	0.5941,	90
4	0.5892,	0.0380,	0.6068,	0.7362,	0.0412,	0.6178,	91
5	0.7596,	0.0440,	0.6240,	0.8068,	0.0476,	0.6310,	92
6	0.8254,	0.0504,	0.6265,	0.8304,	0.0530,	0.6118/	93
END						94	

\$IBFTC IPTDAT DECK

BLOCK DATA						1	
COMMON / ITURB / TFF(15),CN(15,15),DH(15,15),ETA(15,15),N,NP(15)						2	
DATA N,NP/11,9*15,12,9,4*0/						3	
DATA TFF / 70.776, 82.236, 93.468, 103.464, 112.836,						4	
1	116.580,	120.000,	122.676,	125.124,	127.824,	130.536,4*0./	5
DATA (CN(1,J),DH(1,J),ETA(1,J),J=1,15)/						6	
1	0.3522,	0.0016,	0.7120,	0.5104,	0.0023,	0.7300,	7
2	0.7044,	0.0031,	0.7472,	0.9330,	0.0038,	0.7300,	8
3	1.1618,	0.0045,	0.7140,	1.3556,	0.0049,	0.7000,	9
4	1.5497,	0.0052,	0.6850,	1.6905,	0.0054,	0.6730,	10
5	1.9367,	0.0055,	0.6452,	2.1835,	0.0054,	0.6200,	11
6	2.3593,	0.0051,	0.6000,	2.5001,	0.0047,	0.5750,	12
7	2.5941,	0.0038,	0.5310,	2.8175,	0.0031,	0.5000,	13
8	3.1698,	0.0001,	0.3850/				14
DATA (CN(2,J),DH(2,J),ETA(2,J),J=1,15)/						15	
1	0.3522,	0.0023,	0.8000,	0.5278,	0.0035,	0.8100,	16
2	0.7575,	0.0047,	0.8200,	1.0208,	0.0061,	0.8300,	17
3	1.2322,	0.0070,	0.8300,	1.3818,	0.0076,	0.8290,	18
4	1.5201,	0.0084,	0.8100,	1.8130,	0.0089,	0.8000,	19
5	1.9723,	0.0092,	0.7850,	2.1305,	0.0094,	0.7600,	20
6	2.2715,	0.0095,	0.7450,	2.5089,	0.0093,	0.7000,	21
7	2.7471,	0.0089,	0.6800,	2.9227,	0.0083,	0.6450,	22
8	3.1598,	0.0068,	0.5900/				23

DATA (CN(3,J),DH(3,J),ETA(3,J),J=1,15)/							24
1	0.3522,	0.0027,	0.8000,	0.5654,	0.0045,	0.8300,	25
2	0.8279,	0.0063,	0.8600,	1.0296,	0.0076,	0.8630,	26
3	1.1975,	0.0087,	0.8670,	1.3730,	0.0098,	0.8700,	27
4	1.5497,	0.0107,	0.8720,	1.7609,	0.0118,	0.8720,	28
5	1.9367,	0.0126,	0.8700,	2.1479,	0.0134,	0.8670,	29
6	2.3245,	0.0139,	0.8600,	2.4827,	0.0142,	0.8500,	30
7	2.5583,	0.0146,	0.8300,	2.9227,	0.0147,	0.8000,	31
8	3.1598,	0.0145,	0.7600/				32
DATA (CN(4,J),DH(4,J),ETA(4,J),J=1,15)/							33
1	0.3522,	0.0029,	0.7995,	0.4052,	0.0034,	0.8000,	34
2	0.5514,	0.0054,	0.8400,	0.8452,	0.0069,	0.8600,	35
3	1.0567,	0.0084,	0.8680,	1.2322,	0.0097,	0.8730,	36
4	1.4434,	0.0111,	0.8800,	1.6722,	0.0124,	0.8830,	37
5	1.9540,	0.0140,	0.8835,	2.1131,	0.0146,	0.8830,	38
6	2.2715,	0.0153,	0.8800,	2.4915,	0.0161,	0.8740,	39
7	2.7471,	0.0168,	0.8600,	2.9931,	0.0172,	0.8350,	40
8	3.1598,	0.0173,	0.8200/				41
DATA (CN(5,J),DH(5,J),ETA(5,J),J=1,15)/							42
1	0.3522,	0.0031,	0.7750,	0.4844,	0.0043,	0.8000,	43
2	0.7044,	0.0062,	0.8480,	0.9330,	0.0081,	0.8600,	44
3	1.2322,	0.0105,	0.8750,	1.4967,	0.0124,	0.8900,	45
4	1.5548,	0.0136,	0.8912,	1.8834,	0.0152,	0.8940,	46
5	2.0071,	0.0159,	0.8955,	2.1652,	0.0169,	0.8970,	47
6	2.3274,	0.0178,	0.8961,	2.5531,	0.0189,	0.8900,	48
7	2.8175,	0.0199,	0.8790,	3.0461,	0.0207,	0.8671,	49
8	3.1598,	0.0210,	0.8600/				50
DATA (CN(6,J),DH(6,J),ETA(6,J),J=1,15)/							51
1	0.3522,	0.0034,	0.7600,	0.5896,	0.0057,	0.8000,	52
2	0.8008,	0.0076,	0.8450,	1.0567,	0.0100,	0.8600,	53
3	1.2322,	0.0114,	0.8730,	1.4619,	0.0134,	0.8900,	54
4	1.5722,	0.0150,	0.8950,	1.8660,	0.0165,	0.9000,	55
5	2.1171,	0.0184,	0.9005,	2.3245,	0.0199,	0.9010,	56
6	2.5357,	0.0214,	0.9004,	2.7375,	0.0228,	0.9000,	57
7	3.0019,	0.0251,	0.8900,	3.1167,	0.0267,	0.8800,	58
8	3.1598,	0.0280,	0.8735/				59
DATA (CN(7,J),DH(7,J),ETA(7,J),J=1,15)/							60
1	0.3522,	0.0038,	0.7310,	0.7392,	0.0078,	0.8000,	61
2	0.9689,	0.0101,	0.8300,	1.2109,	0.0124,	0.8600,	62
3	1.4089,	0.0142,	0.8750,	1.6056,	0.0159,	0.8900,	63
4	1.7609,	0.0173,	0.8930,	1.9367,	0.0190,	0.8975,	64
5	2.0948,	0.0207,	0.8999,	2.2000,	0.0220,	0.9000,	65
6	2.2889,	0.0233,	0.8980,	2.3949,	0.0250,	0.8937,	66
7	2.4471,	0.0261,	0.8900,	2.5001,	0.0276,	0.8799,	67
8	2.5175,	0.0290,	0.8710/				68
DATA (CN(8,J),DH(8,J),ETA(8,J),J=1,15)/							69
1	0.3522,	0.0042,	0.7100,	0.5808,	0.0069,	0.7450,	70
2	0.7575,	0.0090,	0.7680,	0.9330,	0.0109,	0.8000,	71
3	1.1801,	0.0135,	0.8380,	1.3915,	0.0156,	0.8600,	72
4	1.5571,	0.0177,	0.8712,	1.7609,	0.0199,	0.8780,	73
5	1.9660,	0.0213,	0.8800,	1.9897,	0.0230,	0.8775,	74
6	2.0501,	0.0241,	0.8760,	2.1131,	0.0251,	0.8722,	75
7	2.1552,	0.0263,	0.8660,	2.2009,	0.0276,	0.8600,	76
8	2.2048,	0.0283,	0.8480/				77
DATA (CN(9,J),DH(9,J),ETA(9,J),J=1,15)/							78
1	0.3522,	0.0047,	0.6780,	0.5278,	0.0070,	0.7000,	79
2	0.5340,	0.0084,	0.7125,	0.7922,	0.0104,	0.7350,	80
3	0.9589,	0.0124,	0.7690,	1.1183,	0.0141,	0.8000,	81
4	1.1801,	0.0148,	0.8060,	1.3209,	0.0166,	0.8225,	82
5	1.4519,	0.0184,	0.8395,	1.5497,	0.0196,	0.8450,	83
6	1.5722,	0.0214,	0.8470,	1.7609,	0.0232,	0.8445,	84
7	1.8130,	0.0245,	0.8330,	1.8315,	0.0255,	0.8235,	85
8	1.8401,	0.0267,	0.8080/				86

DATA (CN(10,J),DH(10,J),ETA(10,J),J=1,12)/						87	
1	0.3522,	0.0054,	0.6380,	0.4574,	0.0069,	0.6550,	88
2	0.5167,	0.0092,	0.6700,	0.7218,	0.0107,	0.6850,	89
3	0.8279,	0.0123,	0.7000,	0.9330,	0.0138,	0.7110,	90
4	1.0567,	0.0159,	0.7180,	1.1493,	0.0177,	0.7180,	91
5	1.2148,	0.0191,	0.7170,	1.2505,	0.0202,	0.7140,	92
6	1.2784,	0.0214,	0.7000,	1.2824,	0.0221,	0.5890/	93
DATA (CN(11,J),DH(11,J),ETA(11,J),J=1, 9)/						94	
1	0.3522,	0.0061,	0.6000,	0.4226,	0.0075,	0.6000,	95
2	0.5278,	0.0093,	0.6120,	0.6167,	0.0108,	0.6170,	96
3	0.7044,	0.0124,	0.6210,	0.7922,	0.0140,	0.5258,	97
4	0.8452,	0.0151,	0.6250,	0.8983,	0.0164,	0.6230,	98
5	0.9293,	0.0177,	0.6009/				99
END						100	

\$IBFTC LPTDAT DECK

BLOCK DATA						1	
COMMON / LTURB/TFF(15),CN(15,15),DH(15,15),ETA(15,15),N,NP(15)						2	
DATA N,NP/11,9*15,12,9,4*0/						3	
DATA TFF / 88.470, 102.795, 116.835, 129.330, 141.045,						4	
1	145.725,	150.000,	153.345,	156.405,	159.780,	163.170,4*0./	5
DATA (CN(1,J),DH(1,J),ETA(1,J),J=1,15)/						6	
1	0.3582,	0.0018,	0.7120,	0.5336,	0.0026,	0.7300,	7
2	0.7365,	0.0035,	0.7472,	0.9754,	0.0044,	0.7300,	8
3	1.2146,	0.0051,	0.7140,	1.4173,	0.0056,	0.7000,	9
4	1.5201,	0.0059,	0.6850,	1.7673,	0.0061,	0.6730,	10
5	2.0247,	0.0062,	0.6452,	2.2827,	0.0061,	0.6200,	11
6	2.4665,	0.0057,	0.6000,	2.6137,	0.0053,	0.5750,	12
7	2.3166,	0.0044,	0.5310,	2.9456,	0.0035,	0.5000,	13
8	3.3138,	0.0001,	0.3850/				14
DATA (CN(2,J),DH(2,J),ETA(2,J),J=1,15)/						15	
1	0.3582,	0.0026,	0.8000,	0.5518,	0.0039,	0.8100,	16
2	0.7919,	0.0054,	0.8200,	1.0672,	0.0069,	0.8300,	17
3	1.2882,	0.0080,	0.8300,	1.4446,	0.0087,	0.8290,	18
4	1.5937,	0.0096,	0.8100,	1.8954,	0.0101,	0.8000,	19
5	2.0519,	0.0104,	0.7850,	2.2273,	0.0107,	0.7600,	20
6	2.3747,	0.0108,	0.7450,	2.6229,	0.0106,	0.7000,	21
7	2.8720,	0.0101,	0.6800,	3.0555,	0.0094,	0.6450,	22
8	3.3138,	0.0077,	0.5900/				23
DATA (CN(3,J),DH(3,J),ETA(3,J),J=1,15)/						24	
1	0.3582,	0.0031,	0.8000,	0.5911,	0.0051,	0.8300,	25
2	0.8555,	0.0071,	0.8600,	1.0764,	0.0087,	0.8630,	26
3	1.2519,	0.0099,	0.8670,	1.4354,	0.0111,	0.8700,	27
4	1.5201,	0.0122,	0.8720,	1.8409,	0.0134,	0.8720,	28
5	2.0247,	0.0143,	0.8700,	2.2455,	0.0152,	0.8670,	29
6	2.4302,	0.0157,	0.8600,	2.5956,	0.0162,	0.8500,	30
7	2.7791,	0.0166,	0.8300,	3.0555,	0.0167,	0.8000,	31
8	3.3138,	0.0164,	0.7600/				32
DATA (CN(4,J),DH(4,J),ETA(4,J),J=1,15)/						33	
1	0.3582,	0.0033,	0.7995,	0.4237,	0.0038,	0.8000,	34
2	0.5810,	0.0061,	0.8400,	0.8837,	0.0078,	0.8600,	35
3	1.1047,	0.0096,	0.8680,	1.2882,	0.0110,	0.8730,	36
4	1.5090,	0.0126,	0.8800,	1.7482,	0.0141,	0.8830,	37
5	2.0429,	0.0159,	0.8835,	2.2091,	0.0166,	0.8830,	38
6	2.3747,	0.0174,	0.8800,	2.6047,	0.0183,	0.8740,	39
7	2.8720,	0.0191,	0.8600,	3.1291,	0.0195,	0.8350,	40
8	3.3138,	0.0197,	0.8200/				41

DATA (CN(5,J),DH(5,J),ETA(5,J),J=1,15)/						42	
1	0.3582,	0.0036,	0.7750,	0.5065,	0.0049,	0.8000,	43
2	0.7365,	0.0071,	0.8480,	0.9754,	0.0092,	0.8600,	44
3	1.2882,	0.0119,	0.8750,	1.5647,	0.0141,	0.8900,	45
4	1.7301,	0.0155,	0.8912,	1.9690,	0.0172,	0.8940,	46
5	2.0983,	0.0181,	0.8955,	2.2637,	0.0192,	0.8970,	47
6	2.4332,	0.0202,	0.8961,	2.6691,	0.0214,	0.8900,	48
7	2.9456,	0.0226,	0.8790,	3.1846,	0.0235,	0.8671,	49
8	3.3138,	0.0239,	0.8600/				50
DATA (CN(6,J),DH(6,J),ETA(6,J),J=1,15)/						51	
1	0.3682,	0.0038,	0.7600,	0.6164,	0.0064,	0.8000,	52
2	0.8372,	0.0087,	0.8450,	1.1047,	0.0113,	0.8600,	53
3	1.2382,	0.0130,	0.8730,	1.5283,	0.0152,	0.8900,	54
4	1.7482,	0.0171,	0.8950,	1.9509,	0.0187,	0.9000,	55
5	2.2133,	0.0209,	0.9005,	2.4302,	0.0226,	0.9010,	56
6	2.5510,	0.0244,	0.9004,	2.8619,	0.0259,	0.9000,	57
7	3.1384,	0.0286,	0.8900,	3.2584,	0.0303,	0.8800,	58
8	3.3138,	0.0319,	0.8735/				59
DATA (CN(7,J),DH(7,J),ETA(7,J),J=1,15)/						60	
1	0.3582,	0.0044,	0.7310,	0.7728,	0.0089,	0.8000,	61
2	1.0129,	0.0115,	0.8300,	1.2659,	0.0141,	0.8600,	62
3	1.4729,	0.0162,	0.8750,	1.6785,	0.0181,	0.8900,	63
4	1.8409,	0.0197,	0.8930,	2.0247,	0.0216,	0.8975,	64
5	2.1901,	0.0235,	0.8999,	2.3000,	0.0250,	0.9000,	65
6	2.3929,	0.0265,	0.8980,	2.5038,	0.0284,	0.8937,	66
7	2.5583,	0.0296,	0.8900,	2.6137,	0.0314,	0.8799,	67
8	2.5319,	0.0329,	0.8710/				68
DATA (CN(8,J),DH(8,J),ETA(8,J),J=1,15)/						69	
1	0.3682,	0.0048,	0.7100,	0.6072,	0.0078,	0.7450,	70
2	0.7919,	0.0102,	0.7680,	0.9754,	0.0124,	0.8000,	71
3	1.2337,	0.0153,	0.8380,	1.4548,	0.0177,	0.8600,	72
4	1.5383,	0.0201,	0.8712,	1.8409,	0.0226,	0.8780,	73
5	1.9509,	0.0242,	0.8800,	2.0801,	0.0261,	0.8775,	74
6	2.1537,	0.0274,	0.8760,	2.2091,	0.0285,	0.8722,	75
7	2.2537,	0.0299,	0.8660,	2.3009,	0.0314,	0.8600,	76
8	2.3051,	0.0321,	0.8480/				77
DATA (CN(9,J),DH(9,J),ETA(9,J),J=1,15)/						78	
1	0.3582,	0.0054,	0.6780,	0.5518,	0.0080,	0.7000,	79
2	0.5529,	0.0096,	0.7125,	0.8282,	0.0119,	0.7350,	80
3	1.0129,	0.0141,	0.7690,	1.1691,	0.0160,	0.8000,	81
4	1.2337,	0.0169,	0.8060,	1.3809,	0.0188,	0.8225,	82
5	1.5283,	0.0209,	0.8395,	1.6201,	0.0223,	0.8450,	83
6	1.7482,	0.0244,	0.8470,	1.8409,	0.0263,	0.8445,	84
7	1.8954,	0.0279,	0.8330,	1.9147,	0.0289,	0.8235,	85
8	1.9237,	0.0303,	0.8080/				86
DATA (CN(10,J),DH(10,J),ETA(10,J),J=1,12)/						87	
1	0.3582,	0.0061,	0.6380,	0.4782,	0.0078,	0.6550,	88
2	0.5447,	0.0104,	0.6700,	0.7546,	0.0122,	0.5850,	89
3	0.8555,	0.0139,	0.7000,	0.9754,	0.0157,	0.7110,	90
4	1.1047,	0.0181,	0.7180,	1.2015,	0.0201,	0.7180,	91
5	1.2701,	0.0217,	0.7170,	1.3073,	0.0230,	0.7140,	92
6	1.3365,	0.0244,	0.7000,	1.3407,	0.0251,	0.5890/	93
DATA (CN(11,J),DH(11,J),ETA(11,J),J=1, 9)/						94	
1	0.3582,	0.0069,	0.6000,	0.4418,	0.0086,	0.6000,	95
2	0.5518,	0.0106,	0.6120,	0.5447,	0.0123,	0.5170,	96
3	0.7365,	0.0141,	0.6210,	0.8282,	0.0159,	0.5258,	97
4	0.8837,	0.0172,	0.6250,	0.9391,	0.0186,	0.6230,	98
5	0.9715,	0.0201,	0.6009/				99
END						100	

\$IBFTC ABETTA DECK	
SUBROUTINE ETAAB (FAR,EM6,P6,ETA,ETAADS,ETAASV,P6DS,P6DSAV,AM6DS,A	1
1M6DSV,IDES,FAR7DS,FAR7SV)	2
DIMENSION FART(25),ETABRT(25),EM6T(7),DELM6(7),P6T(14),JELP6(14)	3
DIMENSION X(3),Y(3)	4
DATA FART/.0390,.0585,.0732,.0878,.0976,.1171,.1268,.1453,.1619,	5
1.1834,.1951,.2195,.2439,.2927,.3415,.4146,.4634,.5366,.6341,.7317,	6
2.8293,.9268,1.000,1.0634,1.7/	7
DATA ETABRT/.9400,.9887,1.0193,1.0306,1.0227,.9672,.9377,.9207,	8
1.9354,.9626,.9773,1.0193,1.0532,1.077,1.0781,1.077,1.0747,1.0668,	9
21.0578,1.0510,1.0374,1.0192,1.00,.9626,.9151/	10
DATA EM6T/1.00,1.071,1.190,1.309,1.428,1.547,1.666/	11
DATA DELM6/0,.013,.041,.073,.110,.147,.187/	12
DATA P6T/.220,.2267,.250,.300,.3333,.3767,.4167,.500,.5833,.6667,	13
1.75,.8333,.9167,1.0/	14
DATA JELP6/-.142,-.125,-.10,-.075,-.062,-.05,-.041,-.027,-.019,	15
1-.013,-.008,-.004,-.0021,0./	16
IF (IDES.NE.1) GO TO 5	17
DO 1 K=1,25	18
1 ETABRT(K)=ETABRT(K)*ETAADS/ETAASV	19
DO 2 K=1,25	20
2 FART(K)=FART(K)*FAR7DS/FAR7SV	21
DO 3 K=1,7	22
3 EM6T(K)=EM6T(K)*AM6DS/AM6DSV	23
DO 4 M=1,14	24
4 P6T(M)=P6T(M)*P6DS/P6DSAV	25
ETAASV=ETAADS	26
P6DSAV=P6DS	27
FAR7SV=FAR7DS	28
AM6DSV=AM6DS	29
RETURN	30
5 CONTINUE	31
N=0	32
IF (FAR.GT.0.067) GO TO 8	33
DO 6 J=1,25	34
6 IF (FAR.GE.FART(J)) N=J-1	35
IF (N.EQ.0) N=1	36
IF (N.GE.24) N=23	37
DO 7 I=1,3	38
NN=N-1+I	39
X(I)=FART(NN)	40
7 Y(I)=ETABRT(NN)	41
CALL PARABO (X,Y,FAR,ETA1)	42
GO TO 9	43
8 ETA1=-2.*FAR+.1948	44
9 M=0	45
DO 10 J=1,7	46
10 IF (EM6.GE.EM6T(J)) M=J-1	47
IF (M.EQ.0) M=1	48
IF (M.GE.6) M=5	49
DO 11 I=1,3	50
MM=M-1+I	51
X(I)=EM6T(MM)	52
11 Y(I)=DELM6(MM)	53
CALL PARABO (X,Y,EM6,COR1)	54
L=0	55
DO 12 J=1,14	56
12 IF (P6.GE.P6T(J)) L=J-1	57
IF (L.EQ.0) L=1	58
IF (L.GE.13) L=12	59
DO 13 I=1,3	60
LL=L-1+I	61
X(I)=P6T(LL)	62

13	Y(I)=JELP6(LL)	63
	CALL PARABO (X,Y,P6,COR2)	64
	ETA=ETA1*(1.-COR1)*(1.+COR2)	65
	RETURN	66
	END	67

\$IBFTC INPUT DECK										
SUBROUTINE INPUT(LIUNIT,LOUNIT,IO,D,IT)										1
DIMENSION D(1), IT(1)										2
C										3
C										4
C										5
C	CONSTRUCTION OF THE TABLE									6
C										7
C	BIT STANDARD RETURN RETURN RETURN RETURN RETURN									8
C	NUMBER RETURN 1 2 3 4 5									9
C	0-2 \$ A-Z 0-9()									10
C	3-4 DETPC RI OTH NLSF									11
C	5-6 DER TINL OTH PS CF									12
C	7-8 0-9.+\$)OTH , 'A-Z= -(*/									13
C	9 ,.)A-Z OTH									14
C	9-10 ., A-Z) 0-9+ ' =\$									15
C	10 OTH \$A-Z*)= /									16
C	10-11 0-9.,-OTH +' (\$A-Z*)= /									17
C	12 0-9 A-Z OTH									18
C	12-13 A-Z 0-9 OTH									19
C	13-14 +- OTH 0-9									20
C	15-16 +. DE- OTH									21
C	17-18 r F OTH									22
C	19-20 0 E OTH									23
C	21 OTH .									24
C	21-22 A-Z OTH .									25
C	23 ' OTH									26
C	24 \$ OTH									27
C	25 = OTH									28
C	26-27 (A-Z OTH									29
C	28-29) . OTH									30
C	30 (OTH									31
C	31) OTH									32
C	ERROR CODE DESIGNATIONS									33
C	ROUTINE TYPE									34
C	10 IXQTI 100 ILLEGAL CHARACTER									35
C	20 ITABLI 200 NAME TOO LONG									36
C	30 INMBRI 300 TABLE FULL OR BAD									37
C	40 INAMEI 400 SCALING ERROR									38
C	50 IARITI 500 NAME NOT IN TABLE									39
C	60 INAMEN 600 \$DATA() INCOMPLETE									40
C	70 INPJT 700 FORMULA ILL FORMED									41
C	800 FUNCTION UNDEFINED									42
C	MCNVRT = TYPE OF LEFT HAND VARIABLE									43
C	KCNVRT = TYPE OF CURRENT VARIABLE									44
C	ITYPE MEANING									45
C	1 REAL									46
C	2 INTEGER									47
C	3 DOUBLE PRECISION									48
C	4 TYPELESS OR NO CONVERSION									49
C	5 SUBROUTINE									50
C	6 FUNCTION									51
C	FORMAT OF TABLE									
C	1ST WORD 012 3456									
										7-31

C	TYPE	NUMBER OF WORDS	ADDRESS	52			
C	NEXT 1 TO 15 WORDS	THE NAME, 4 CHARACTERS TO THE WORD		53			
	DIMENSION IFT(31), IPTAB(21), ITAB(65)			54			
	DIMENSION ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)			55			
	• ,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)			56			
C	COMMON			57			
	./ICOMVI/ VALUE	,ICOMP	,IFNTYP	,IMAGE1	,IRADIX	,ISUB	58
	• ,KCH	,KCNVRT	,KCOUNT	,KDIF	,KFLD1	,KFLD2	59
	• ,LCOMP	,LCNVRT	,LEVEL	,LFRT	,LOOK		60
	• ,MCNVRT	,MDIF	,MODALL	,MSTOR			61
	• ,NAME	,NERROR	,NONEW	,NOTARG			62
	• ,SMCHR	,TEST	,ERMARX				63
	./ICNSTI/ BLANK	,BLANKS		,EDS			64
	• ,ICOMMA	,IDOLAR	,IFT	,IPTAB	,ITAB		65
	• ,KAMIO	,KBPC	,KBPW	,KCPCD	,KERTYP		66
	• ,KZERO	,NOPRNT	,TAB1				67
	./IPARAM/ ABORT	,KIUNIT	,KOUNIT	,LIMALF	,LOCK	,LOCX	68
	• ,NOLIST	,NSTDIR	,TRACE				69
	./ISTAKI/ STACK	,ISTDIM	,KSTACK	,LEVLIM			70
C	INTEGER BLANK	,BLANKS	,EDS	,IDOLAR	,TAB1	,TRACE	71
	•,CONTYP ,STORED						72
	DOUBLE PRECISION STACK, VALUE						73
	LOGICAL ABORT						74
C		• STOP FOR ERROR					75
	LOGICAL DOLLAR						76
C		• TRUE IF A \$T,\$P, OR \$C IS BEING PROCESSED					77
	LOGICAL ERMARX						78
C		• .TRUE. IF ANY ERROR HAS OCCURED					79
	LOGICAL LIMALF						80
C		• TERMINATE ALF CONSTANTS AT 15 WORDS.					81
	LOGICAL LOCK						82
C		• DO NOT STORE ANY INPUT DATA					83
	LOGICAL NOLIST						84
C		• DO NOT LIST INPUT CARDS					85
	LOGICAL NONEW						86
C		• DO NOT READ A NEW CARD. SET IN INPUT					87
	LOGICAL MODALL						88
C		• .TRUE. RETURN ALL CHARACTERS					89
C		• .FALSE. DELETE BLANKS AND SKIP FOR E					90
	LOGICAL SHORT						91
C		• DO NOT STORE ALPHABETIC AND RADIX CO					92
C		DIRECTLY INTO D ARRAY					93
	LOGICAL SMCHR						94
C		• SAME CHARACTER IS TO BE USED OVER BY					95
	LOGICAL TEST						96
C		• STORE COMMA AT END OF CARD IF NEEDED					97
C	END	*****					98
	EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE),(NAME,ANAME)						99
	EQUIVALENCE (ICOMNI,ISUB),(IMAGE,IMAGE1),(IPARAM,ABORT)						100
	LOGICAL END2						101
C		• .TRUE. IF A \$END CARD HAS ALREADY BEEN FOUND					102
	KIUNIT=LIUNIT						103
	KOUNIT=LOUNIT						104
C	CALL DEBUGX						105
	LIMALF = .TRUE.						106
	DOLLAR = .FALSE.						107
	NOLIST = .FALSE.						108
	END2 = .FALSE.						109
	MODALL = .FALSE.						110
	ERMARX = .FALSE.						111
							112
							113

	LOCK = .FALSE.	114
	KERTYP = 0	115
	IFNTYP = -1	116
	LOCX=1	117
	L=LEVLIM+3	118
	DO 16 I=1,L	119
16	KSTACK(I)=0	120
	LEVEL=0	121
	LFRT=0	122
	NOTARG = ISTDIM	123
	LCNVRT = 3	124
C	* DOUBLE PRECISION, STANDARD	125
	MCNVRT = 1	126
	KDIF = 1	127
	MDIF = 1	128
1	NONEW = .FALSE.	129
	SMCHR = .TRUE.	130
	CALL ICHAR2(\$9470,24)	131
C	* ERROR IF \$ IS NOT 1ST CHARACTER READ	132
	LDOLAR = KCOUNT	133
C	* CARD COLUMN OF LAST DOLLAR SIGN	134
	NONEW=.TRUE.	135
C	* \$DATA MUST BE ON A SINGLE CARD	136
C	D E OTH	137
	CALL ICHAR1(\$3,\$9470,19)	138
C	I A-Z OTH	139
4	CALL ICHAR1(\$4,\$9470,26)	140
	CALL ISUBI	141
	CALL ICHAR2(\$9470,31)	142
C	* ERROR IF NO)	143
	NONEW=.FALSE.	144
	IT(1)=ISUB	145
C	CALL DEBUG 2 (5HSETNO,IT(1))	146
C	CALL DEBUG 2(6HKERSIN,KERSIN)	147
	IF (ID.NE. IT(1)) GO TO 99	148
C	*RETURN BECAUSE WRONG DATA SET	149
	IF (IT(2).LT. 0)GO TO 9370	150
C	* TABLE FULL OR BAD	151
	GO TO 19	152
3	IF (END2) GO TO 99	153
	END2 = .TRUE.	154
	NONEW = .FALSE.	155
5	CALL ICHAR2(\$1,9)	156
C	* PASS LETTERS AND ,.)	157
	GO TO 5	158
9470	KERTYP = 470	159
	GO TO 999	160
99	CONTINJE	161
C	CALL DEBUG 2(6HSTATMT,99)	162
	KCH = IDOLAR	163
	KCOUNT = LDOLAR	164
C	*NEXT CALL BEGINS PROCESSING AT THIS CARD COLUMN	165
	IF(ERMARK) IT(1) = -IABS(IT(1))	166
C	* WARN PROGRAMER OF POSSIBLE ERRO	167
	IF(ERMARK.AND.ABORT) GO TO789	168
	RETURN	169
789	WRITE(6,787)	170
787	FORMAT(44H ERROR HAS OCCURED AND ABORT IS .TRUE., STOP)	171
	STOP	172
C	BEGIN TO INPUT THE DATA	173
C	STATEMENTS 19 AND 20 ARE THE SWITCHHOUSE	174
C	CONTRL COMES HERE FOR NEW DIRECTION.	175
C	LFRT = 0 INDICATES THAT THE PREVIOUS TASK WAS COMPLETED.	176

19	CONTINUE	177
C	CALL DEBUG2(6HSTATMT,19)	178
	TEST=.FALSE.	179
	GO TO 21	180
120	DOLLAR = .FALSE.	181
20	CONTINUE	182
C	CALL DEBUG 2(6HSTATMT,20)	183
	TEST=.TRUE.	184
C	ENTRY ICHAR4 LOADS LCOMP WITH TAB NO 10 AND RETURNS ON TAB NO 7	185
C	OTH , 'A-Z= -(*/	186
21	CALL ICHAR4(\$31, \$32, 7,10)	187
C	NOW WE TEST LCOMP ON EACH OF THE 3 POSSIBLE RETURNS. THIS IS A 12 WAY	188
C	0-9. + \$)	189
	GO TO (203,510, 64,430),LCOMP	190
C	, ' A-Z =	191
31	GO TO (460,202,530,450),LCOMP	192
C	- (* /	193
32	GO TO(520,420,440,441),LCOMP	194
64	IF(LFRT.NE.0) GO TO 9770	195
	LDOLAR = KCOUNT	196
C	* CARD COLUMN OF LAST DOLLAR SIGN	197
	DOLLAR = .TRUE.	198
C	CALL DEBUG 2 (6HSTATMT,64)	199
	TEST=.FALSE.	200
	CALL ICHAR4(\$9170,\$9170,3,5)	201
C	DE T P C	202
	GO TO(99,100,150,630),LCOMP	203
100	CALL ITABLI(IT)	204
	IF (KERTYP) 999,120,999	205
C		206
C		207
C	\$PARAMETER	208
C	(A-Z OTH	209
150	CALL ICHAR1(\$150,\$9170,26)	210
151	CALL ICHAR2(\$152,0)	211
C	* GET FIRST CHAR OF NAME ..DON'T BRAN	212
152	CALL INAMEN	213
	CALL ILOOKI(\$153,IPTAB)	214
9570	KERTYP = 570	215
	GO TO999	216
153	CALL ICHAR2(\$9170,29)	217
C	* ERROR IF NO =	218
	CALL ICHAR2(\$9170,10)	219
C	* GET CHARACTER,SEPERATE PART OF	220
	CALL INMBRI	221
	KCNVRT=IFLD(0,3,IPTAB(LOOK))	222
	CALL ICNVRTI(3,KCNVRT)	223
C	* CONVERT FROM DP TO TYPE OF VARIABLE	224
	IPT=IFLD(7,25,IPTAB(LOOK))	225
	IPARAM(IPT) = KVALUE	225
C	, OTH	227
	CALL ICHAR1(\$151,\$9170,28)	228
C	IF(TRACE.GT.0) CALL DEBUGX	229
C	* IF TRACE TURNED ON PUT ON HEAD	230
	GO TO 120	231
C	* GO TO SWITCH HOUSE TEST FOR INSERTIO	232
C		233
C		234
C	CONSTANTS = LOGICAL,NUMERIC,ALPHAMERIC,RADIX.	235
201	ASSIGN 220 TO CNTYP	236
C	* RADIX	237
	IFNTYP = -1	238

	GO TO 210	239
202	ASSIGN 230 TO CONTYP	240
C		241
	GO TO 210	242
203	ASSIGN 250 TO CONTYP	243
C		244
	NUMERIC, LOGICAL	245
C		246
C		247
C	ALL CONSTANTS 210	248
210	LOP = 0	249
	SHORT = .TRUE.	250
	MSTOR = 0	251
C	CALL DEBUG2(6HST CON, 210)	252
	IF (LEVEL.EQ.0) LEVEL=3	253
	IF (KSTACK(LEVEL).EQ.1) GO TO 9770	254
C		255
	GO TO CONTYP, (220, 230, 250)	256
C		257
C		258
C	RADIX CONSTANTS 220	259
C	(A-Z OTH	260
220	CALL ICHAR2(\$9170, 30)	261
	CALL ISUBI	262
	NAME (2) = 0	263
221	CALL ICHAR2(\$2215, 29)	264
C		265
	GO TO 9170	266
2215	CALL ICHAR2(\$9170, 12)	267
C		268
222	IRADIX = ISUB	269
	SMCHR = .TRUE.	270
223	CALL ISUBI	271
	MSTOR = MSTOR + 1	272
	NAME(MSTOR) = ISUB	273
	IF (MSTOR.GE.15) GO TO 225	274
C		275
) , OTH	276
224	CALL ICHAR1(\$223, \$9170, 28)	277
	IRADIX = 10	278
	GO TO 240	279
225	ASSIGN 224 TO NEXT	280
	GO TO 241	281
C		282
C		283
C	HOLLERITH CONSTANTS 230	284
230	MODALL = .TRUE.	285
	TEST = .FALSE.	286
	NAME(2) = BLANKS	287
231	CALL INAMEN	288
	IF (.NOT. MODALL) GO TO 240	289
C		290
	IF (LIMLF) GO TO 265	291
C		292
	LJVG CONSTANT GOES TO 234	293
234	ASSIGN 231 TO NEXT	294
	GO TO 241	295
C		296
C		297
C	STORE ALF + RADIX 240	298
240	ASSIGN 260 TO NEXT	299
	IF (MSTOR-2) 242, 2405, 241	300
2405	IF (LFRT.EQ.1) GO TO 242	301
	CALL ICHAR2(\$2406, 29)	
	SMCHR = .TRUE.	
	GO TO 242	

2406	SMCHR = .TRUE.	302
241	SHORT = .FALSE.	303
	IF (LFRT.NE.0) GO TO 265	304
C	CALL DEBUG2(6HST 241,0)	305
242	KVALUE = NAME(1)	306
	RVALUE(2) = ANAME(2)	307
	IF (NOTARG.GT.LEVEL) LCNVRT = 4	308
C	CALL DEBUG2(6HST 242,0)	309
C	CALL DEBUG2(6HSHORT ,SHORT)	310
C	CALL DEBUG2(6HLFRT ,LFRT)	311
243	IF (SHORT) GO TO 255	312
	DO 245 I= 1,MSTOR	313
	IF (LOCK) GO TO 245	314
	D(LOCK) = ANAME(I)	315
245	LOCK = LOCK + 1	316
246	GO TO NEXT, (224,231,260)	317
C	SEE NEXT AND MEANING	318
C	250 * ALF OR) AND RADIX	319
C	224 MORE THAN 15 ELEMENTS IN RADIX FIELD	320
C	231 MORE THAN 15 ELEMENTS IN ALF FIELD	321
C		322
C		323
C	NUMERIC + LOGICAL 250	324
250	TEST = .TRUE.	325
	CALL INMBRI	326
255	CALL IARITI	327
	LFRT = 1	328
	IF (KERTYP) 999,20,999	329
C		330
C		331
C	RESET STACK BECAUSE IT WAS NOT USED	332
260	LEVEL = 0	333
	TEST = .TRUE.	334
	CALL ICHAR2(\$19,29)	335
C	* SKIP COMMA	336
	KERTYP = 171	337
	GO TO 999	338
265	KERTYP = 270	339
	GO TO 999	340
C		341
C	TEST EMPTY PARENTHESES	342
400	IF ((KSTACK(LEVEL-1)+KSTACK(LEVEL-2)).NE.0) GO TO 997	343
C	CALL DEBUG 2(6HSTA (),400)	344
C	CALL STACKP	345
C	PRINT STACK	346
401	LEVEL=LEVEL-3	347
C	EMPTY FUNCTION ARGUMENT IS NOT A CURRENT LEFT SIDE	348
	IF (KSTACK(LEVEL).LT.6) GO TO 404	349
405	VALUE=0.	350
	GO TO 403	351
404	RVALUE(1) = D (LOCK)	352
	RVALUE(2) = D(LOCK+1)	353
	CALL ICNVTI(MCNVRT,3)	354
403	CALL IARITI	355
	GO TO 20	356
C	((((((((((((((((357
420	IF (LEVEL.EQ.0) LEVEL=3	358
C	CALL DEBUG 2(6HSTAT (,420)	359
422	LOP=0	360
	LFRT = 1	361
	LEVEL=LEVEL+3	362
C	CALL STACKP	363

C		PRINT STACK	364
	IF (KSTACK(LEVEL-3)-1) 20, 997, 20		365
	C))))))))))))))		366
430	IF (LEVEL.LT.6) GO TO 997		367
C	CALL DEBUG 2(6HSTAT),430)		368
C	CALL STACKP		369
C		PRINT STACK	370
	LFRT = 1		371
431	IF (KSTACK(LEVEL)-1) 400, 432, 997		372
432	DO 433 I=1,3		373
	VALUE=STACK(LEVEL)		374
	KSTACK(LEVEL)=0		375
	LEVEL=LEVEL-1		376
433	CALL IARITI		377
	IF (KERTYP) 999, 20, 999		378
C*****	(KOP=2)		379
440	KOP=2		380
C	CALL DEBUG 2(6HSTAT *,440)		381
	GO TO 445		382
C////////	(KOP=3)		383
441	KOP = 3		384
C			385
C	CALL DEBUG 2(6HSTAT /,441)		385
445	IF (LJP.NE.0) GO TO 997		387
C	CALL DEBUG 2(6HSTATMT,445)		388
C	CALL STACKP		389
C		PRINT STACK	390
	LOP = 1		391
	LFRT = 1		392
	IF (LEVEL.EQ.0) LEVEL=3		393
	IF (KSTACK(LEVEL).NE.1) GO TO 997		394
444	VALUE=STACK(LEVEL)		395
	KSTACK(LEVEL)=0		396
	LEVEL=LEVEL-1		397
	CALL IARITI		398
	IF (KERTYP.NE.0) GO TO 999		399
446	KSTACK(LEVEL)=KOP		400
	LEVEL=LEVEL+1		401
C	CALL STACKP		402
C		PRINT STACK	403
	GO TO 19		404
C=====			405
450	IF (LEVEL.NE.0) GO TO 997		406
C	CALL DEBUG 2(6HSTAT =,450)		407
451	LOP=0		408
	LOCX = LOOK		409
	MCNVRT = KCNVRT		410
	MDIF = KDIF		411
	LCNVRT = 3		412
C		* DOUBLE PRECISION, STANDARD	413
	LEVEL=3		414
	LFRT=0		415
	KSTACK(3)=0		416
C	CALL DEBUG2(6HLOCX ,LOCX)		417
C	CALL DEBUG2(6HMCNVRT,MCNVRT)		418
	GO TO 19		419
C,,,,,			420
460	CONTINJE		421
C	CALL DEBUG 2(6HSTAT ,,460)		422
C	CALL STACKP		423
C		PRINT STACK	424
	IF (LEVEL - 3) 461,463, 600		425
461	IF (LEVEL.NE.0) GO TO 997		426

462	LEVEL=3	427
463	IF(LFRT.EQ.0)GO TO 480	428
C	* INCREMENT LOCX WITHOUT STORING	429
C	LFRT WILL BE ZERO AFTER \$ EXPRESSION OR FOR CONSECUTIVE COMMAS	430
470	IF (KSTACK(3).NE.1) GO TO 997	431
471	DO 475 I = 1,2	432
	VALUE=STACK(LEVEL)	433
	KSTACK(LEVEL)=0	434
	LEVEL=LEVEL-1	435
475	CALL IARITI	436
	IF (KERTYP.NE.0) GO TO 999	437
476	KSTACK(1)=0	438
	CALL ICNVTI (LCNVRT,MCNVRT)	439
474	IF(LOCK) GO TO 480	440
	D(LOCK) = RVALUE(1)	441
	IF (MCNVRT .EQ. 3) D(LOCK+1) = RVALUE(2)	442
480	LOP=0	443
C	CALL STACKP	444
C	PRINT STACK	445
	LFRT=3	446
	LOCX = LOCX + MDIF	447
	LEVEL=0	448
	LCNVRT = 3	449
C	* DOUBLE PRECISION, STANDARD	450
	GO TO 19	451
C+++++++(KOP=4)		452
510	KOP=4	453
C	CALL DEBUG 2(6HSTAT +,510)	454
	GO TO 521	455
C------(KOP=5)		456
520	KOP=5	457
C	CALL DEBUG 2(6HSTAT -,520)	458
521	IF (LOP.NE.0) GO TO 997	459
522	LOP=1	460
	LFRT = 1	461
	IF (LEVEL.EQ.0) LEVEL=3	462
524	IF (KSTACK(LEVEL)-1) 525, 526, 997	463
525	STACK(LEVEL)=0. DO	464
C	CALL DEBUG 2(6HSTATMT,525)	465
526	DO 528 I=1,2	466
C	CALL DEBUG 2(6HSTATMT,526)	467
	VALUE=STACK(LEVEL)	468
	KSTACK(LEVEL)=0	469
	LEVEL=LEVEL-1	470
528	CALL IARITI	471
	IF (KERTYP.NE.0) GO TO 999	472
527	KSTACK(LEVEL)=KOP	473
	LEVEL=LEVEL+2	474
	GO TO 19	475
CABCDEFGHIJKLMNOPQRSTUVWXYZABCDEFGHIJKLMNOPQRS		476
530	LOP=0	477
C	CALL DEBUG 2(6HSTAT A,530)	478
	IF ((LEVEL.NE.0).AND.(KSTACK(LEVEL).EQ.1)) GO TO 997	479
535	TEST = .TRUE.	480
	CALL INAMEI(\$999,D,IT)	481
	IF (I=NTYP) 531,621,640	482
531	LFRT = 1	483
	IF(LEVEL.NE.0)GO TO 540	484
532	STACK(3)=VALUE	485
C	CALL DEBUG 2(6HSTATMT,532)	486
	KSTACK(3)=1	487
C	CALL STACKP	488

C		PRINT STACK	489
	GO TO 20		490
540	CONTINUE		491
C	CALL DEBUG 2(6HSTATMT,540)		492
	CALL IARITI		493
	IF(KERTYP) 999,20,999		494
C			495
C	PROCESS SUBROUTINES AND FUNCTIONS		496
C			497
C	COMMA SEPARATING FUNCTION ARGUMENTS		498
600	IF (KSTACK(LEVEL-3).LT.6) GO TO 997		499
C	CALL DEBUG 2(6HSTAT F,600)		500
601	IF (KSTACK(LEVEL)-1) 602, 603, 997		501
602	STACK(LEVEL)=0.		502
603	DO 610 I=1,2		503
	VALUE = STACK(LEVEL)		504
	KSTACK(LEVEL)=0		505
	LEVEL=LEVEL-1		506
610	CALL IARITI		507
	IF (KERTYP.NE.0) GO TO 999		508
611	KSTACK(LEVEL)=KSTACK(LEVEL-1)+1		509
	LEVEL=LEVEL+3		510
C	CALL STACKP		511
C		PRINT STACK	512
	GO TO 19		513
C	\$CALL		514
630	DO 631 I = 1,4		515
C	SKIP 'ALL' IN \$CALL		516
C	OTH OPERATORS		517
	CALL ICHAR2(\$9170,12)		518
631	CONTINUE		519
	DOLLAR = .FALSE.		520
	TEST = .TRUE.		521
	CALL INAMEI (\$999,D,IT)		522
	IF (IFNTYP .NE. 1) GO TO 640		523
	KVALUE = -KVALUE		524
C		* INDICATE THAT NO RESULT IS TO BE STORED FOR SJB	525
C	FUNCTION NAME		526
621	CONTINUE		527
C		* FUNCTIONS	528
C	CALL DEBUG2(6HST FNC,621)		529
C	CALL DEBUG2(6HIFNTYP,IFNTYP)		530
C	CALL DEBUG2(6HKVALUE,KVALUE)		531
	IFNTYP = -1		532
	IF (LEVEL.EQ.0) LEVEL = 3		533
	IF(NOTARG .GT. LEVEL) NOTARG = LEVEL		534
	LFRT = 1		535
	STACK(LEVEL)=KVALUE		536
	KSTACK(LEVEL)=6		537
	LEVEL = LEVEL + 3		538
C	CALL STACKP		539
C		PRINT STACK	540
C	(OTH		541
	CALL ICHAR2(\$622,30)		542
	GO TO 19		543
C		* IF THERE ARE ARGUMENTS	544
622	SMCHR = .TRUE.		545
	IF (KVALUE .LT. 0) LEVEL = LEVEL -3		546
	GO TO 405		547
640	IF (IFNTYP - 2) 9770,201,700		548
C	LOCK THE SUBROUTINE THAT STORES LOCK THE SUBSCRIPT OF D ARR		549
700	IF (IFNTYP .NE. 3) GO TO 9770		550
	IFNTYP = -1		551

	IF(LEVEL.NE.0) GO TO 9770	552
	LFRT = 0	553
	CALL ICHAR2(\$9170,30)	554
C	' ERROR IF NO (AFTER LOCX	555
	ASSIGN 704 TO NEXT	556
C	A-Z OTH	557
702	CALL ICHAR2(\$9170,9)	558
	CALL INAMEI(\$999,D,IT)	559
	IF (IFNTYP .GT. -1) GO TO 9770	560
	GO TO NEXT,(704,708)	561
704	LOOKX = LOOK	562
C	OTH ,	563
	CALL ICHAR2(\$706,29)	564
C	' ERROR IF NO COMMA	565
	GO TO 9170	566
706	ASSIGN 708 TO NEXT	567
	GO TO 702	568
708	KVALUE = LOCX - LOOKX + 1	569
	CALL ICNVTI(2,KCNVRT)	570
C	' CONVERT FROM INT TO TYPE OF 2ND ARG	571
	IF(LOCK) GO TO 710	572
	D(LOCK) = RVALUE(1)	573
710	CALL ICHAR2(\$9170,31)	574
C	' ERROR IF NO)	575
	TEST = .TRUE.	576
C	OTH ,	577
	CALL ICHAR2(\$19,29)	578
C	' SKIP THE COMMA THAT MUST FOLLOW	579
9170	KERTYP=170	580
	GO TO 999	581
9370	KERTYP=370	582
	GO TO 999	583
997	CONTINUE	584
9770	KERTYP = 770	585
	GO TO 999	586
999	CALL IERRORI	587
	LFRT = 0	588
	LOP = 0	589
	NAME (1)=0	590
	GO TO 19	591
C	' NOW GO TO SWITCH HOUSE .. GOOD	592
	END	593

\$IBFTC	BLOCK	DECK	
	BLOCK DATA		1
C	ICOMVI	BLOCK DATA PROGRAM	2
	DIMENSION	IFT(31), IPTAB(21), ITAB(65)	3
	DIMENSION	ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)	4
	.	,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)	5
C	COMMON		6
	./ICOMVI/	VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB	7
	.	,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2	8
	.	,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK	9
	.	,MCNVRT ,MDIF ,MODALL ,MSTOR	10
	.	,NAME ,NERROR ,NONEW ,NOTARG	11
	.	,SMCHR ,TEST ,ERMARK	12
	./ICNSTI/	BLANK ,BLANKS ,DOLLAR ,EOS	13
			14

```

.      ,ICOMMA      ,IDOLAR      ,IFT      ,IPTAB      ,ITAB      15
.      ,KAM10      ,KBPC      ,KBPW      ,KCPCD      ,KERTYP      16
.      ,KZERO      ,NOPRNT      ,TAB1      17
./IPARAM/ ABORT      ,KIUNIT      ,KOUNIT      ,LIMAF      ,LOCK      ,LOCKX      18
.      ,NOLIST      ,NSTDIR      ,TRACE      19
./ISTA<I/ STACK      ,ISTDIM      ,KSTACK      ,LEVLIM      20
C
INTEGER BLANK      ,BLANKS      ,EOS      ,IDOLAR      ,TAB1      ,TRACE      21
DOUBLE PRECISION STACK, VALUE      22
LOGICAL ABORT,DOLLAR,ERMARK,LIMAF,LOCK,NOLIST,NEW,MODALL,      23
.NSTDIR,SMCHR,TEST      24
DATA VERROR,TRACE/0,0/      25
DATA ITAB/ 65* 20572347824/      26
DATA ITAB( 49)/-11890146992/      27
. , ITAB( 44)/ 11923699376/, ITAB( 61)/ 29355228304/      28
. , ITAB( 29)/ 29053238816/, ITAB( 12)/ -3484760752/      29
. , ITAB( 28)/-11821997744/, ITAB( 17)/-11903778480/      30
. , ITAB( 33)/-12155951008/, ITAB( 45)/-12192136880/      31
. , ITAB( 50)/-12208914096/, ITAB( 13)/-12041137840/      32
. , ITAB( 60)/-11957255792/, ITAB( 11)/-11890146992/      33
. , ITAB( 1)/ 29063724720/, ITAB( 2)/ 29063724720/      34
. , ITAB( 3)/ 29063724720/, ITAB( 4)/ 29063724720/      35
. , ITAB( 5)/ 29063724720/, ITAB( 6)/ 29063724720/      36
. , ITAB( 7)/ 29063724720/, ITAB( 8)/ 29063724720/      37
. , ITAB( 9)/ 29063724720/, ITAB( 10)/ 29063724720/      38
. , ITAB( 18)/ 20572347824/, ITAB( 20)/ 18961735088/      39
. , ITAB( 21)/ 17350532528/, ITAB( 22)/ 17350565296/      40
. , ITAB( 23)/ 23256571312/, ITAB( 26)/ 20035476912/      41
. , ITAB( 36)/ 22182960560/, ITAB( 38)/ 22182960560/      42
. , ITAB( 40)/ 18424854176/, ITAB( 42)/ 19498606000/      43
. , ITAB( 51)/ 22719831472/, ITAB( 52)/ 17887731120/      44
DATA BLANK,BLANKS,EOS,ICOMMA,IDOLAR,KAM10,KBPC,KBPW,KCH,KCPCD,      45
. KZERO,NOPRNT,TAB1,IMAGE1(2),IMAGE1(3)/      46
.      48,-17997958192,      10,      59,      43,      47
.      7,      6,      36,      43,      80,      48
.      0, 11555507248,      64, 11555507248, 11555507248/      49
DATA ABORT,IMAGE1(1),IMAGE1(81),IRADIX,ISTDIM,      50
. KCOUNT,KDIF,KSTACK,LEVLIM,NSTDIR,STACK/      51
. .FALSE.,1HE,1H ,10,27,0,1,27*0,24,.TRUE.,27*0.D0/      52
DATA IFT/      53
.      0,      31,-17716740112, 21051935792,-17716740128,      54
. 23481748528,-17716740144, -3864988720,-17716740160,-19756682288,      55
. -17715740176,-20009401392,-17716740192, 19113857200,-17716740208,      56
. -8210767088,-17716740224, -8428915760,-17716740240, -8433110054,      57
. -17715740256, 18568907824,-17716740272, 21907455024, -9126805536,      58
. -9954237936, -9126805552, -3863968816, -9126805696, -3863883248,      59
.      0/      60
DATA IPTAB/      61
.      0,      21, -536870928, 18565733616, -536870976,      62
. -3650164950, -536870992, -3863882800, -536871024, -6015524019,      63
. -535871040, -6221022825, 17716740128, -2580698739, 17716740144,      64
. -2793302547, 17716740192, -3863968816, 17716740240,-21093496175,      65
.      0/      66
END      67
      68

```

\$IBFTC IARITI DECK

C ARITHMETIC OPERATIONS FOR INPUT R. U. A. S. 2
SUBROUTINE IARITI

1
2

C	CALLED FROM INPUT	3
	DIMENSION IFT(31), IPTAB(21), ITAB(65)	4
	DIMENSION ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)	5
	• ,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)	6
C	COMMON	7
	./ICOMNI/ VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB	8
	• ,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2	9
	• ,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK	10
	• ,MCNVRT ,MDIF ,MODALL ,MSTOR	11
	• ,NAME ,NERRDR ,NOVEW ,NOTARG	12
	• ,SMCHR ,TEST ,ERMARK	13
	./ICNSTI/ BLANK ,BLANKS ,DOLLAR ,EOS	14
	• ,ICOMMA ,IDOLAR ,IFT ,IPTAB ,ITAB	15
	• ,KAM10 ,KBPC ,KBPW ,KCPCD ,KERTYP	16
	• ,KZERO ,NOPRNT ,TAB1	17
	./IPARAM/ ABORT ,KIUNIT ,KOUNIT ,LIMLF ,LOCK ,LOCX	18
	• ,NOLIST ,NSTDIR ,TRACE	19
	./ISTACKI/ STACK ,ISTDIM ,KSTACK ,LEVLIM	20
C	INTEGER BLANK ,BLANKS ,EOS ,IDOLAR ,TAB1 ,TRACE	21
	DOUBLE PRECISION STACK, VALUE	22
	LOGICAL ABORT,DOLLAR,ERMARK,LIMLF,LOCK,NOLIST,NOVEW,MODALL,	23
	• NSTDIR,SMCHR,TEST	24
	EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE),(NAME,ANAME)	25
	EQUIVALENCE (ICOMNI,ISUB),(IMAGE,IMAGE1),(IPARAM,ABORT)	26
C	CALL DEBUG(6HIARITI)	27
C	CALL STACKP	28
C	PRINT STACK	29
C		30
	IF (KERTYP.NE.0) GO TO 100	31
	IF ((LEVEL.LE.0).OR.(LEVEL.GT.LEVLIM)) GO TO 120	32
C	BRANCH ON KSTACK(LEVEL)= -, 0, 1, 2, 3, 4, 5, 6, 7, 8 AND UP	33
2	K=MAXJ(1,KSTACK(LEVEL)+2)	34
	IF(K.GE.8) GO TO 60	35
C	• 0 1 2 3 4 5 =KSTACK(LEVEL)	36
3	GO TO (120,90, 120, 20, 30, 40, 50),K	37
20	VALUE=STACK(LEVEL)*VALUE	38
	GO TO 90	39
30	VALUE=STACK(LEVEL)/VALUE	40
	GO TO 90	41
40	VALUE=STACK(LEVEL)+VALUE	42
	GO TO 90	43
50	VALUE=STACK(LEVEL)-VALUE	44
	GO TO 90	45
60	LEVEL1= LEVEL-K+9	46
	DO 61 I=LEVEL1,LEVEL	47
61	KSTACK(I)=0	48
	LEVEL = LEVEL1-1	49
C	CALL DEBUG 2(5HLEVEL,LEVEL)	50
C	CALL DEBUG 2(5HVALUE,VALUE)	51
	IF (LOCK) GO TO 62	52
	CALL IXQTI(VALUE,STACK(LEVEL))	53
62	CONTINUE	54
C	CALL DEBUG 2(5HVALUE,VALUE)	55
	IF (LEVEL .LE. NOTARG) NOTARG = ISTDIM	56
	IF (STACK(LEVEL).LT.0.DO) GO TO 110	57
	GO TO 90	58
90	STACK(LEVEL)=VALUE	59
	KSTACK(LEVEL)=1	60
100	CONTINUE	61
C	CALL STACKP	62
		63
		64

C		PRINT STACK	65
C	CALL DEBUGR		66
	RETURN		67
110	LFRT=0		68
C		* SPECIAL TREATMENT FOR SUBROUTINES	69
C	CALL DEBUG2(6HSTATMT,110)		70
	KSTACK(LEVEL)=0		71
	LEVEL=0		72
C	OTH		73
	CALL ICHAR2(\$100,29)		74
	KERTYP = 150		75
	GO TO 100		76
120	KERTYP=750		77
	GO TO 100		78
	END		79

\$IBFTC	ICHA4	DECK		1
	SUBROUTINE	ICHA4(*,*,LIST1,LIST2)		2
	DIMENSION	IFT(31), IPTAB(21), ITAB(65)		3
	DIMENSION	ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)		4
		,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)		5
C	COMMON			6
	./ICOMNI/	VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB		7
		,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2		8
		,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK		9
		,MCNVRT ,MDIF ,MODALL ,MSTOR		10
		,NAME ,NERROR ,NONEW ,NOTARG		11
		,SMCHR ,TEST ,ERMARK		12
	./ICNSTI/	BLANK ,BLANKS ,DOLLAR ,EOS		13
		,ICOMMA ,IDOLAR ,IFT ,IPTAB ,ITAB		14
		,KAM10 ,KBPC ,KBPW ,KCPCD ,KERTYP		15
		,KZERO ,NOPRNT ,TAB1		16
	./IPARAM/	ABORT ,<IUNIT ,KOUNTIT ,LIMALF ,LOCK ,LOCX		17
		,NOLIST ,VSTDIR ,TRACE		18
	./ISTAKI/	STACK ,ISTDIM ,KSTACK ,LEVLIM		19
C	INTEGER	BLANK ,BLANKS ,EOS ,IDOLAR ,TAB1 ,TRACE		20
	.,COMPR			22
	DOUBLE	PRECISION STACK, VALUE, IDEBUA, DEBGNA(2)		23
	DATA	DEBGNA/6HICHAR ,5HITCHR/		24
	LOGICAL	ABORT,DOLLAR,ERMARK,LIMALF,LOCK,NOLIST,NONEW,MODALL,		25
		NSTDIR,SMCHR,TEST		26
	EQUIVALENCE	(STACK,ISTACK), (VALUE,KVALUE,RVALUE),(NAME,ANAME)		27
	EQUIVALENCE	(ICOMNI,ISUB),(IMAGE,IMAGE1),(IPARAM,ABORT)		28
	LOGICAL	GOTCD		29
C		* TRUE IF NEW CARD WAS READ		30
	LOGICAL	SMCHR		31
C		* RE-PROCESS THE SAME CHARACTER AS LAS		32
	KFLD2=	LIST2		33
	IBITS=	2		34
	ASSIGN	36 TO COMPR		35
	IDEBJN =	4		36
	GO TO	10		37
	ENTRY	ICHARI(*,*,LIST1)		38
	IBITS =	2		39
	IDEBJN =	3		40
	GO TO	4		41

	ENTRY ICHAR2(*,LIST1)	42
	IBITS=1	43
	IDEBUN = 2	44
4	KFLD2 = -1	45
	ASSIGN 37 TO COMPR	46
	IF(MJDALL)GO TO 40	47
10	ASSIGN 20 TO IGETR	48
12	KFLD1=LIST1	49
	IDEBUA = DEBGNA(1)	50
	ASSIGN 110 TO NEXT	51
	GOTCD=.FALSE.	52
	IF(SMCHR) GO TO 35	53
C	GO TO 200	54
	* SKIP GETTING NEW CHARACTER	55
C	GO TO 200	56
	* GET CHARACTER RETURN TO 20	57
20	IF(KCH.EQ.BLANK)GO TO 200	58
C	IF (KCH.NE.EOS) GO TO 30	59
	* BYPASS BLANKS	60
C	ASSIGN 200 TO INEWR	61
	* EOS= END OF STATEMENT CHARACTER	62
C	GO TO 300	63
	* PROVIDE TO GET FIRST CHARACTER FROM NEXT C	64
30	IF(TEST.AND.GOTCD)GO TO 100	65
35	ITEMP = ITAB(KCH+1)	66
	ICOMP=IFLD(LIST1,IBITS,ITEMP)	67
	GO TO COMPR,(36,37)	68
36	LCOMP=IFLD(LIST2,2,ITEMP)+1	69
37	CONTINUE	70
	IF(TRACE .LT.4) GO TO 38	71
C	CALL DEBUG2(IDEBUA,IDEBUG)	72
38	SMCHR = .FALSE.	73
	IF(ICOMP-1) 381,382,383	74
381	RETURN	75
382	RETURN1	76
383	RETURN2	77
40	ASSIGN 35 TO IGETR	78
	GO TO 12	79
C	- - - - - ICHR - - - - -	80
C	SPECIAL ROUTINE TO INSERT COMMA AT END OF CARD IF	81
C	THE NEXT CARD BEGINS WITH \$ OR A LEFT MEMBER	82
C	TO THIS ROUTINE, A LEFT MEMBER BEGINS WITH A-Z	83
C	FOLLOWED BY ANY OF 0-9()A-Z FOLLOWED BY =	84
100	IDEBUA = DEBGNA(2)	85
	ITEMP = ITAB(KCH+1)	86
	ICOMP=IFLD(0,3,ITEMP)	87
	GO TO NEXT,(110,120)	88
C	* 110 FOR 1ST CHR ON NEW CARD, 120 FOR FOLL	89
110	NONEW=.TRUE.	90
	ASSIGN 120 TO NEXT	91
C	\$ A-Z 0-9() = OTH	92
	GO TO(130,200,140,140,140),ICOMP	93
C	\$ A-Z 0-9() = OTH	94
120	GO TO(140,200,200,130,140),ICOMP	95
130	KCOUNT=0	96
	KCH=ICOMMA	97
	NONEW=.FALSE.	98
	GOTCD = .FALSE.	99
	GO TO 35	100
C	* COMMA IS CHARACTER RETURNED	101
140	KCOUNT=0	102
	NONEW=.FALSE.	103
	GOTCD = .FALSE.	104

C	GO TO 200	105
C	* FIRST NON-BLANK CHARACTER ON CARD IS RETURNED	106
C	- - - - -	107
C	ROUTINE TO GET NEXT CHARACTER	108
C		109
200	IF(KCOUNT.LT.KCPCD)GO TO 210	110
	ASSIGN 210 TO INEWR	111
	GO TO 300	112
210	KCOUNT=KCPCD+1	113
C	* CARD COLUMN OF NEW CHARACTER	114
	KCH=IFLD(0,KBPC,IMAGE(KCOUNT))	115
	GO TO IGETR,(20,35)	116
C	* 35 IS USED ONLY FOR MODALL .TRU	117
C	- - - - -	118
C	ROUTINE TO PRINT OLD CARD AND READ NEXT	119
C	IF NONEW IS TRUE, STORE TAB1 IN KCH AND RETURN	120
300	IF(NONEW)GO TO 310	121
	KCH=IMAGE(KCOUNT+1)	122
C	* GET CARRIAGE CONTROL (IMAGE(81) IS A BLANK)	123
	IF((KCH.EQ.NOPRNT).OR.NOLIST) GO TO 305	124
	WRITE(KOUNIT,398)KCH,IMAGE	125
398	FORMAT(1A1,5X,80A1)	126
305	READ(KIUNIT,399)IMAGE	127
399	FORMAT(80A1)	128
	GOTCD=.TRUE.	129
	KCOUNT=0	130
	GO TO INEWR,(200,210)	131
310	KCH=TAB1	132
	GO TO 30	133
C	* TAB1 IS RETURNED, INDICATES END OF C	134
	END	135

\$IBFTC	ICNVTI	DECK	
	SUBROUTINE	ICNVTI	(IFROM,ITO)
	DIMENSION	IFT(31), IPTAB(21), ITAB(65)	
	DIMENSION	ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)	
	.	,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)	
C			
	COMMON		
	./ICOMNI/	VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB	
	.	,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2	
	.	,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK	
	.	,MCNVRT ,MDIF ,MODALL ,MSTOR	
	.	,NAME ,NERROR ,NONEW ,NOTARG	
	.	,SMCHR ,TEST ,ERMARK	
	./ICNSTI/	BLANK ,BLANKS ,DOLLAR ,EOS	
	.	,ICJMA ,IDOLAR ,IFT ,IPTAB ,ITAB	
	.	,KAM10 ,KBPC ,KBPW ,KCPCD ,KERTYP	
	.	,KZERO ,NOPRNT ,TAB1	
	./IPARAM/	ABORT ,KIUNIT ,KOUNIT ,LIMLFF ,LOCK ,LOCX	
	.	,NOLIST ,NSTDIR ,TRACE	
	./ISTACK/	STACK ,ISTDIM ,KSTACK ,LEVLIM	
C			
	INTEGER	BLANK ,BLANKS ,EOS ,IDOLAR ,TAB1 ,TRACE	
	DOUBLE PRECISION	STACK, VALUE	
	LOGICAL	ABORT,DOLLAR,ERMARK,LIMLFF,LOCK,NOLIST,NONEW,MODALL,	
	.	NSTDIR,SMCHR,TEST	
	EQUIVALENCE	(STACK,ISTACK), (VALUE,<VALUE,RVALUE),(NAME,ANAME)	

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EQUIVALENCE (ICOMNI,ISUB),(IMAGE,IMAGE1),(IPARAM,ABORT)      26
DIMENSION NTYPE(4)                                           27
DOUBLE PRECISION NTYPE                                         28
DATA NTYPE / 6HREAL ,6HINT ,6HDP ,5HNOCONV/                 29
IF((IFROM.LE.0).OR.(ITO.LE.0)) GO TO 100                      30
IF (IFROM - 4) 1,99,100                                       31
1 IF (ITO - 4) 2,99,100                                       32
2 ITOM2 = ITO - 2                                             33
IF (IFROM-2) 3,4,5                                           34
3 IF (ITOM2) 99,10,20                                         35
4 IF (ITOM2) 30,99,40                                         36
5 IF (ITOM2) 50,60,99                                         37
10 KVALUE = RVALUE(1)                                         38
GO TO 99                                                       39
20 VALUE = RVALUE(1)                                          40
GO TO 99                                                       41
30 RVALUE(1) = KVALUE                                         42
GO TO 99                                                       43
40 VALUE = KVALUE                                             44
GO TO 99                                                       45
50 RVALUE(1) = VALUE                                          46
GO TO 99                                                       47
60 KVALUE = VALUE                                             48
99 CONTINUE                                                  49
C CALL DEBUG3 (6HICNVTI,0.D0,3)                               50
C CALL DEBUG3(NTYPE(IFROM),NTYPE(ITO) ,3)                     51
RETURN                                                        52
100 WRITE(KOUNIT,101) IFROM,ITO                               53
101 FORMAT(35H ARGUMENTS OF ICNVTI BAD. IFROM = ,I13,8H, ITO = ,I13,
. 41H(1 TO 4 ALLOWABLE). CHECK IPTAB IN COMNJ)                55
GO TO 99                                                       56
END                                                            57

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$IBFTC IERORI DECK
SUBROUTINE IERORI
DIMENSION IFT( 31), IPTAB( 21), ITAB( 65)
DIMENSION ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)
. ,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)
C
COMMON
./ICOMVI/ VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB
. ,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2
. ,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK
. ,MCNVRT ,MDIF ,MODALL ,MSTOR
. ,NAME ,NERROR ,NONEW ,NOTARG
. ,SMCHR ,TEST ,ERMARK
./ICNSTI/ BLANK ,BLANKS ,DOLLAR ,EOS
. ,ICOMMA ,IDOLAR ,IFT ,IPTAB ,ITAB
. ,KAM10 ,KBCPC ,KBPW ,KCPCD ,KERTYP
. ,KZERO ,NOPRNT ,TAB1
./IPARAM/ ABORT ,KIUNIT ,KOUNIT ,LIMLFF ,LOCK ,LOCKX
. ,NOLIST ,NSTDIR ,TRACE
./ISTACK/ STACK ,ISTDIM ,KSTACK ,LEVLIM
C
INTEGER BLANK ,BLANKS ,EOS ,IDOLAR ,TAB1 ,TRACE
DOUBLE PRECISION STACK, VALUE
LOGICAL ABORT,DOLLAR,ERMARK,LIMLFF,LOCK,NOLIST,NONEW,MODALL,
.NSTDIR,SMCHR,TEST

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	EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE), (NAME,ANAME)	25
	EQUIVALENCE (ICOMNI,ISUB), (IMAGE,IMAGE1), (IPARAM,ABORT)	26
	INTEGER H(4,7)	27
	EQUIVALENCE (H(1,1),H1), (H(1,2),H2), (H(1,3),H3), (H(1,4),H4)	28
	., (H(1,5),H5), (H(1,6),H6), (H(1,7),H7)	29
	DATA <A/1H*/	30
	DATA <B/1H /	31
	DATA KC/1H, /	32
	DATA <D/1H' /	33
	INTEGER MSGTYP(2,2), H1(4), H2(4), H3(4), H4(4), H5(4), H6(4), H7(4)	34
	DATA MSGTYP(1,1)/24H*DIAGNOSTIC**ERROR** /	35
	DATA +1(1) /24HINAPPROPRIATE CHARACTER /	36
	DATA +2(1) /24HNAME TOO LONG /	37
	DATA +3(1) /24HTABLE FULL OR DESTROYED /	38
	DATA +4(1) /24H\$D INCOMPLETE OR MISSING /	39
	DATA +5(1) /24HNAME NOT IN TABLE /	40
	DATA +6(1) /24HFUNCTN OR SUB NOT ABOARD /	41
	DATA +7(1) /24HFORMJLA ILL-FORMED /	42
C	CALL DEBUGC(6HIERORI)	43
	NONEW = .FALSE.	44
	NERROR = NERROR + 1	45
C		46
	EMARX = .TRUE.	47
C		48
	IOP = 0	49
C		50
	MODE = 1	51
	IF(NOTARG.LT.LEVEL) MODE = 2	52
	KC1=KB	53
	IF(KCJUNT.EQ.0) KC1=KC	54
	L = KCJUNT + 1	55
	MODALL = .FALSE.	56
	TEST = .TRUE.	57
	M = IABS(KERTYP)	58
	J = M/100	59
	K = 2	60
	IF(DOLLAR) MODE = 2	61
	IF(KERTYP.GT.0) LOCK = .TRUE.	62
	IF(KERTYP.LT.0) K=1	63
11	WRITE(KOUNT,90)(MSGTYP(I,K),I=1,2),M,(H(I,J),I=1,4),KC1,	64
	IMAGE,(KB,I=1,L),KA,(KB,I=L,81),NERROR,LOCK	65
90	FORMAT(1H ,2A6,2H (,I3,2H) ,4A6,2X,81A1/45X,83A1,	66
	./14X,11HERROR COUNT,I4,13H LOCK = .,L1,1H.)	67
	IF(NERROR.GT.64)STOP	68
	IF(KERTYP.LT.0) GO TO 99	69
	SMCHX = .TRUE.	70
30	CALL ICHAR4(\$31,\$65,7,10)	71
C	0-9. + \$)	72
	GO TO (30,65,39,80),LCOMP	73
C	, ' A-Z =	74
31	GO TO (60,65,30,60),LCOMP	75
39	SMCHX = .TRUE.	76
40	L = KCJUNT + 1	77
	KC1 = KB	78
	IF(KCJUNT.EQ.0)KC1=KC	79
	WRITE(KOUNT,91)KC1,IMAGE,(KB,I=1,L),KD,(KB,I=L,81)	80
91	FORMAT(14X,43HSKIP AHEAD AND GUESS AT NEXT GOOD STATEMENT,	81
	./14X,32HBEGIN PROCESSING AT SYMBOL ,81A1/45X,83A1	82
	./14X,3HLOOK FOR DIAGONISTICS BUT DON'T STORE ANYTHING.//)	83
	IFNTYP = -1	84
	DO 1 I=1,ISTDIM	85

1	KSTACK(I) = 0	86
	LEVEL = 0	87
	NOTARG = ISTDIM	88
99	CONTINUE	89
	KERTYP = 0	90
C	CALL DEBUGR	91
	RETURN	92
C	,,=,=,=,=,=,=,=,=,=,=,=,=,=,=,=	93
60	IF(MODE.EQ.1) GO TO 40	94
C	ALL OPERATORS	95
65	IOP = IOP + 1	96
	IF(IOP.LE.16) GO TO 30	97
C	IF YOU FIND 16 OPERATORS BEFORE) GIVE UP	98
C		99
C))))))))))))))))))))))	100
80	MODE = 1	101
	DOLLAR = .FALSE.	102
	GO TO 30	103
C		104
C	MODE ACTION	105
C		106
C	1 SKIP TO NEXT \$ OR = OR , (UNCONDITIONAL)	107
C	2 SKIP TO NEXT \$ (UNCONDITIONAL) OR TO = OR , AFTER)	108
	END	109

```

1      IF(IT(LOOK) .EQ.0) GO TO 7          33
C                                     * RETURN WITH NEXT AVAILABLE ENTR 34
      J=IFLD(3,4,IT(LOOK))                35
C                                     * MAX POSSIBLE J IS 15          36
      IF (J .NE.MSTOR) GO TO 8             37
C                                     * B IF NAME IS WRONG LENGTH, TRY 38
      DO 4 <=1,J                          39
C                                     * CHECK NAME FOR MATCH          40
      LOOKK = LOOK + K                     41
      IF (NAME(K).NE.IT(LOOKK)) GO TO 8    42
4      CONTINUE                           43
C      CALL DEBUG2(6H*LOOKF,LOOK)          44
C                                     * ENTRY WAS FOUND              45
      RETURN 1                             46
7      CONTINUE                           47
C      CALL DEBUG2(6H*LOOKN,LOOK)          48
C                                     * NO ENTRY WAS FOUND            49
      RETURN                               50
8      LOOK = LOOK+J+1                    51
                                           52
                                           53
C                                     * TRY NEXT ENTRY                54
      GO TO 1                              55
      END                                  56

```

```

$IBFTC INAMEI DECK
      SUBROUTINE INAMEI(*,D,IT)            1
C      CALLED FROM INPUT                    2
C      CALL DEBUGC(6HINAMEI)              3
C      SUBROUTINE TO OBTAIN LOCATION, MODE, AND CONTENTS OF A NAMED CELL 4
C                                           5
      DIMENSION IFT( 31), IPTAB( 21), ITAB( 65) 6
      DIMENSION ANAME(15), IMAGE(80), IMAGE1(81), IPARAM(9) 7
      *      ,KSTACK(27),NAME(15),RVALUE(2),STACK(27) 8
C                                           9
      COMMON                               10
      ./ICOMNI/ VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB 11
      *      ,KCH ,KCNVRT ,KCJUNT ,KDIF ,KFLD1 ,KFLD2 12
      *      ,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK 13
      *      ,MCNVRT ,MDIF ,MODALL ,MSTOR 14
      *      ,NAME ,NERROR ,NONEW ,NOTARG 15
      *      ,SMCHR ,TEST ,ERMARK 16
      ./ICNSTI/ BLANK ,BLANKS ,DOLLAR ,EOS 17
      *      ,ICOMMA ,IDOLAR ,IFT ,IPTAB ,ITAB 18
      *      ,KAM10 ,KBPC ,KBPW ,KCPCD ,KERTYP 19
      *      ,KZERO ,NOPRNT ,TAB1 20
      ./IPARAM/ ABORT ,KIUNIT ,KOUNTIT ,LIMLF ,LOCK ,LOCKX 21
      *      ,NQLIST ,NSTDIR ,TRACE 22
      ./ISTACK/ STACK ,ISTDIM ,KSTACK ,LEVLIM 23
C                                           24
      INTEGER BLANK ,BLANKS ,EOS ,IDOLAR ,TAB1 ,TRACE 25
      DOUBLE PRECISION STACK, VALUE 26
      LOGICAL ABORT,DOLLAR,ERMARK,LIMLF,LOCK,NQLIST,NONEW,MODALL, 27
      NSTDIR,SMCHR,TEST 28
      EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE), (NAME,ANAME) 29
      EQUIVALENCE (ICOMNI,ISUB), (IMAGE,IMAGE1), (IPARAM,ABORT) 30
      DIMENSION IT(1) 31

```

C		* TABLE OF NAMES PROVIDED BY USER	32
	DIMENSION D(1)		33
C		* USERS VARIABLES ARE IN D ARRAY	34
	ISUB = 1		35
C		* UNDERSTOOD SUBSCRIPT	36
	KDIF = 1		37
	CALL INAMEN		38
	CALL ILOOKI(\$1,IT)		39
	CALL ILOOKI(\$64,IFT)		40
9540	KERTYP=540		41
C		* NAME NOT IN TABLE	42
	GO TO 99		43
1	CONTINUE		44
C	CALL DEBUG2(6HSTATMT,1)		45
	ITL = IT(LOOK)		46
	KCNVRT=IFLD(0,3,ITL)		47
C	CALL DEBUG 2 (6HKCNVRT,KCNVRT)		48
C	R I D T S F		49
	GO TO(12,12,10,12,62,61),KCNVRT		50
10	KDIF = 2		51
C	(OTH		52
12	CALL ICHAR2(\$90,30)		53
C		* GO TO 90 FOR NO SUBSCRIPT	54
C	A-Z 0-9 OTH		55
	CALL ICHAR1(\$82,\$9140,12)		56
	CALL INAMEN		57
	CALL ILOOKI(\$83,IT)		58
	GO TO 9540		59
82	SMCHR = .TRUE.		60
	CALL ISUBI		61
	GO TO 84		62
83	ITYPE=IFLD(0,3,IT(LOOK))		63
	LOC=IFLD(7,25,IT(LOOK))		64
	RVALUE(1) = D(LOC)		65
	RVALUE(2) = D(LOC+1)		66
	CALL ICNVTI (ITYPE,2)		67
	ISUB = KVALUE		68
C) OTH		69
84	CALL ICHAR2(\$9140,31)		70
C		* ERROR IF NO)	71
	GO TO 91		72
90	SMCHR = .TRUE.		73
91	LOC=IFLD(7,25,ITL)		74
C	CALL DEBUG 2 (5HLOC 1,LOC)		75
	LOC = LOC + (ISUB-1)*KDIF		76
50	RVALUE(1) = D(LOC)		77
	RVALUE(2) = D(LOC+1)		78
C	CALL DEBUG 3(5HIT(L),ITL,2)		79
	CALL ICNVTI (KCNVRT,3)		80
52	LOOK = LOC		81
98	CONTINUE		82
C	CALL DEBUG 3(5HVALUE,VALUE,4)		83
C	CALL DEBUG 2 (6HLOOK ,LOOK)		84
C	CALL DEBUG 2 (5HLOC 2,LOC)		85
C	CALL DEBUG 2(6HKCNVRT,KCNVRT)		86
C	CALL DEBUGR		87
	RETURN		88
61	IFNTYP = 0		89
C		* FUNCTION	90
	GO TO 63		91
62	IFNTYP = 1		92

C		* SUBROUTINE	93
63	KVALJE=IFLD(7,25,ITL)		94
	GO TO 98		95
64	IFNTYP = 0		96
C		* LIBRARY FUNCTION OR SUBROUTINE	97
	KVALJE=IFLD(7,25,IFT(LOOK))		98
C		* PROGRAM NUMBER (USED BY IXQTI)	99
	MSTOR = 0		100
	KCNVRT=IFLD(0,3,IFT(LOOK))		101
C		* 5 FOR SUBROUTINES, 6 FOR FUNCTIONS	102
	IF(KCNVRT.EQ.5)IFNTYP=KVALUE		103
C		* PROVIDE FOR EXECUTION OF INPUT SUBRO	104
C	CALL DEBUG2(6HLIBF ,KVALUE)		105
C	MEANING OF IFNTYP		106
C	VALUE NAME IS		107
C	-1(NORMAL) AN ORDINARY VARIABLE		108
C	0 FUNCTION (USER OR FORTRAN MATH)		109
C	1 USER SUBROUTINE		110
C	2 RADIX (INPUT FUNCTION)		111
C	3 LOCK (INPUT SUBROUTINE)		112
C	\$CALL LOCK(Y,I) CAUSES I TO BE SET SO THAT Y(I)REFERS TO THE CUR		113
C	LEFT SIDE.		114
	GO TO 98		115
9140	KERTYP =140		116
99	CONTINUE		117
C	CALL DEBUG2(6HRETURN,1)		118
C	CALL DEBUGR		119
C		* REGISTER RETURN	120
	RETURN 1		121
	END		122

\$IBFTC	INAMEN	DECK	
	SUBROUTINE	INAMEN	1
C	CALL	FROM	2
C		INAMEI	3
C		ITABLI	4
C		INPUT	5
C	CALL	DEBUGC(6HINAMEN)	6
C			7
	DIMENSION	IFT(31), IPTAB(21), ITAB(65)	8
	DIMENSION	ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)	9
	.	,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)	10
C			11
	COMMON		12
.	/ICOMVI/	VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB	13
.		,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2	14
.		,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK	15
.		,MCNVRT ,MODIF ,MODALL ,MSTOR	16
.		,NAME ,NERROR ,NONEW ,NOTARG	17
.		,SMCHR ,TEST ,ERMARK	18
.	/ICNSTI/	BLANK ,BLANKS ,DOLLAR ,EOS	19
.		,ICOMMA ,IDOLLAR ,IFT ,IPTAB ,ITAB	20
.		,KAM10 ,KBPC ,KBPW ,KCPCD ,KERTYP	21
.		,KZERO ,NOPRNT ,TAB1	22
.	/IPARAM/	ABORT ,KIUNIT ,KOUNIT ,LIMAF ,LOCK ,LOCKX	23
.		,NOLIST ,NSTDIR ,TRACE	24
.	/ISTAKI/	STACK ,ISTDIM ,KSTACK ,LEVLIM	25

C	INTEGER BLANK ,BLANKS ,EOS ,IDOLAR ,TAB1 ,TRACE	26
	DOUBLE PRECISION STACK, VALUE, DNAME	27
	LOGICAL ABORT,DOLLAR,ERMARK,LIMLF,LOCK,VOLIST,NONEW,MODALL,	28
	.NSTDIR,SMCHR,TEST	29
	EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE),(NAME,ANAME)	30
	EQUIVALENCE (ICOMNI,ISUB),(IMAGE,IMAGE1),(IPARAM,ABORT)	31
	EQUIVALENCE(NAME,DNAME)	32
C	COLLECTS NAME (UP TO 15 WORDS) TERMINATED BY ANY SPECIAL CHAR	33
	ASSIGN 6 TO NEXT	34
	IF(MODALL)ASSIGN 2 TO NEXT	35
	MSTOR = 0	36
	J = KBPW	37
	NAME(2) = BLANKS	38
1	GO TO NEXT,(2,6,7)	39
C	OTH	40
2	CALL ICHAR2(\$8 ,23)	41
	TEST = .TRUE.	42
	MODALL = .FALSE.	43
	CALL ICHAR2(\$99,23)	44
C	* PAIR OF APOSTROPHYS DOES NOT EN	45
	TEST = .FALSE.	46
	MODALL = .TRUE.	47
	GO TO 8	48
6	ASSIGN 7 TO NEXT	49
	GO TO 8	50
C	A-Z-0-9 OTH	51
7	CALL ICHAR2(\$99,12)	52
8	IF(J.LT.KBPW) GO TO 9	53
	IF(MSTOR.EQ.15) GO TO 10	54
	MSTOR = MSTOR + 1	55
	NAME(MSTOR) = BLANKS	56
	J = 0	57
9	NAME(MSTOR)=IFLD4(KCH,J,KBPC,NAME(MSTOR))	58
	J=J+KBPC	59
	GO TO 1	60
10	IF(MODALL) GO TO 99	61
	KERTYP = -260	62
	CALL IERORI	63
95	CALL ICHAR2(\$99,12)	64
C	* SKIP REST OF NAME	65
	GO TO 95	66
99	CONTINUE	67
	SMCHR = .TRUE.	68
C	CALL DEBUG3(6HNAME ,DNAME,3)	69
C	CALL DEBUG2(5HMSTOR,MSTOR)	70
C	CALL DEBUGR	71
	RETURN	72
	END	73
		74

\$IBFTC INMBRI DECK

C	SUBROJTINE TO TRANSLATE A NUMERIC FIELD	1
	SUBROJTINE INMBRI	2
C	CALLED FROM INPUT	3
C	INMBRI IS CALLED WITH FIRST DIGIT IN KCH	4
C	CALL DEBUGC(6HINMBRI)	5
	DIMENSION IFT(31), IPTAB(21), ITAB(65)	6
	DIMENSION ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)	7
.	,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)	8

C	COMMON						9	
	./ICOMNI/	VALUE	,ICOMP	,IFNTYP	,IMAGE1	,IRADIX	,ISUB	10
	.	,KCH	,KCNVRT	,KCOUNT	,KDIF	,KFLD1	,KFLD2	11
	.	,LCOMP	,LCNVRT	,LEVEL	,LFRT	,LOOK		12
	.	,MCVVRT	,MDIF	,MODALL	,MSTOR			13
	.	,NAME	,NERROR	,NONEW	,NOTARG			14
	.	,SMCHR	,TEST	,ERMARK				15
	./ICNSTI/	BLANK	,BLANKS	,DOLLAR	,EOS			16
	.	,ICOMMA	,IDOLAR	,IFT	,IPTAB	,ITAB		17
	.	,KAM10	,KBPC	,KBPW	,KPCPD	,KERTYP		18
	.	,KZERO	,NOPRNT	,TAB1				19
	./IPARAM/	ABORT	,KIUNIT	,KOUNIT	,LIMALF	,LOCK	,LOCX	20
	.	,NOLIST	,NSTDIR	,TRACE				21
	./ISTAKI/	STACK	,ISTDIM	,KSTACK	,LEVLIM			22
C								23
	INTEGER	BLANK	,BLANKS	,EOS	,IDOLAR	,TAB1	,TRACE	24
	DOUBLE PRECISION	STACK, VALUE, DNBR, H, FD(4)						25
	LOGICAL	ABORT,DOLLAR,ERMARK,LIMALF,LOCK,NOLIST,NONEW,MODALL,						26
	.NSTDIR,SMCHR,TEST							27
	EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE), (NAME,ANAME)							28
	EQUIVALENCE (ICOMNI,ISUB), (IMAGE,IMAGE1), (IPARAM,ABORT)							29
	LOGICAL LVALUE							30
	EQUIVALENCE (KVALUE,LVALUE)							31
	LOGICAL SWITCH							32
	DIMENSION LD(4)							33
	DATA LD(1),LD(2),LD(3),LD(4)/8,4,2,1/							34
	DATA FD(1),FD(2),FD(3),FD(4)/1.0D8,1.0D4,1.0D2,10.0D0/							35
	DNBR=0							36
C		* THE NUMBER COLLECTED SO FAR						37
	ICSC=0							38
C		* THE CHARACTERISTIC SCALE FACTOR						39
	IPF = 1							40
C		* SIGN OF EXPONENT						41
	IESE = 0							42
C		* THE EXPONENT						43
	ASSIGN 1 TO NEXT							44
	SWITCH = .FALSE.							45
	SMCHR = .TRUE.							46
	CALL ICHAR2(\$70,21)							47
C		* GO TO 70 FOR LOGICAL CONSTANTS						48
	SMCHR = .TRUE.							49
C		+ - OTH 0-9						50
1	CALL ICHAR4(\$2, \$3, 13,15)							51
	GO TO 50							52
3	DNBR = DNBR*10.0D0+FLOAT(KCH-KZERO)							53
	GO TO NEXT,(1,15)							54
15	ICSC = ICSC - 1							55
	GO TO 1							56
C		. DE OTH						57
2	GO TO(20,30,50),LCOMP							58
19	SMCHR=.TRUE.							59
C		*ENTER HERE FOR INITIAL DECIMAL POINT						60
20	IF (SWITCH) GO TO 9130							61
C	CALL DEBUG2(6HSTAT .,20)							62
	ASSIGN 15 TO NEXT							63
	SWITCH = .TRUE.							64
	GO TO 1							65
37	IPF=-1							66
	GO TO 36							67
30	CONTINUE							68
C	CALL DEBUG 2(6HSTAT E,30)							69
								70

C	+ - OTH 0-9	71
	CALL ICHAR4(\$50,\$35, 13,15)	72
C	+ -	73
	GO TO(36,37),LCOMP	74
35	SMCHR=.TRUE.	75
36	CALL ISUBI	76
C	' TEST WILL BE TRUE	77
44	ICSC = ICSC+ISIGN(ISUB,IPF)	78
C	' RESOLVE SCALE FACTORS	79
50	CONTINUE	80
C	CALL DEBUG 2(6HSTAT ,50)	81
C	CALL DEBUG3 (5HDNBR1,DNBR ,4)	82
	H = 1.00	83
	IESC=IABS(ICSC)	84
	DO 63 I=1,4	85
61	IF (IESC.LT. LD(I)) GO TO 63	86
62	IESC=IESC-LD(I)	87
	H=H*FD(I)	88
	GO TO 61	89
63	CONTINUE	90
	IF (ICSC.LT.0) GO TO 65	91
64	DNBR = DNBR*H	92
	GO TO 98	93
C	T F OTH	94
70	CALL ICHAR1(\$73,\$19,17)	95
	LVALJE=.TRUE.	96
71	IF(NOTARG.GT.LEVEL)LCNVRT=4	97
C	' LOGICAL CONSTANTS NOT CONVERTED	98
	RVALJE(2)=0.	99
C	A-Z OTH .	100
72	CALL ICHAR1(\$9130,\$99,21)	101
C	'DISCARD REST OF WORD, MUST FIND	102
	GO TO 72	103
73	LVALJE=.FALSE.	104
	GO TO 71	105
65	DNBR = DNBR/H	106
98	VALUE = DNBR	107
	SMCHR = .TRUE.	108
99	CONTINUE	109
C	CALL DEBUG3(6HDNBR 2,DNBR,4)	110
C	CALL DEBUG3(5HVALUE,VALUE,4)	111
C	CALL DEBUG2(6HICSC ,ICSC)	112
C	CALL DEBUG2(6HIPF ,IPF)	113
C	CALL DEBUG 2(6HIESC ,IESC)	114
C	CALL DEBUGR	115
	RETURN	116
9130	KERTYP = 130	117
	GO TO 99	118
	END	119

\$IBFTC ITABLI DECK

C	SUBROJTINE TO CONSTRUCT TABLE ENTRIES	1
	SUBROJTINE ITABLI(IT)	2
C	CALLED FROM INPUT	3
C	CALL DEBUGC(6HITABLI)	4


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C      DIMENSION IFT( 31), IPTAB( 21), ITAB( 65)
      DIMENSION ANAME(15) , IMAGE(80) , IMAGE1(81), IPARAM(9)
      *      , KSTACK(27), NAME(15) , RVALUE(2) , STACK(27)
C
      COMMON
      ./ICOMNI/ VALUE      , ICOMP      , IFNTYP      , IMAGE1      , IRADIX      , ISUB
      *      , KCH          , KCNVRT      , KCOUNT      , KDIF          , KFLD1      , KFLD2
      *      , LCOMP      , LCNVRT      , LEVEL          , LFRT          , LOOK
      *      , MCNVRT      , MDIF          , MODALL          , MSTORE
      *      , NAME          , NERROR      , NONEW          , NOTARG
      *      , SMCHR      , TEST          , ERMARK
      ./ICNSTI/ BLANK      , BLANKS      , DOLLAR      , EOS
      *      , ICOMMA      , IDOLAR      , IFT          , IPTAB      , ITAB
      *      , KAM10      , KBPC          , KBPW          , KCPCD      , KERTYP
      *      , KZERO      , NOPRNT      , TAB1
      ./IPARAM/ ABORT      , KIUNIT      , KOUNTIT      , LIMAF      , LOCK      , LOCK
      *      , NOLIST      , NSTDIR      , TRACE
      ./ISTAKI/ STACK      , ISTDIM      , KSTACK      , LEVLIM
C
      INTEGER BLANK      , BLANKS      , EOS      , IDOLAR      , TAB1      , TRACE
      DOUBLE PRECISION STACK, VALUE
      LOGICAL ABORT, DOLLAR, ERMARK, LIMAF, LOCK, NOLIST, NONEW, MODALL,
      * NSTDIR, SMCHR, TEST
      EQUIVALENCE (STACK, ISTACK), (VALUE, KVALUE, RVALUE), (NAME, ANAME)
      EQUIVALENCE (ICOMNI, ISUB), (IMAGE, IMAGE1), (IPARAM, ABORT)
      DIMENSION IT (1)
C
      KDIF = 1
      ISUBX = 1
      ITYPE = 1
C
      * ALWAYS INITIALIZE TO .REAL.
C      ( A-Z OTH
1  CALL ICHAR1 ($1, $9120, 26)
C      * SKIP 'ABLE' IN TABLE
3  CONTINUE
C  CALL DEBUG2(6HSTATMT, 3)
  CALL ICHAR4($4, $9120, 7, 9)
C      * 0-9+ OTH
  GO TO (30, 98, 10, 9120), LCOMP
C      * A-Z OTH OTH
4  GO TO (3, 20, 9120, 9120), LCOMP
9120 KERTYP = 120
C      * ILLEGAL CHARACTER
  GO TO 98
10  CONTINUE
C  CALL DEBUG2(6HST 0-9, 10)
  SMCHR = .TRUE.
  CALL ISUBI
  ISUBX = ISUB
  CALL ICHAR2($9120, 25)
C      * ERROR IF NO =
  GO TO 3
30  CONTINUE
C  CALL DEBUG 2(6HSTATMT, 30)
31  KDIF = 1
C  SPLIT TYPES (INT, REAL, DP, NO CONVERSION, FUNCTION, SUBROUTINE)
  CALL ICHAR4($32, $33, 3, 5)
  IF(LCOMP.GT.1) GO TO 9120
  KDIF = 2
  GO TO 33

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32	IF (LCOMP.EQ.3) GO TO 9120	65
	ITYPE = LCOMP	67
	GO TO 34	68
33	ITYPE = LCOMP+2	69
C	A-Z OTH .	70
34	CALL ICHAR1(\$9120,\$3,21)	71
	GO TO 34	72
20	CONTINUE	73
C	CALL DEBUG2(6HSTATMT,20)	74
	CALL INAMEN	75
50	CONTINUE	76
C	CALL DEBUG2(6HSTATMT,50)	77
	ITBUFF=IFLD4(ITYPE,0,3,ITBUFF)	78
	ITBUFF=IFLD4(MSTOR,3,4,ITBUFF)	79
	ITBUFF=IFLD4(ISUBX,7,25,ITBUFF)	80
C	CALL DEBUG2 (5HITYPE,ITYPE)	81
C	CALL DEBUG2 (5HMSTOR,MSTOR)	82
C	CALL DEBUG2 (5HISUBX,ISUBX)	83
C	CALL DEBUG4 (6HITBUFF,ITBUFF,2)	84
	ISUBX = ISUBX + KDIF	85
	CALL ILOOKI(\$56,IT)	85
	IF((IT(2).NE.0).AND.((LOOK+MSTOR +2).GT.IT(2))) GO TO 9320	87
	IT(LOOK)=ITBUFF	88
	DO 55 <=1,MSTOR	89
	LOOK=LOOK+1	90
55	IT(LOOK)=NAME(K)	91
C	CALL DEBUG2(6HSTATMT,55)	92
	LOOK=LOOK+1	93
	IT(LOOK)=0	94
	GO TO 3	95
56	IT(LOOK)=ITBUFF	96
C	CALL DEBUG2(6HSTATMT,56)	97
	GO TO 3	98
98	CONTINUE	99
C	CALL DEBUGR	100
	RETURN	101
9320	KERTYP =-320	102
	CALL IERORI	103
	GO TO 3	104
	END	105

\$18FTC	ISUBI DECK	
	SUBROUTINE ISUBI	1
C	ISUBI FINDS SUBSCRIPTS AND INTEGER CONSTANTS	2
C	CALLED FROM	3
C	INAMEI	4
C	INMBRI	5
C	INPUT	6
C	ITABLI	7
C	ISJB BEGINS PROCESSING WITH THE NEXT CHARACTER.	8
C	CALL DEBUGC(5HISUBI)	9
C	COLLECTS INTEGER OF BASE IRADIX TERMINATED BY A SPECIAL CHARACTER	10
	DIMENSION IFT(31), IPTAB(21), ITAB(65)	11
	DIMENSION ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)	12
	. ,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)	13
C		14
	COMMON	15
	./ICOMVI/ VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB	16

.	,KCH	,KCNVRT	,KCOUNT	,KDIF	,KFLD1	,KFLD2	17	
.	,LCOMP	,LCNVRT	,LEVEL	,LFRT	,LOOK		18	
.	,MCNVRT	,MDIF	,MODALL	,MSTOR			19	
.	,NAME	,NERROR	,NONEW	,NOTARG			20	
.	,SMCHR	,TEST	,ERMARK				21	
./ICNSTI/	BLANK	,BLANKS	,DOLLAR	,EOS			22	
.	,ICOMMA	,IDOLAR	,IFT	,IPTAB	,ITAB		23	
.	,KAM10	,KBPC	,KBPW	,KCPCD	,KERTYP		24	
.	,KZERO	,NOPRNT	,TAB1				25	
./IPARAM/	ABORT	,KIUNIT	,KOUNIT	,LIMLF	,LOCK	,LOCX	25	
.	,NOLIST	,NSTDIR	,TRACE				27	
./ISTAKI/	STACK	,ISTDIM	,KSTACK	,LEVLM			28	
C							29	
	INTEGER	BLANK	,BLANKS	,EOS	,IDOLAR	,TAB1	,TRACE	30
	DOUBLE	PRECISION	STACK,	VALUE				31
	LOGICAL	ABORT,DOLLAR,ERMARK,LIMLF,LOCK,NOLIST,NONEW,MODALL,						32
	,NSTDIR,SMCHR,TEST							33
	EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE), (NAME,ANAME)							34
	EQUIVALENCE (ICOMNI,ISUB), (IMAGE,IMAGE1), (IPARAM,ABORT)							35
	ISUB = 0							36
C	A-Z	0-9	OTH					37
80	CALL	ICHARI(\$10,\$99,12)						38
	IDIGIT=KCH-KAM10							39
C								40
30	IF(IDIGIT .GE. IRADIX)	GO TO 99						41
35	ISUB = ISUB * IRADIX + IDIGIT							42
C								43
								44
99	GO TO 80							45
	SMCHR = .TRUE.							45
C								46
								47
C	CALL	DEBUG2(6H*ISUBI,ISUB)						48
C	CALL	DEBUGR						49
	RETURN							50
10	IDIGIT=KCH-KZERO							51
C								52
								53
	GO TO 30							
	END							

\$IBFTC	IXQTI	DECK						
	SUBROJTIME	IXQTI (ARGL,ARGS)						1
C	USER	MAY PUT HIS OWN COMMON STATEMENTS IN THIS ROUTINE AND						2
C	USE	THEM TO SUPPLY ARGUMENTS TO HIS CALLS IF HE DESIRES						3
	COMMON							4
./IPARAM/	ABORT ,KIUNIT ,KOUNIT, LIMLF ,LOCK, LOCX, NOLIST, NSTDIR							5
.,	TRACE							6
	DOUBLE	PRECISION	ARGS(27), ARGL, ARG22					7
	DIMENSION	ARG2(2)						8
	EQUIVALENCE	(ARG2,ARG22)						9
C								10
	M =	DABS(ARGS(1))						11
	IF (M.LT.1 .OR. M.GT.16)	GO TO 99						12
	GO TO (1,2,3,4,5,6,7,8,9,10,11,12,13,14,15,16),M							13
1	ARGL =	COS(ARGL)						14
	GO TO	100						15
2	ARGL =	EXP(ARGL)						16
	GO TO	100						17
3	ARGL =	ALOG(ARGL)						18
	GO TO	100						19

4	ARGL = SIN(ARGL)	20
	GO TO 100	21
5	ARGL = SQRT(ARGL)	22
	GO TO 100	23
6	ARGL = ATAN2(ARGS(2),ARGL)	24
	GO TO 100	25
C	PRINT FUNCTION	26
7	ARG22 = ARGS(2)	27
	WRITE (KOUNIT,101) ARG2,ARGL	28
101	FORMAT(1H 2A6,3H = ,D26.17)	29
	GO TO 100	30
8	ARGL = SNGL(ARGS(2))*IFIX(SNGL(ARGL+DSIGN(.5D0,ARGL)))	31
	GO TO 100	32
9	ARGL = ABS(SNGL(ARGS(2)))*SNGL(ARGL)	33
	GO TO 100	34
10	ARGL = ABS(ARGL)	35
	GO TO 100	36
C	DISCRIMINANT) FUNCTION	37
11	IF (ARGL) 102,103,104	38
102	ARGL = ARGS(2)	39
	GO TO 100	40
103	ARGL = ARGS(3)	41
	GO TO 100	42
104	ARGL = ARGS(4)	43
	GO TO 100	44
12	CALL LOCKX(ARGL)	45
	GO TO 100	46
13	CONTINUE	47
	GO TO 100	48
14	CONTINUE	49
	GO TO 100	50
15	CONTINUE	51
16	CONTINUE	52
100	CONTINUE	53
C	CALL DEBUGR	54
	RETURN	55
99	KERTYP =-610	56
	CALL IERORI	57
	GO TO 100	58
	END	59

\$IBFTC	LOCKX	DECK						1
	SUBROJTIME	LOCKX(J)						2
	COMMON							3
	./IPARAM/	ABORT	,KIUNIT	,KOUNIT	,LIMALF	,LOCK	,LOCKX	4
	.	,NOLIST	,NSTDIR	,TRACE				5
	LOGICAL	J,LOCK						6
	LOCK = J							7
	RETURN							8
	END							

```

$IBFTC DEBUGX DECK
SUBROUTINE DEBUGX
C
C          *INITIALIZATION
C          DIMENSION ISUBN(20)
C          TRACE = 0 NO PRINTING
C          TRACE = 1 PRINT DEBUG 2+3 CALLS ONLY
C          TRACE = 2 PRINT DEBUG 2+3 CALLS ONLY
C          TRACE = 3 PRINT DEBUG 2+3 AND STACK PRINT
C          TRACE = 4 PRINT DEBUG 2+3 AND STACK PRINT AND CALLS FROM CHAR
C          DIMENSION IFT(27), IPTAB(21), ITAB(65)
C          DIMENSION ANAME(15), IMAGE(80), IMAGE1(81), IPARAM(9)
C          , KSTACK(27), NAME(15), RVALUE(2), STACK(27)
C
C          COMMON
C          ,/ICOMNI/ VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB
C          , ,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2
C          , ,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK
C          , ,MCNVRT ,MDIF ,MODALL ,MSTOR
C          , ,NAME ,NERROR ,NONEW ,NOTARG
C          , ,SMCHR ,TEST ,ERMARK
C          ,/ICNSTI/ BLANK ,BLANKS ,DOLLAR ,EOS
C          , ,ICOMMA ,IDOLLAR ,IFT ,IPTAB ,ITAB
C          , ,KAM10 ,KBPC ,KBPW ,KCPCD ,KERTYP
C          , ,KZERO ,NOPRNT ,TAB1
C          ,/IPARAM/ ABORT ,<IUNIT ,KOUNIT ,LIMLF ,LOCK ,LOCX
C          , ,NOLIST ,VSTDIR ,TRACE
C          ,/ISTACK/ STACK ,ISTDIM ,KSTACK ,LEVLIM
C
C          INTEGER BLANK ,BLANKS ,EOS ,IDOLLAR ,TAB1 ,TRACE
C          DATA BADCAL/6HBADCAL/
C          DOUBLE PRECISION ALFARG, ISUBN, STACK, VALUE, DBLANK, DARG
C          LOGICAL ABORT,DOLLAR,ERMARK,LIMLF,LOCK,NOLIST,NONEW,MODALL,
C          ,NSTDIR,SMCHR,TEST
C          EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE), (NAME,ANAME)
C          EQUIVALENCE (ICOMNI,ISUB), (IMAGE,IMAGE1), (IPARAM,ABORT)
C          , (DBLANK,BLANK)
C          ISUBC = 0
C          DO 10 I = 1,10
10      ISUBN(I) = DBLANK
C          IF(TRACE.EQ.0) GO TO 99
C          WRITE (KOUNIT,410)
C          GO TO 99
C          ENTRY DEBUGC(ISUBNA)
C          DOUBLE PRECISION ISJBNA
C          , NEW SUBROUTINE CALLED
C          IF (ISUBC.GT.10) GO TO 98
C          ALFARG = DBLANK
C          NUMARG = -1
C          ISUBC = ISUBC+1
C          ISUBN(ISUBC) = ISUBNA
C          IF(TRACE .GE.4) GO TO 400
C          GO TO 99
C
C          ALFARG = BADCAL
C          , CALLS MIGHT GET OUT OF RANGE
C          NUMARG = ISUBC
C          GO TO 400
C
C          ENTRY DEBUGR
C          , CALL AT RETURN
C          IF (ISUBC.LT.1) GO TO 98
C          ALFARG = DBLANK
C          NUMARG = -1

```

	ISUBN(ISUBC) = DBLANK	63
	ISUBC = ISUBC - 1	64
	IF(TRACE .GE.4) GO TO 400	65
	GO TO 99	66
C		67
	ENTRY DEBUG2(ALFAR,NUMAR)	68
	DOUBLE PRECISION ALFAR	69
	NUMARG = NUMAR	70
	ALFARG = ALFAR	71
20	IF (TRACE .LT.1) GO TO 99	72
400	WRITE (KOUNIT,405) (ISUBN(I),I=1,4),ALFARG,NUMARG,	73
	.KCOUNT,SMCHR,KCH,KFLD1,ICOMP,KFLD2,LCOMP	74
99	RETURN	75
C		76
	ENTRY DEBUG3(ALFAR,DNUMAR,ITYPE)	77
	DOUBLE PRECISION DNUMAR	78
	ENTRY DEBUG4(ALFAR,DNUMAR,ITYPE)	79
	DARG = DNUMAR	80
	NUMARG=NUMAR	81
	ALFARG = ALFAR	82
	IF (TRACE .LT.1) GO TO 99	83
	GO TO (30,40,50,60), ITYPE	84
30	WRITE (KOUNIT,406) (ISUBN(I),I=1,4),ALFARG,NUMARG,	85
	.KCOUNT,SMCHR,KCH,KFLD1,ICOMP,KFLD2,LCOMP	86
	GO TO 99	87
40	WRITE (KOUNIT,407) (ISUBN(I),I=1,4),ALFARG,NUMARG,	88
	.KCOUNT,SMCHR,KCH,KFLD1,ICOMP,KFLD2,LCOMP	89
	GO TO 99	90
50	WRITE (KOUNIT,408) (ISUBN(I),I=1,4),ALFARG,NUMARG,	91
	.KCOUNT,SMCHR,KCH,KFLD1,ICOMP,KFLD2,LCOMP	92
	GO TO 99	93
60	WRITE (KOUNIT,409) (ISUBN(I),I=1,4),ALFARG,NUMARG,	94
	.KCOUNT,SMCHR,KCH,KFLD1,ICOMP,KFLD2,LCOMP	95
	GO TO 99	96
405	FORMAT(1H ,29X,5(A6,1X),I24,	97
	.1H(,I3,L2,1H),A6,2H(,I3,I2,1H),2H(,I3,I2,1H))	98
406	FORMAT(1H ,29X,5(A6,1X),E24.8,	99
	.1H(,I3,L2,1H),A6,2H(,I3,I2,1H),2H(,I3,I2,1H))	100
407	FORMAT(1H ,29X,5(A6,1X),11X,012,1X,	101
	.1H(,I3,L2,1H),A6,2H(,I3,I2,1H),2H(,I3,I2,1H))	102
408	FORMAT(1H ,29X,5(A6,1X),16X,A8,	103
	.1H(,I3,L2,1H),A6,2H(,I3,I2,1H),2H(,I3,I2,1H))	104
409	FORMAT(1H ,29X,5(A6,1X),024.18,	105
	.1H(,I3,L2,1H),A6,2H(,I3,I2,1H),2H(,I3,I2,1H))	106
410	FORMAT(29X,11HDEBUG TRACE,52X,9HI--KCOUNT,5X,8HI--LIST1/	107
	.68X,94D ARG OR,15X,9HI I-SMCHR,5X,10HI I--ICOMP /	108
	.29X,19HSUBROUTINES CALLED,11X,7HALF ARG,7X,11HNUMERIC ARG,9X,	109
	.3HI I,1X,5HKCH-I,5X,16HI I I--LIST2 /	110
	.92X,34I I,5X,1HI,5X,18HI I I I--LCJMP)	111
	END	112

\$IBFTC STACKP DECK

	SUBROJTIME STACKP	1
	DIMENSION IFT(27), IPTAB(21), ITAB(65)	2
	DIMENSION ANAME(15), IMAGE(80), IMAGE1(81), IPARAM(9)	3
	. , KSTACK(27), NAME(15), RVALUE(2), STACK(27)	4
C		5
	COMMON	6

	./ICOMVI/	VALUE	,ICOMP	,IFNTYP	,IMAGE1	,IRADIX	,ISUB	7	
	.	,KCH	,KCNVRT	,KCOUNT	,KDIF	,KFLD1	,KFLD2	8	
	.	,LCOMP	,LCNVRT	,LEVEL	,LFRT	,LOOK		9	
	.	,MCNVRT	,MDIF	,MODALL	,MSTOR			10	
	.	,NAME	,NERROR	,NONEW	,NOTARG			11	
	.	,SMCHR	,TEST	,ERMARK				12	
	./ICNSTI/	BLANK	,BLANKS	,DOLLAR	,EOS			13	
	.	,ICOMMA	,IDOLAR	,IFT	,IPTAB	,ITAB		14	
	.	,KAM10	,KBPC	,KBPW	,KCPCD	,KERTYP		15	
	.	,KZERO	,NOPRNT	,TAB1				16	
	./IPARAM/	ABORT	,KIUNIT	,KOUNIT	,LIMALF	,LOCK	,LOCKX	17	
	.	,NOLIST	,NSTDIR	,TRACE				18	
	./ISTACKI/	STACK	,ISTDIM	,KSTACK	,LEVLIM			19	
C		INTEGER	BLANK	,BLANKS	,EOS	,IDOLAR	,TAB1	,TRACE	20
		DOUBLE PRECISION	STACK, VALUE						21
		LOGICAL	ABORT,DOLLAR,ERMARK,LIMALF,LOCK,NOLIST,NONEW,MODALL,						22
		,NSTDIR,SMCHR,TEST							23
		EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE), (NAME,ANAME)							24
		EQUIVALENCE (ICOMNI,ISUB), (IMAGE,IMAGE1), (IPARAM,ABORT)							25
		IF (TRACE .LT. 3) GO TO 99							26
		LEVELX = LEVEL + 3							27
		WRITE(KOUNIT,85) LEVEL,VALUE,KVALUE,RVALUE,LOCKX							28
		WRITE(KOUNIT,89) (STACK(I),I=1,LEVELX)							29
		WRITE(KOUNIT,86) (KSTACK(I),I=1,LEVELX)							30
89		FORMAT(11D12.4)							31
86		FORMAT(11I12)							32
85		FORMAT(7H LEVEL=I3,7H VALUE=D25.17,8H KVALUE=I3,3H ,D,2(1X,D12),							33
		.6H LOCKX=I6)							34
99		RETURN							35
		END							36
									37

APPENDIX B

HUFF INPUT ROUTINE

The first version of the Huff Input Routine was reported in reference 5. The Huff Input Routine provides more versatility in reading input data into the computer than the NAMELIST feature in FORTRAN. The Huff Input Routine has the ability to make simple arithmetic manipulations (such as conversion of units) during loading and to load alphanumeric data. While not an indispensable feature, it has been found to be quite convenient. The Huff Input Routine also allows for the automatic printout of data cards at execution time.

The following section contains a general description of the Huff Input Routine and its usage.

Usage

The programmer transfers control to the INPUT routine with a standard FORTRAN IV call

CALL INPUT (5, 6, 1, X, ITABLE)

Argument 1 is the system input tape number (5 on the Lewis System). Argument 2 is the system output tape number (6 on the Lewis System). Argument 3 is the identifying number of a data group. This value is compared with an identification number occurring on an input card (\$DATA card). If the values agree, the data are processed until another \$DATA or end-of-data (\$END) card is encountered. If the values do not agree, no data are processed and control is returned to the calling program. Argument 4 is the array X, which serves as a reference point for the storing of input data. Since all data are stored relative to X, the programmer must provide fixed relations between the location of X and other locations to be loaded (e. g. , through the use of common blocks and/or equivalence statements). In this code, X is 'WORD,' the first name in the labeled common block 'ALL.' Common blocks ALL, DESIGN, FRONT, SIDE, BACK, DUMMY, and SPOOL 2 are in all routines; hence, they are loaded sequentially so that the location of all variables is known. Argument 5 is the array ITABLE, which contains the names of the variables used on the cards and their subscript location relative to X. Sufficient space must be provided in the calling program for storing the table of names. This is done by a DIMENSION statement. The dimension of ITABLE should be stored in ITABLE(2). ITABLE(3) must initially be zero.

Types of Input Statements

\$DATA statement. - The \$DATA statement identifies a group of data with an identification number. It must be the first statement on a card. For example, \$DATA(1) or \$D(1) on the first card of a data group causes the value 1 to be compared with argument 3 in the calling sequence. If unequal, control is returned to the calling program. If equal, data are loaded until the next \$DATA or \$END statement is reached.

\$TABLE statement. - The \$TABLE statement makes a list of names needed for loading data. Consider for example that the real variable names VELOCITY, MASS, and RADIUS are to be assigned to memory locations X(1), X(2), and X(3), respectively. The card would be punched \$TABLE (.REAL., 1 = VELOCITY, 2 = MASS, 3 = RADIUS). These variables will be treated as real in any subsequent loading of data. A limit of 15 computer words is placed on the length of a name. Since .REAL. is what designates the mode of the name, a name may begin with any alphabetic letter. For example, the statement \$TABLE (.INTEGER., 20 = INDEX, SUBSCRIPT, I) will place these names in the table and any values subsequently loaded will be stored in X(20), X(21), and X(22), respectively, as integers. In a similar manner, \$TABLE(.DOUBLE PRECISION., 10 = RADIUS DOUBLE, .LOGICAL., 12 = SWITCH 1, SWITCH 2) causes the name RADIUS DOUBLE to be stored in the table as a double-precision variable equivalent to X(10) and X(11), and the logical variables SWITCH 1 and SWITCH 2 will be equivalent to X(12) and X(13).

Note that \$TABLE statements are loaded as Number = Name to avoid confusion with loading statements.

Loading Statement

The loading statement loads data by taking the name of a variable previously appearing in a \$TABLE statement and setting it equal to a value which may be of several forms.

Numeric values. - Standard FORTRAN language is used; for example, VELOCITY = 3.4, MASS = 32 (no decimal point is needed and MASS will have the REAL value 32), RADIUS = 4E21, and INDEX = 3. Data can be continued from one card to another; for example, SUBSCRIPT may appear at the end of one card and = 47 on the next card.

Subscripts may be used. Since 3 = RADIUS, RADIUS(2) = 6, 10, 12, , 14 will put real numbers in X(4), X(5), X(6), and X(8) and leave X(7) unchanged because of the double comma.

If new values are assigned to a variable before the next \$D(1) card, the new value

will override the previous one. For example, $RADIUS(2) = 8$ will override the $RADIUS(2) = 6$ card.

Internally addressed values. - An internally addressed value is one that refers to the contents of memory by name. $RADIUS(7) = RADIUS(3)$, $RADIUS(INDEX)$ causes $RADIUS(7)$ to assume the value of $RADIUS(3)$ and $RADIUS(8)$ to also assume the value of $RADIUS(3)$ since $INDEX = 3$.

The statement $RADIUS(7) = RADIUS(INDEX + 1)$ however is ILLEGAL.

Arithmetic expressions. - Provisions have been made to allow arithmetic operations to be performed on data at execution time. The operations + (addition), - (subtraction), * (multiplication), and / (division) and the functions, included among which are SQRT, EXP, SIN, COS, and $PWR(X, Y) (= X**Y)$, may be used with name or numbers (or any expression that has a value) to compute the value of an arithmetic expression. Parentheses may be used to indicate the order of performing the operations. The computations are analyzed from left to right and any intermediate results are stored in up to 24 locations in the core (the stack) which is sufficient for fairly complex expressions. All numeric operations are carried out in double-precision floating-point FORTRAN arithmetic. As an example, $RADIUS(2) = RADIUS(2)*SQRT(RADIUS(2))$ or $RADIUS(2) = PWR(RADIUS(2), 1.5)$ will set $RADIUS(2) = 8^{3/2}$.

Alphanumeric expressions. - Alphanumeric data may be entered by placing the variable name on the 'REAL' list and then setting the variable equal to the data by first enclosing in parentheses the length of the word to be read in. As an example,

$Q = (A39) \text{ THIS IS AN EXAMPLE OF ALPHANUMERIC DATA}$

The (A39) gives the length of the data including imbedded blanks. Of course, since on the IBM 7094 there are six characters per word, Q must internally be dimensioned to at least 7.

Printing Input Cards

Each input card processed will normally be written on the tape specified by the second argument of the calling sequence. An end-of-statement symbol read on the card will cause interpretation of the card to stop at that point and permit comments to be placed on the remainder of the card to be printed with the output. In order to avoid printing the card at all, the nonprint character is placed in the next column following the end-of-statement character. The developers of the routine selected the sign \neq for both characters. This is punched as a colon on an IBM Model 29 Key punch and corresponds to a 2-8 punch.

If the character following the end-of-statement symbol is other than a nonprint character, it is inserted as the printer control character in the first position of the output format before the card is written on output tape. If no end-of-statement character occurs on the card, a blank printer control character is used. Comment cards having the end-of-statement character as the first nonblank character will be printed and may be placed anywhere except in a continued alphabetic field.

In summary, the end-of-statement character has the effect of moving the end of the card forward to the column ahead of the end-of-statement character and the column following it is printer control.

If the control parameter NOLIST is true, printing is suppressed for all cards.

APPENDIX C

SYMBOLS

General Symbols Internal to Program

Variables in program are formed by combining these symbols.

Station Numbers

See figures 1 to 9 for each type of engine.

Thermodynamic Properties

AM	Mach number
FAR	fuel-air ratio, f/a
H	enthalpy, Btu/lbm
P	total pressure, atm
PS	static pressure, atm
S	entropy, Btu/ ^o R/lbm
T	total temperature, ^o R
TS	static temperature, ^o R
V	velocity, ft/sec

Component Symbols

A, AFT	afterburner
B	burner
C	inner compressor
COM	combustor
D	fan duct
F	first or fan compressor

I	intermediate (middle) compressor
M	core nozzle
MAIN	all but wing
NOZ	nozzle
OB	overboard
T	total
THP	inner (high pressure) turbine
TIP	middle (intermediate pressure) turbine
TLP	outer (low pressure) turbine
WDUCT	wing (third stream) duct
WING, WNG	wing (third stream)

Engine Symbols

BL	bleed, lbm/sec
CN	ratio of corrected speed to design corrected speed
DHT	turbine delta enthalpy, Btu/lbm
DHTC	turbine delta enthalpy (temperature corrected), $(H_{in} - H_{out})/T_{in}$, Btu/ $^{\circ}$ R/lbm
DP	pressure drop, $\Delta P/P$
DT	temperature change, $^{\circ}$ R
ETA	efficiency
ETAR	ram recovery, P_2/P_1
HPEXT	horsepower extracted
PCBL	fractional bleed
PCN	percent of design shaft speed
PR	pressure ratio
TFF	turbine flow function, $\text{lbm} \sqrt{^{\circ}\text{R}}/(\text{psia})(\text{sec})$
WA	airflow, lbm/sec
WF	fuel flow, lbm/sec

WG	gas flow, lbm/sec
Z	ratio of pressure ratios

Miscellaneous Symbols

A	area, ft
ALTP	altitude, ft
AM	Mach number of aircraft
BPRINT	bypass ratio (wing duct air/core air)
BYPASS	bypass ratio (fan duct air/air entering intermediate compressor)
C	when following component symbol signifies "corrected"
CF	correction factor, when used following component symbol
CS	ambient speed of sound, ft/sec
CV	nozzle velocity coefficient
DEL	delta degradation coefficient
DS	design value
DUM	dummy value
FCOVFN	ratio of core thrust to net thrust
FFOVFN	ratio of fan thrust to net thrust
FG	gross thrust, lbf
FGM	momentum thrust, lbf
FGP	pressure thrust, lbf
FMOVFN	ratio of fan plus core thrust to net thrust
FN	net thrust, lbf
FNOVFD	ratio of net thrust to design-point net thrust
FRD	ram drag, lbf
GU	initial or guessed values
ITRYS	number of loops through engine before quitting
LOOP	variable counter
LOOPER	number of loops through engine counter

SFC	specific fuel consumption, lbm/lbm/hr
TOLALL	tolerance on convergence
VA	velocity of aircraft, ft/sec
VJ	jet velocity, ft/sec

Input Symbols

AFTFAN	logical control for an aft-fan engine
ALTP	altitude, ft
AM	Mach number of aircraft
AM6	design afterburner entrance Mach number
AM23	design duct-burner entrance Mach number
AM55	design low-pressure-turbine exit Mach number
A6	area of afterburner entrance (calculated from AM6), ft ²
A8	main nozzle throat area (can be changed at off-design), ft ²
A28	fan duct nozzle throat area (see A8), ft ²
A38	wing duct nozzle throat area (see A8), ft ²
CNHPDS	design corrected speed - inner turbine
CNIPDS	design corrected speed - middle turbine
CNLPDS	design corrected speed - outer turbine
COLDDAY	factor for correcting corrected airflows for cold day, T2 = -19° F
CVDNOZ	nozzle thrust coefficient (DUCT)
CVDWNG	nozzle thrust coefficient (WING)
CVMNOZ	nozzle thrust coefficient (CORE)
DELFG	gross-thrust delta degradation multiplier
DELFN	net-thrust delta degradation multiplier
DELSFC	specific-fuel-consumption delta degradation multiplier
DPAFDS	afterburner design pressure drop, $\Delta P/P$
DPCODS	combustor design pressure drop, $\Delta P/P$
DPDUDS	duct design pressure drop, $\Delta P/P$

DPWGDS	wing duct design pressure drop, $\Delta P/P$
DTCODS	combustor design temperature increase (automatically set to $T_4 - T_3$), $^{\circ}\text{R}$
DUMMYSPOOL	logical control for spool which does not change temperature or pressure of air
ETAA	afterburner efficiency (not required)
ETAADS	afterburner efficiency at design
ETABDS	combustor efficiency at design
ETACDS	inner-compressor adiabatic efficiency at design
ETAD	duct-burner combustor efficiency
ETAFDS	front (outer) compressor adiabatic efficiency at design
ETAIDS	intermediate (middle) compressor adiabatic efficiency at design
ETAR	inlet pressure recovery (ram recovery), P_2/P_1
ETHPDS	high-pressure- (inner) turbine design adiabatic efficiency
ETIPDS	intermediate-pressure- (middle) turbine design adiabatic efficiency
ETLPDS	low-pressure- (outer) turbine design adiabatic efficiency
FIXFANTOMIDDLE	logical control for boosted fan
FIXMIDDLETOCOMP	logical control for supercharged compressor
HOTDAY	factor for correcting corrected airflows to hot day, $T_2 = 44^{\circ}\text{F}$
HPEXT	horsepower extraction
IAFTBN	index on afterburning desired
IAMTP	index on ram or inlet operation desired
IDBURN	index on duct burning desired
IDCD	duct nozzle convergent-divergent when $IDCD = 1$ (design or off-design)
IDES	index for design point; must be set equal to 1 to design engine; zeroed automatically
IDUMP	index for dumping of error matrix
IGASMX	index for mixed-flow or non-mixed-flow turbofans

IMCD	main nozzle convergent-divergent when IMCD = 1 (design or off-design)
ITRYS	index for maximum number of iterations
MODE	independent variable designator for engine operation
NOZFLT	index for floating main or duct nozzle
PCBLC	ratio of compressor bleed to turbines to compressor airflow
PCBLDU	ratio of compressor bleed leaked into fan duct to total compressor bleed flow
PCBLF	ratio of bleed from outer compressor to fan airflow dumped overboard (i. e. , leakage)
PCBLHP	fraction of PCBLC used for high-pressure (inner) turbine (cooling)
PCBLIDS	ratio of design value of air into wing to air into core; zero for two-stream engine
PCBLIP	fraction of PCBLC used for intermediate-pressure turbine (cooling)
PCBLLP	fraction of PCBLC used for low-pressure (outer) turbine (cooling)
PCBLOB	inner-compressor bleed compressor airflow (overboard for customer use)
PCNC	inner-compressor shaft speed as a percent of design shaft speed
PCNCDS	design inner-compressor shaft speed
PCNF	outer-compressor shaft speed as a percent of design
PCNFDS	design outer-compressor shaft speed as a percent of design shaft speed
PCNI	intermediate-compressor shaft speed as a percent of design
PCNIDS	design intermediate-compressor shaft speed as a percent of design shaft speed
POLARDAY	factor for correcting corrected airflow to polar day, $T_2 = -75^{\circ} \text{ F}$
PRCDS	design inner-compressor pressure ratio
PRFDS	design outer-compressor pressure ratio

PS55	static pressure at low-pressure-turbine exit
P2	compressor-face total pressure (for nonstandard days only), atm
TFHPDS	design inner-turbine flow function
TFIPDS	design intermediate-turbine flow function
TFLPDS	design outer-turbine flow function
TOLALL	tolerance on error matrix
TROPICALDAY	factor for correcting corrected airflows for tropical day, $T2 = 31^{\circ} \text{ F}$
T2	compressor-face total temperature (for nonstandard days only), $T1 + T2$
T24	duct-burner exit temperature, $^{\circ}\text{R}$
T4	combustor exit temperature, $^{\circ}\text{R}$
T4DS	design combustor exit temperature, $^{\circ}\text{R}$
T7	afterburner exit temperature, $^{\circ}\text{R}$
T7DS	design afterburner exit temperature, $^{\circ}\text{R}$
WAFCDs	design outer-compressor corrected airflow, lbm/sec
WAICDS	design intermediate-compressor corrected airflow, lbm/sec
WFA	fuel flow rate to afterburner (IAFTBN = 2 only), lbm/sec
WFB	fuel flow rate to main burner (MODE = 2 only), lbm/sec
WFBDS	design fuel flow rate to main burner (MODE = 2 only), lbm/sec
ZCDS, ZFDS, ZIDS	design ratio of inner compressor, fan compressor, and middle compressor pressure ratios, respectively; equals pressure ratio at design point on design speed line minus value of pres- sure ratio at lowest point on speed line divided by high (surge) value minus low value of pressure ratio on the design speed line

Output Symbol List¹

A area, ft^2

¹Some symbols, such as T4, are followed by station numbers; see appropriate figure for each engine in order to determine station numbers.

ALTP	altitude, ft
AM	Mach number
BLC	bleed flow out of compressor, lbm/sec
BLF	bleed flow out of fan (dumped overboard), lbm/sec
BLHP	bleed flow into high-pressure turbine, lbm/sec
BLI	airflow into third stream, lbm/sec
BLIP	bleed flow into intermediate-pressure turbine, lbm/sec
BPRINT	ratio of airflow into wing duct to airflow into core
BYPASS	ratio of airflow into fan duct to airflow into intermediate compressor
CNC	corrected shaft speed - inner compressor
CNF	corrected shaft speed - fan
CNHP	corrected shaft speed - high-pressure turbine, $PCNC/\sqrt{T_{in}}$
CNHPCF	corrected speed - high-pressure-turbine correction factor
CNI	corrected shaft speed - intermediate compressor
CNIP	corrected shaft speed - intermediate-pressure turbine, $PCNI/\sqrt{T_{in}}$
CNIPCF	corrected speed - intermediate-pressure-turbine correction factor
CNLP	corrected speed - low-pressure turbine, $PCNF/\sqrt{T_{in}}$
CNLPCF	corrected speed - low-pressure-turbine correction factor
CVDNOZ	velocity coefficient of fan nozzle
CVDWNG	velocity coefficient of wing nozzle
CVMNOZ	velocity coefficient of core nozzle
DHHPCF	high-pressure-turbine delta enthalpy correction factor
DHIPCF	intermediate-pressure-turbine delta enthalpy correction factor
DHLPCF	low-pressure-turbine delta enthalpy correction factor
DHTC	work done by high-pressure turbine, Btu/lbm
DHTCHP	enthalpy change temperature corrected - high-pressure turbine, Btu/ $^{\circ}$ R/atm/lbm
DHTCIP	enthalpy change temperature corrected - intermediate-pressure turbine, Btu/ $^{\circ}$ R/atm/lbm

DHTCLP	enthalpy change temperature corrected - low-pressure turbine, Btu/ ⁰ R/atm/lbm
DHTF	work done by low-pressure turbine, Btu/lbm
DHTI	work done by intermediate-pressure turbine, Btu/lbm
DPCOM	$(\Delta P/P)_{\text{combustor}}$
DPDUC	$(\Delta P/P)_{\text{fan duct}}$
DPWING	$(\Delta P/P)_{\text{wing duct}}$
DTCCOF	temperature-rise-across-combustor correction factor
ETAB	combustor efficiency
ETABCF	combustor efficiency correction factor
ETAC	inner-compressor adiabatic efficiency
ETACCF	inner-compressor efficiency correction factor
ETAD	duct-burner efficiency
ETAF	fan adiabatic efficiency
ETAFCF	fan efficiency correction factor
ETAI	intermediate-compressor adiabatic efficiency
ETAICF	intermediate-compressor efficiency correction factor
ETATHP	high-pressure-turbine adiabatic efficiency
ETATIP	intermediate-pressure-turbine adiabatic efficiency
ETATLP	low-pressure-turbine adiabatic efficiency
ETHPCF	high-pressure-turbine efficiency correction factor
ETIPCF	intermediate-pressure-turbine efficiency correction factor
ETLPCF	low-pressure-turbine efficiency correction factor
FAR	fuel-air ratio, f/a
FCOVFN	ratio of core thrust to net thrust
FFOVFN	ratio of fan thrust to net thrust
FG	gross thrust, lbf
FGM	momentum thrust of all but wing, lbf
FGMWNG	momentum thrust of wing, lbf
FGP	pressure thrust of all but wing, lbf

FGPWNG	pressure thrust of wing, lbf
FMOVFN	ratio fan thrust plus core thrust to net thrust
FN	net thrust, lbf
FNMAIN	net thrust of all but wing, lbf
FNOVFD	ratio of net thrust to design-point net thrust
FNWING	net thrust of wing, lbf
FRD	ram drag, lbf
FWOVFN	ratio of net wing thrust to net thrust
HPEXT	horsepower extracted, hp
P	pressure, atm
PCBLC	fraction of compressor exit air bled for cooling or lost to cycle
PCBLDU	fraction of bleed air out of compressor which leaks into fan duct
PCBLF	fraction of fan exit airflow lost overboard
PCBLHP	fraction of compressor bleed air put into high-pressure turbine
PCBLI	fraction of intermediate-compressor air which goes into third stream
PCBLIP	fraction of compressor bleed air put into intermediate-pressure turbine
PCBLLP	fraction of compressor bleed air put into low-pressure turbine
PCNC	inner-compressor shaft speed as a percent of design
PCNF	fan-compressor shaft speed as a percent of design
PCNI	intermediate-compressor shaft speed as a percent of design
PRC	pressure ratio of inner compressor
PRCCF	pressure-ratio-of-inner-compressor correction factor
PRF	pressure ratio of fan
PRFCF	pressure-ratio-of-fan correction factor
PRI	pressure ratio of intermediate compressor
PRICF	pressure-ratio-of-intermediate-compressor correction factor
PS	static pressure, atm
SFC	specific fuel consumption, lbm/lbf/hr
T	temperature, °R
T2DS	design exit temperature of fan, °R

T21DS	design exit temperature of inner compressor, $^{\circ}\text{R}$
T22DS	design exit temperature of intermediate compressor, $^{\circ}\text{R}$
TFFHP	high-pressure-turbine flow function, $(\text{lbm})(\sqrt{^{\circ}\text{R}})(\text{in.}^2)/(\text{sec})(\text{lbf})$
TFFIP	intermediate-pressure-turbine flow function, $(\text{lbm})(\sqrt{^{\circ}\text{R}})(\text{in.}^2)/(\text{sec})(\text{lbf})$
TFFLP	low-pressure-turbine flow function, $(\text{lbm})(\sqrt{^{\circ}\text{R}})(\text{in.}^2)/(\text{sec})(\text{lbf})$
TFHPCF	high-pressure-turbine flow function correction factor
TFIPCF	intermediate-pressure-turbine flow function correction factor
TFLPCF	low-pressure-turbine flow function correction factor
V	velocity, ft/sec
VA	velocity of aircraft, ft/sec
VJD	fan duct exhaust velocity, ft/sec
VJM	core exhaust velocity, ft/sec
VJW	wing duct exhaust velocity, ft/sec
WA	airflow, lbm/sec
WA3CDS	corrected airflow in combustor at design, lbm/sec
WAC	inner-compressor airflow, lbm/sec
WACC	inner-compressor corrected airflow, lbm/sec
WACCF	inner-compressor corrected airflow correction factor
WACI	intermediate-compressor corrected airflow, lbm/sec
WAD	fan duct airflow, lbm/sec
WAF	fan airflow, lbm/sec
W AFC	fan corrected airflow, lbm/sec
WAFCF	fan corrected airflow correction factor
WAI	intermediate-compressor airflow, lbm/sec
WAICF	intermediate-compressor corrected airflow correction factor
WFB	fuel flow rate to combustor, lbm/sec
WFD	fuel flow rate to duct burner, lbm/sec
WFT	total fuel flow rate, lbm/sec
WG	gas flow rate, lbm/sec
WGT	total gas flow rate, lbm/sec

ZC	ratio of inner-compressor pressure ratios
ZF	ratio of fan pressure ratios
ZI	ratio of intermediate-compressor pressure ratios

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4. Keenan, Joseph H. ; and Kaye, Joseph: Gas Tables. John Wiley & Sons, Inc., 1948.
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TABLE I. - VARIABLES AND ERRORS

	Engine designation								
	a	b	c	d	e	f	g	h	i
	Number of spools								
	3	2	2	3	2	3	2	2	3
	Number of streams								
	3	3	3	2	2	3	3	2	2
	Turbofan	Boosted fan	Supercharged compressor	Turbofan		Aft fan	Supercharged compressor	Aft fan	
Variable 1	ZF	ZF	ZF	ZF	ZF	ZF	ZF	ZF	ZF
Variable 2	PCNF	PCNF	PCNF	PCNF	PCNF	PCNF	PCNF	PCNF	PCNF
Variable 3	ZC	ZC	ZC	ZC	ZC	ZC	ZC	ZC	ZC
Variable 4	PCNC	PCNC	PCNI	PCNC	PCNC	PCNC	PCNI	PCNC	PCNC
Variable 5	TFFHP	TFFHP	TFFIP	TFFHP	TFFHP	TFFHP	TFFIP	TFFHP	TFFHP
Variable 6	TFFLP	TFFLP	TFFLP	TFFLP	TFFLP	TFFLP	TFFLP	TFFLP	TFFLP
Variable 7	ZI	ZI	ZI	ZI	-----	ZI	ZI	-----	ZI
Variable 8	PCNI	-----	-----	PCNI	-----	PCNI	-----	-----	PCNI
Variable 9	TFFIP	-----	-----	TFFIP	-----	TFFIP	-----	-----	TFFIP
Error 1	$\frac{\text{TFHCAL} - \text{TFFHP}}{\text{TFHCAL}}$	(a)	$\frac{\text{TFICAL} - \text{TFFIP}}{\text{TFICAL}}$	(a)	(a)	(a)	(b)	(a)	(a)
Error 2	$\frac{\text{DHTCC} - \text{DHTCHP}}{\text{DHTCC}}$	(a)	$\frac{\text{DHTIC} - \text{DHTCIP}}{\text{DHTIC}}$	(a)	(a)	(a)	(b)	(a)	(a)
Error 3	$\frac{\text{TFLCAL} - \text{TFFLP}}{\text{TFLCAL}}$	(a)	(a)	(a)	(a)	(a)	(a)	(a)	(a)
Error 4	$\frac{\text{DHTCF} - \text{DHTCLP}}{\text{DHTCF}}$	(a)	(a)	(a)	(a)	(a)	(a)	(a)	(a)
Error 5	$\frac{\text{P25R} - \text{P25}}{\text{P25R}}$	(a)	(a)	(a)	(a)	(a)	(a)	(a)	(a)
Error 6	$\frac{\text{P7R} - \text{P7}}{\text{P7R}}$	(a)	(a)	(a)	(a)	(a)	(a)	(a)	(a)
Error 7	$\frac{\text{P38R} - \text{P38}}{\text{P38R}}$	(a)	(a)	$\frac{\text{WAC} - \text{WAI}}{\text{WAC}}$	-----	(a)	(a)	-----	(c)
Error 8	$\frac{\text{TFICAL} - \text{TFFIP}}{\text{TFICAL}}$	-----	-----	(a)	-----	(a)	-----	-----	(a)
Error 9	$\frac{\text{DHTIC} - \text{DHTCIP}}{\text{DHTIC}}$	-----	-----	(a)	-----	(a)	-----	-----	(a)
Matrix size	9 × 9	7 × 7	7 × 7	9 × 9	6 × 6	9 × 9	7 × 7	6 × 6	9 × 9

^aSame as error for engine a.^bSame as error for engine c.^cSame as error for engine d.

TABLE II. - INPUTS REQUIRED FOR BASIC CYCLES

Variable	Units or type	Definition	Engine designation								
			a	b	c	d	e	f	g	h	i
			Number of spools								
			3	2	2	3	2	3	2	2	3
			Number of streams								
			3	3	3	2	2	3	3	2	2
			Turbo-fan	Boosted fan	Super-charged compressor	Turbfan		Aft fan	Super-charged compressor	Aft fan	
PRFDS	-----	Fan pressure ratio	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes	Yes
WAFCDs	lb/sec	Fan corrected airflow	↓	↓	↓	↓	↓	↓	↓	↓	↓
ETAFDS	-----	Fan efficiency	↓	↓	↓	↓	↓	↓	↓	↓	↓
ZFDS	-----	Design Z of fan	↓	↓	↓	↓	↓	↓	↓	↓	↓
PCNFDS	-----	Corrected speed of fan	↓	↓	↓	↓	↓	↓	↓	↓	↓
PRIDS	-----	Intermediate pressure ratio	↓	↓	↓	↓	↓	↓	↓	↓	↓
WAICDS	lb/sec	Intermediate corrected airflow	↓	↓	↓	↓	↓	↓	↓	↓	↓
ETAIDS	-----	Intermediate efficiency	↓	↓	↓	↓	↓	↓	↓	↓	↓
ZIDS	-----	Design Z of intermediate compressor	↓	↓	↓	↓	↓	↓	↓	↓	↓
PCNIDS	-----	Corrected speed of intermediate compressor	↓	↓	↓	↓	↓	↓	↓	↓	↓
PRCDS	-----	Compressor pressure ratio	↓	↓	↓	↓	↓	↓	↓	↓	↓
PCBLIDS	-----	Fraction of air into third duct	↓	↓	↓	↓	↓	↓	↓	↓	↓
ETACDS	-----	Compressor efficiency	↓	↓	↓	↓	↓	↓	↓	↓	↓
ZCDS	-----	Design Z of compressor	↓	↓	↓	↓	↓	↓	↓	↓	↓
PCNCDS	-----	Corrected speed of compressor	↓	↓	↓	↓	↓	↓	↓	↓	↓
ETABDS	-----	Combustor efficiency	↓	↓	↓	↓	↓	↓	↓	↓	↓
DPCODS	-----	Combustor pressure drop, $\Delta P/P$	↓	↓	↓	↓	↓	↓	↓	↓	↓
T4DS	$^{\circ}R$	Turbine inlet temperature	↓	↓	↓	↓	↓	↓	↓	↓	↓
T FHPDS	$\frac{lb\sqrt{^{\circ}R}}{(sec)(psia)}$	High-pressure-turbine flow function	↓	↓	↓	↓	↓	↓	↓	↓	↓
CNHPDS	-----	High-pressure-turbine corrected speed	↓	↓	↓	↓	↓	↓	↓	↓	↓
ETHPDS	-----	High-pressure-turbine efficiency	↓	↓	↓	↓	↓	↓	↓	↓	↓
T FIPDS	$\frac{lb\sqrt{^{\circ}R}}{(sec)(psia)}$	Intermediate-turbine work function	↓	↓	↓	↓	↓	↓	↓	↓	↓
CNIPDS	-----	Intermediate-pressure-turbine corrected speed	↓	↓	↓	↓	↓	↓	↓	↓	↓
ETIPDS	-----	Intermediate-pressure-turbine efficiency	↓	↓	↓	↓	↓	↓	↓	↓	↓
T FL PDS	$\frac{lb\sqrt{^{\circ}R}}{(sec)(psia)}$	Low-pressure-turbine flow function	↓	↓	↓	↓	↓	↓	↓	↓	↓
CNL PDS	-----	Intermediate-pressure-turbine corrected speed	↓	↓	↓	↓	↓	↓	↓	↓	↓
ETLPDS	-----	Intermediate-pressure-turbine efficiency	↓	↓	↓	↓	↓	↓	↓	↓	↓
DPDUDS	-----	Fan pressure drop, $\Delta P/P$	↓	↓	↓	↓	↓	↓	↓	↓	↓
DPWGDS	-----	Wing duct pressure drop, $\Delta P/P$	↓	↓	↓	↓	↓	↓	↓	↓	↓
DPAFDS	-----	Afterburner pressure drop, $\Delta P/P$	↓	↓	↓	↓	↓	↓	↓	↓	↓
FIXFANTOMIDDLE	Logical	Boosted fans	. F.	. T.	. F.	. F.	. F.	. F.	. F.	. F.	. F.
FIXMIDDLETOCOMP	Logical	Supercharged compressors	. F.	. F.	. T.	. F.	. F.	. F.	. T.	. F.	. F.
DUMMYSPOOL	Logical	No intermediate spool	. F.	. F.	. F.	. T.	. F.	. F.	. F.	. T.	. F.
AFT FAN	Logical	Aft-fan engines	. F.	. F.	. F.	. F.	. T.	. T.	. T.	. T.	. T.

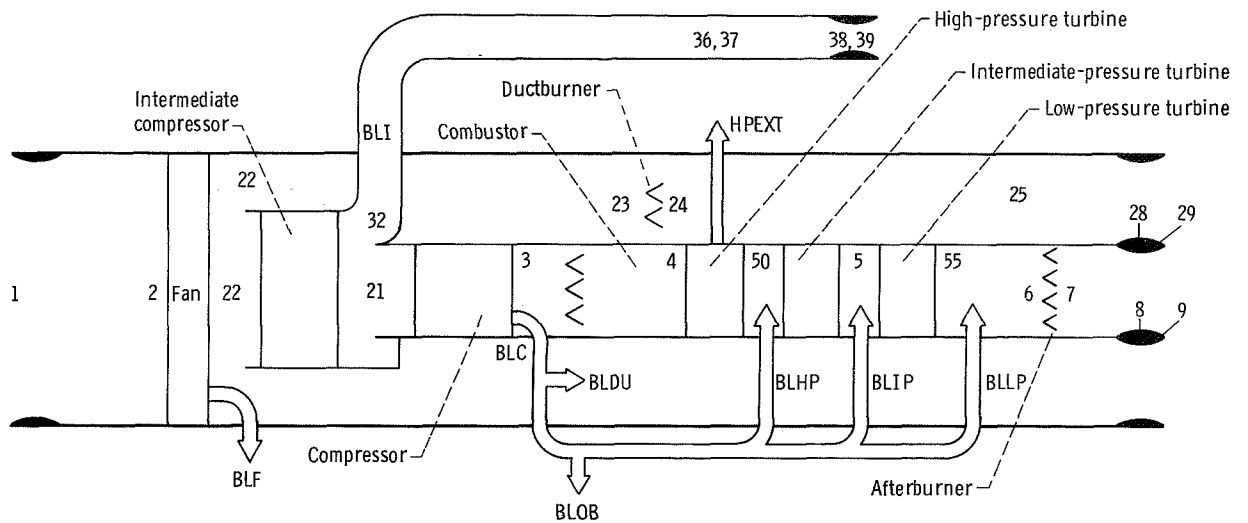


Figure 1. - Three-spool, three-stream turbofan engine (type a).

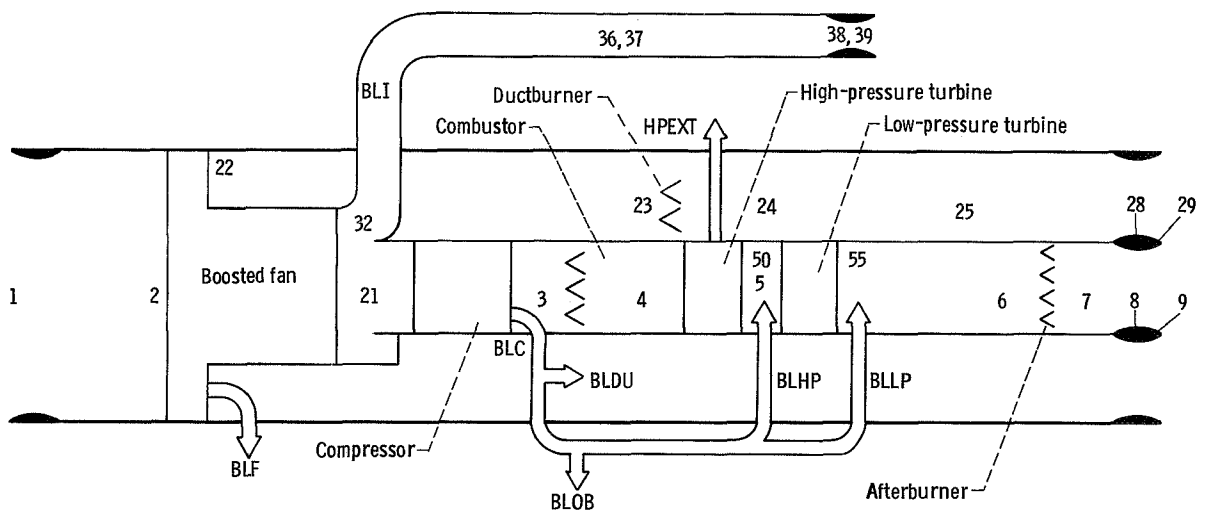


Figure 2. - Two-spool, three-stream boosted-fan engine (type b).

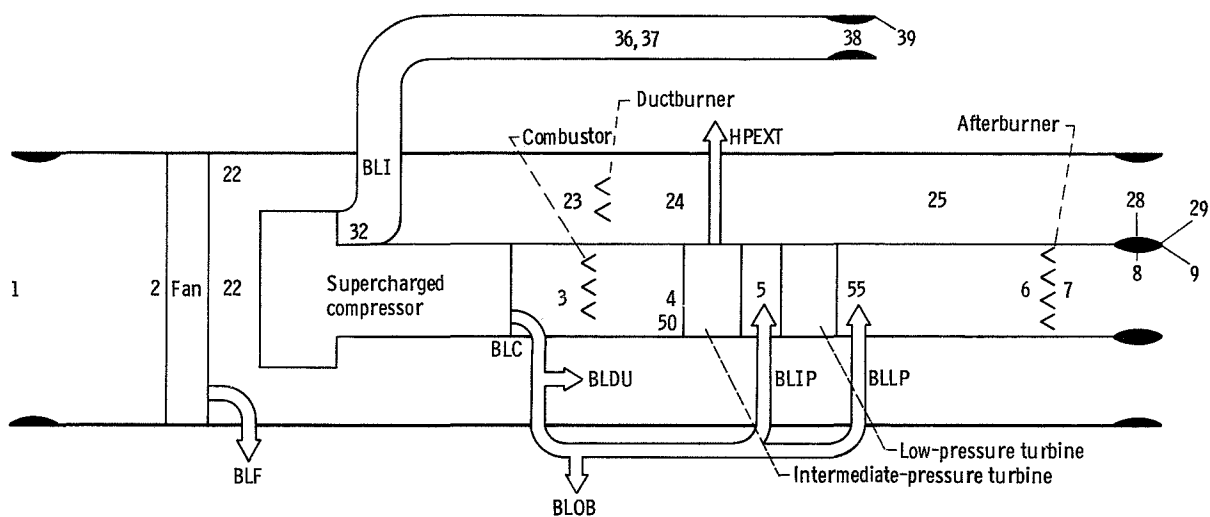


Figure 3. - Two-spool, three-stream, supercharged-compressor engine (type c).

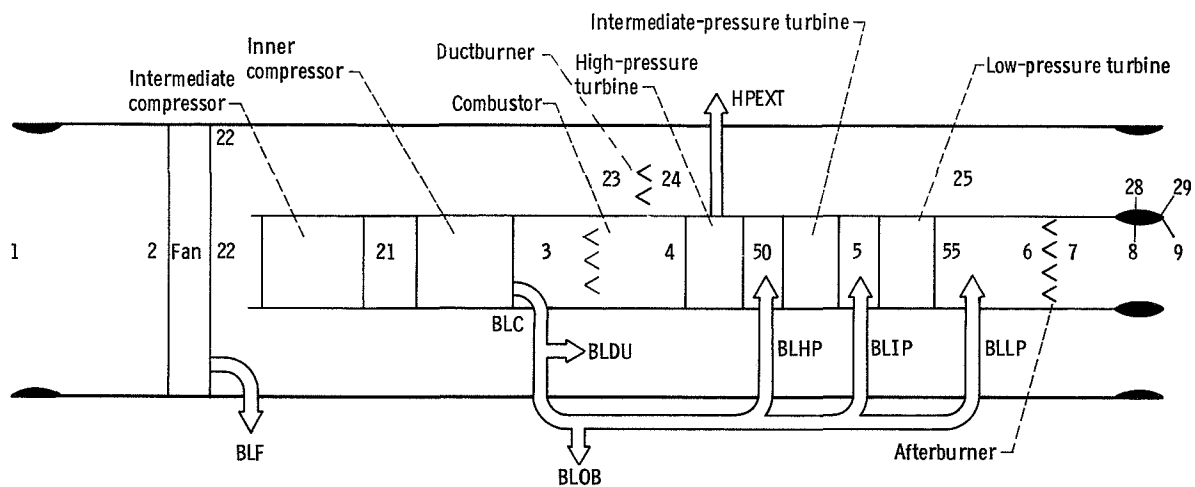


Figure 4. - Three-spool, two-stream engine (type d).

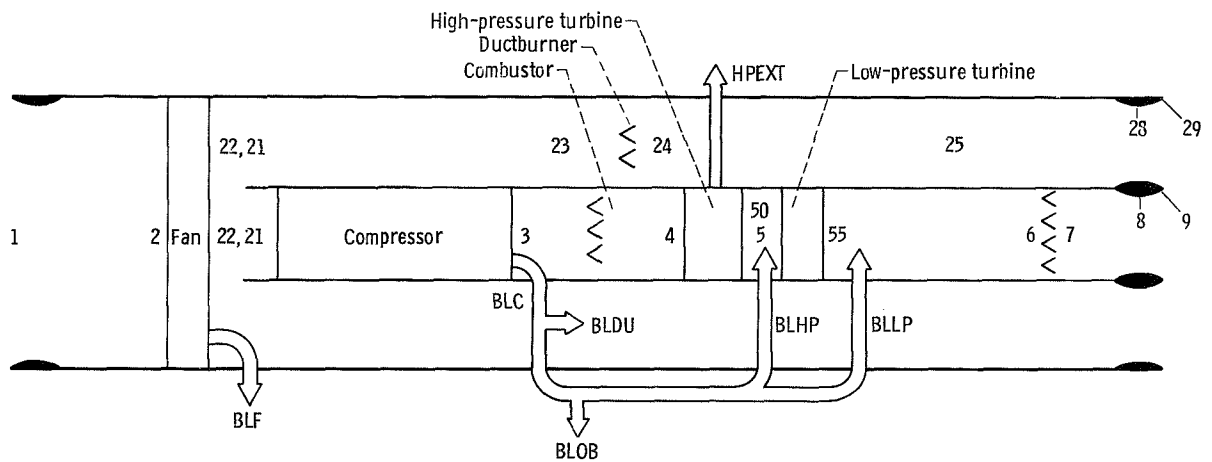


Figure 5. - Two-spool, two-stream turbofan engine (type e).

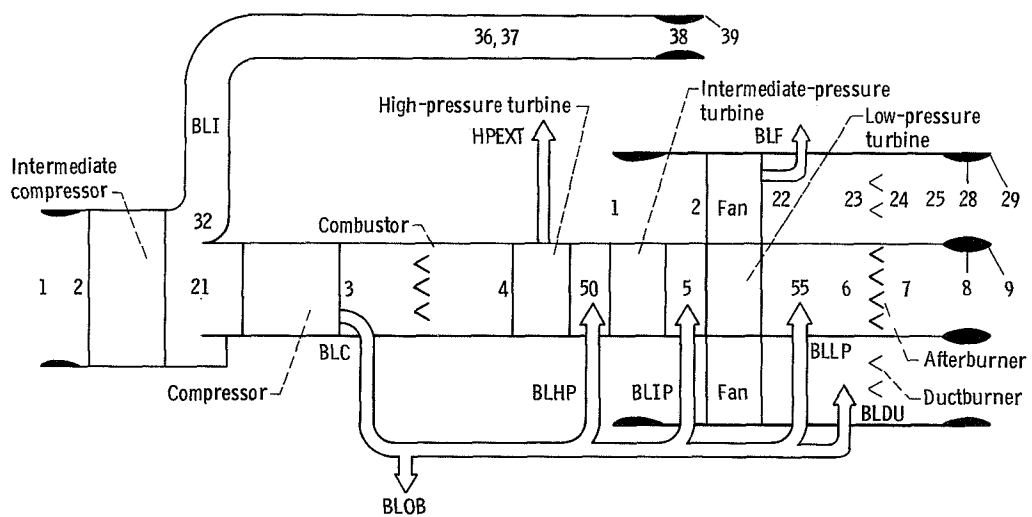


Figure 6. - Three-spool, three-stream, aft-fan engine (type f).

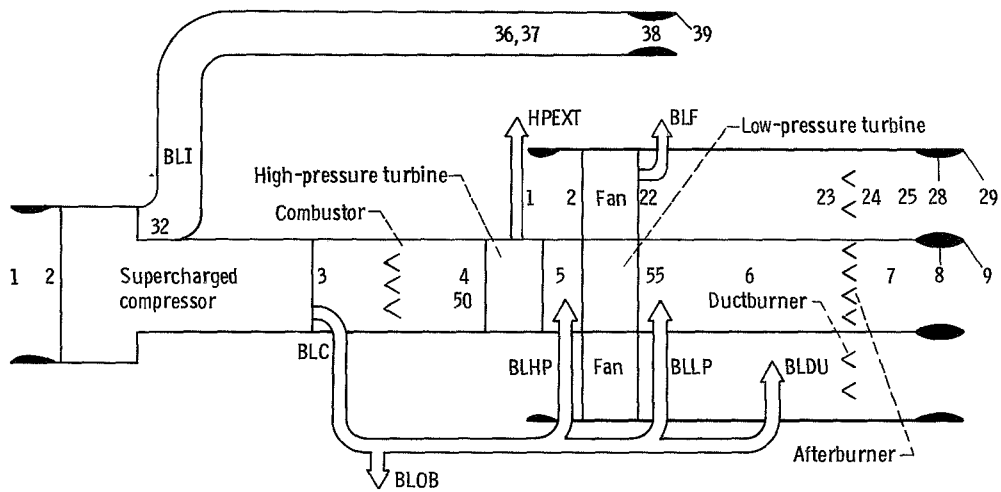


Figure 7. - Two-spool, three-stream, aft-fan engine (type g).

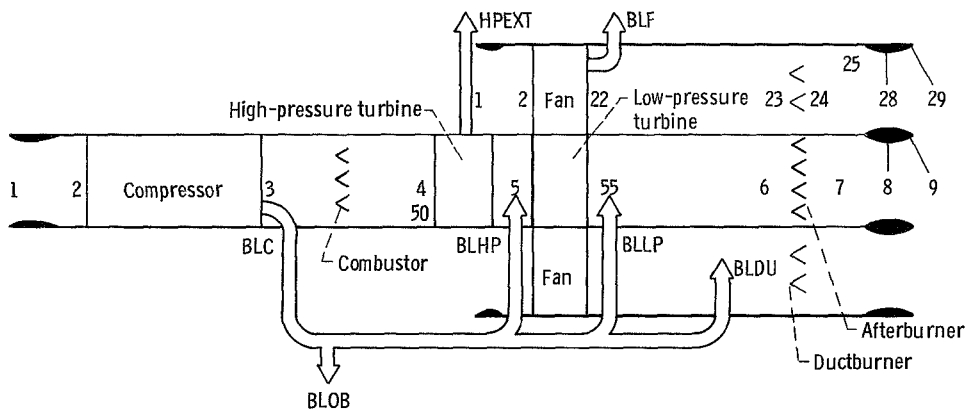


Figure 8. - Two-spool, two-stream aft-fan engine (type h).

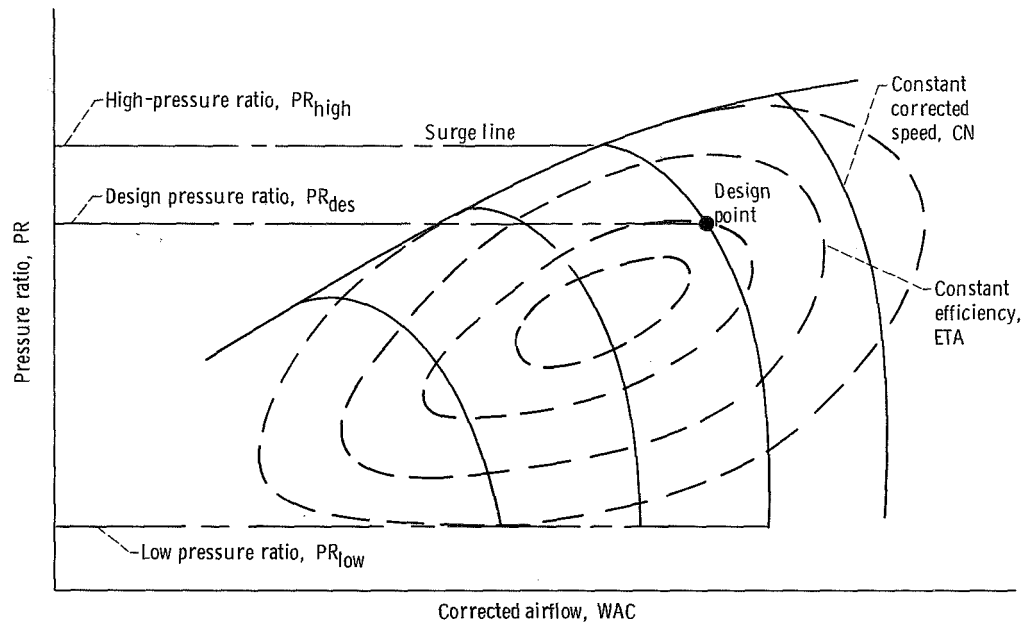


Figure 11. - Example of a specific fan-compressor map. $Z = (PR_x - PR_{low}) / (PR_{high} - PR_{low})$.

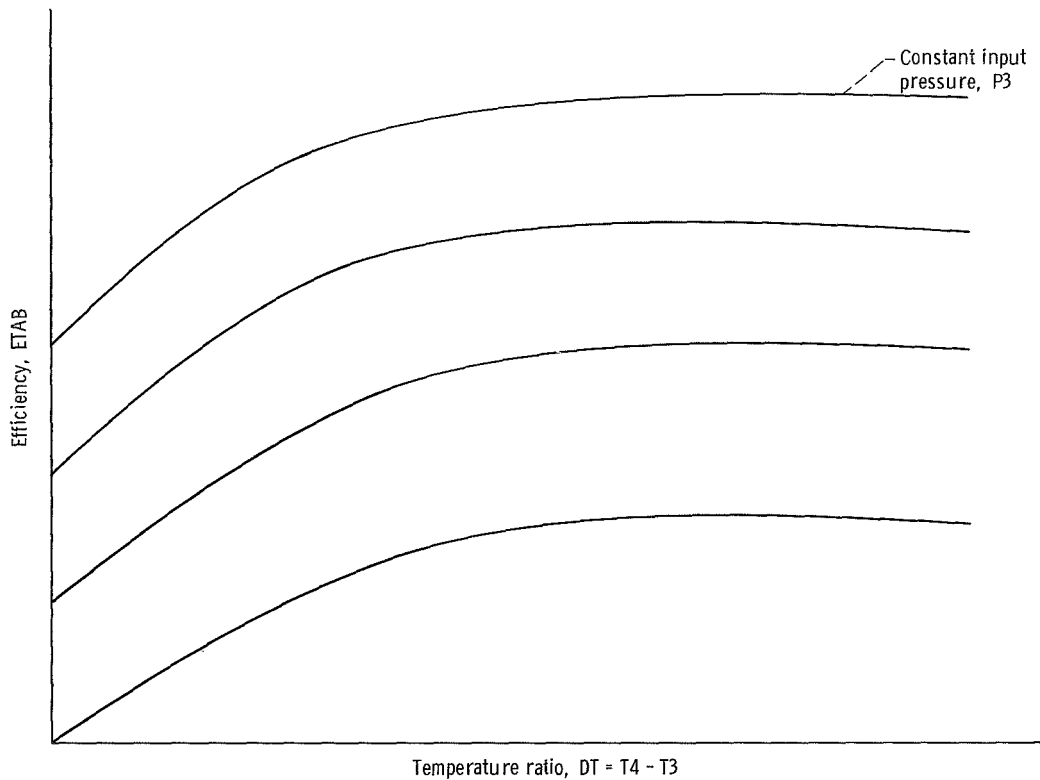


Figure 12. - Example of combustor map.

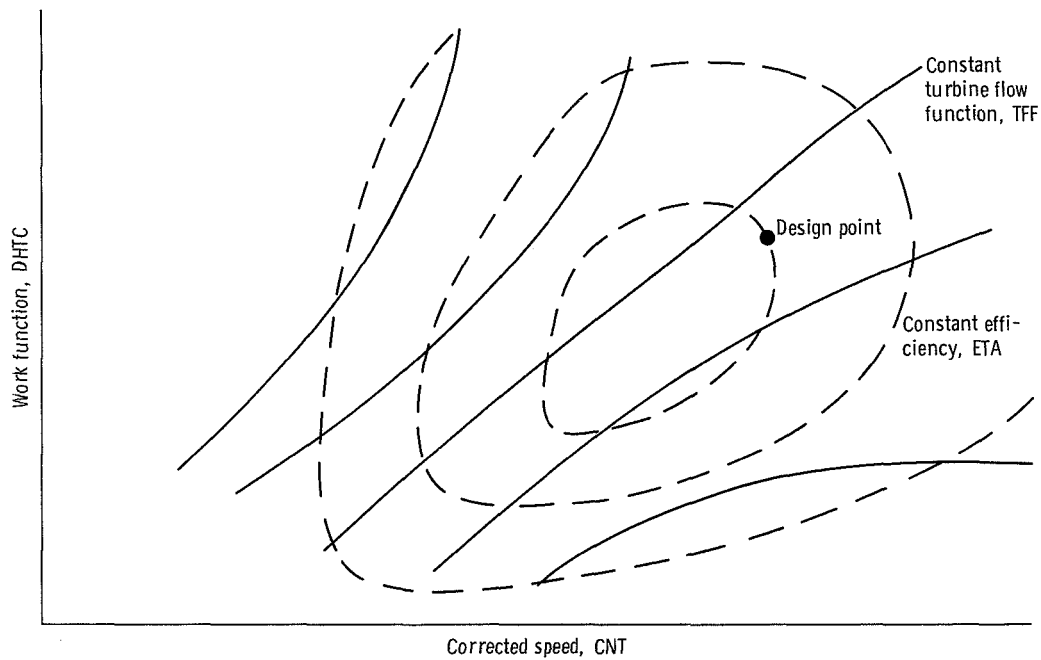
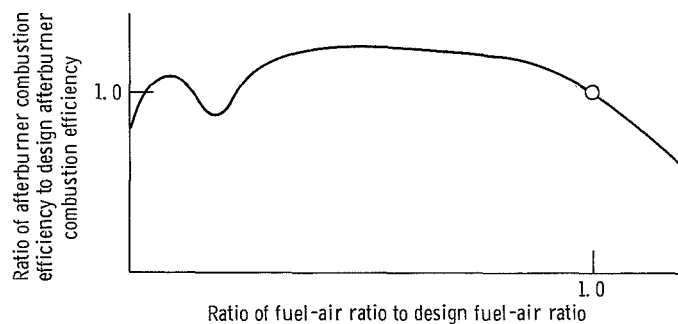
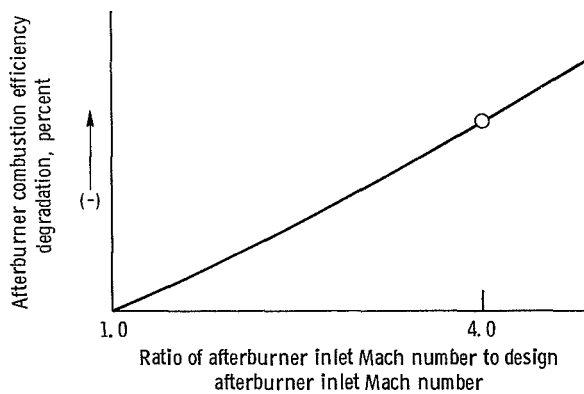


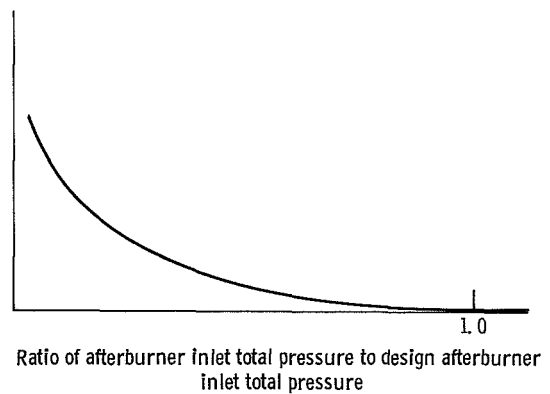
Figure 13. - Example of specific turbine map.



(a) Generalized afterburner combustion efficiency as function of fuel-air ratio.



(b) Efficiency correction factor against afterburner inlet Mach number.



(c) Efficiency correction factor against afterburner inlet total pressure.

Figure 14. - Example of a generalized afterburner combustion efficiency performance map.



POSTMASTER: If Undeliverable (Section 158
Postal Manual) Do Not Return

"The aeronautical and space activities of the United States shall be conducted so as to contribute . . . to the expansion of human knowledge of phenomena in the atmosphere and space. The Administration shall provide for the widest practicable and appropriate dissemination of information concerning its activities and the results thereof."

— NATIONAL AERONAUTICS AND SPACE ACT OF 1958

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